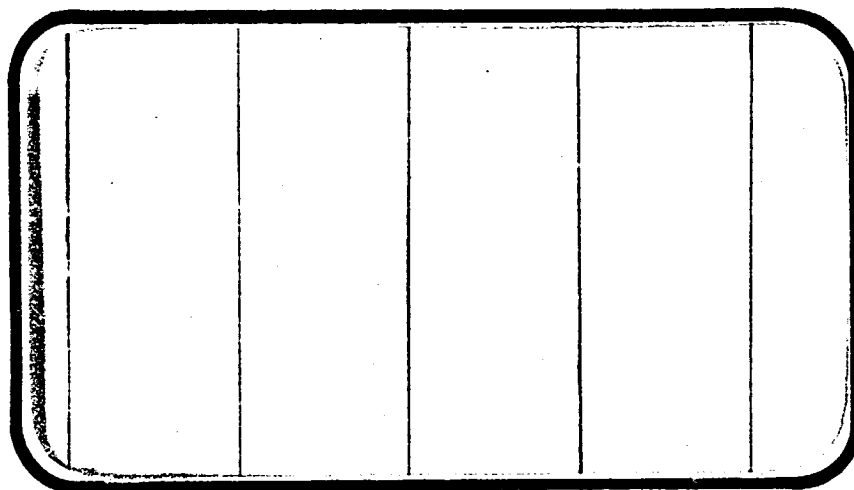


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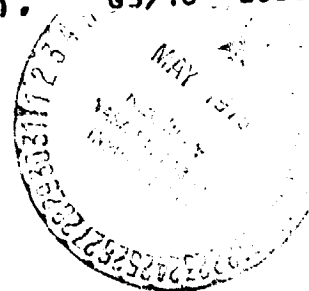
134448



(NASA-CR-134448) AIRLOADS INVESTIGATION OF  
AN 0.030-SCALE MODEL OF THE SPACE SHUTTLE  
VEHICLE 140A/B LAUNCH CONFIGURATION (MODEL  
47-OTS) IN THE ARC 11-FOOT UNITARY PLAN WIND  
TUNNEL FOR MACH RANGE 0.6 TO 1.4 (IA14A).

N75-23660

Unclas  
G3/18 20535



**SPACE SHUTTLE**

**AEROTHERMODYNAMIC DATA REPORT**

**JOHNSON SPACE CENTER**

**HOUSTON, TEXAS**

**DATA Management services**

**SPACE DIVISION**



**CHRYSLER  
CORPORATION**

March, 1975

DMS-DR-2084  
NASA CR-134,448

VOLUME 6 OF 11

AIRLOADS INVESTIGATION OF AN 0.030-SCALE MODEL  
OF THE SPACE SHUTTLE VEHICLE  
140A/B LAUNCH CONFIGURATION (MODEL 47-OTS)  
IN THE ARC 11-FOOT UNITARY  
PLAN WIND TUNNEL FOR MACH RANGE 0.6 TO 1.4 (IA14A)

by

R. L. Gillins  
Rockwell International Space Division

Prepared under NASA Contract Number NAS9-13247

by

Data Management Services  
Chrysler Corporation Space Division

for

Engineering Analysis Division

Johnson Space Center  
National Aeronautics and Space Administration  
Houston, Texas

WIND TUNNEL SPECIFICS:

Test Number: ARC 11-716  
NASA Series No.: IA14A  
Model Number: 47-OTS  
Test Dates: 4 through 13 September 1973

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Chrysler Corporation Space Division assumes no responsibility for the data presented other than display characteristics.

AIRLOADS INVESTIGATION OF AN 0.030-SCALE MODEL  
OF THE SPACE SHUTTLE VEHICLE  
140A/B LAUNCH CONFIGURATION (MODEL 47-OTS)  
IN THE ARC 11-FOOT UNITARY  
PLAN WIND TUNNEL FOR MACH RANGE 0.6 TO 1.4 (IA14A)  
VOLUME 6

By R. L. Gillins, Rockwell International Space Division

ABSTRACT

This report presents results of tests conducted on an 0.030-scale launch configuration model of the Space Shuttle Vehicle 140A/B in the NASA/ARC 11-Foot Unitary Plan Wind Tunnel. Aerodynamic loads data were obtained at Mach numbers from 0.6 to 1.4.

Surface pressure distributions were obtained simultaneously with six-component stability and control force data on the complete launch configuration. The configuration consisted of the orbiter, an external tank, two solid rocket boosters, and associated intercomponent attach hardware. Angles of attack and sideslip from -10 degrees to +10 degrees were investigated. The tests, designated IA14A, were conducted from 4 September 1973 through 13 September 1973.



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## INTRODUCTION

The 0.030-scale aero loads Space Shuttle Model was tested in the ARC Unitary Plan Wind Tunnels as follows:

IA14A	4 thru 13 Sept. 1973
IA14B	17 thru 19 Sept. 1973
OA22A	13 thru 14 Sept. 1973
OA22B	19 thru 20 Sept. 1973

For tests IA14B, OA22A, and OA22B, see reference 34, 35, and 36, respectively.

The testing was conducted in the 11-foot and the 9- by 7-foot tunnels of the ARC Unitary Plan Wind Tunnels. The IA14A/B tests were for the launch configurations at Mach numbers from 0.6 to 2.2. The OA22A/B tests were for the orbiter alone configuration at Mach numbers from 0.6 to 2.2. The effects of control surface deflections were also investigated in tests OA22A/B.

This report for test IA14A consists of one volume of force data and ten volumes of pressure data for a total of eleven volumes arrayed in the following manner:

Volume No.	Contents	Page
1.	IA14A force data	
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# NOMENCLATURE General

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
a		speed of sound; m/sec, ft/sec
C <sub>p</sub>	CP	pressure coefficient; $(p_1 - p_\infty)/q$
M	MACH	Mach number; $V/a$
p		pressure; N/m <sup>2</sup> , psf
q	Q(NSM) Q(PSF)	dynamic pressure; $1/2\rho V^2$ , N/m <sup>2</sup> , psf
RN/L	RN/L	unit Reynolds number; per m, per ft
V		velocity; m/sec, ft/sec
$\alpha$	ALPHA	angle of attack, degrees
$\beta$	BETA	angle of sideslip, degrees
$\psi$	PSI	angle of yaw, degrees
$\phi$	PHI	angle of roll, degrees
$\rho$		mass density; kg/m <sup>3</sup> , slugs/ft <sup>3</sup>

## Reference & C.G. Definitions

A <sub>b</sub>		base area; m <sup>2</sup> , ft <sup>2</sup>
b	BREF	reference span; m, ft
c.g.		center of gravity
$\bar{c}$		reference length or wing mean aerodynamic chord; m, ft
S	SREF	wing area or reference area; m <sup>2</sup> , ft <sup>2</sup>
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis

## SUBSCRIPTS

b	base
l	local
s	static conditions
t	total conditions
$\infty$	free stream

# NOMENCLATURE (Continued)

## Body-Axis System

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
$C_N$	CN	normal-force coefficient; $\frac{\text{normal force}}{qS}$
$C_A$	CA	axial-force coefficient; $\frac{\text{axial force}}{qS}$
$C_Y$	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
$C_{A_b}$	CAB	base-force coefficient; $\frac{\text{base force}}{qS}$ $-A_b(P_b - P_\infty)/qS$
$C_{A_f}$	CAF	forebody axial force coefficient, $C_A - C_{A_b}$
$C_m$	CLM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS l_{REF}}$
$C_n$	CYN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qS b}$
$C_l$	CLB	rolling-moment coefficient; $\frac{\text{rolling moment}}{qS b}$

## Stability-Axis System

$C_L$	CL	lift coefficient; $\frac{\text{lift}}{qS}$
$C_D$	CD	drag coefficient; $\frac{\text{drag}}{qS}$
$C_{D_b}$	CDB	base-drag coefficient; $\frac{\text{base drag}}{qS}$
$C_{D_f}$	CDF	forebody drag coefficient; $C_D - C_{D_b}$
$C_Y$	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
$C_m$	CLM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS l_{REF}}$
$C_n$	CLN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qS b}$
$C_l$	CLB	rolling-moment coefficient; $\frac{\text{rolling moment}}{qS b}$
$L/D$	L/D	lift-to-drag ratio; $C_L/C_D$

NOMENCLATURE (Continued)  
Additions To Standard List

<u>Symbol</u>	<u>SADSAC Symbol</u>	<u>Definition</u>
$A( )$		model base area, subscript is base orifice number and identifies location
$C_{A_b}$	CAB	model base axial-force coefficient
$C_{p( )}$		model static pressure coefficient, subscript is orifice number, $[p( ) - p_\infty]/q$
$C_{AU}$	CA	axial-force coefficient, unadjusted
$C_{AF}$	CAF	forebody axial-force coefficient, $C_{AU}$ adjusted for base terms
ET		external tank
IV		integrated vehicle, consists of orbiter, external tank, and two solid rocket motors
$L_{REF}$	LREF	reference length, inches
MRC		moment reference center
OMS		orbital maneuvering system
$\delta_e$	ELEVON	elevon, surface deflection angle, positive deflection trailing edge down, degrees
$\delta_f$	BDFLAP	orbiter body flap deflection angle, positive deflection angle is trailing edge down, degrees
$\delta_R$	RUDDER	rudder, surface deflection angle, positive deflection trailing edge to the left, degrees
$\delta_{SB}$	SPDBRK	speed brake deflection angle, left split rudder trailing edge left and right split rudder trailing edge right, $\delta_{SB} = (\delta_{RL} + \delta_{RR})/2$ , positive deflection, degrees
$i_0$	ORBINC	incidence angle between the orbiter and external tank, $i_0 = \alpha_0 - \alpha_T$ , degrees



# NOMENCLATURE (Continued)

$\beta_T$	BETAT	angle of sideslip of external tank, degrees
$\alpha_T$	ALPHAT	angle of attack of external tank, degrees
$l_B$	LB	length of orbiter body, in
$l_T$	LT	length of external tank, in
$l_s$	LS	length of SRM booster, in
$l_{NM}$	LNM	length of OMS nozzle, positive direction forward of exit plane, in
$l_{NP}$	LNP	length of MPS nozzle, positive direction forward of exit plane, in
$b/2$	BW	wing semi-span, in
$b_v$	BV	vertical tail span, in
$x$	X	distance from component nose, in
$y$	Y	lateral distance from centerline, in
$z$	Z	vertical distance measured from W.L. 500 (vertical tail reference root chord), in
$c_w$	CW	local wing chord, in
$c_v$	CV	local vertical tail chord, in
$x/l_B$	X/LB	longitudinal position/orbiter body length
$x/l_T$	X/LT	longitudinal position/external tank length
$x/l_s$	X/LS	longitudinal position/booster length
$x/l_{NM}$	X/LNM	longitudinal position/OMS nozzle length

# NOMENCLATURE (Concluded)

$x/l_{NP}$	X/LNP	longitudinal position/MPS nozzle length
$x/c_w$	X/CW	local chordwise position/local wing chord length
$x/c_v$	X/CV	local chordwise position/local vertical tail chord length.
$\eta$	Y/BW	local spanwise position/wing semi-span
$\eta_v$	Z/BV	local spanwise position/vertical tail span
$x_{CP}/l$	XCP/L	center of pressure distance from MRC, expressed as a fraction of body length
$\beta_0$	BETA0	angle of sideslip of orbiter
$\alpha_0$	ALPHA0	angle of attack of orbiter

## CONFIGURATIONS INVESTIGATED

The 0.030-scale Aero Loads Model, 47-OTS, was configured after the Shuttle Vehicle MCR 0200 Baseline R1, as defined in drawing number VL70-000088B. The orbiter configuration was a combination of the VL70-000140A orbiter and a VL70-000140B wing and midbody, from which the 140A/B designation was derived. The basic launch configuration consisted of the orbiter, an external tank with simulated fuel and vent lines, and two solid rocket boosters, designated  $O_1$   $T_{12}$   $S_{12}$   $N_{25}$ .

Three launch configurations were tested. One was the basic configuration described above mounted on a dual balance and sting arrangement, illustrated in figure 2d. A second contained attach hardware, designated  $AT_{10}$ , mating the orbiter with the external tank and mounted on a single sting and balance in the orbiter, illustrated in figure 2b. The third utilized a similar attach hardware configuration, designated  $AT_{11}$ , which was attached to the orbiter but not to the external tank and was mounted on the same dual sting and balance arrangement as the basic configuration (figure 2c). In all three configurations, the SRB-to-ET attach hardware was simulated at the forward attach location but not at the aft attach location. Model and component general arrangements are shown in figures 2e through 2o.

Component	Description
$O_1$	140A/B orbiter minus the main propulsion system nozzles
$T_{12}$	324-inch diameter external tank with ogive nose and external fuel and vent lines
$S_{12}$	142.3-inch diameter solid rocket boosters

N <sub>25</sub>	Nozzles for S <sub>12</sub> boosters
AT <sub>10</sub>	Orbiter-to-ET attach hardware, fixed to both vehicles
AT <sub>11</sub>	Orbiter-to-ET attach hardware, fixed to orbiter only
LV	O <sub>1</sub> T <sub>12</sub> S <sub>12</sub> N <sub>25</sub>
LVA	O <sub>1</sub> T <sub>12</sub> S <sub>12</sub> N <sub>25</sub> AT <sub>10</sub>
LVAP	O <sub>1</sub> T <sub>12</sub> S <sub>12</sub> N <sub>25</sub> AT <sub>11</sub>

The orbiter O<sub>1</sub>, consisted of the following components:

B<sub>26</sub> C<sub>9</sub> F<sub>8</sub> M<sub>7</sub> N<sub>28</sub> V<sub>8</sub> R<sub>5</sub> W<sub>116</sub> E<sub>26</sub>.

B <sub>26</sub>	Double delta wing fuselage, 140A/B
C <sub>9</sub>	Canopy, 140A
F <sub>8</sub>	Body flap, 140A
M <sub>7</sub>	OMS pods, 140A
N <sub>28</sub>	OMS nozzles, 140A
V <sub>8</sub>	Vertical tail, 140A
R <sub>5</sub>	Rudder, 140A
W <sub>116</sub>	Double delta wing, 140B
E <sub>26</sub>	Elevons, 140B

Parametric investigations were limited to angles of attack and sideslip with all orbiter control surfaces at 0° deflection.

## INSTRUMENTATION DESCRIPTION

The left side of the orbiter and the external tank and the left hand SRB were extensively instrumented with pressure orifices for measurement of surface static pressure distributions. Additionally, there were clusters of orifices around inter-component attach structure locations on the right hand side of the orbiter and external tank. The orbiter contained 471 operational orifices, of which 83 were clustered around attach structure. The external tank contained 270 operational orifices, of which 127 were clustered around attach structure. The SRB contained 124 operational orifices. A three-tube total pressure rake was installed in the opening between the orbiter and external tank. Tables and sketches defining orifice locations are included in this report. All model pressures were measured by model mounted Scanivalve, Inc., S-type scanivalve modules - twelve in the orbiter, seven in the external tank, and five in the SRB.

Force instrumentation consisted of a six-component internal force balance in both the orbiter and external tank for the LV and LVAP configurations, and a single six-component internal force balance in the orbiter for the attached LVA configuration.

### TEST FACILITY DESCRIPTION

The tests were conducted in the Ames 11- by 11-Foot Transonic Wind Tunnel which is a variable density, closed return, continuous flow type. This tunnel has an adjustable nozzle (two flexible walls) and a slotted test section to permit transonic testing over a Mach number range continuously variable from 0.4 to 1.4.

## DATA REDUCTION

Data were reduced to coefficient form about body axes using the following reference constants:

$S_{REF} = 2.421 \text{ ft}^2$	reference area for force and moment coefficients
$L_{REF} = 38.709 \text{ in}$	reference length for moment coefficients
$A_1 = 0.07670 \text{ ft}^2$	Orbiter sting cavity
$A_2 = 0.21340 \text{ ft}^2$	Orbiter heat shield base
$A_3 = 0.08560 \text{ ft}^2$	Orbiter OMS base (2)
$A_4 = (\text{see table below})$	Orbiter speed brake base
$A_{501} = 0.07266 \text{ ft}^2$	Tank sting cavity
$A_{502} = 0.44264 \text{ ft}^2$	Tank base
$A_{801} = 0.19600 \text{ ft}^2$	SRM nozzle base (2)
$A_{802} = 0.16590 \text{ ft}^2$	SRM skirt base (2)
$\delta_{SB} = 0 \text{ deg}$	$A_4 = 0 \text{ ft}^2$
14.92	0.02327
24.92	0.03866
34.92	0.05370
54.92	0.08252
84.92	0.12083
$X_{MRP} = 0 \text{ in}$	
$Y_{MRP} = 0 \text{ in}$	
$Z_{MRP} = 9.99 \text{ in}$	

The incidence angle between the orbiter and the external tank is equal to zero for angle of attack and angle of sideslip. Therefore, the angle of attack, ALPHA, used in the force plots is equal to ALPHA0. Also the angle of sideslip, BETA, used in the force plots is equal to BETA0.

The force and moment data recorded by the orbiter balance for configuration LV and LVAP are identified as RB10XX datasets. Force and moment data recorded by the tank balance for configuration LV and LVAP and by the orbiter balance for LVA (composite) are identified by RB11XX.

The pressure data were recorded for each component. The fourth character in each dataset identifier (i.e. RB1BXX, B for fuselage) represents the individual component. The following list indicates the symbol for each component.

SYMBOL	COMPONENT
B	Orbiter fuselage
C	Orbiter base
E	OMS nozzle
F	Body flap
M	OMS pod outside
L	Lower wing surface
U	Upper wing surface
R	Right vertical tail surface
V	Left vertical tail surface
S	SRM booster
T	External tank
X	SRM nozzle



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**SYMBOL**

**COMPONENT**

Y	External tank base & SRM booster base
1	Orbiter attach points
2	External tank attach points
3	External tank base rake

## REFERENCES

1. Orbiter - Lines and Configuration Control Drawings
2. VL70-000140A, Orbiter Configuration Control Drawing MCR 0200 Baseline
3. VL70-000143A, Lines Control, Vehicle 4 Forward Body - Cabin - Canopy MCR 0200 Baseline
4. VL70-000200, Lines Control, Midbody - Wing - Boot Fairing MCR 200 R3 dated 7-2-73
5. VL70-000145, Lines Control - Aft Body - OMS/RCS Pods. MCR 0200 - R1 baseline
6. VL70-000146A, Lines Control (Vehicle 4) Vertical Tail MCR 0200 Baseline
7. External Oxygen Hydrogen Tank (EOHT) - Lines and Configuration Control Drawings
8. VL78-000041B, External Tank - Configuration Control MCR 0200 Baseline R2
9. VL78-000024A, Structural Assy - External Tank MCR 0200 R2
10. VL78-000031A, Thermal Protection - External Tank, MCR 0200 Baseline
11. Solid Rocket Boosters (SRB) - Lines and Configuration Control Drawings
12. VL77-000036A, SRB Configuration Control MCR 0200 R1
13. VL77-000041, SRB Booster Assy, MCR 0200 R1
14. Integrated Vehicle - Lines and Configuration Control Drawings
15. VL72-000088A, Shuttle Configuration MCR 0200 Baseline R1
16. VL72-000089, SRB-ET-Orbiter Interface Disconnects MCR 0200 Baseline
17. VL72-000075, External Tank to SRB Attach Interface MCR 0074 Baseline
18. Aero Loads Model 47-OTS - Model Fabrication, Assembly and Installation Drawings

19. SS-A00119, Orbiter Assy - .030 Scale Pressure/Loads Model (140A/B Lines)
20. SS-A00120, Assy & Details - EOHT - .030 Scale Pressure/Loads Model (140A Lines)
21. SS-A00121, Orbiter/EOHT Attachments .030 Scale Pressure/Loads Model (140A Lines)
22. SS-A00122, Assy & Details - SRM - .030 Scale Pressure/Loads Model (140A Lines)
23. SS-A00123, Assy & Details - Forebody - .030 Scale Pressure/Loads Model (140A Lines)
24. SS-A00124, Assy & Details - Aft Fuselage - .030 Scale Pressure/Loads Model (140A Lines)
25. SS-A00125, Assy & Details - Wing Splice Plate & Cuff - .030 Scale Pressure/Loads Model (140A Lines)
26. SS-A00126, Assy & Details - Vertical Stabilizer - .030 Scale Pressure/Loads Model (140A Lines)
27. SS-A00127, Ames 11-ft x 11-ft Wind Tunnel Installation - .030 Scale Pressure/Loads Model (140A/B Lines)
28. SS-A00128, Ames 9-ft x 7-ft Wind Tunnel Installation - .030 Scale Pressure/Loads Model (140A/B Lines)
29. SS-A00130, Lines Control - Profile VL70-000140A - .030 Scale Pressure/Loads Model (140A/B Lines)
30. W-1104S Sting - Ames MK II 4" Balance (Male End), Ames MK XX 2.5" Balance
31. W-1105S, Sting - Ames MK II 4" Balance (Male End), RI MK I 2.75 Balance
32. W-1106A, Adapter - Ames MK II, 4" Balance (Male & Female)
33. W-1107A, 13.5° Bent Sting Adapter Ames MK II 4" Balance (Male & Female)

34. (DMS-DR-2129), "Airloads Investigation of an 0.030-Scale Model of the Space Shuttle Vehicle 140A/B Launch Configuration (Model 47-OTS) in the ARC 9- by 7-foot Unitary Plan Wind Tunnel for Mach Range 1.55 and 2.2 (IA14B)"
35. (DMS-DR-2130), "Airloads Investigation of an 0.030-Scale Model of the Space Shuttle Vehicle 140A/B Orbiter Configuration (Model 47-0) in the ARC 11-foot Unitary Plan Wind Tunnel for Mach Range 0.6 and 0.9 (OA22A)"
36. (DMS-DR-2131), "Airloads Investigation of an 0.030-Scale Model of the Space Shuttle Vehicle 140A/B Orbiter Configuration (Model 47-0) in the ARC 9- by 7-foot Unitary Plan Wind Tunnel for Mach Range 1.55 and 2.2 (OA22B)"

TABLE I.

TEST : IA-14A		DATE : 9-13-73	
TEST CONDITIONS			
MACH NUMBER	REYNOLDS NUMBER (per unit length)	DYNAMIC PRESSURE (pounds/sq. ft.)	STAGNATION TEMPERATURE (degrees Fahrenheit)
0.60	$4.0 \times 10^6$	480	120
0.75	$3.75 \times 10^6$	540	120
0.85	$3.5 \times 10^6$	550	120
0.90	$3.5 \times 10^6$	580	120
0.95	$3.25 \times 10^6$	610	120
0.975	$3.0 \times 10^6$	530	120
1.00	$3.0 \times 10^6$	535	120
1.025	$3.0 \times 10^6$	540	120
1.05	$3.0 \times 10^6$	545	120
1.10	$3.0 \times 10^6$	550	120
1.15	$3.0 \times 10^6$	575	120
1.25	$2.75 \times 10^6$	540	120
1.40	$2.75 \times 10^6$	570	120

BALANCE UTILIZED:      LVA: 2.5-in MK XX (ORBITER)  
                                  LVAP: 2.5-in MK XX (ORB.), 2.75-in MK I (ET)

	CAPACITY:		ACCURACY:		COEFFICIENT TOLERANCE:
	MK XX	MK I	MK XX	MK I	
NF	6000	7500			_____
SF	3000	3750	0.2%	0.2%	_____
AF	600	700	0.2%	0.2%	_____
PM					_____
RM	4000	4000	0.2%	0.2%	_____
YM					_____

COMMENTS: Test conditions for LVA and LVAP model configurations

TABLE I. - Concluded.

TEST : IA-14A	DATE : 9-13-73		
TEST CONDITIONS			
MACH NUMBER	REYNOLDS NUMBER (per unit length)	DYNAMIC PRESSURE (pounds/sq. ft)	STAGNATION TEMPERATURE (degrees Fahrenheit)
0.60	$4.0 \times 10^6$	480	120
0.75	$4.25 \times 10^6$	610	120
0.85	$4.5 \times 10^6$	710	120
0.90	$4.5 \times 10^6$	750	120
0.95	$4.5 \times 10^6$	780	120
0.975	$4.25 \times 10^6$	750	120
1.05	$4.25 \times 10^6$	790	120
1.10	$4.0 \times 10^6$	760	120
1.15	$3.75 \times 10^6$	720	120
1.25	$2.75 \times 10^6$	735	120
1.40	$3.0 \times 10^6$	620	120

BALANCE UTILIZED: 2.5-in MK XX (ORB.), 2.75-in MK I (ET)

	CAPACITY:	ACCURACY:	COEFFICIENT TOLERANCE:
	MK XX MK I	MK XX MK I	
NF	6000 7500	0.2% 0.2%	_____
SF	3000 3750	0.2% 0.2%	_____
AF	600 700	0.2% 0.2%	_____
PM			_____
RM	4000 4000	0.2% 0.2%	_____
YM			_____

COMMENTS: Test conditions for LV model config.

TABLE II

TEST : IA-4A

DATE : 13 Sept 1973

DATA SET/RUN NUMBER COLLATION SUMMARY

DATA SET IDENTIFIER	CONFIGURATION	SCHD.			CONTROL DEFLECTION			NO. OF RUNS	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE)		
		a	B	C	D	E	F		G	H	I
RBI x 17	Q1+T12+S12+N25+ATTI	B	O	O	O	O	O		0.9	0.92	1.1
T 18	+ATTI	O	B	T	T	T	T		40	93	89
24	Q1+T12+S12+N25+ATTI	-10	Z						41	94	90
25		-8							8*		
26		-6	↓						9*		
27		-4	-10						10*		
28		O	Y						11*		
29		-10	O								
Y 30	Q1+T12+S12+N25+ATTI	O	O	Y	Y	Y	Y		38		
									39*		
											P/C2

TEST RUN NUMBERS

1 7 13 19 25 31 37 43 49 55 61 67 75 76

CM RM CA CAF CY CYN CBL IDVAR (1) IDVAR (2) NDV  
 $\alpha(B) = \beta(B) = -8, -4, 0, 8$   
 $\beta(Y) = -10, -8, -6, -4, -2, 0$   
SCH DOLES MACH(D) = 0.975, 1.0, 1.02

\* FORCE DATA NOT AVAILABLE.

NASA-MSFC-MAF

TABLE II - Continued

TEST: I A 14 A										DATA SET/RUN NUMBER COLLATION SUMMARY										DATE: 13 Sept 1975														
DATA SET IDENTIFIER		CONFIGURATION	SCAD. PARAMETERS/VALUES					NO. OF RUNS		IL. V/H (OR ALTERNATE INDEPENDENT VARIABLE)										TEST RUN NUMBERS														
			B	M	Se	SR	SSR	0		-10	-8	-6	-4	0	2	4	6	8	10															
RBI 31		$\phi_1 + T_{12} + S_{12} N_{25} + A_{10}$	A	.9	C	C	0			1*	2*	3*	4*	5	6	7	12	13	14	15														
32			T	1.1						16*	17*	18*	19*	20	21	22	23	24	25	26														
33			V	1.25						27*	28*	29*	30*	31	32	33	34	35	36	37														
34		$+A_{11}$	B	.6							47		43	49			50		51															
35			T	.75							56		55	54			53		52															
36				.85							42		43	44			45		46															
37				.95							57		58	59			60		61															
38				1.05							63		64	65			66		67															
39				1.15							68		69	70			71		72															
40				1.4							84		85	86			87		88															
41				.6							149		148	147			146		145															
42				.75							140		141	142			143		144															
43				.85							135		136	137			138		139															
44				.96							130		131	132			133		134															
45				.75							125		126	127			128		129															
46				.575							120		121	122			123		124															
47				1.35							115		116	117			118		119															
V 48		V		1.10							110		111	112			113		114															
1		7	13	19	25	31	37	43	49	55	61	67	75	76																				



**TABLE II - Concluded**

[illegible]

TABLE III. - MODEL DIMENSIONAL DATA

MODEL COMPONENT: ATTACH STRUCTURE - AT<sub>10</sub>GENERAL DESCRIPTION: Attach structure for Integrated Vehicle Configuration4 per VL72-000088B and VL72-000089, modified as follows: RemovedET-to-SRM aft attach struts (4) and left orbiter to right ET aftattach crossover rod.MODEL SCALE: 0.030

DRAWING NO.: SEE DESCRIPTION

DIMENSIONS:	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
<b>FORWARD ATTACH</b>		
Orbiter to Tank		
Number of struts	<u>2</u>	<u>2</u>
Diameter - In.	<u>6.000</u>	<u>0.180</u>
Location - In.		
$X_O$	<u>382.000</u>	<u>11.460</u>
$X_T$	<u>1078.000</u>	<u>32.340</u>
<b>DRAG LINK ATTACH</b>		
Orbiter to Tank		
Number of struts	<u>2</u>	<u>2</u>
Diameter, In.	<u>15.000</u>	<u>0.450</u>
Location, In.		
$X_O$	<u>1307.000</u>	<u>39.210</u>
$X_T$	<u>1859.000</u>	<u>55.770</u>
<b>AFT ATTACH</b>		
Orbiter to Tank		
Number of struts	<u>2</u>	<u>2</u>
Diameter - In.	<u>12.000</u>	<u>0.360</u>
Location - In.		
$X_O$	<u>1307.000</u>	<u>39.210</u>
$X_T$	<u>2058.000</u>	<u>61.740</u>
<b>CROSSOVER ROD (RIGHT ORBITER TO LEFT ET)</b>		
Diameter, In.	<u>8.000</u>	<u>0.240</u>
Location - In.		
$X_O$	<u>1307.000</u>	<u>39.210</u>
$X_T$	<u>2058.000</u>	<u>61.740</u>

TABLE III. - Continued.

MODEL COMPONENT: ATTACH STRUCTURE - AT<sub>11</sub>

GENERAL DESCRIPTION: Attach structure, same as AT<sub>10</sub> except the forward attach struts are rotated to the vertical, and the structure extends from the orbiter but is not attached to the tank.

MODEL SCALE: 0.030

DIMENSIONS:	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
FORWARD ATTACH		
Orbiter to Tank		
Location - In.		
$X_O$	<u>382.000</u>	<u>11.460</u>
$X_T$	<u>1133.000</u>	<u>33.990</u>
Clearance, tank to strut - In.	<u>16.667</u>	<u>0.500</u>
DRAG LINK ATTACH		
Orbiter to Tank		
Clearance, tank to strut - In.	<u>8.333</u>	<u>0.250</u>
AFT ATTACH		
Orbiter to Tank		
Clearance, Tank to strut - In.	<u>8.333</u>	<u>0.250</u>
Crossover Rod		
Clearance, tank to strut - In.	<u>8.333</u>	<u>0.250</u>

TABLE III. - Continued.

MODEL COMPONENT: BODY - B<sub>26</sub>GENERAL DESCRIPTION: Orbiter Fuselage Configuration 140 A/BNOTE: B<sub>26</sub> identical to B<sub>24</sub> except underside of fuselage refaired to accept W<sub>116</sub>.Model Scale = .030DRAWING NUMBER: VL70-000193  
VL70-000140A

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length (Body Fwd Sta $X_0 = 238$ ) - in.	<u>1293.3</u>	<u>38.799</u>
Max. Width (at $X_0 = 1520$ ) - in.	<u>262.0</u>	<u>7.860</u>
Max. Depth (at $X_0 = 1464$ ) - in.	<u>250.0</u>	<u>7.500</u>
Fineness Ratio	<u>0.26357</u>	<u>0.26357</u>
Area - ft <sup>2</sup>		
Max. Cross-Sectional	<u>340.88462</u>	<u>0.30679</u>
Planform	<u>                    </u>	<u>                    </u>
Wetted	<u>                    </u>	<u>                    </u>
Base	<u>                    </u>	<u>                    </u>

TABLE III. - Continued.

MODEL COMPONENT: CANOPY - Cg

GENERAL DESCRIPTION: Configuration 3A

Model Scale = .030

DRAWING NUMBER

VL70-000140A

VL70-000143A

DIMENSION:

FULL SCALE

MODEL SCALE

Length ( $X_0=434.643$  to  $670$ )

235.357

7.06071

Max Width ( $\phi X_0=513.127$ )

152.412

4.57236

Max Depth ( $\phi X_0=485.0$ )

25.000

0.75000

Fineness Ratio

Area

Max Cross-Sectional

Planform

Wetted

Base

ORIGINAL PARTS  
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TABLE III. - Continued.

MODEL COMPONENT: ELEVON - E<sub>26</sub>GENERAL DESCRIPTION: Configuration 4NOTE: VL70-000400 data for (1) of (2) sides. Identical to E<sub>25</sub> except  
airfoil thicknessModel Scale = .030DRAWING NUMBER:VL70-000200  
VL70-000140 BDIMENSIONS:FULL-SCALEMODEL SCALE

Area	<u>223.5814</u>	<u>0.20122</u>
Span (equivalent)	<u>368.34</u>	<u>11.05020</u>
Inb'd equivalent chord	<u>119.623</u>	<u>3.58869</u>
Outb'd equivalent chord	<u>55.1922</u>	<u>1.65577</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>0.2096</u>	<u>0.2096</u>
At Outb'd equiv. chord	<u>0.4004</u>	<u>0.4004</u>
Sweep Back Angles, degrees		
Leading Edge	<u>0.00</u>	<u>0.00</u>
Tailing Edge	<u>-10.056</u>	<u>-10.056</u>
Hingeline	<u>0.00</u>	<u>0.00</u>
Area Moment (Normal to hinge line)	<u>851.1502</u>	<u>0.76604</u>

TABLE III. - Continued.

MODEL COMPONENT: Body Flap - F<sub>8</sub>

GENERAL DESCRIPTION: Configuration 4

Model Scale - .030  
DRAWING NUMBER VL70-000140B, VL70-000200

<u>DIMENSION:</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Length in.	<u>84.7</u>	<u>2.541</u>
Max Width in.	<u>262.308</u>	<u>7.86924</u>
Max Depth in.	<u>23.000</u>	<u>0.69000</u>
Fineness Ratio		
Area - ft <sup>2</sup>		
Max Cross-Sectional		
Planform	<u>158.85350</u>	<u>0.14297</u>
Wetted		
Base	<u>41.89642</u>	<u>0.03771</u>

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TABLE III. - Continued.

MODEL COMPONENT: OMS POD - M7

GENERAL DESCRIPTION: Configuration 3A

Model Scale = .030

DRAWING NUMBER VL70-000140A  
VL70-000145

DIMENSION:

FULL SCALE

MODEL SCALE

Length (OMS Fwd Sta  $X_0=1233.0$ ) - IN.

327.000

9.810

Max Width (@  $X_0=1450.0$ ) - IN.

94.5

2.8350

Max Depth (@  $X_0=1493.0$ ) - IN.

109.000

3.270

Fineness Ratio

Area

Max Cross-Sectional

Planform

Wetted

Base



TABLE III. - Continued.

MODEL COMPONENT: BSRM NOZZLES - N25GENERAL DESCRIPTION: Configuration 3A BSRM Nozzles

Model Scale = .030

DRAWING NO. VL72-000038A  
VL77-000036A

DIMENSIONS	FULL-SCALE	MODEL SCALE
MACH NO. _____		
DIAMETER DEX ~ IN (One Nozzle)	<u>141.3</u>	<u>4.2390</u>
DIAMETER DT ~ IN	<u>          </u>	<u>          </u>
DIAMETER DIN ~ IN	<u>          </u>	<u>          </u>
ON ~ DEGREES	<u>          </u>	<u>          </u>
AREA - FT <sup>2</sup> (One Nozzle)		
MAX CROSS-SECTIONAL	<u>108.89595</u>	<u>0.09801</u>
GIMBAL ORIGIN	<u>X<sub>o</sub></u>	<u>Y<sub>o</sub></u> <u>Z<sub>o</sub></u>
LEFT NOZZLE ~ IN. F.S.	<u>1825.3</u>	<u>-243</u> <u>400</u>
RIGHT NOZZLE ~ IN. FS	<u>1825.3</u>	<u>+243</u> <u>400</u>
NULL POSITION - DEG.	<u>PITCH</u>	<u>YAW</u>
LEFT NOZZLE	<u>+8</u>	<u>+8</u>
RIGHT NOZZLE	<u>+8</u>	<u>+8</u>

TABLE III. - Continued.

MODEL COMPONENT: NOZZLES - N28GENERAL DESCRIPTION: Configuration 3A OMS NozzleModel Scale = .030DRAWING NO. VL70-000140A

## DIMENSIONS

## FULL-SCALE

## MODEL SCALE

MACH NO. \_\_\_\_\_

DIAMETER DEX ~ IN (One nozzle)

DIAMETER DT ~ IN

DIAMETER DIN ~ IN

ON ~ DEGREES

AREA - Ft<sup>2</sup> (one nozzle)

MAX CROSS-SECTIONAL

## GIMBAL ORIGIN

X<sub>o</sub>Y<sub>o</sub>Z<sub>o</sub>

LEFT NOZZLE ~ IN.

1518.0-88.0492.0

RIGHT NOZZLE ~ IN.

1518.0+88.0492.0

## NULL POSITION

PITCHYAW

LEFT NOZZLE (Null Pitch 15°49'; Yaw 12°17' OUTB'D)

±8°13°17' OUTB'D2°30' INB'D

RIGHT NOZZLE (Null Pitch 15°49'; Yaw 12°17' OUTB'D)

±8°13°17' OUTB'D2°17' INB'D

TABLE III. - Continued.

MODEL COMPONENT: RUDDER - R5GENERAL DESCRIPTION: 2A, 3 and 3A Configuration per Rockwell LinesVL70-000095Model Scale = .030DRAWING NUMBER: VL70-000095DIMENSIONS:FULL-SCALEMODEL SCALEArea - FT<sup>2</sup>106.380.33574

Span (equivalent) - IN.

201.06.0300

Inb'd equivalent chord

91.5852.74755

Outb'd equivalent chord

50.8331.52499Ratio movable surface chord/  
total surface chord

At Inb'd equiv. chord

0.4000.400

At Outb'd equiv. chord

0.4000.400

Sweep Back Angles, degrees

Leading Edge

34.8334.83

Tailing Edge

26.2526.25

Hingeline

34.8334.83Area Moment (Normal to hinge line)- FT<sup>3</sup>526.130.01420

Product of Area and Mean Chord

TABLE III. - Continued.

MODEL COMPONENT: BOOSTER SOLID ROCKET MOTOR - S<sub>12</sub>GENERAL DESCRIPTION: Configuration 3A, Data for (1) of (2) sides,per Rockwell Lines VL77-000036AModel Scale = .030

DRAWING NUMBER

VL72-000088AVL77-000036ADIMENSION:FULL SCALEMODEL SCALE

Length (Includes Nozzle) - IN.	<u>1741.0</u>	<u>52.2300</u>
Max Width (Tank Dia) - IN.	<u>142.3</u>	<u>4.2690</u>
Max Depth (Aft Shroud) - IN.	<u>192.0</u>	<u>5.7600</u>
Fineness Ratio	<u>9.06771</u>	<u>9.06771</u>
Area - FT <sup>2</sup>		
Max Cross-Sectional	<u>201.06193</u>	<u>0.18096</u>
Planform	<u>                    </u>	<u>                    </u>
Wetted	<u>                    </u>	<u>                    </u>
Base	<u>                    </u>	<u>                    </u>
WP of BSRM Centerline (Z <sub>T</sub> ) - IN.	<u>400</u>	<u>12.000</u>
PS of BSRM Nose (X <sub>T</sub> ) - IN.	<u>200</u>	<u>6.000</u>

TABLE III. - Continued.

MODEL COMPONENT: EXTERNAL TANK - T12GENERAL DESCRIPTION: External Oxygen Hydrogen TankNOTE: Identical to T11 with external fuel lines addedModel Scale = 030

DRAWING NUMBER

VL78-000031AVL78-000041ADIMENSION:FULL SCALEMODEL SCALELength - IN. (Nose @  $X_T = 309$ )186557.629

Max Width (Dia) - IN.

3249.72

Max Depth

Fineness Ratio

5.756175.75617Area - FT<sup>2</sup>

Max Cross-Sectional

572.55517.177

Planform

Wetted

Base

WP of Tank Centerline ( $X_T$ ) - IN.400.0

TABLE III. - Continued.

MODEL COMPONENT: VERTICAL - V<sub>8</sub>GENERAL DESCRIPTION: Configuration 3aNOTE: Similar to V5 with radius on TE upper corner and LE lower corner  
where vertical meets fuselage.Model Scale = .030

DRAWING NUMBER:

VL70-000140a  
VL70-000146ADIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATAArea (Theo) Ft<sup>2</sup>413.2530.37193

Planform

Span (Theo) In

315.7209.47160

Aspect Ratio

1.6751.675

Rate of Taper

0.5070.507

Taper Ratio

0.403990.40399

Sweep Back Angles, degrees

Leading Edge

45.0045.00

Trailing Edge

25.94725.947

0.25 Element Line

41.13041.1300

## Chords:

Root (Theo) WP

268.5008.05500

Tip (Theo) WP

108.4703.25410

MAC

199.807565.99423

Fus. Sta. of .25 MAC

1163.5043.9050

W. P. of .25 MAC

635.52219.06566

B. L. of .25 MAC

0.000.00

## Airfoil Section

Leading Wedge Angle Deg

10.0010.00

Trailing Wedge Angle Deg

14.92014.920

Leading Edge Radius (Min) - IN.

2.000.060

Void Area

13.170.01185

Blanketed Area

0.000.00ORIGINAL PARTIAL  
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TABLE III. - Concluded.

MODEL COMPONENT: WING-W<sub>116</sub>GENERAL DESCRIPTION: Configuration 4NOTE: Identical to W<sub>114</sub> except airfoil thickness. Dihedral angle is along trailing edge of wing.Model Scale = .030

TEST NO.

DWG. NO. VL70-000140B  
VL70-000200DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATAArea (Theo.)  $\text{Ft}^2$ 

Planform

Span (Theo) In.

Aspect Ratio

Rate of Taper

Taper Ratio

Dihedral Angle, degrees (at  $X_0=1506.623, Y_0=$ Incidence Angle, degrees 105,  $Z_0=282.75$ )

Aerodynamic Twist, degrees

Sweep Back Angles, degrees

Leading Edge

Trailing Edge

0.25 Element Line

Chords:

Root (Theo) B.P.O.O.

Tip, (Theo) B.P.

MAC

Fus. Sta. of .25 MAC

W.P. of .25 MAC

B.L. of .25 MAC

EXPOSED DATAArea (Theo)  $\text{Ft}^2$ 

Span, (Theo) In. BP108

Aspect Ratio

Taper Ratio

Chords

Root BP108

Tip 1.00  $\frac{b}{2}$ 

MAC

Fus. Sta. of .25 MAC

W.P. of .25 MAC

B.L. of .25 MAC

Airfoil Section (Rockwell Mod NASA)  
XXXX-64Root  $\frac{b}{2} = 0.425$ Tip  $\frac{b}{2} = 1.00$ 

Data for (1) of (2) Sides

Leading Edge Cuff

Planform Area  $\text{Ft}^2$ 

Leading Edge Intersects Fus M. L. @ Sta

Leading Edge Intersects Wing @ Sta

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TABLE IV. - ORBITER FUSELAGE PRESSURE ORIFICE LOCATIONS

ORBITER X <sub>0</sub> IN.			RADIAL LOCATION Ø DEGREES																		
FULL	MODEL	X <sub>0</sub> /10	0	20	40	55	70	90	105	110	120	135	140	150	151	156	162	165	169	174	180
235	7.05	0	6					8													9
245	7.35	.008	7					15			16			17							13
265	7.95	.023	10	11	12	13	14	15			25			26							27
295	8.85	.047	19	20	21	22	23	24			34			35							36
325	9.75	.070	28	29	30	31	32	33			43			44							45
380	11.40	.112	37	38	39	40	41	42												46	
440	13.20	.159	47	48	49	50	51	52			53				57	54	58		55		56
450	13.50	.167																			
465	13.95	.178																			
500	15.00	.205	59	60	61	62	63	64			65		66	67			68				69
560	16.80	.252	70		71		72	73			74			75			76				77
625	18.75	.301	78		79		80	81			82			83			84				85
725	21.75	.379	86		87		88	89			90			91			92				93
880	26.40	.499	94		95		96	97			98			99			100				101
980	29.40	.576	102		103		106	107			108			109			110				111
1080	32.40	.653	104		105		114	115			116			117							118
1180	35.40	.730	112		113		121	122	123		124	125		126			127				128
1245	37.35	.781	119		120		131	132	133		134	135		136							137
1300	39.00	.823	129		130		140	141	142		143	144		145			146				
1375	41.25	.882	138		139		149	150	151		152	153		154			155				
1430	42.90	.923	147		148		158	159	160		161	162		163			164				
1480	44.40	.963	156		157						165	166									
1530	45.90	1.002								167	168										
1530	45.90	1.002																			
1555	46.65	1.021	169		170																
1590	47.70	1.048	171		172																
1590	47.70	1.048	173		174																

X<sub>0</sub> = 1293.3 full scaleX<sub>0</sub> = 38.799 model

data in datasets RB18XX

a: OMS pod, inside c: Body flap lower surface

b: OMS pod, outside d: Body flap upper surface



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OF FOUR

TABLE V. - ORBITER WING PRESSURE ORIFICE LOCATIONS

ORBITER WING

ORBITER D.P.A.		X/C % LOCAL WING CHORD																							
CHORD	WING	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160	170	180	190	200	210	220	230
140	4.20	.299	U	L	E	179	181	191	202	207	214	216	218	220	222	224	226	228	230	232	234	236	238	240	242
170	5.10	.364	U	L	E	200	206	211	215	219	223	227	231	235	239	243	247	251	255	259	263	267	271	275	279
200	6.00	.427	U	L	E	211	217	221	225	229	233	237	241	245	249	253	257	261	265	269	273	277	281	285	289
250	7.50	.534	U	L	E	236	242	246	250	254	258	262	266	270	274	278	282	286	290	294	298	302	306	310	314
315	9.45	.675	U	L	E	261	267	271	275	279	283	287	291	295	299	303	307	311	315	319	323	327	331	335	339
365	10.95	.780	U	L	E	282	288	292	296	300	304	308	312	316	320	324	328	332	336	340	344	348	352	356	360
415	12.45	.887	U	L	E	299	305	311	317	323	329	335	341	347	353	359	365	371	377	383	389	395	401	407	413

- 1 X/C = .19
- 2 X/C = .34
- 3 X/C = .03
- 4 X/C = .045

data in data sets RB1LX (lower surface and RB1LXX (upper surface)

TABLE VI. - ORBITER VERTICAL TAIL PRESSURE ORIFICE LOCATIONS

ORBITER VERTICAL TAIL

VERTICAL $W_L \sim Z_0$		$X/C_V$										
FULL	MODEL	$\eta V$		0	.025	.05	.15	.30	.52	.685	.775	.90
550	16.50	.158	RH LE LH	316		324	325	326	327	328	329	
600	18.00	.316	RH LE LH	330	317	318	319	320	321	322	323	
690	20.70	.600	RH LE LH	346		339	340	341	342	343	344	345
765	22.95	.840	RH LE LH	362		355	356	357	358	359	360	361
792	23.76	.925	RH LE LH	378		348	349	350	351	352	353	354
					347							
						363						
							387					
					379							
							381					
								382				
									383			
										384		
											385	
												386

data in datasets R3LVXX (left side) and RBIRXX (right side)

TABLE VII. - ORBITER BASE, BODYFLAP, AND OMS NOZZLE PRESSURE ORIFICE LOCATIONS

ORBITER BASE

LOCATION	ORIFICE NUMBER
Orbiter Sting Cavity	1
Orbiter Base (Lower Left Corner)	2
OMS Nozzle Base	3

data in datasets RB1CXX

RUDDER FLARE BASE

RUDDER $\eta_L \sim Z_O$		X/C <sub>V</sub>
FULL	MODEL	.75
725	18.75	4
625	21.75	5

data in datasets RB1CXX

BODY FLAP

ORBITER $\sim X_O$		$\phi \sim \text{Deg}$	
FULL	MODEL	0	40
1555	46.65	169	170
1590	47.70	173	174
1590	47.70	171	172

data in datasets RB1FXX

LEFT OMS NOZZLE SURFACE

X $\sim$ IN. FWD. NOZZLE EXIT		$\phi \sim \text{DEG.}$		
FULL	MODEL	135	180	225
10	.30	175	176	177
20	.60		178	

data in datasets RB1EXX

TABLE VIII. - EXTERNAL TANK PRESSURE ORIFICE LOCATIONS

EXTERNAL TANK

TANK STATION $\sim X_T$			$\theta \sim \text{DEG.}$									
FULL SCALE	MODEL SCALE	$\frac{X_T}{L_T}$	0	30	60	90	120	135	150	165	180	270
309	9.27	0	503			505			513		506	507
324	9.72	.008	504			511	512		520		514	
400	12.00	.049	508	509	510	518	519		528	521	522	
520	15.60	.113	515	516	517	526	527		536	529	530	
640	19.20	.178	523	524	525	534	535		544	537	538	
670	20.10	.194	531	532	533	542	543		553	545	546	
710	21.30	.215	539	540	541	550	551	552	561	554	555	
760	22.80	.242	547	548	549	559	560	569	570	562	563	
850	25.50	.290	556	557	558		568		578	571	572	
950	28.50	.344	565	566	567		577		587	579	580	
1050	31.30	.394	573	574	575	576		586		588	589	
1150	34.50	.451	581	582	583	584	585		595	596	597	
1250	37.50	.505	590	591	592	593	594		604	605	606	
1350	40.50	.558	598	599	600	601	602	603	612	613	614	
1500	45.00	.638	607	608	609	610	611		621	622	623	
1700	51.00	.746	615	616	617	618	619	620	630	631	632	
1900	57.00	.853	624	625	626	627	628	629	638	639	640	
2040	61.20	.928	633	634	635		636	637				
TANK BASE			501								502	
STING CAVITY												

$L_T = 1865 \text{ IN. FULL SCALE}$   
 $55.950 \text{ IN. MODEL SCALE}$

data in datasets RB1TXX

TABLE IX. - SRM PRESSURE ORIFICE LOCATIONS

## LEFT SRM

SRM STATION ~ X <sub>s</sub>			Ø ~ DEG.								
FULL SCALE	MODEL SCALE	X <sub>s</sub> T <sub>s</sub>	0	45	90	135	180	225	270	315	
200	6.00	0	805	807	808	809	810	811	812	813	
260	7.80	.034	806	815	816	817	818	819	820	821	
370	11.10	.098	814	823	824	825	826	827	828	828	
400	12.00	.115	822	830	831	832	833	834	835	836	
450	13.50	.144	829	838	839	840	841	842	843	844	
550	16.50	.201	837				847	848	849	850	
700	21.00	.287	845				853		854		
850	25.50	.373	851				857				
1050	31.50	.488	855				860				
1250	37.50	.603	858				863		864		
1450	43.50	.718	861				867		868		
1650	49.50	.833	865				873	874	875	876	
1750	52.50	.890	869	870	871	872	873	882	883	884	
1796	53.88	.917	877	878	879	880	881	890	891	892	
1835	55.05	.939	885	886	887	888	889	898	899	900	
1868	56.04	.958	893	894	895	896	897	804			
SKIRT BASE			802			803					
NOZZLE BASE			801								
NOZZLE EXTERNAL PRESSURES											
1850	55.50	.948	901	902	903	904	905	906	907	908	
1905	57.15	.979	909	910	911	912	913	914	915	916	
1928	57.84	.993	917	918	919	920	921	922	923	924	

$L_s = 1741$  IN. FULL SCALE  
 52.53 IN. MODEL SCALE

data in datasets RBISXX

TABLE X. - ORBITER ATTACH POINT PRESSURE ORIFICE LOCATIONS

ORBITER ATTACH POINT ORIFICE LOCATIONS																	
FULL X <sub>O</sub> SCALE	347	357	367	377	387	397	407	1252	1262	1272	1282	1292	1302	1312	1322	1332	
X <sub>O</sub> MODEL	10.41	10.71	11.01	11.31	11.61	11.91	12.21	37.56	37.86	38.16	38.46	38.76	39.06	39.36	39.96	40.26	
X <sub>O</sub> /ℓ <sub>O</sub>	.087	.095	.102	.110	.118	.126	.133	.788	.796	.804	.811	.819	.827	.835	.850	.858	
F.S. MODEL	394	397					412				436	447		468	474	480	
Y <sub>O</sub>	Y <sub>O</sub>																
.021 10	.30	396	399	403	407	411	415				435	446	457	467	473	479	
.043 20	.60	395	398	402	406	410	414				434	445	456	466	472	478	
.064 30	.90			401	405	409	413				433	444	455	465	471	477	
.085 40	1.20										432	443	454	464	470	476	
.107 50	1.50														469	475	
.149 69.75	2.09										431	442	453	463			
.170 79.75	2.39									424	430	441	452	462			
.192 89.75	2.69								419	423	429	440	451	461			
.213 99.75	2.99							416	418	422	428						
.234 109.75	3.29								417	421	427	439	450	460			
.256 119.75	3.59									420	426	438	449	459			
.277 129.75	3.89										425	437	448	458			

data in datasets RB11XX

TABLE XI. - EXTERNAL TANK ATTACH POINT PRESSURE ORIFICE LOCATIONS

X <sub>T</sub> Full Scale	1103	1093	1083	1073	1063	1053	1043	
X <sub>T</sub> Model Scale	33.09	32.79	32.49	32.19	31.89	31.59	31.29	
X <sub>T</sub> /l <sub>T</sub>	.424	.419	.413	.408	.402	.397	.391	
								Ø DEG.
FWD ATTACH POINT (ORBITER TO E-T)	684	676	668	660				182.84
	685	677	669	661				186.38
	686	678	670	662	655			189.92
	687	679	671	663	656	652		193.46
	688	680			657	653	651	197.0
	689	681	673	665	658	654		200.54
	690	682	674	666	659			204.08
	691	683	675	667				207.62

data in datasets RB12XX

TABLE XI. - EXTERNAL TANK ATTACH POINT PRESSURE ORIFICE LOCATIONS  
(CONTINUED)

FWD DRAG LINK ATTACH POINT	$X_T$ FULL SCALE	1874	1864	1854	1844	1834	1824	1814	
	$X_T$ MODEL SCALE	56.22	55.92	55.62	55.32	55.02	54.72	54.42	
	$X_T/l_T$	.839	.834	.828	.823	.818	.812	.807	
									$\phi \sim$ DEG.
		719	713	707					222.84
		720	714	708	701				226.38
		721	715	709	702	696			229.92
		722		710	703	697	693		233.46
					704	698	694	692	237.00
						699	695		240.54
		723	718	712	706	700			244.08

data in datasets RB12XX



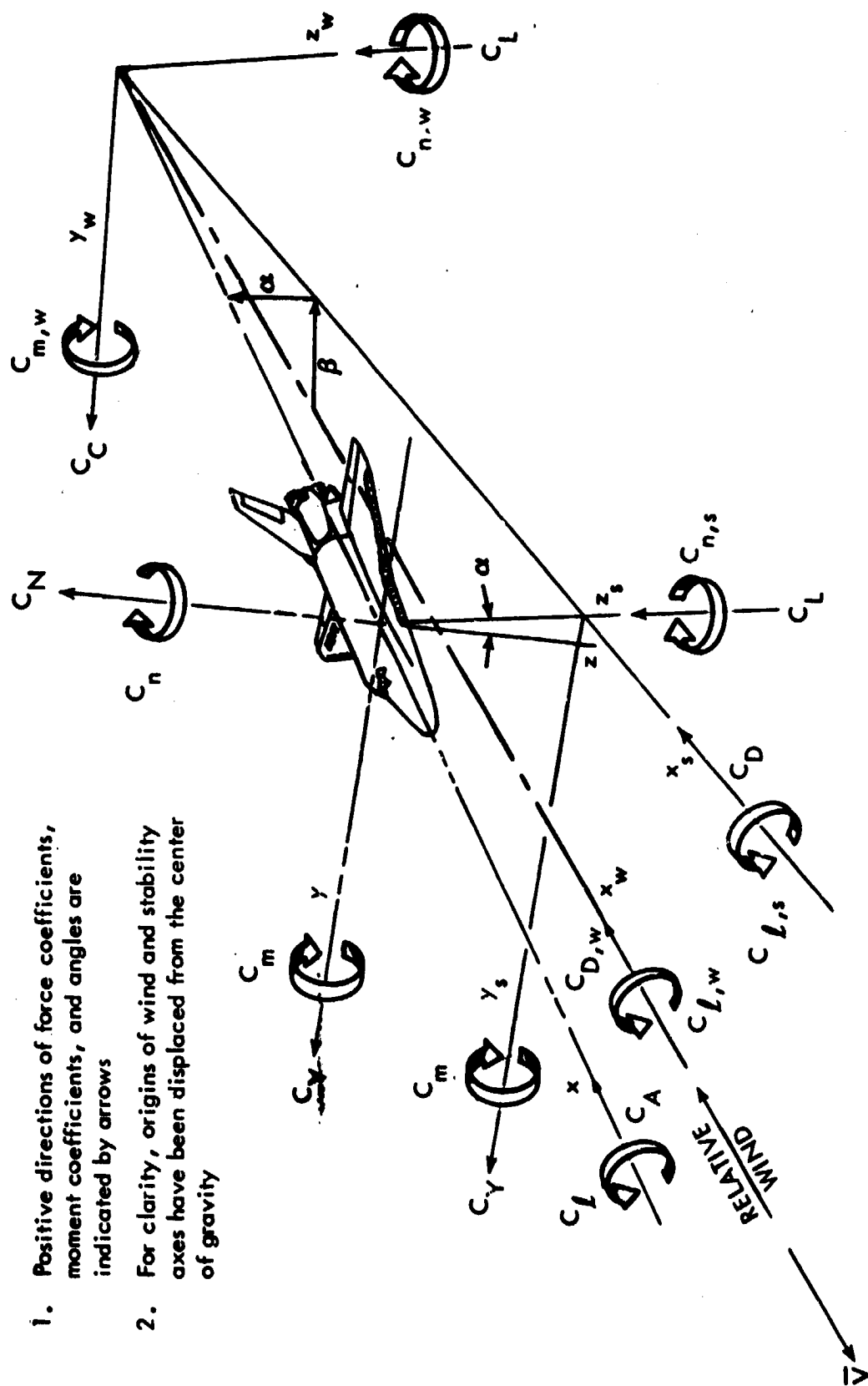
TABLE XI. - EXTERNAL TANK ATTACH POINT PRESSURE ORIFICE LOCATIONS  
(CONCLUDED)

$x_T$ FULL SCALE	2078	2068	2058	2048	2038	2028	2018	
$x_T$ MODEL SCALE	62.34	62.04	61.74	61.44	61.14	60.84	60.54	
$x_T/l_T$	.948	.943	.938	.932	.927	.921	.916	
								$\emptyset \sim \text{DEG.}$
AFT UPPER ATTACH	777	766	754					234.04
	778	767	755	742				237.58
	779	768	756	743	732			241.12
	780	769		744	733	726		244.66
	781	770		745	734	727	724	248.2
				746	735	728		251.74
		771	759	747	736			255.28
AFT LOWER ATTACH	782	772	760					323.51
	783	773	761	748				327.05
	784	774	762	749	737			330.59
	785	775		750	738	729		334.13
	786	776		751	739	730	725	337.67
				752	740	731		341.21
			765	753	741			344.75

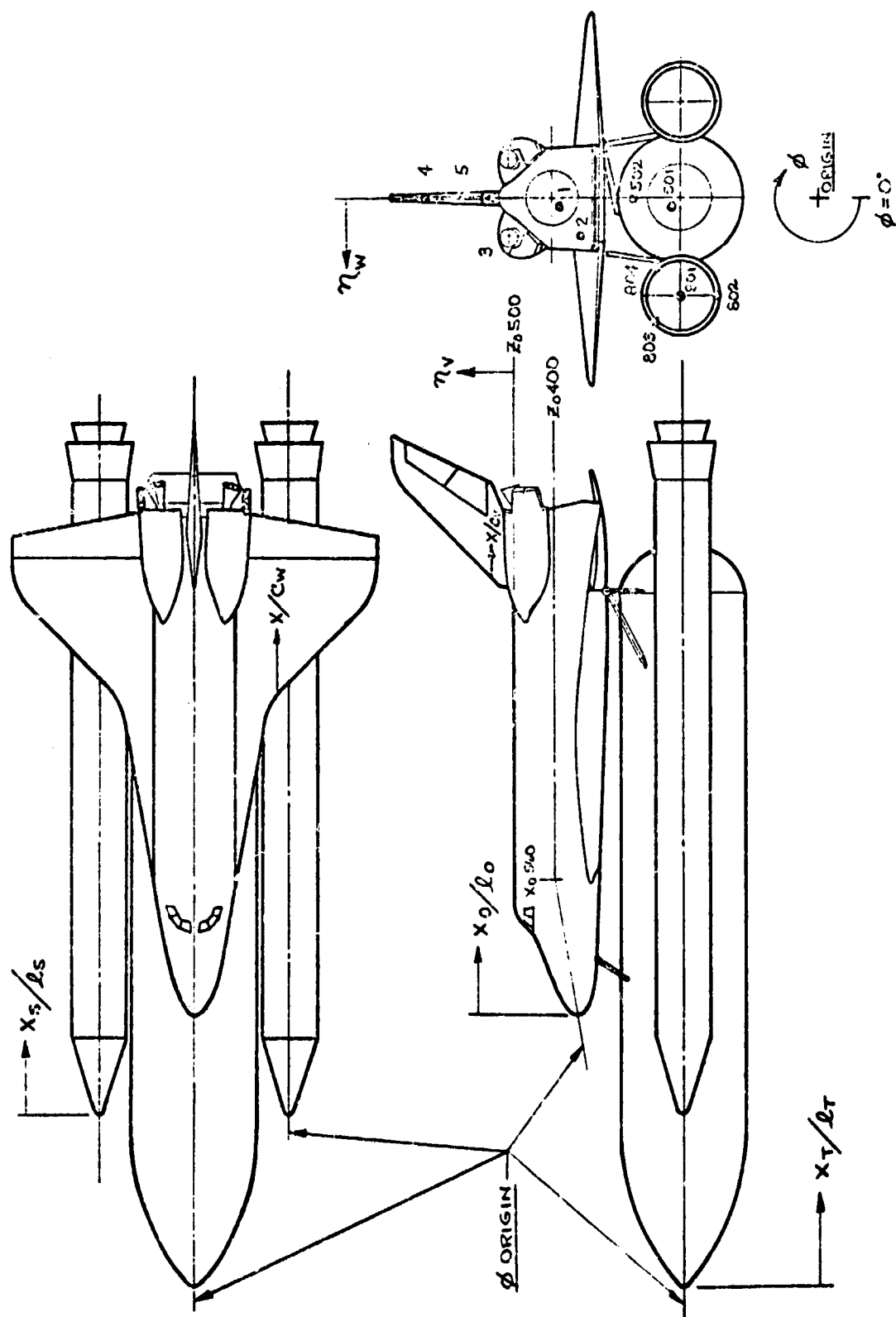
data in datasets RB12XX

# **Notes**

1. Positive directions of force coefficients, moment coefficients, and angles are indicated by arrows
2. For clarity, origins of wind and stability axes have been displaced from the center of gravity



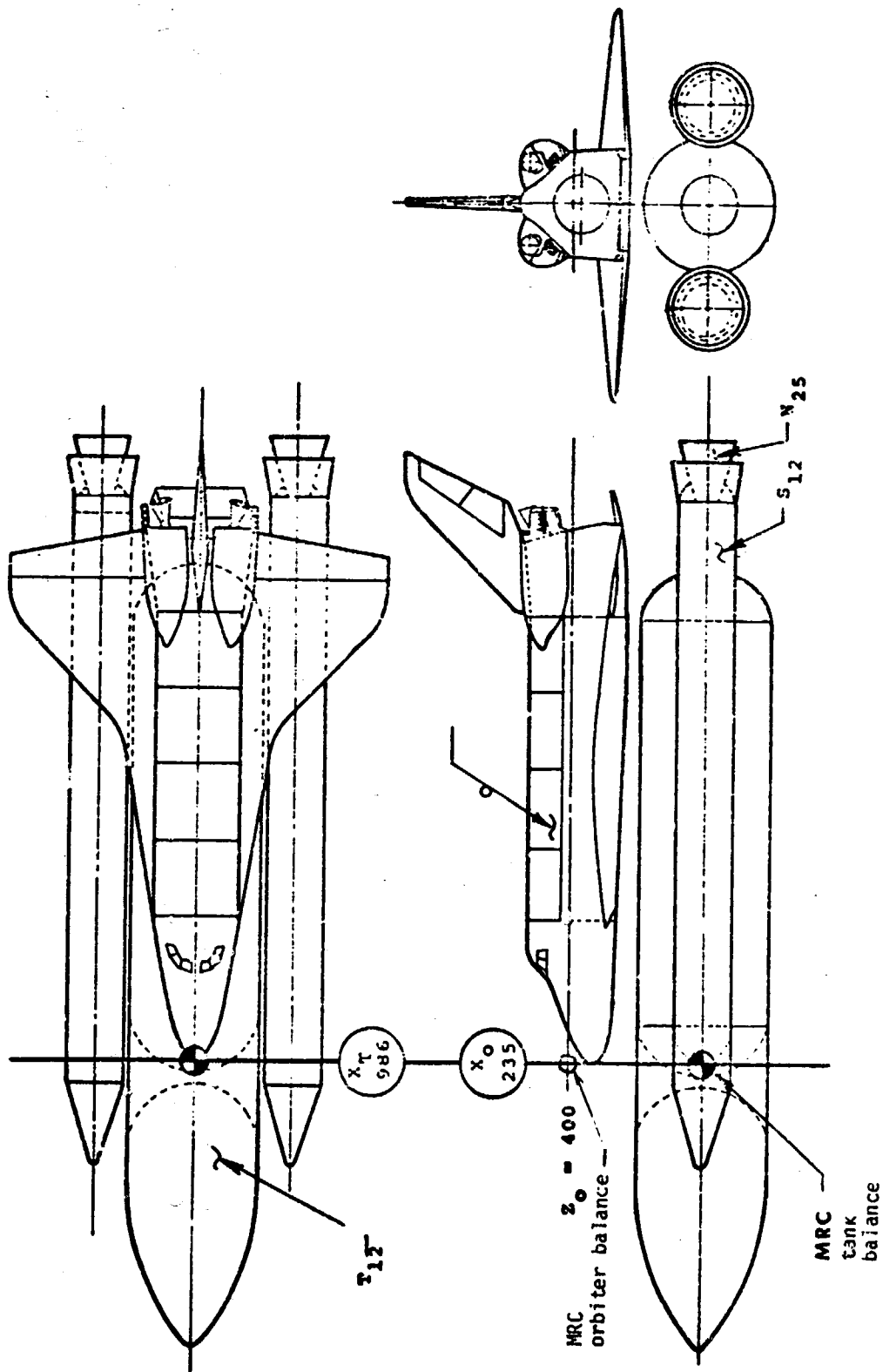
a. Stability and body axis systems  
Figure 1. - Axis Systems



b. Orifice location nomenclature diagram

Figure 1. - Concluded

LV



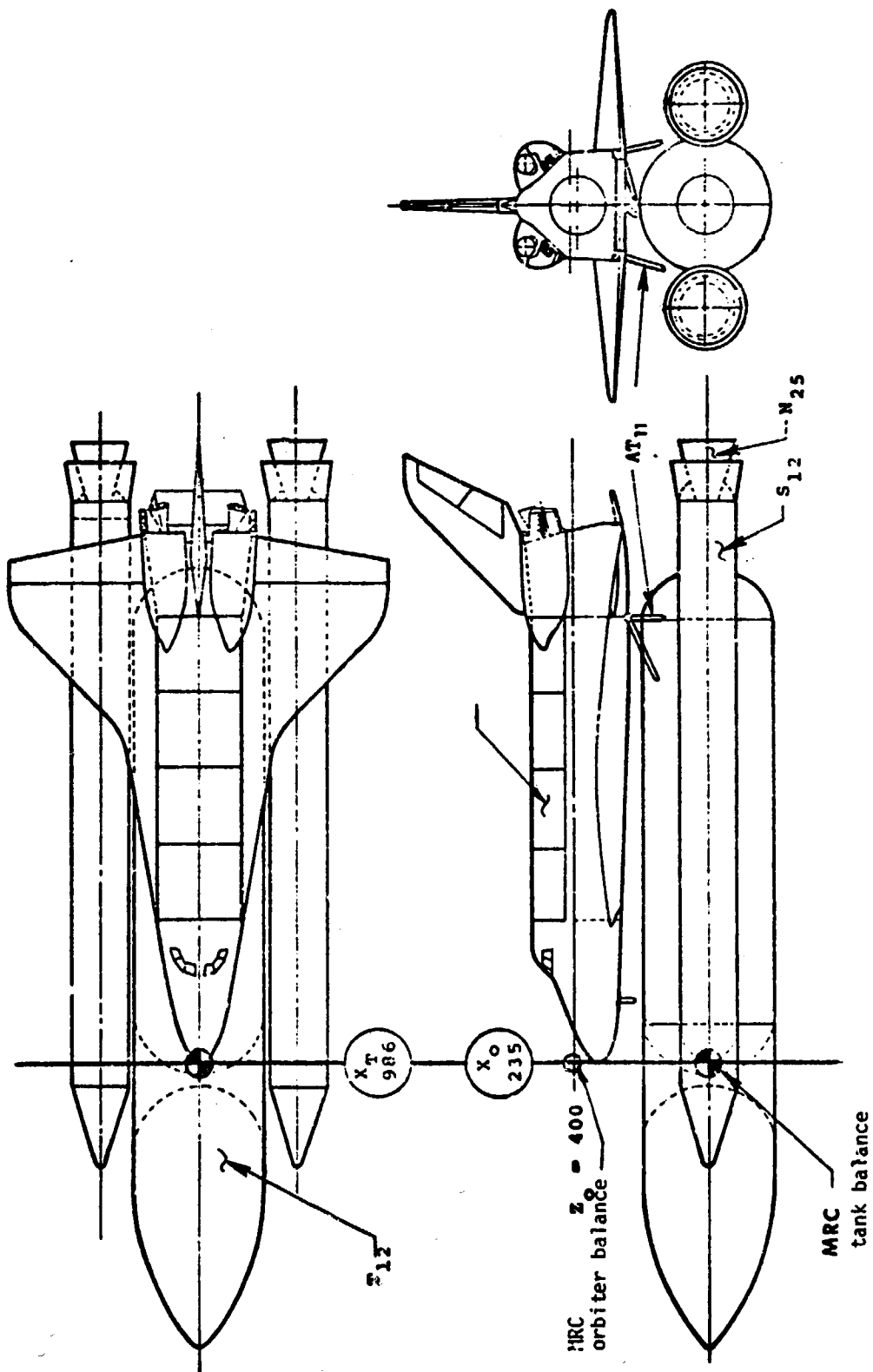
a. Integrated vehicle - 2 balances, no attach structure

Figure 2. - Model Sketches

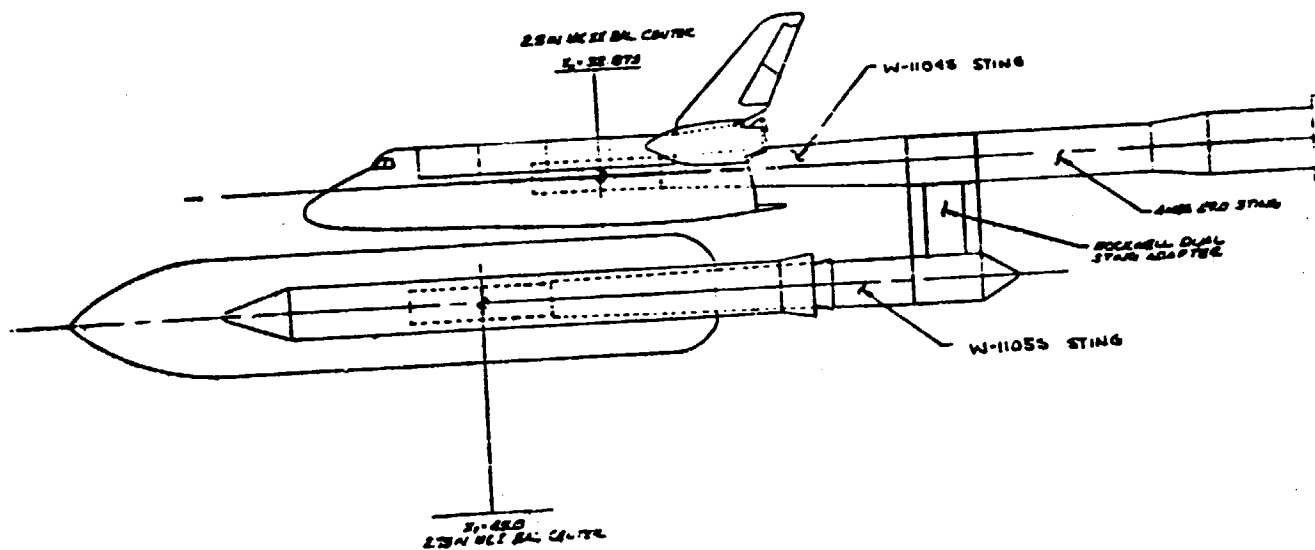
b. Integrated vehicle - 1 balance with attach structure

**Figure 2. - Continued**

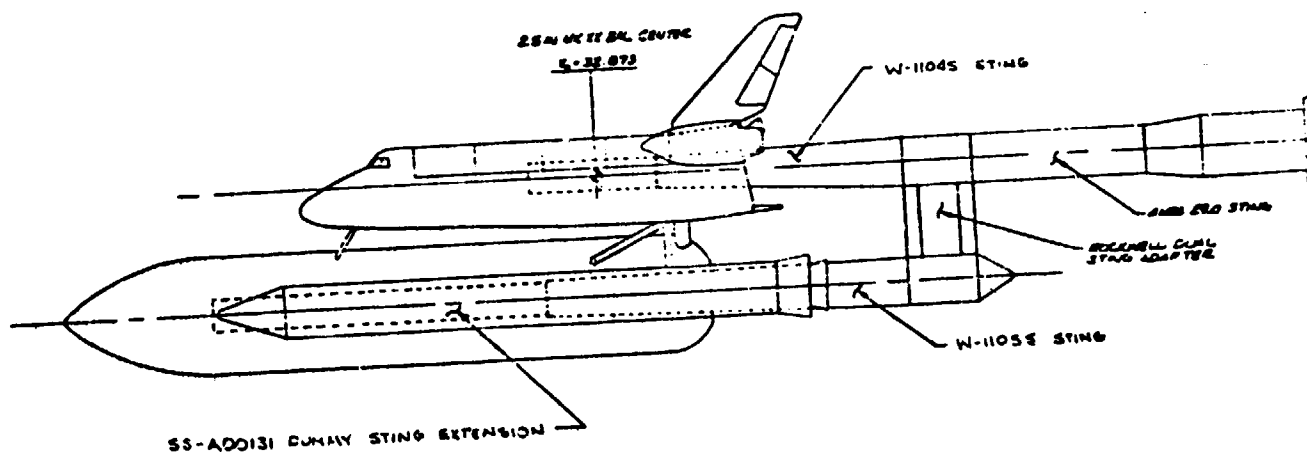
LVAP



c. Integrated vehicle - 2 balances with attach structure  
Figure 2. - Continued



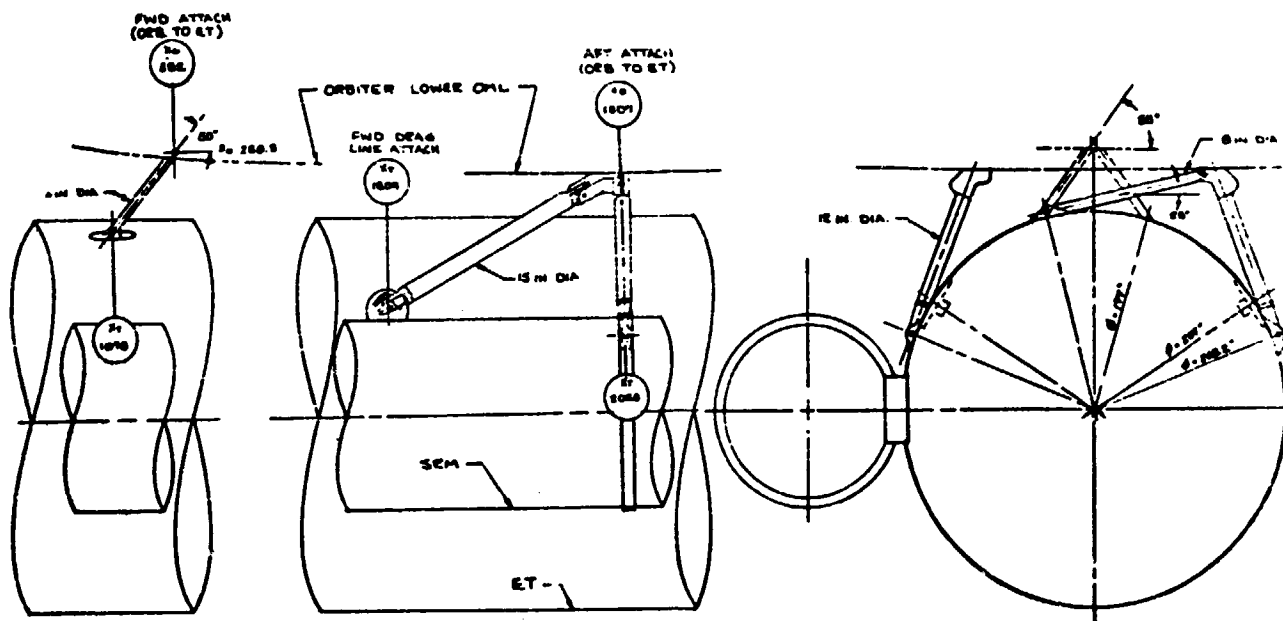
DUAL BALANCE CONFIGURATION ~ LV ± LYAP



SINGLE BALANCE CONFIGURATION ~ LVA

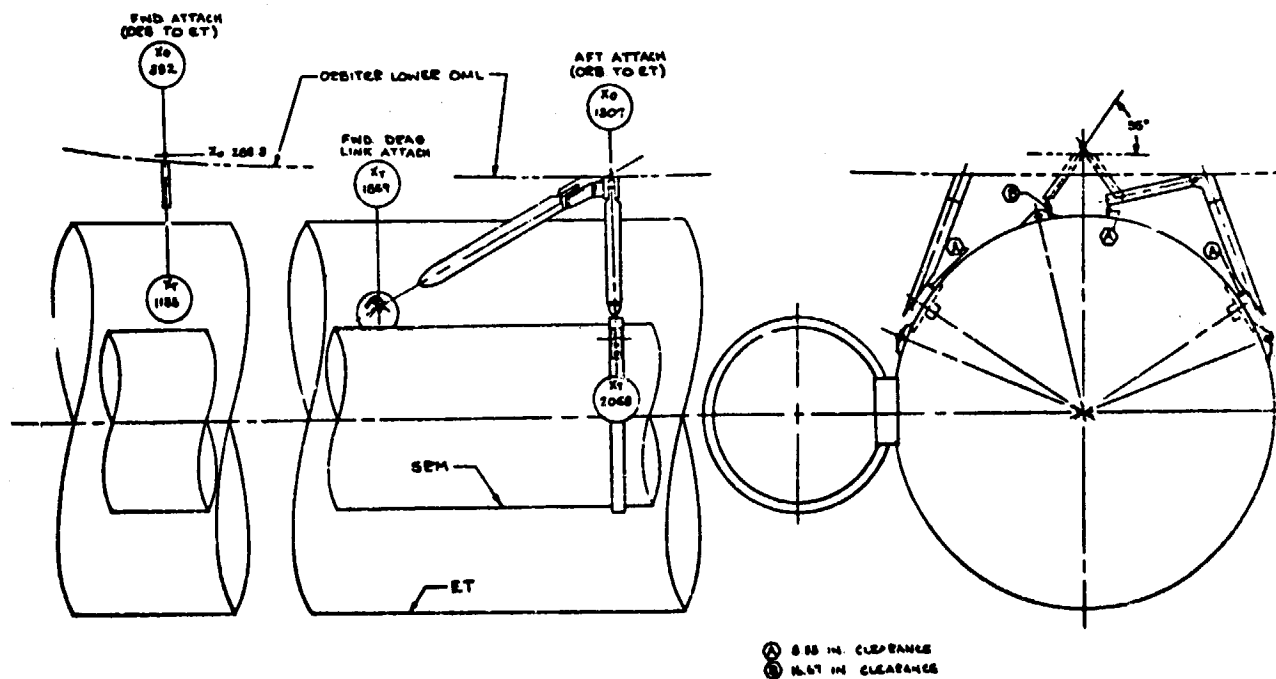
d. Installation side views

Figure 2. - Continued



ATTACH HARDWARE CONFIGURATION - AT10

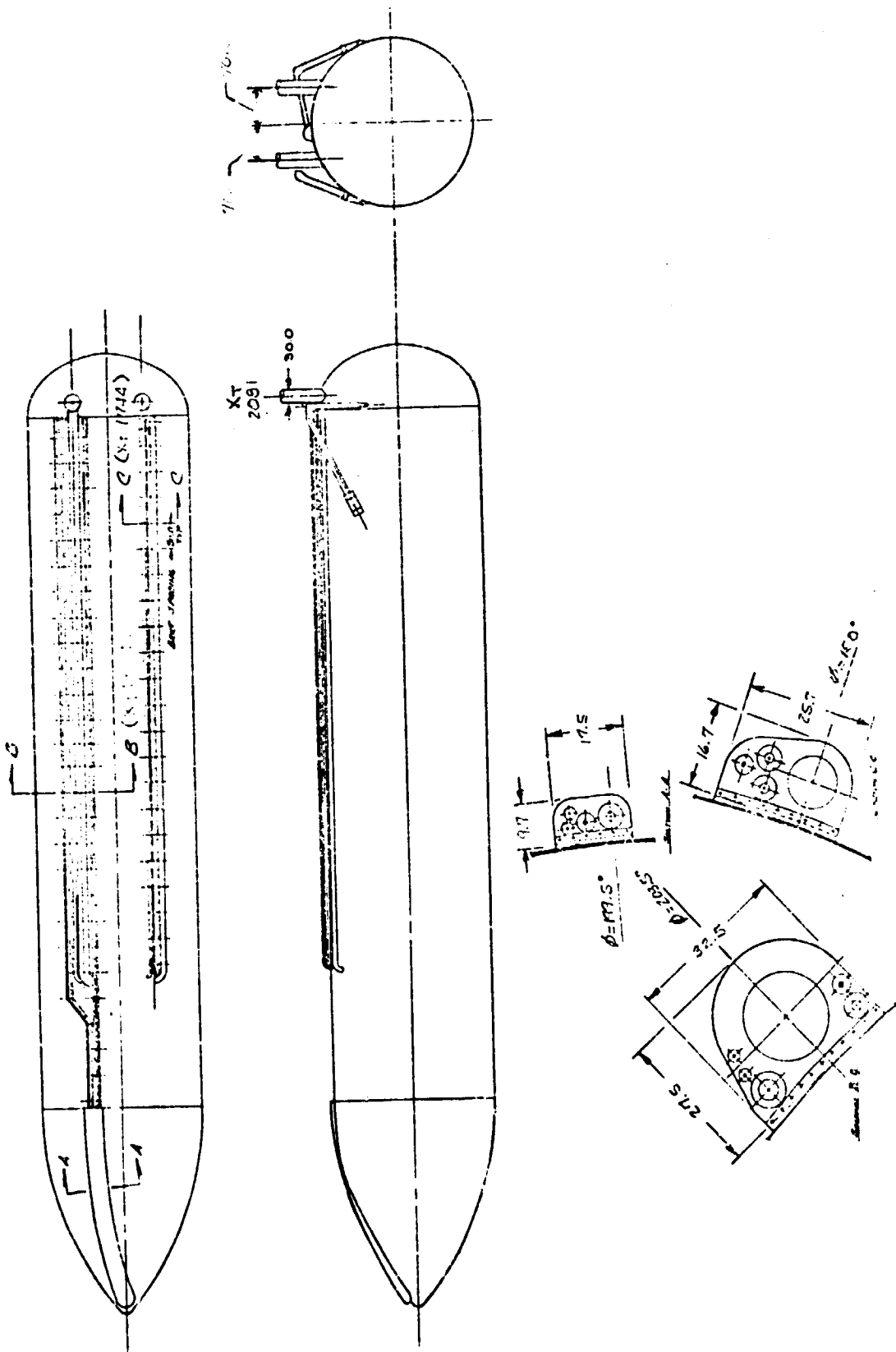
ATTACH HARDWARE CONFIGURATION - AT11



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e. Attach hardware  
Figure 2. Continued





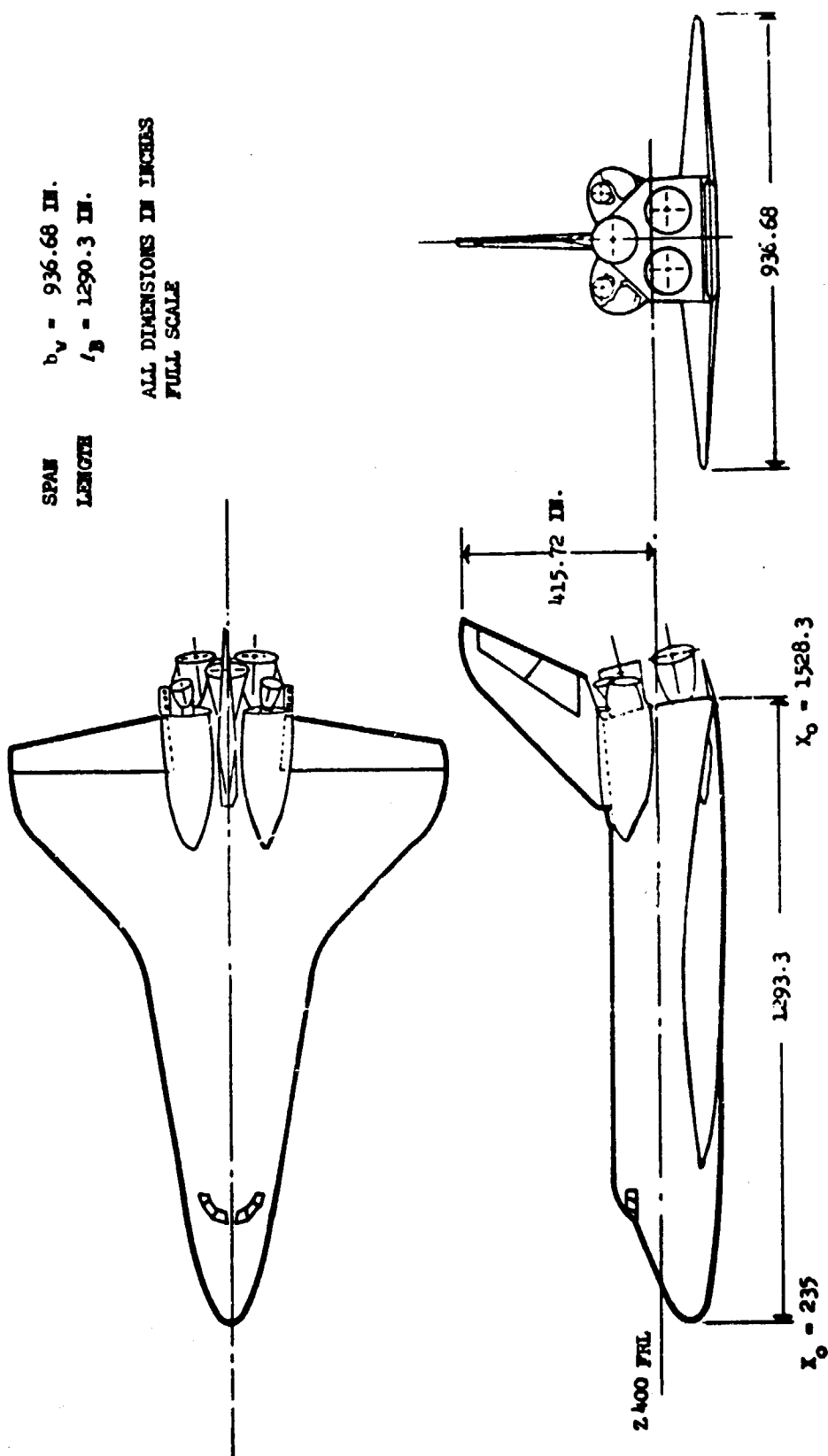
f. External tank protuberances  
Figure 2. - Continued

REFERENCE      DIMENSIONS (FS)

AREA       $S_v = 2600 \text{ FT}^2$   
 MAC       $C = 474.8 \text{ IN.}$

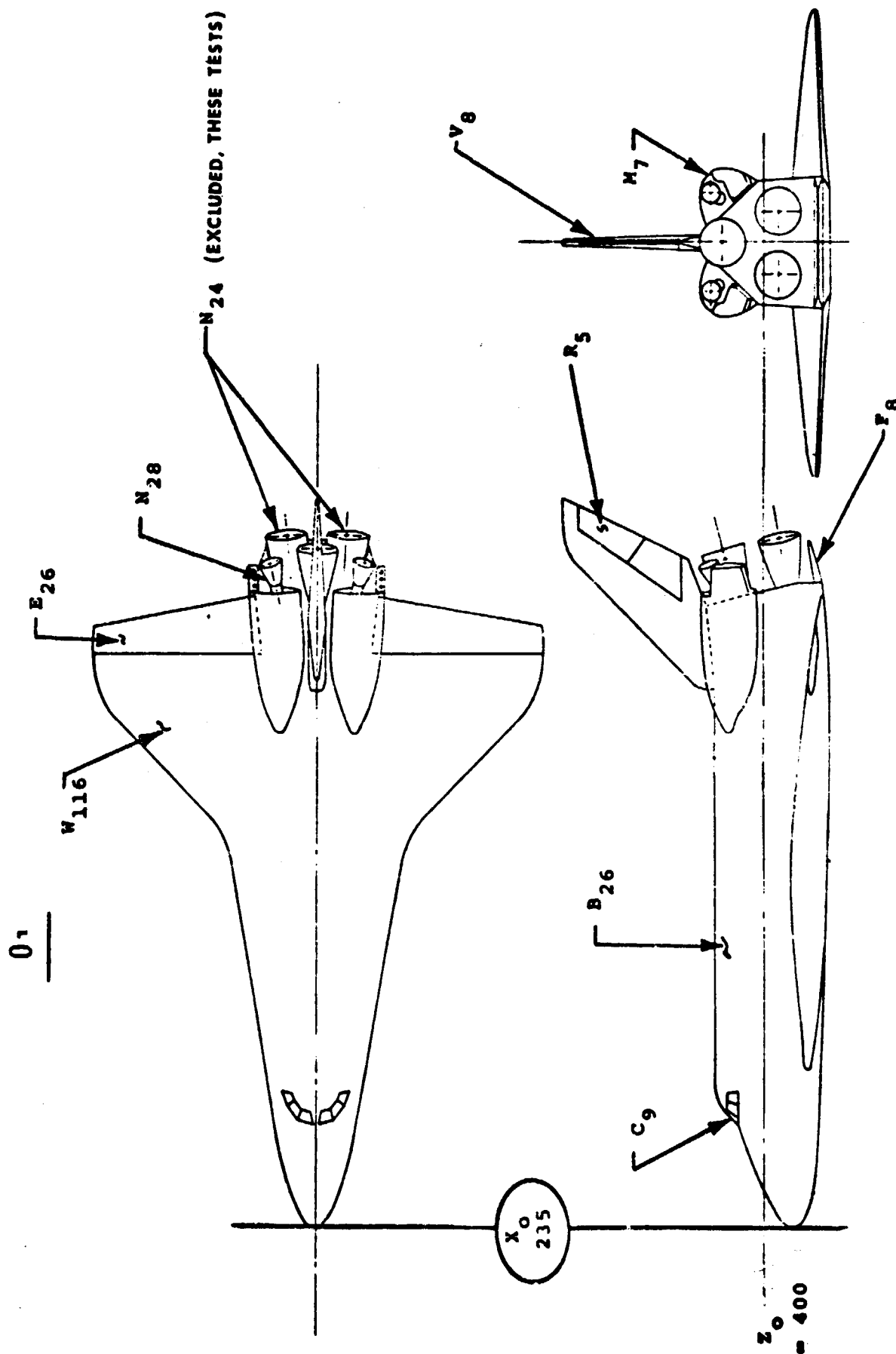
SPAN       $b_v = 936.68 \text{ IN.}$   
 LENGTH       $l_B = 1290.3 \text{ IN.}$

ALL DIMENSIONS IN INCHES  
 FULL SCALE



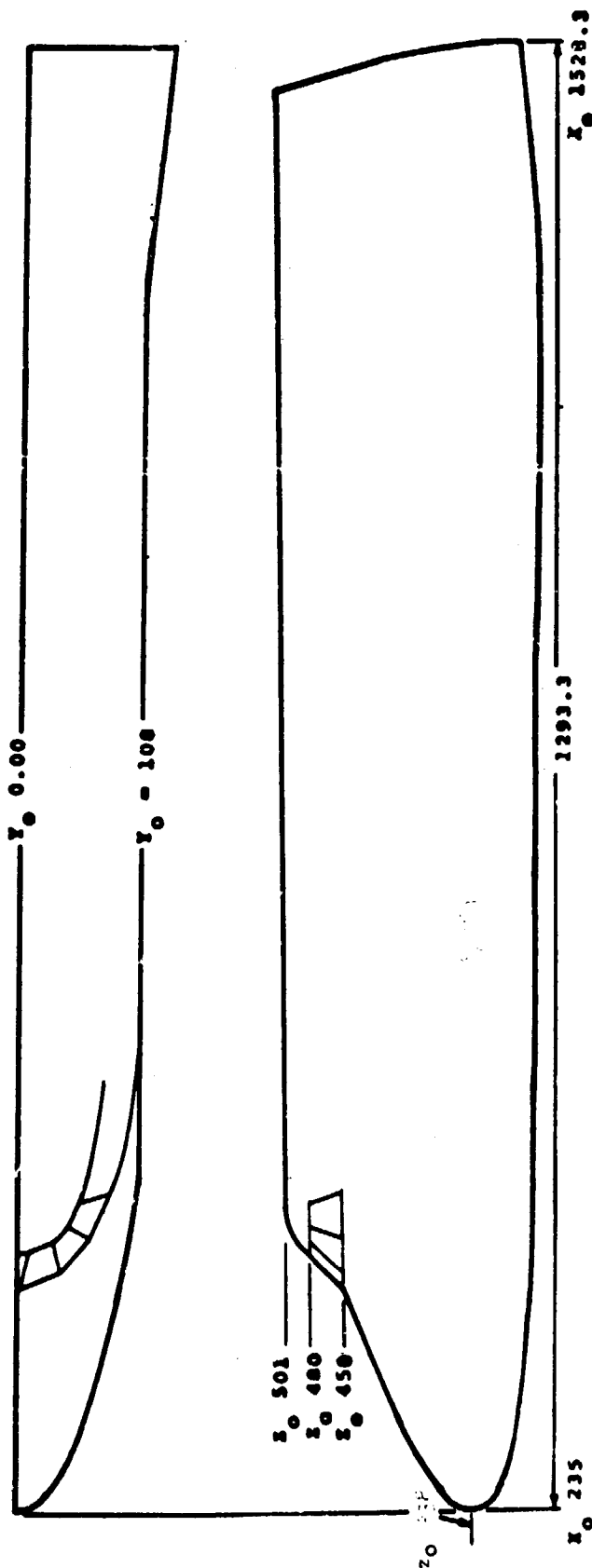
g. SSV orbiter configuration 140A/B

Figure 2. - Continued

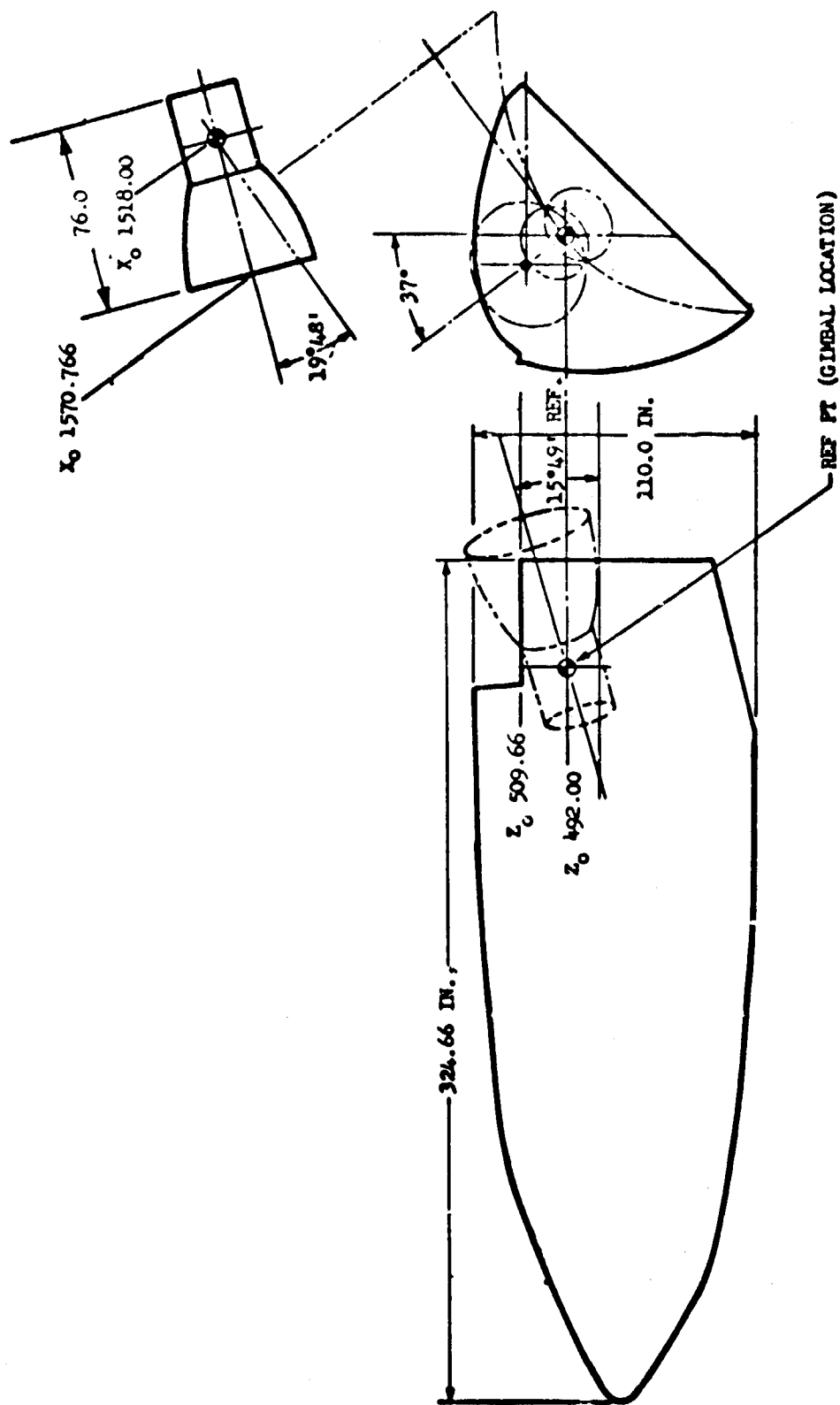


h. Orbiter nomenclature

Figure 2. - Continued

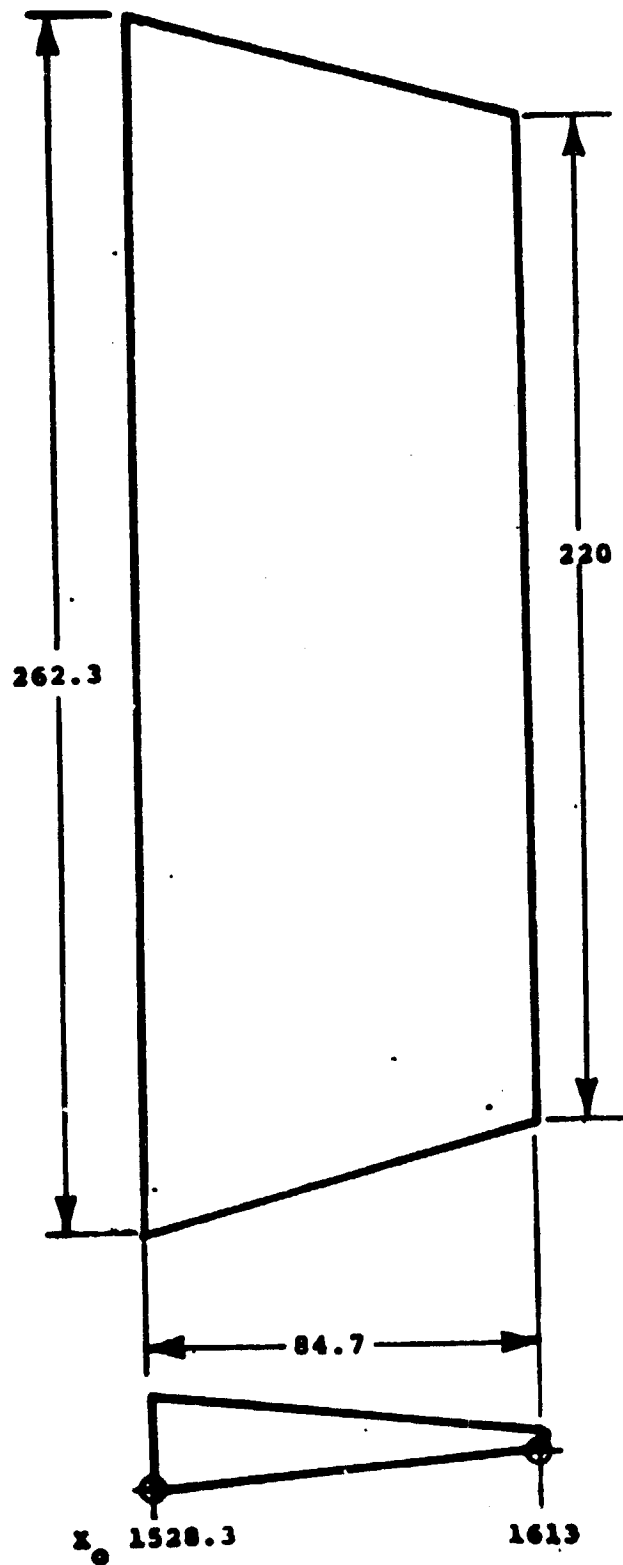


i. Canopy, Cg, and body, B26, lines drawing VL70-00193 and VL70-000140A/B  
Figure 2. - Continued



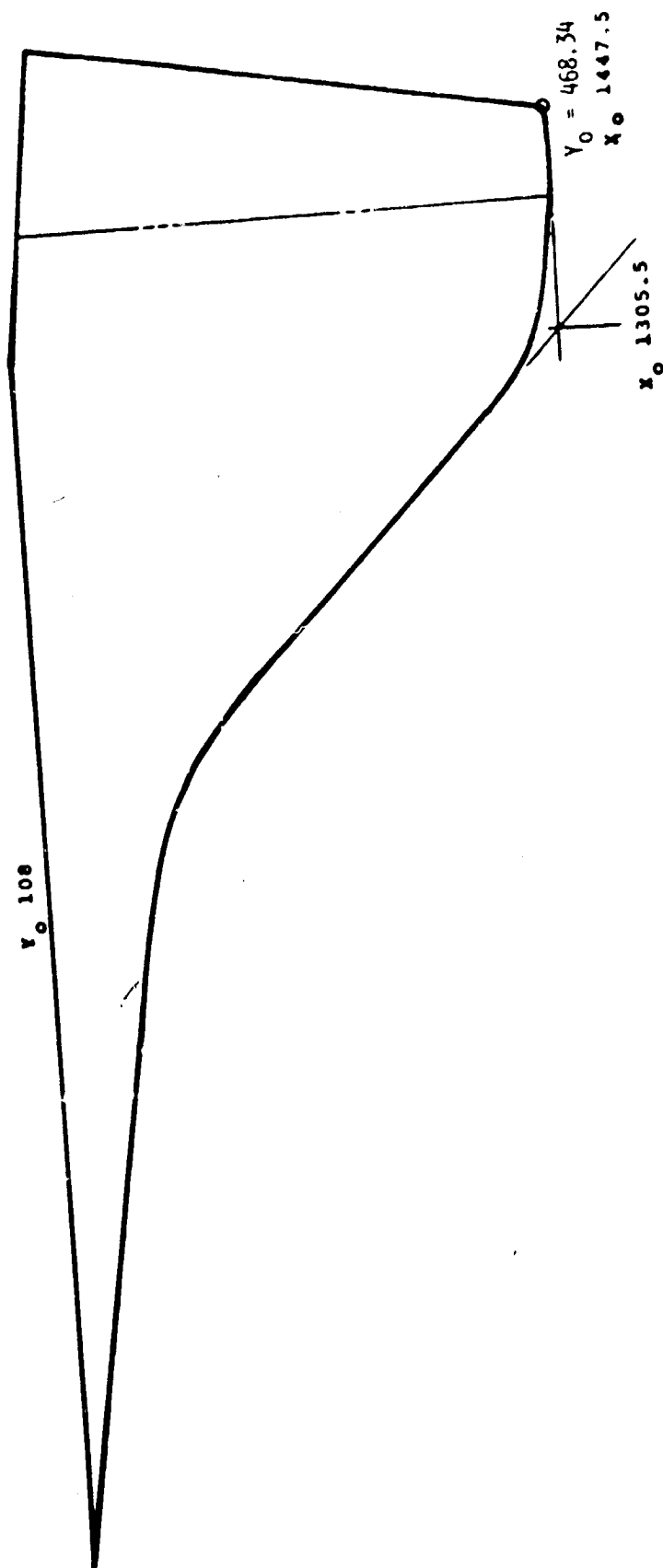
J. M7 - OMS Pod

Figure 2. - Continued



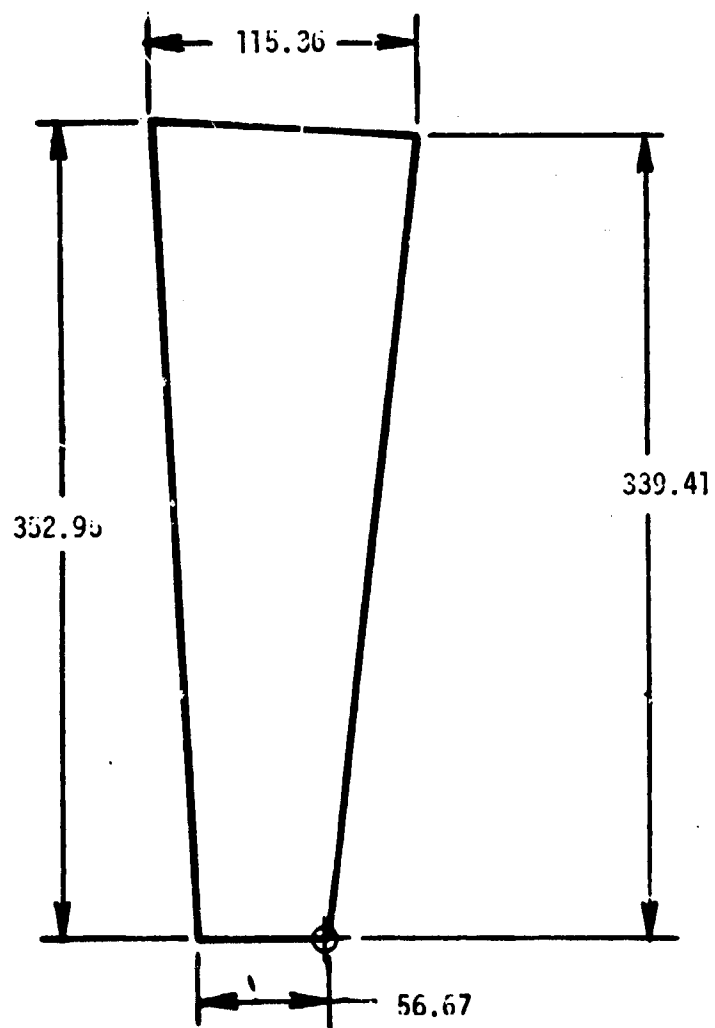
k. Body flap,  $F_8$ , lines drawing no. VL70-000140A/B

Figure 2. - Continued



1. Wing, W116, lines drawing no. VL70-000200

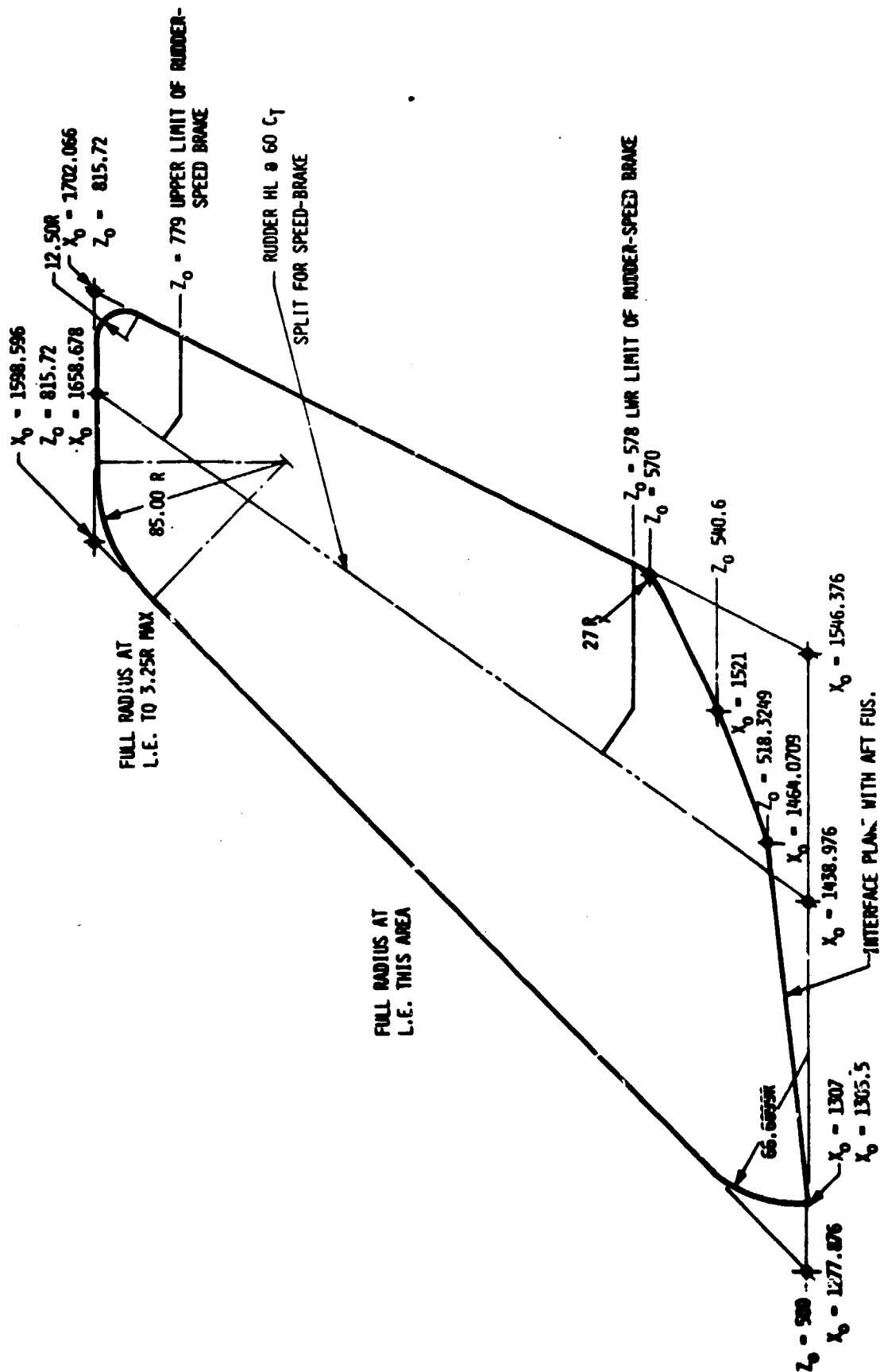
Figure 2. - Continued



m. Elevon, E<sub>26</sub>, lines drawing no. VL70-000200, VL70-000140A/8

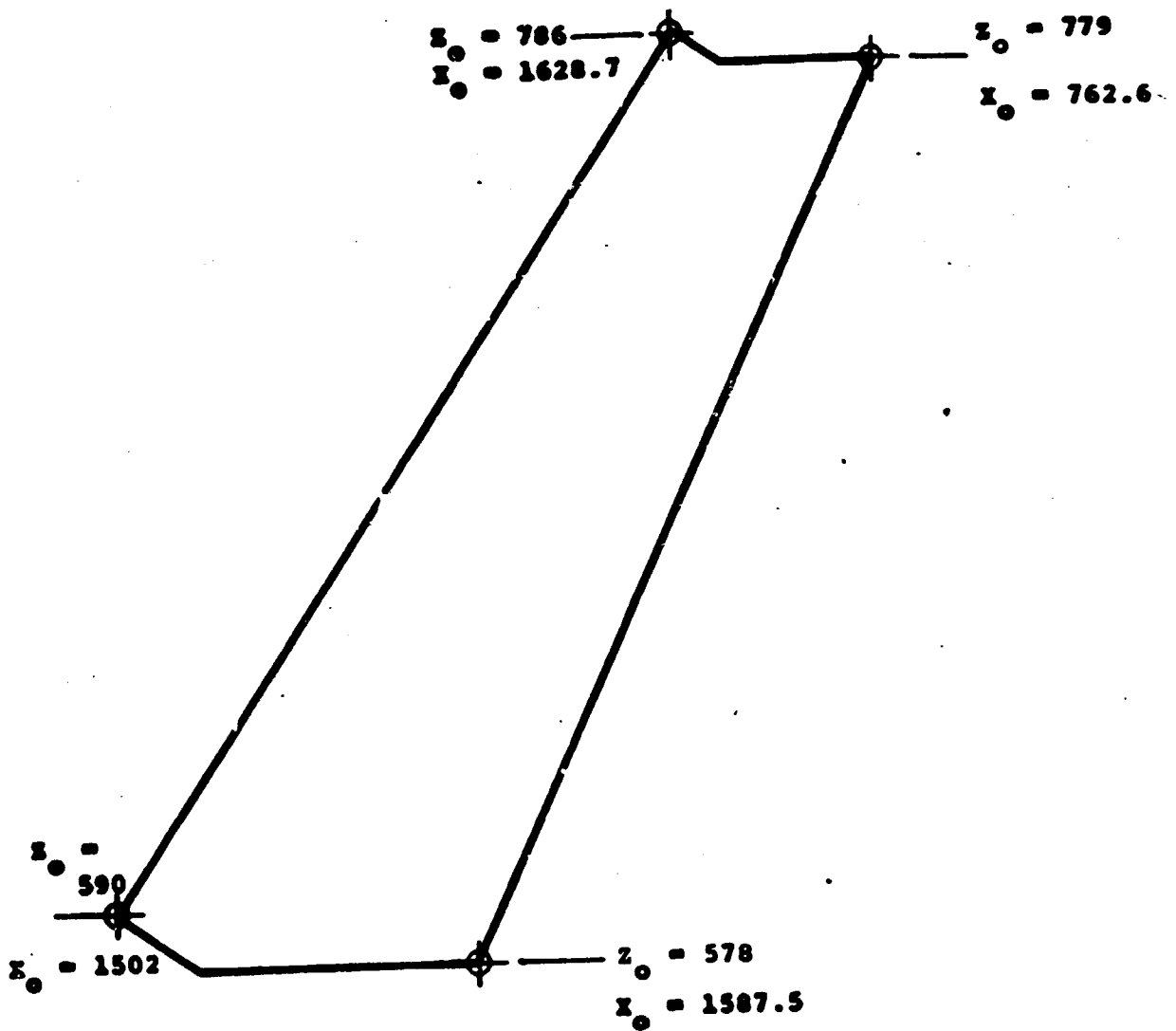
Figure 2. - Continued





n. Vertical tail,  $V_8$ , and rudder,  $R_5$ , lines drawing no. VL70-000146A

Figure 2. - Continued



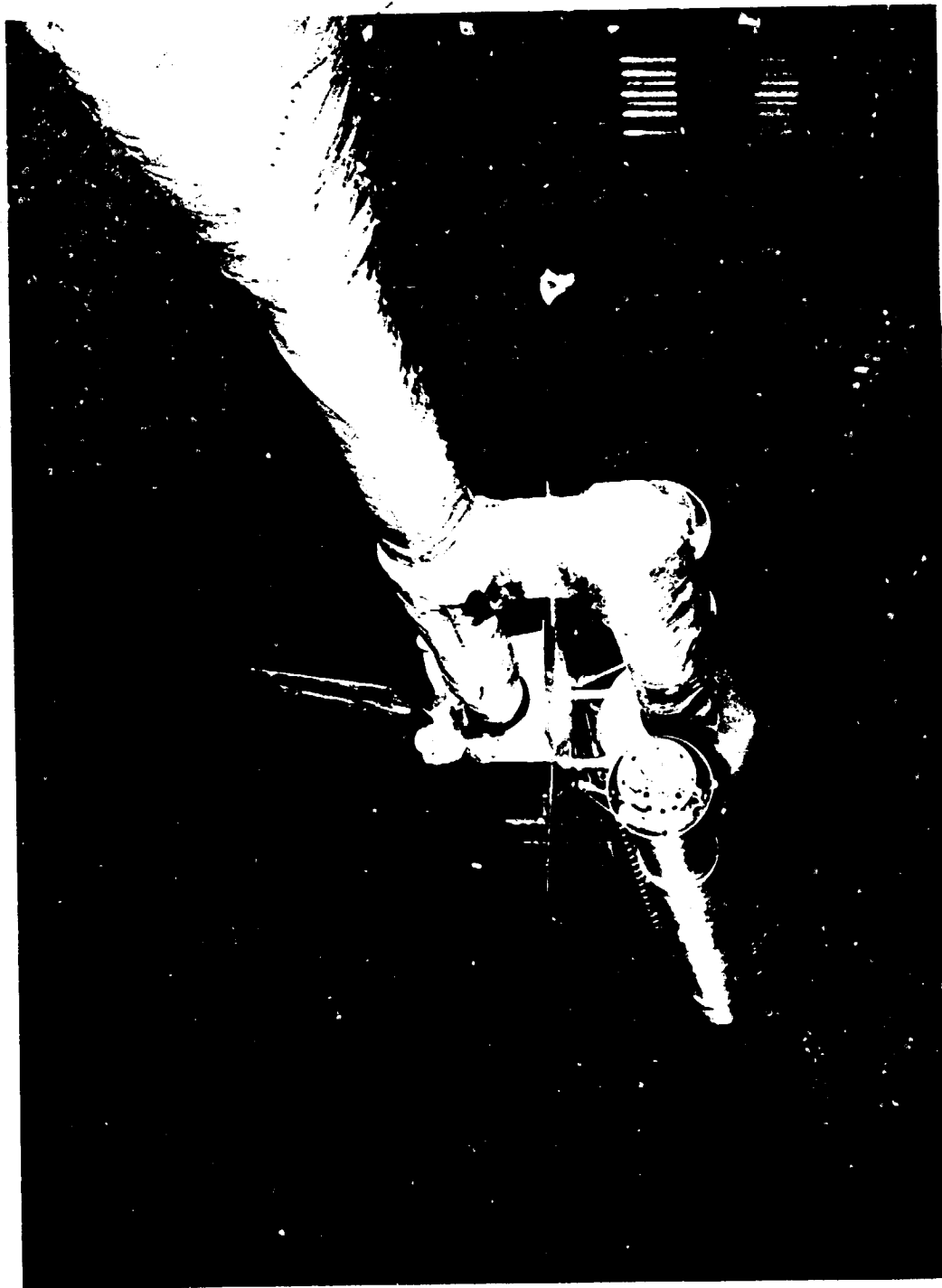
o. Rudder, R5, lines drawing no. VL70-000095

Figure 2. - Concluded



a. Front view of model installed in tunnel

Figure 3. - Model photographs.



b. Rear view of model installed in tunnel

Figure 3. - Concluded.

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TABULATED PRESSURE DATA

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

## REFERENCE DATA

YREF =	2.4210 INCHES	YMRP =	29.9800 INCHES
LEF =	30.7300 INCHES	YMRP =	.0000 INCHES
ORF =	30.7300 INCHES	ZMRP =	.0000 INCHES
SCALE =	.0300 SCALE		

**MACM ( 1 ) = .098 ALPHA( 1 ) = -0.173**

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP				
$\gamma/\delta W$	.2999	.3640	.4270	.5340	.6730 .7800 .8870
$x/CW$					
.000	.0196	.0368	.1916	.1751	.0105 -.1124 -.0653
.020				.4727	.0000 .4186 .3650
.025			.3590		
.045		.1698			
.045			.3161		
.050	.0321			.3120	.3021 .2926 .2307
.060			.2293	.1796	
.085		.2479			
.096					
.094	.0640			-.0167	-.0142 -.0110 -.0025
.150		.0000			
.163		.1823	.0444		
.177					
.193					
.229	.1464		-.0733		
.250				-.1758	-.2045 -.2427 -.3457
.274					
.339		-.0516			
.362	.1454			-.2083	-.2947 -.4896
.400			-.1716		
.492	.0000			-.2399	-.3491
.497			.0000		
.550					
.565					
.600					
.650					
.700	-.1509			-.1926	-.2631
.725					
.750			-.1811		
.760				.0343	.0441
.775			.0196		
.808					
.834	-.1145				
.850					
.857			.0313	.0044	.0675
.900	.0176				
.925	-.0044			.0051	.0055
			-.0109		.0321

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(R81U17)

ARC11-716 IA14 01-T12-S12M23-AT11 UPPER WING

MACH (1) = .898 ALPHA(1) = -8.170

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3840 .4270 .5340 .6730 .7800 .8470

X/CW

.930 -.0384 .0312

.933

-.0489

.965 -.0732

MACH (1) = .836 ALPHA(2) = -4.140

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3840 .4270 .5340 .6730 .7800 .8870

X/CW

.000 .0132 .0969 .3134 .3472 .2320 .1543 .1921

.020 .025 .025 .2886

.045 .045 .045 .2147

.050 .0224 .1468 .1310 .1088 .0437

.080 .083 .083 .0846

.086 .086 .086 .2148

.094 .094 .094 .0348

.190 .163 .163 .0000

.177 .193 .193 .0772

.229 .1114

.250 .274

.339 .339 .339 .1923

.382 .382 .382 .0686

.403 .403 .403 .2941

.402 .497 .497 .0000

.530 .543 .543 .0000

.600 .600 .600 .0000

.650 .650 .650 .0000

.700 .700 .700 .1123

.723 .723 .723 .1976

.750 .750 .750 .1976

.780 .780 .780 .1231

.773 .773 .773 .0394

.808 .808 .808 .0033

.834 .834 .834 .0224

.850 .850 .850 .0677



DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N25+AT11 UPPER WING

(RBIU17)

MACH ( 1 ) = .896 ALPHA( 2 ) = -4.140

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.897  
 .863  
 .803  
 .903  
 .905  
 .933  
 .963  
 -.0092  
 -.0340  
 .0177  
 -.0065  
 -.0192  
 -.0406  
 .0877  
 .0187  
 .0481

MACH ( 1 ) = .896 ALPHA( 3 ) = -.220

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000  
 .020  
 .025  
 .040  
 .045  
 .090  
 .093  
 .085  
 .086  
 .094  
 .150  
 .163  
 .177  
 .193  
 .229  
 .250  
 .274  
 .339  
 .362  
 .400  
 .402  
 .497  
 .550  
 .565  
 .600  
 .650  
 .700  
 .725  
 .750  
 .760  
 -.0434  
 -.0302  
 .0698  
 .1431  
 .0648  
 -.0852  
 .1508  
 -.0096  
 .0000  
 -.0464  
 .0699  
 .274  
 .339  
 .362  
 .400  
 .402  
 .497  
 .550  
 .565  
 .600  
 .650  
 .700  
 .725  
 .750  
 .760  
 .3974  
 .4813  
 .1705  
 .1635  
 .0648  
 -.0852  
 .0648  
 .0000  
 -.0464  
 .0699  
 .274  
 .339  
 .362  
 .400  
 .402  
 .497  
 .550  
 .565  
 .600  
 .650  
 .700  
 .725  
 .750  
 .760  
 .4224  
 .0900  
 .0809  
 .0241  
 -.1226  
 -.1623  
 -.2532  
 -.2553  
 -.4317  
 -.4939  
 -.4915  
 -.5538  
 -.6093  
 -.6486  
 -.7149  
 -.7590  
 -.3732  
 -.5445  
 -.6603  
 -.4809  
 -.4148  
 -.5145  
 .0000  
 -.3346  
 -.1993  
 -.6030  
 -.7101  
 .1715  
 .0960



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 04+712+512N23+AT11 UPPER WING (RBU17)

MACH ( 1 ) = .896 ALPHA( 2 ) = -.220

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.8990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.775				.0010	.0960		
.808			-.0040				
.834	-.2031						
.850				.0110	.0875	.1359	
.857			.0056				
.865	-.0551						.1216
.900	-.0275		.0059	.0089	.0316		
.905				-.0262	.0653		
.950			-.0177				
.953							
.965	-.0569						

MACH ( 1 ) = .896 ALPHA( 4 ) = 3.830

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.8990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1312	-.0331	.3319	.5251	.4630	.4165	.3779
.020				-.0356	.0000	-.2455	-.3658
.025		.0050	.0179				
.040			-.0927				
.045				-.2042	-.3925	-.4930	-.6360
.050	-.0095			-.4825			
.080			-.2490				
.085							
.086		.0781					
.094	-.0484			-.6819	-.7960	-.7936	-.8339
.158		.0000					
.163			-.4543				
.177							
.193		-.1488					
.229	.0395			-.7537	-.8920	-.9903	-.8974
.250			-.5527				
.274							
.339		-.4875					
.362	-.0616			-.8016	-.9878		-.7244
.400			-.6880				
.402							
.497	.0000						
.550			.0000	-.5938	-.4197		
.565							-.5812
.600							-.5082
.690							





(R81U17)

ARC11-716 1A14 01+T12+312N25+AT11 UPPER WING

MACH ( 1 ) = .898 ALPHA( 5 ) = 8.03D

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.590				-.3549	-.6786		
.565			.0000				
.600						-.5865	
.650					-.5836		
.700	-.2349			-.2223			
.725				-.3203			
.750						-.4915	-.5506
.760			-.2247				
.775				-.2131	-.4225		
.806			-.6872				
.834	-.1789						
.850				-.1464	-.2950	-.4258	
.857			-.0589				
.865	-.0502			-.0956			-.5128
.900	-.0508		-.0415	-.1605			
.905				-.0722		-.3558	
.950			-.0518				
.953							
.965	-.0470						

MACH ( 2 ) = .977 ALPHA( 1 ) = -7.92D

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.050	-.1291	.1308	.3025	.2972	.1246	.0020	.0283
.020				.5418	.4679	.4736	.4256
.025		.4419					
.040	.2525						
.045		.4024		.3846	.3597	.3416	.2924
.050	.0265			.2332			
.060		.3113					
.065							
.066	.3276						
.074				.0647	.0615	.0714	.0035
.150		.0000					
.163							
.177		.1316					
.193		.2682					
.229	.2201			-.1852	-.1119	-.1483	-.2475
.250							
.274							
.339		.0364					



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 OL+T12+S12N23+AT11 UPPER WING

(R01U17)

MACH ( 2 ) = .977 ALPHA( 1 ) = -7.920

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.362	.2338					
.400						
.402						
.497	.0000					
.553						
.565						
.600						
.850						
.700	.0369					
.725						
.750						
.760						
.775						
.808						
.834	-.1886					
.850						
.857						
.865	.0203					
.900	.0375					
.905						
.950						
.953						
.965	-.0135					

MACH ( 2 ) = .976 ALPHA( 2 ) = -3.880

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2663	.1731	.4052	.4128	.3237	.2499	.2843
.020				.4173	.3634	.3676	.3392
.025							
.040							
.045							
.050	.0330						
.080							
.085							
.086							
.094	.0744						
.150							
.165							
.177							
.193							

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 OI+T12+S12M23+AT11 UPPER WING (R01U17)

MACH ( 2 ) = .976 ALPHA( 2 ) = -3.880

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.229	.1888					
.250						
.274						
.339						
.362						
.400						
.402						
.497						
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.806						
.834						
.850						
.857						
.865						
.900						
.905						
.905						
.950						
.955						
.965						

MACH ( 2 ) = .977 ALPHA( 3 ) = .090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000						
.020						
.025						
.040						
.045						
.090						
.095						
.096						
.894						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U17)

ARC11-716 1A14 01+712+912N25+AT111 UPPER WING

MACH ( 2 ) = .977 ALPHA( 3 ) = .090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.150						
.163						
.177						
.193						
.229						
.250						
.274						
.339						
.362						
.400						
.402						
.497						
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.953						
.965						

MACH ( 2 ) = .975 ALPHA( 4 ) = 4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000						
.020						
.025						
.040						
.045						
.050						

(R81017)

ARC11-716 1A14 01-T12-S12N2-JAT11 UPPER WING

MACH (2) = .973 ALPHA(4) = 4.920

SECTION (3) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.8990	.9640	.4270	.3340	.6730	.7800	.8979
X/CW							
.000							
.005							
.006							
.007							
.008							
.009							
.010							
.011							
.012							
.013							
.014							
.015							
.016							
.017							
.018							
.019							
.020							
.021							
.022							
.023							
.024							
.025							
.026							
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.030							
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.092							
.093							
.094							
.095							
.096							
.097							
.098							
.099							
.100							





MACH (3) = 1.102 ALPHA(1) = -7.940

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BL	.2990	.3840	.4270	.5340	.6750	.7000	.8070
X/CW							
.02C	-.2140	-.0837	.3032	.3723	.2386	.1327	.1520
.02D			.4002	.5419	.4969	.5202	.5025
.02A							
.040		.04C1					
.045			.3597				
.050	-.1182			.3789	.3911	.3929	.3797
.080				.2496			
.085			.2759				
.086		.2080					
.094	-.0544						
.150				.0864	.1112	.1511	.1194
.163		.0000					
.177			.1193				
.193		.2075					
.229	-.0298			-.1000	-.0423	-.0331	-.1017
.290			.0509				
.274		.0304					
.339							
.362	.1547			-.0249	-.0344		-.2318
.400			-.0101				
.402				-.0260	-.0857		
.497	.0000		.0200				
.550						-.2911	
.565					-.1913		
.600				-.1282			
.650						-.1545	-.1774
.700	.1801						
.725			-.1003				
.750				-.1013	-.1010		
.760			-.0270				
.775							
.808				-.0605	-.0709	-.0556	
.834	-.0327		.0112				
.850							
.857							
.865	.0229						.0458
.900	.0457			.0258			
.905			.0741		.0917		.0775
.925				.0668			
.950			.0793				
.953							
.969	.0473						



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 TABULATED PRESSURE DATA - IA14A - VOL. 6  
 ARC11-716 IA14 OI+T12+S12N23+AT11 UPPER WING

(R01017)

MACH (3) = 1.101 ALPHA(2) = -3.890

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP		
Y/BW	2990	.3640	.4270 .5340 .6750 .7800 .8870
X/CW			
.000	-.1879	-.0176	.3969 .4863 .3809 .3035 .3440
.020			.4291 .3743 .3602 .4046
.025		.0987	
.040			
.045		.2689	
.050	-.1905		.2270 .2022 .1803 .2075
.060			.0722
.085		.1569	
.086		.2010	
.094	-.1213		-.1255 -.1370 -.1010 -.0760
.150		.0000	
.163		.1268	
.177		-.0158	
.193			
.229	-.0298		-.2608 -.2654 -.2708 -.3040
.250		-.1313	
.274		-.1116	
.339	.1135		-.2796 -.3015 -.4916
.362		-.2554	
.400	.0000		
.497		-.0117 .1034	
.550		.0900	-.2276
.565			-.2309
.600			
.650	.1169		-.1090
.700		-.1031	
.725			-.1570 -.0712
.750		-.1182	
.760		-.0843 -.1248	
.775		-.0537	
.808			
.834	-.0810	-.0806 -.0817 .0309	
.850		-.0519	
.857			
.865	-.0016		.1205
.900	-.0037	-.0423 .0657	
.905		-.0037 .0441	.1081
.930		.0158	
.955			
.965	-.0210		

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(R01U17)

ARC11-716 1A14 C11T12+SIGNES+AT11 UPPER WING

MACH (3) = 1.103 ALPHA(3) .090

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6792	.7800	.8870
X/CW							
.000	-.1972	-.0639	.4200	.5737	.5153	.5056	.5491
.020				.2737	.2527	.2430	.2214
.025			.2280				
.040		.0744					
.045			.1307				
.050	-.2518			.0606	.0367	.0032	-.0287
.060				-.1043			
.065			-.0022				
.066		.1613					
.094	-.1928			-.2922	-.3021	-.2740	-.2776
.130			.0000				
.163			-.1480				
.177		.0407					
.193				-.4702	-.4303	-.4519	-.4740
.229	-.0489		-.2436				
.250		-.2042					
.274							
.339				-.5670	-.5456		-.6502
.362	.0682						
.400				-.1327	-.3449		
.402			-.3984				
.497	.0000		.0000				
.550							
.56							
.56							
.600							
.650					-.5310		-.6321
.700	.0363			-.1927			
.725				-.1783			
.750			-.1475		-.0190	-.2633	
.760							
.775			-.0713	-.1489	-.1063		
.808							
.834	-.0987						
.850				-.1449	-.0850	.0737	
.857			-.0835				
.865	-.0282						
.900	-.0384			-.1015	.0503		-.0864
.925			-.0960				
.930				-.0213		.1258	
.953			-.0232				
.963	-.0603						



TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81017)

ARC11-716 1A14 01+Y12+S12N2S+A111 UPPER WING

$$\text{ALPHA}(3) = 1.199 \quad \text{ALPHA}(4) = 4.069$$

SECTION ( 1 ) UPPER WING

	.2995	.5640	.4270	.5340	.6735	.7895	.8870
--	-------	-------	-------	-------	-------	-------	-------

0/1

.000	-.2123	-.2456	.3721	.6045	.5085	.5452	.5375
.020				.0057	.0804	-.0002	-.0072
.025			.0040				
.040		-.0263					
.045			-.0260				
.050	-.2740			-.1336	-.1664	-.2240	-.3095
.060				-.2901			
.085			-.1470				
.086		.0017					
.094	-.2745			-.4554	-.5065	-.4036	-.5022
.150		.0000					
.163							
.177			-.2725				
.193		-.0001					
.229	-.0015			-.6340	-.6263	-.6430	-.6375
.250			-.3693				
.274		-.2924					
.339							
.362	.0291			-.6986	-.7165		-.7029
.400			-.4990				
.402							
.497	.0000			-.3013	-.7036		
.550			.0000				
.565							
.600							
.650							-.7413
.700	-.1544			-.2060		-.7553	
.725					-.2104		
.750			-.1903			-.5678	-.5759
.760				-.2095	-.1417		
.775			-.0062				
.800							
.834	-.0004						
.850				-.1015	-.0729	-.3775	
.897			-.1112				
.865	-.0601						
.900	-.0710			-.1327			-.3921
.935			-.1035		-.0056		
.950				-.1005		-.0546	
.955			-.0060				
.955	-.1291						

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U17)

RC11-716 1A14 01+712-312M25-AT11 UPPER WING

MACH (3) = 1.099 ALPHA(3) = 8.020

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.7260	-.5191	.2073	.6033	.5983	.5112	.4487
.020				-.0849	-.0948	-.2331	-.3663
.025				-.0555			
.040		-.1709					
.045			-.1591				
.050	-.2843			-.2770	-.3432	-.4367	-.5644
.060				-.4303			
.065			-.2902				
.086		-.0231					
.094	-.2802			-.6276	-.6561	-.6678	-.7015
.150		.0000					
.163			-.3858				
.177		-.1906					
.193							
.229	-.1377			-.8035	-.7784	-.7898	-.7948
.250			-.4609				
.274		-.3536					
.339				-.8334	-.9468		-.8340
.382	.0017						
.400			-.5636				
.402				-.7839	-.8233		
.497	.0000						
.590							
.565			.0000				
.600							
.650							
.700	-.4090			-.3689		-.7320	-.7731
.725							
.790			-.2317			-.6214	-.7106
.760				-.3060	-.3039		
.775			-.1116				
.800							
.634	-.0904			-.2507	-.2309	-.5559	
.850			-.1417				
.857							
.865	-.0671						
.900	-.1164			-.2056			-.6770
.905			-.1592	-.0901			
.950				-.1712		-.5523	
.955			-.1392				
.965	-.1793						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+S12N23+AT11 UPPER WING (R01U17)

MACH (4) = 1.240 ALPHA(1) = -7.940

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000	-.1566	-.1952	.1018	.3690	.2759	.2187	.2521
.020				.5267	.4768	.5050	.3009
.025			.3128				
.040		-.0662					
.045			.2840				
.050	-.1204			.5695	.3773	.3750	.3640
.080				.2324			
.085			.1886				
.086		.0469					
.094	-.0704			.0457	.0813	.1163	.1164
.150		.0003					
.163			.0914				
.177		.1273					
.193	-.0390						
.229				-.2107	-.0848	-.0791	-.1064
.250			-.0179				
.274		-.0269					
.339	.0798						
.362				-.3559	-.2370		-.2996
.400			-.1788				
.402	.0000						
.497				-.0364	-.1821		
.557			.0000				
.565							
.600							
.650							
.720	.1.45						
.725				-.0367			
.750							
.760		.0118					
.775				-.0077	.0597		
.808			.0697				
.834	.0437						
.850				-.0536	.0563	.0778	
.857		.0607					
.865	.0974						
.900	.1000			-.0437			.1635
.925			.0894		.0816		
.950				.0023			.1502
.953		.1124					
.965	.0958						



TABULATED PRESSURE DATA - IA14A - VOL. 6

DATE 10 JAN 75

ARC11-716 IA14 OA+12+S12N23+AT11 UPPER WING (RB1U17)

MACH (4) = 1.244 ALPHA(3) = .050

SECTION (1) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.1282	-.1179	.3679	.6015	.5435	-.4879	.5171
.020				.3193	.5055	.2878	.2895
.040			.2113				
.060		-.0423					
.080			.1257				
.100	-.1942			.1966	.0960	.0745	.0732
.120				-.0488			
.140			.0159				
.160		.0199					
.180	-.1873			-.2317	-.2478	-.1986	-.1691
.200		.0000					
.220			-.1287				
.240		.0080					
.260	-.0961						
.280				-.4002	-.3617	-.3600	-.3566
.300			-.2568				
.320		-.1697					
.340	.0067						
.360				-.4933	-.4746		-.5229
.380		-.3533					
.400				-.5100	-.4875		
.420		.0000					
.440			.0000				
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.990							

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(R01U17)

ARC11-716 1A14 OR-T12-S12N3-AT11 UPPER WING

MACH (4) = 1.249 ALPHA(4) = 4.010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.020	-.1269	-.2138	.3567	.6458	.6094	.5669	.6082
.020			.0946	.1419	.1678	.1125	.0783
.025		-.0986					
.040							
.045			-.0041				
.050	-.1815			-.0554	-.0654	-.0938	-.1361
.060				-.2003			
.085			-.1390				
.086		-.0396					
.094	-.1829						
.130				-.3892	-.3847	-.3446	-.3308
.163		.0000					
.177		-.0760	-.2476				
.193							
.229	-.1182						
.250				-.5936	-.5104	-.4855	-.4696
.274			-.3239				
.339		-.2443					
.362	-.0304						
.400				-.6135	-.5960		-.5975
.402		-.4151					
.497	.0000						
.550				-.6303	-.5970		
.565		.0000					
.600							
.650							-.5890
.700	-.3349				-.2010		
.725				-.2913			
.750							
.760		-.1456					
.775				-.2522	-.3344		
.800		-.0489					
.834	-.0278						
.850				-.2061	-.2042	-.4223	
.857		-.0427					
.865	-.0009						
.900	-.0120			-.1720			.4477
.925		-.0440			-.1294		
.930				-.1406			-.3803
.935		-.0503					
.965	-.0917						



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI4A - VOL. 6

(RBLU17)

ARC11-716 IAI4 01-112-512N25-AT11 UPPER WING

MACH (4) = 1.249 ALPHA(5) = 7.930

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

X/CW Y/SW .2990 .3640 .4270 .5340 .6730 .7600 .8870

.000	-.1947	-.3104	.2667	.6329	.6330	.3741	.5450
.020				-.0385	.0157	-.0764	-.1673
.025			-.0674				
.040		-.2277					
.045			-.1625				
.050	-.1080		-.2105	-.2225	-.2687	-.3582	
.080			-.3447				
.085			-.2807				
.086		-.1422					
.094	-.1945						
.150		.0000		-.5257	-.5072	-.4906	-.4988
.163			-.3743				
.177		-.1662					
.193	-.1774			-.6583	-.6149	-.6029	-.5928
.229			-.4331				
.274		-.3123					
.339	-.0633			-.6828	-.6973		-.6869
.352			-.4594				
.400				-.6095	-.6761		-.6613
.497	.0000		.0000			-.6730	
.550				-.2320			
.565				-.2940			
.600						-.6137	-.6308
.650			-.2372	-.2478	-.5864		
.700	-.4023		-.1367				
.725							
.750				-.2500	-.5199	-.5480	
.760			-.1041				
.775				-.2330			-.3101
.808	-.0341			-.0819	-.3487		-.4656
.834				-.2073			
.850				-.0898			
.857							
.865	-.0099						
.900	-.0317						
.905							
.950							
.953							
.965	-.0997						

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## DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U18)

ARC11-716 1A14 01+112+512M25+AT11 UPPER WING

MACH ( 1 ) = .898 CETAO ( 1 ) = -8.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5140 .6730 .7800 .8370

X/CW

.930 -.0017 .1096

.953 .0250

.965 .0228

MACH ( 1 ) = .898 BETAO ( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.950 .0173 .1337 .4275 .5682 .5213 .4772 .4821

.920 .020 .025 .1899 .1791 .0000 .0710 .0179

.945 .040 .1877 .0792

.945 .050 -.0006 .0792 -.0824 -.1544 -.1969 -.2993

.980 .080 -.2876

.985 .085 -.1038

.986 .086 .1840

.994 .0144

.150 .0000 .0000

.163 .177 -.3415

.193 .0527

.229 .0930

.250 .274 -.4541

.339 -.4116

.362 -.0147

.400 .402 -.6591 -.8594 -1.0060

.497 .0000

.550 .550 .3020 -.4549

.565 .0000

.600 .650 .0000

.700 -.0997

.725 .750 .0826 -.1777 -.2538

.760 .775 .0834 .0901 .0993

.775 .0762

.938 .934 -.0649

.850 .850 .0665 .0932 .1416

(R81U19)

ARC11-716 1A14 01+12+512M25+AT11 UPPER WING

MACH ( 1 ) = .898 BETAO ( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000			.0563				
.025	.0716						
.050	.0374		.0551				.0767
.075		.0432		.0526			
.100	.0195				.0880		
.125	.0006						
.150							
.175							
.200							
.225							
.250							
.275							
.300							
.325							
.350							
.375							
.400							
.425							
.450							
.475							
.500							
.525							
.550							
.575							
.600							
.625							
.650							
.675							
.700							
.725							
.750							
.775							
.800							
.825							
.850							
.875							
.900							
.925							
.950							
.975							
.1000							

MACH ( 1 ) = .897 BETAO ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000			.3582	.4822	.4233	.3787	.3921
.025			.1679	.0000	.0798	.0281	
.050			.1637				
.075		.1408					
.100		.0672					
.125							
.150							
.175							
.200							
.225							
.250							
.275							
.300							
.325							
.350							
.375							
.400							
.425							
.450							
.475							
.500							
.525							
.550							
.575							
.600							
.625							
.650							
.675							
.700							
.725							
.750							
.775							
.800							
.825							
.850							
.875							
.900							
.925							
.950							
.975							
.1000							



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01-712-912N25+AT11 UPPER WING (R81U10)

MACH ( 1 ) = .897 BETAO ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.775 .708 .634 -.1961 .650 .657 .665 -.0513 .900 -.0213 .905 .950 .955 .965 -.0533

.0073

.0037 .0806 .1166

.0027 .0272 .0586

.1094

MACH ( 1 ) = .896 BETAO ( 4 ) = 4.103

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000 -.1224 .0116 .2436 .4166 .3356 .2864 .3059

.020 .025 .040 .045 .050 .080 .085 .086 .094 .150 .163 .177 .193 .229 .250 .274 .359 .362 -.0160 .400 .402 .497 .550 .565 .600 .650

.1160

.0777

.0474

.0570

-.1090

-.1447

-.2464

-.2206

-.0717

.0965

-.0290

.0000

-.2486

-.0324

-.3285

-.2800

-.4086

-.5420

-.8875

-.3940

.0000

-.3721

-.5346

.0000

-.6755

-.6452

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01US0)

ARC11-716 1A14 01-712-812N25-AT11 UPPER WING

MACH ( 1 ) = .806 BETAO ( 4 ) = 4.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .9670

X/CW						
.700	-.2136		-.4137		-.2712	
.725					-.1626	-.1171
.750						
.760		-.3603		-.2413	-.1691	
.775						
.806		-.2178				
.834	-.3461			-.0446	-.0416	.0011
.850						
.857				-.1076		
.865	-.2652					
.900	-.1605			-.0244	.0054	.0446
.905						
.930		-.0603		-.0156		.0169
.953		-.0566				
.965	-.1111					

MACH ( 1 ) = .806 BETAO ( 5 ) = 0.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .9670

X/CW						
.002	-.2535	-.0228	.0690	.2432	.2799	.1963
.060				.1003	.0000	.0622
.025			.0325			
.040		.0106				
.045						
.050	-.1360		-.0007	-.0366	-.0999	-.1090
.060				-.1539		
.065			-.0499			
.066		.0131				
.094	-.0699			-.2997	-.3462	-.3696
.120		.0000				
.163			-.1628			
.177		-.0206				
.193						
.229	.0316			-.3725	-.4752	-.5621
.250						
.274			-.2436			
.339	-.2062					
.362	-.0190					
.400				-.3706	-.5035	-.7623
.402						
.497	.0020			-.3114		



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBU101)

ARC11-716 1A14 OI+T12+312M3+AT11 UPPER WING

MACH ( 1 ) = .890 BETAO ( 5 ) = 0.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.550	-.3985	-.5342				
.565	.0000					-.6980
.600						-.6584
.650						
.700	-.2481					
.725						
.750						
.760						
.775						
.808						
.834	-.3127					
.850						
.857						
.865						
.900	-.1874					
.905						
.930						
.935						
.965						

MACH ( 2 ) = .970 BETAO ( 1 ) = -0.050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .9670

X/CW

.000	-.0799	.2109	.3367	.7019	.6666	.6277	.6229
.020				.2389	.2042	.1522	.1109
.025				.2497			
.040							
.045							
.050	-.0760						
.080							
.085							
.086							
.094							
.150							
.165							
.177							
.193							
.229							
.250							
.274							
.339							

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U10)

ARC11-716 1A14 01+712+812N23+AT11 UPPER WING

MACH ( 2 ) = .976 BETAO ( 1 ) = -0.050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.0000	.362	.0902
.0000	.400	-.0364
.0000	.402	-.5712
.0000	.497	.0000
.0000	.590	-.2076
.0000	.585	-.7251
.0000	.600	.0000
.0000	.650	-.3391
.0000	.700	-.0162
.0000	.725	-.1242
.0000	.750	-.3967
.0000	.780	-.4280
.0000	.775	.0729
.0000	.808	.1016
.0000	.834	.1474
.0000	.850	.1336
.0000	.857	.0661
.0000	.865	.1368
.0000	.900	-.2037
.0000	.905	.1826
.0000	.930	.1336
.0000	.950	.0524
.0000	.953	.1113
.0000	.963	-.0140
.0000	.965	-.0184
.0000	.965	.0590

MACH ( 2 ) = .976 BETAO ( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.0000	.362	.0902
.0000	.400	.6368
.0000	.402	.5892
.0000	.497	.2631
.0000	.590	.2148
.0000	.585	.1694
.0000	.600	.1172
.0000	.650	.2673
.0000	.700	.1598
.0000	.725	.0106
.0000	.750	-.0454
.0000	.780	-.1116
.0000	.775	-.1728
.0000	.808	-.0117
.0000	.834	.2511
.0000	.850	.0512
.0000	.857	-.3966
.0000	.865	-.4502
.0000	.900	-.4433
.0000	.905	-.4397
.0000	.930	-.2281
.0000	.950	-.0000
.0000	.953	.0393
.0000	.963	.0393
.0000	.965	.0393



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 OR-T12-S12N85+AT11 UPPER WING

(RB1U18)

MACH ( 2 ) = .976 BETAO ( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.289	.1683					
.290						
.274						
.339						
.362	.0762					
.400						
.402						
.497	.0000					
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.925						
.950						
.953						
.965						

MACH ( 2 ) = .975 BETAO ( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.2952	.1311	.4262	.5517	.4871	.4429	.4593
.020				.2683	.2136	.1831	.1340
.025				.2513			
.040			.2114				
.045			.1605				
.050	.0067			.0337	-.0198	-.0726	-.1380
.080							
.085							
.086							
.094	.0427		.0169				



(R81U16)

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

$\alpha_{CH} (2) = .976$   $\beta_{TAO} (4) = 4.079$

SECTION ( ) UPPER WING

**DEPENDENT VARIABLE CP**

Y/8W	.2993	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

AS/X

[illegible]

$$\text{WACH} (2) = .974 \quad \text{DETAO} (5) = 8.120$$

SECTION, NUMBER AND  
DEPENDENT VARIABLE OF

Y/8W	.2945	.3640	.4270	.5340	.6730	.7800	.8670
------	-------	-------	-------	-------	-------	-------	-------

**NO/K**

Year	1992	1996	1997
0.020	.1692	.1670	.1297

**.343** **.0343**

Variable	Mean	Standard Deviation	Minimum	Maximum
Age	35.2	12.5	18	65
Gender	0.52	0.50	0	1
Marital Status	0.68	0.47	0	1
Education	12.8	1.5	9	16
Income	45000	15000	20000	80000
Health	0.75	0.43	0	1
Exercise	0.35	0.48	0	1
Stress	0.65	0.48	0	1
Depression	0.25	0.44	0	1
Life Satisfaction	0.70	0.46	0	1

	1960	1961
... ..	.085	.5169

.794 -.1617

	1960	1961	1962	1963	1964	1965	1966	1967	1968	1969	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	2101	2102	2103	2104	2105	2106	2107	2108	2109	2110	2111	2112	2113	2114	2115	2116	2117	2118	2119	2120	2121	2122	2123	2124	2125	2126	2127	2128	2129	2130	2131	2132	2133	2134	2135	2136	2137	2138	2139	2140	2141	2142	2143	2144	2145	2146	2147	2148	2149	2150	2151	2152	2153	2154	2155	2156	2157	2158	2159	2160	2161	2162	2163	2164	2165	2166	2167	2168	2169	2170	2171	2172	2173	2174	2175	2176	2177	2178	2179	2180	2181	2182	2183	2184	2185	2186	2187	2188	2189	2190	2191	2192	2193	2194	2195	2196	2197	2198	2199	2200	2201	2202	2203	2204	2205	2206	2207	2208	2209	2210	2211	2212	2213	2214	2215	2216	2217	2218	2219	2220	2221	2222	2223	2224	2225	2226	2227	2228	2229	2230	2231	2232	2233	2234	2235	2236	2237	2238	2239	2240	2241	2242	2243	2244	2245	2246	2247	2248	2249	2250	2251	2252	2253	2254	2255	2256	2257	2258	2259	2260	2261	2262	2263	2264	2265	2266	2267	2268	2269	2270	2271	2272	2273	2274	2275	2276	2277	2278	2279	2280	2281	2282	2283	2284	2285	2286	2287	2288	2289	2290	2291	2292	2293	2294	2295	2296	2297	2298	2299	2300	2301	2302	2303	2304	2305	2306	2307	2308	2309	2310	2311	2312	2313	2314	2315	2316	2317	2318	2319	2320	2321	2322	2323	2324	2325	2326	2327	2328	2329	2330	2331	2332	2333	2334	2335	2336	2337	2338	2339	2340	2341	2342	2343	2344	2345	2346	2347	2348	2349	2350	2351	2352	2353	2354	2355	2356	2357	2358	2359	2360	2361	2362	2363	2364	2365	2366	2367	2368	2369	2370	2371	2372	2373	2374	2375	2376	2377	2378	2379	2380	2381	2382	2383	2384	2385	2386	2387	2388	2389	2390	2391	2392	2393	2394	2395	2396	2397	2398	2399	2400	2401	2402	2403	2404	2405	2406	2407	2408	2409	2410	2411	2412	2
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**Est. 4,190,000**

**.259**      **-1.347**      **-1.442**      **-1.343**

**656** **-1158**

600°  
-2.502  
-1.000

467 .0000  
- 2428 - 1000

[illegible][illegible]

	.723	- .3104
		- .4475 - .3912

763	-3330
764	-3078
765	-4071

1000

**.930**

**- .2775**

**.657**

900	- 2478	- .2109	- .1480
905	- 1296		

.999		-1.471	-1.395
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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U18)

ARC11-716 1A14 01-112-312N23-AT11 UPPER WING

DATE 10 JAN 75

MACH (3) = 1.102 BETA (1) = -0.090

SECTION (1) UPPER WING

DEPENDENT VARIABLE CF

Y/B	X/CW	0.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	.0182	.0906	.5611	.7416	.7198	.6944	.7046	
.020			.2839	.3080	.3139	.2834	.2569	
.040		.1071						
.060	.1230		.1608	.0350	.0478	.0066	-.0284	
.080				-.1430				
.100			-.0351					
.120	.2041							
.140	-.1134			-.3618	-.3661	-.3041	-.2873	
.160	.0000							
.180	.177		-.2221					
.200	.193		.0134					
.220	-.0465			-.5339	-.5241	-.5155	-.4878	
.240	.274		-.3446					
.260	.339	-.2831						
.280	.362	.0629		-.5855	-.6396		-.6836	
.300	.400		-.4545					
.320	.432	.0000		-.5311	-.6764			
.340	.497	.0000						
.360	.550		.0000				-.6614	
.380	.565				-.6679			
.400	.600			-.1670	-.2280			
.420	.650					-.4400	-.5893	
.440	.720	-.0373						
.460	.725		-.1300					
.480	.750		-.0251	-.0844				
.500	.760		.0432					
.520	.775			.0492	.1292	-.3342		
.540	.808	-.0227						
.560	.834		.0024					
.580	.850							
.600	.865	.0075		.1048	.1588		-.3226	
.620	.900	.0928	.1634					
.640	.905		.1026					
.660	.950		.1532					
.680	.955							
.700	.965	.1478						

(R011110)

ABC11-716 1A14 04+T12+S12N25+AT11 UPPER WING

WACH ( 3 ) = 1.190  
BETA0 ( 2 ) = -4.010

**SECTION ( UPPER WING**

DEPENDENT VARIABLE: GP

[illegible]

**ADIX**

0000	-.0654	-.0628	.4826	.6641	.6152	.5670	.6802
0020				.5048	.2860	.2685	.2366
0025		.0353	.2622				
0045			.1567				
0060	-.1909			.0692	.0435	.0059	-.0275
0080				-.1164			
0085			-.0066				
0086		.1349					
0094	-.1570						
0150		.0000		-.3829	-.3536	-.3001	-.2677
0163			-.1755				
0177		.0210					
0193							
0229	-.0705			-.5227	-.4816	-.4900	-.4894
0290			-.2856				
0274		-.2273					
0339							
0362	.0552						
0400			-.4430	-.6168	-.6072		-.6737
0402							
0497	.0000			-.2605	-.6261		
0550			.0000				
0565							-.6747
0600						-.6238	
0650					-.2010		
0700	.0034			-.1946		-.4176	-.4672
0725							
0750			-.1399	-.0930	-.0337		
0760							
0775			-.0235				
0808							
0834	-.0682			-.0374	.0560	-.1646	
0850			-.0300				
0857							
0865	.0803			.0525	.1041		-.3166
0900	.0029		.0232				
0925				.0757		.1516	
0950							
0953			.0796				
0963	.0396						



(R81U18)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

DATE 10 JAN 75

MACH (3) = 1.102 BETAO (3) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/B	.2990	.3640	.4270	.5340	.6730	.7802	.8870
X/CW							
.000	-.1803	-.0609	.4190	.5714	.5000	.4823	.5394
.020				.2897	.2579	.2619	.2435
.040		.0459	.2383				
.060			.1504				
.080	-.2497		.0777	.0443	.0241	-.0026	
.100			-.0876				
.120		.0151					
.140		.1628					
.160	-.1865		-.2837	-.2971	-.2642	-.2619	
.180		.0700					
.200		.0426	-.1398				
.220	-.0559		-.2369	-.4005	-.4250	-.4413	-.4624
.240		-.1928					
.260	.0675			-.5053	-.5445		-.6386
.280			-.3957				
.300	.0000		.0020	-.1166	-.3026		-.6277
.320					-.5030		
.340	.0270			-.1314			
.360			-.1482	-.1048	-.1010	-.0353	-.2654
.380		-.0715					
.400		-.0018		-.1037	-.0187	.0736	
.420	-.0994						-.0679
.440		-.0260		-.0609		.0557	
.460	-.0554		-.0482	.2136		.1246	
.480			-.0148				
.500		-.0613					

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DATE 19 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2491

ARC11-716 1A14 01+T12+512N23+AT11 UPPER WING

(RB1U18)

MACH (3) = 1.100 BET/J (5) = 0.130

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3840 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.3700	-.0706	.0983	.2802	.3730	.3243	.3524
.020				.1862	.2288	.2257	.2217
.025			.0701				
.040		-.0176					
.045			.0589				
.050	-.3448			.0698	.0737	.0314	.0306
.080				-.0189			
.085			.0261				
.086		-.0447					
.094	-.2122						
.150		.0000		-.1396	-.1819	-.1638	-.1837
.163							
.177		.0838	-.0532				
.193							
.229	.0426			-.2501	-.2861	-.2965	-.3667
.250							
.274		-.1156	-.1236				
.339							
.362	.0568			-.1467	-.2104		-.4495
.400			-.1323				
.402				-.1240	-.2043		
.497	.0000		.0000				
.550							
.565							
.600							
.650							
.700	.0106			-.2209	-.2219	-.3209	-.3798
.725							
.750			-.2047			-.2541	-.2253
.760				-.1904	-.2147		
.775			-.1602				
.808							
.834	-.1686			-.2103	-.1972	-.1446	
.850			-.1797				
.857							
.865	-.1386						
.900	-.1501		-.2711				-.0079
.905			-.1596		-.1188		
.950			-.1743			-.0265	
.953			-.1 15				
.965	-.1699						



ARC11-716 1A14 Q1+712+912N23+AT11 UPPER WING

(R31V18)

WACH ( 4 ) = 1.244 BETA0 ( 2 ) = -4.030

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP					
Y/BW	.2990	.3640	.4270	.5340	.6730	.7000	.8070
X/CW							
.000	-.0558	-.0414	.4148	.6883	.6395	.5763	.6755
.020			.2594	.3975	.3748	.3503	.3552
.025		-.0026					
.045			.1586	.1613	.1434	.1172	.1165
.050	-.0973			-.0151			
.080			.0367				
.086		.0462					
.094	-.1285			-.2339	-.2332	-.1744	-.1439
.150		.0000					
.163			-.1457				
.177		.0127					
.193							
.229	-.0810		-.2553	-.4832	-.3757	-.3641	-.3431
.250							
.274		-.2016					
.339			-.3699	-.5071	-.5101		-.5133
.362	.0216						
.400							
.402							
.457	.0000						
.550			.0000	-.5454	-.5412		
.565							
.600							
.650							-.5432
.700	-.2886					-.5463	
.750				-.2287	-.1822		
.790			-.0905			-.5280	-.4771
.760				-.1546	-.2989		
.775			.0435				
.806	-.0065						
.834							
.850							
.857			.0371	-.0701	-.1973	-.3025	
.865	.0635						
.900	.0680		.0366	-.0308			-.3655
.905							
.950				-.0044	-.0840	-.2484	
.953			.0165				
.965	-.0023						

(R01U18)

ARC11-71C 1A14 01+712+512M25+AT11 UPPER WING

MACH (4) = 1.249 BETAO (3) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.000	-.1173	.3633
.020	-.1121	.6007
.040		.5399
.060		.4837
.080		.3205
.100	.3205	.3057
.120	.2219	
.140		
.160	-.0459	.1353
.180		.1283
.200		.1106
.220		.0884
.240		.0690
.260		-.0336
.280	.0392	
.300		
.320	.0230	
.340		
.360	-.1195	-.2183
.380		-.2266
.400		-.1833
.420		-.1539
.440		
.460		
.480		
.500		
.520		
.540		
.560		
.580		
.600		
.620		
.640		
.660		
.680		
.700		
.720		
.740		
.760		
.780		
.800		
.820		
.840		
.860		
.880		
.900		
.920		
.940		
.960		
.980		
.995		



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2433

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

(R81018)

MACH (4) = 1.246 BETAO (4) = 4.090

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2993	.3640	.4270	.5340 .6730 .7600 .8870
X/CW				
.000	-.2320	-.1344	.2681	.5139 .4430 .4012 .4833
.025			.1514	.2770 .2620 .2706 .2743
.050		-.0217		
.075			.0960	
.100	-.2818			.0903 .0736 .0764 .0740
.125			.0028	-.0439
.150		.0232		
.175	-.2149			-.1910 -.2143 -.1663 -.1518
.200		.0000		
.225		.0334	-.0891	
.250	-.0678			
.275			-.1699	
.300		-.1430		-.3280 -.3236 -.3176 -.3342
.325	.0419			
.350			-.3164	-.4156 -.4183
.375	.0000			
.400			.0000	-.3917 -.4348
.425				
.450				-.4869
.475	-.0960		-.1640	
.500				-.0845
.525				
.550				-.0981
.575				-.0115 -.0623
.600				
.625				-.0039
.650				
.675	-.0562			-.0301 -.0103 -.1595
.700				
.725				-.0263
.750				
.775				-.0622
.800				-.0202
.825				-.0636
.850				
.875				-.0250
.900				
.925				-.1749
.950				.0416 .0804
.975				
.995				
.965	-.0314			

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U18)

ARC11-716 1A14 01+112+512M25+AT11 UPPER WING

MACH ( 4 ) = 1.248 BETAO ( 5 ) = 0.150

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
X/CW			
.000	-.4841	-.0935	.1065
.020			.3905
.025			.2145
.040			.0798
.045		-.0436	.0586
.050	-.3583		.0811
.080			-.0267
.085			.0206
.086		-.0290	
.094	-.2307		-.1326
.150			-.1836
.165		.0000	-.1694
.177		-.0695	-.1647
.193		.0662	
.229	.0390		-.2853
.250			-.2779
.274		-.1468	-.3080
.339		-.1434	-.3523
.352	.0560		
.400		-.2710	-.4767
.402			
.497	.0000		
.550		-.1554	-.3704
.585		.0000	
.600			-.4714
.650			-.4252
.700	-.0925	-.1116	
.725			-.1795
.750			-.3522
.760		-.1201	
.775		-.0492	-.0622
.808		-.0470	
.834	-.0626		-.0668
.850		-.0632	-.0448
.857			-.0496
.865	-.0335		
.900	-.0478		-.1157
.905		-.0776	
.905		-.0235	
.950		-.0650	-.0235
.955		-.0814	.0615
.965		-.0683	
.965	-.0951		

(R01V24) ( 27 SEP 73 )

### PARAMETRIC DATA

ALPHA0 :	-10.000	ELEVON :	.000
BUNDEF :	.000	SPDRBK :	.000

## REFERENCE DATA

YREF =	2.4210 30.FT.	YMRP =	29.9900 INCHES
ZREF =	38.7090 INCHES	ZMRP =	.0000 INCHES
SCALE =	38.7090 INCHES		.0000 INCHES
	.0000 SCALE		

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MACH ( 1 ) = .902   BETAO ( 1 ) = -9.090

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DEPENDENT VARIABLE CF

SECTION ( UPPER WING

	1967	1968	1969	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	2101	2102	2103	2104	2105	2106	2107	2108	2109	2110	2111	2112	2113	2114	2115	2116	2117	2118	2119	2120	2121	2122	2123	2124	2125	2126	2127	2128	2129	2130	2131	2132	2133	2134	2135	2136	2137	2138	2139	2140	2141	2142	2143	2144	2145	2146	2147	2148	2149	2150	2151	2152	2153	2154	2155	2156	2157	2158	2159	2160	2161	2162	2163	2164	2165	2166	2167	2168	2169	2170	2171	2172	2173	2174	2175	2176	2177	2178	2179	2180	2181	2182	2183	2184	2185	2186	2187	2188	2189	2190	2191	2192	2193	2194	2195	2196	2197	2198	2199	2200	2201	2202	2203	2204	2205	2206	2207	2208	2209	2210	2211	2212	2213	2214	2215	2216	2217	2218	2219	2220	2221	2222	2223	2224	2225	2226	2227	2228	2229	2230	2231	2232	2233	2234	2235	2236	2237	2238	2239	2240	2241	2242	2243	2244	2245	2246	2247	2248	2249	2250	2251	2252	2253	2254	2255	2256	2257	2258	2259	2260	2261	2262	2263	2264	2265	2266	2267	2268	2269	2270	2271	2272	2273	2274	2275	2276	2277	2278	2279	2280	2281	2282	2283	2284	2285	2286	2287	2288	2289	2290	2291	2292	2293	2294	2295	2296	2297	2298	2299	2300	2301	2302	2303	2304	2305	2306	2307	2308	2309	2310	2311	2312	2313	2314	2315	2316	2317	2318	2319	2320	2321	2322	2323	2324	2325	2326	2327	2328	2329	2330	2331	2332	2333	2334	2335	2336	2337	2338	2339	2340	2341	2342	2343	2344	2345	2346	2347	2348	2349	2350	2351	2352	2353	2354	2355	2356	2357	2358	2359	2360	2361	2362	2363	2364	2365	2366	2367	2368	2369	2370	2371	2372	2373	2374	2375	2376	2377	2378	2379	2380	2381	2382	2383	2384	2385	2386	2387	2388	2389	2390	2391	2392	2393	2394	2395	2396	2397	2398	2399	2400	2401	2402	2403	2404	2405	2406	2407	2408	2409	2410	2411	2412	2413	2414	2415	2416	2417	2418	2419	2
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**MJ/CW**

.090	.0334	.0240	.2687	.3275	.2186	.1037	.1340
.020				.6113	.4917	.5903	.5513
.040		.2607	.5013				
.045			.4402				
.090	.1362			.4165	.4259	.4306	.3813
.080				.2541			
.085			.3115				
.086		.3537					
.094	.1319			.0302	.0530	.0709	.0357
.150		.0000					
.177			.0913				
.193		.2526					
.229	.2323						
.250							
.274							
.339		-.0016	-.0328	-.1336	-.1440	-.1787	-.2496
.362	.2175						
.400			-.0635	-.0984	-.1765		-.3238
.402							
.497	.0000						
.550							
.565							
.600							
.650			.0000	-.0424	-.1154		-.1737
.700	.1400						
.725							
.750				-.1529			
.760							
.775			.0041				
.808			.1634	.1710	.1468		
.834	.0826					.1496	.0842
.850				.1200	.1444	.1516	
.897			.1327				
.865	.1980						
.900	.1452			.1096			.1193
.905			.1028		.0975		



(R01024)

ARC11-716 IAL14 Q1+T12+S12H25+AT10 UPPER WING

MACH ( 1 ) = .902 BETAO ( 1 ) = -9.890

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.950	.0310	.1199
.953	.0402	
.965	.0340	

MACH ( 1 ) = .899 BETAO ( 2 ) = 10.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1145	-.0055	.0183	-.0461	-.3235	-.4090	-.4973
.020			.3288	.4633	.3467	.2804	.1845
.025		.0812					
.040			.3266				
.045				.3873	.3189	.2633	.1669
.050	-.0112			.2939			
.080			.2784				
.085		.1897					
.086							
.094	.0286			.1320	.0984	.0638	-.0426
.150		.0000	.1323				
.163							
.177		.1785					
.193							
.229	.1096			-.0371	-.0779	-.1328	-.2700
.250			.0210				
.274		-.0174					
.339	.0887						
.362				-.1553	-.2270		-.4400
.400			-.1320				
.402			.0000	-.2834	-.3610		
.497	.0000						
.550							
.565							
.600							
.630							
.700	-.1836						
.725				-.3713	-.2425		
.750							
.760							
.775				-.3521	-.1622	-.1574	
.808							
.834	-.2807						
.850				-.1032	-.0414	-.0044	



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+S12N23+AT10 UPPER WING  
 (R01024)

MACH ( 1 ) = .699 BETA0 ( 2 ) = 10.090		DEPENDENT VARIABLE CP	
SECTION ( 1 ) UPPER WING			
V/BW			
.2990	.3640	.4270	.5340
		.6730	.7500
			.8670
X/CW			
.897		-.1930	
.865	-.1932		
.900	-.1676	-.0643	.0486
.905		.0035	
.920		-.1292	.0371
.923		-.0626	
.965	-.1715	-.1446	



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01\*112\*512\*25\*AT10 UPPER WING  
 (RBLU25)

MACH ( 1 ) = .899 BETAO ( 1 ) = -9.930

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW  
 .930 .0661 .1266  
 .933 .0510  
 .945 .0311

MACH ( 1 ) = .899 BETAO ( 2 ) = 10.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW  
 .000 -.1205 .0186 .0484 .0232 -.2020 -.3004 -.3846  
 .020 .4285 .2571 .2860 .2020  
 .025 .2755  
 .040 .0725 .2684  
 .045 .3245 .2735 .2273 .1386  
 .050 -.0337 .2286  
 .060 .2234  
 .065 .1568  
 .086 .0108  
 .094 .0572 .0289 .0023 -.0950  
 .150 .0000  
 .163 .0822  
 .177 .1902  
 .193 .0943  
 .229 -.0757 -.1484 -.1963 -.3316  
 .250 .0327  
 .274 -.0415  
 .339 .0631  
 .362 -.1808 -.2011 -.4934  
 .400  
 .402 -.1703  
 .497 .0000  
 .550  
 .565 .2993 -.4010  
 .603  
 .650  
 .700 -.1993  
 .725  
 .750  
 .760  
 .775  
 .808  
 .834 -.3076  
 .850  
 .3928  
 .2560  
 .5471  
 .2740 -.1350  
 .2127 -.2443  
 .1977  
 .0917 -.0566 -.0333

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(081025)

ARC11-716 1A14 C1+T12+S12N23+AT10 UPPER WING

MACH ( 1 ) = .898 BETAO ( 2 ) = 10.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W .8990 .9640 .4270 .5340 .6730 .7800 .8870

X/CW

.897

-.1710

.865

-.2247

.900

-.1719

-.0515

.0481

.905

-.1362

.0023

.990

-.0531

.0393

.953

-.1390

.963

-.1571



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2403

ARC11-716 1A14 01-112-312MS+AT10 UPPER WING (R01UR6) ( 27 SEP 75 )

## REFERENCE DATA:

SREF = 2.4210 SQ.FT. XMRP = 29.9000 INCHES  
 LREF = 38.7090 INCHES YMRP = .0000 INCHES  
 BRFP = 38.7090 INCHES ZMRP = .0000 INCHES  
 SCALE = .0300 SCALE

## PARAMETRIC DATA

ALPHA = -6.000 ELEVON = .000  
 RUDDER = .000 SPOBRK = .000

MACH ( 1 ) = .897 BETA0 ( 1 ) = -9.940

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
.2990	.3640	.4270	.5340 .6750 .7800 .8870
X/CU			
.000	.0901	.1504	.4214 .4969 .4207 .3362 .3682
.020			.4693 .4222 .4596 .4430
.025		.4115	
.040		.2774	
.045		.3108	
.050	.1069		.2189 .2191 .2227 .1783
.060			.0259
.085		.1489	
.086		.3065	
.094	.1117		
.150			-.2015 -.1977 -.1764 -.1876
.163	.0000		
.177		-.1145	
.193	.1196		
.229	.1793		
.250			-.3448 -.4066 -.4519 -.4904
.274		-.2208	
.339	-.1747		
.362	.1132		
.400		-.2100 -.3359	-.6086
.432		-.1930	
.497	.0000		
.550		-.0687 -.1732	
.565		.0000	
.600			-.1990
.650	.0455		-.1126
.725		-.1637	
.750		-.0004	
.760		.0084	.1355 .0509
.775		.1660	.1410
.806		.1980	
.834	.0685		
.850		.1256 .1420	.1506
.857	.1308		
.865	.1793		
.900	.1425	.1164	.1237
.905		.1020	.0904

(R81026)

ARC11-716 1A14 OL+T12+S12N25+AT1D UPPER WING

MACH ( 1 ) = .897 BETAO ( 1 ) = -9.940

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .950 .953 .965 .0344 .0661 .1222

MACH ( 1 ) = .896 BETAO ( 2 ) = 10.070

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .000 -.1770 .0203 .0613 .0875 -.0703 -.2472 -.2092  
 .020 .025 .040 .045 .050 .060 .065 .066  
 .094 .120 .163 .177 .193 .229 .250 .274  
 .339 .362 .400 .402 .487 .590 .593 .650  
 .700 .723 .730 .760 .775 .800 .834 .850  
 .0226 .2143 .2559 .2195 .1793 .0927  
 .1460 .1781 .1342 -.0043 .0000 .1165  
 -.0740 -.0644 -.2196 -.3256 -.5320  
 -.2012 -.3176 -.4316 .0000 -.6341  
 -.2556 -.3963 -.3741 -.2727 -.3132  
 -.2535 -.1184 -.0829 -.0635



(R01026)

TABULATED PRESSURE DATA - IA14A - VOL. 6

ARC11-716 IA14 Q1+T12+312N25+AT10 UPPER WING

DATE 10 JAN 75

MACH ( 1 ) = .800 BETA ( 2 ) = 10.070

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CF

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

-.1804

.837

.865

.900

.935

.953

.965

.0044

.0004

-.0633

-.0332

-.1398

-.1353

-.1750

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(R01027) ( 27 SEP 73 )

ARC11-716 1A14 01+T12+S12N23+AT10 UPPER WING

## PARAMETRIC DATA

ALPHAO =	-4.000	ELEVON =	.000
BUDDER =	.000	SPDRK =	.000

## REFERENCE DATA

3REF =	2.4210 30.FT.	XMAP =	20.9000 INCHES
4REF =	30.7090 INCHES	YMAP =	.0000 INCHES
5REF =	30.7090 INCHES	ZMAP =	.0000 INCHES
SCALE =	.0000 SCALE		

WACH ( 1 ) = .099      BETAO ( 1 ) = -9.990

SECTION (1) UPPER WING

## DEPENDENT VARIABLE CP

Y/BW	X/C4	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000		.0977	.1888	.4715	.5821	.5246	.4820	.4766
.020				.3467	.3780	.3748	.3475	.3319
.040			.2716					
.045				.2317				
.050		.0889			.1016	.0877	.0680	.0364
.060					-.1083			
.085				.0447				
.086			.2780					
.094		.0933						
.130			.0000		-.3413	-.3486	-.3269	-.3191
.163				-.2213				
.177			.0326					
.193								
.229		.1577			-.4734	-.5373	-.5867	-.6001
.230				-.3206				
.274			-.2635					
.339								
.362		.0287			-.2536	-.3571		-.6228
.400				-.2744				
.452								
.497		.0000			-.1106	-.1859		-.0986
.530				.0000				
.563								
.600								
.690								
.700		.0668			.0068			
.723						-.1569		
.750				.0103			.1400	.0546
.760					.1638	.1448		
.775				.1567				
.808								
.834		.0687						
.850				.1312	.1308	.1417	.1496	
.857								
.865		.1840						.1170
.900		.1469			.1167		.0942	
.903				.1037				



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2467

ARC11-716 1A14 01+712+812N25+A710 UPPER WING

(R81027)

MACH ( 1 ) = .899 CETAO ( 1 ) = -9.990

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.950

.953

.965

.0684

.3519

.1076

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(N8102C) ( 27 SEP 75 )

ARC11-716 1A14 01+712+S12N25+AT10 UPPER WING

## REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
 LREF = 39.7090 INCHES YMRP = .0000 INCHES  
 BRP = 39.7090 INCHES ZMRP = .0000 INCHES  
 SCALE = .0300 SCALE

MACH ( 1 ) = 1.246 BETAO ( 1 ) = -10.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.000	.0481	.5478
.020		.7921
.023		.7296
.040	.0770	.6848
.045		.7411
.050	.0169	.4519
.060		.4597
.075		.4189
.086	.1125	
.094		.3289
.100		.2517
.103		.2042
.107		.1877
.117		.1069
.133		.1807
.150	.0251	.0107
.163		.0955
.177	.0449	
.193		.0050
.209		.1167
.230		.0449
.274		.0050
.339		.1167
.402		.0449
.450		.0050
.497		.1167
.550		.0449
.565		.0050
.600		.1167
.650		.0449
.700		.0050
.725		.1167
.750		.0449
.760		.0050
.775		.1167
.808		.0449
.834		.0050
.850		.1167
.857		.0449
.865		.0050
.900		.1167
.905		.0449

## PARAMETRIC DATA

ALPHA0 = .000 ELEVON = .000  
 RUDDER = .000 SPDRK = .000



C

**DATE 10 JAN 75**

TABLED PRESSURE DATA - 1A14A - VOL. 6

(R51120)

ARCI-716 1A14 01+729+312+25+AT10 UPPER WING

RACH ( 1 ) = 1.246      BETAO ( 1 ) = -10.080

## DEPENDENT VARIABLE CP

1/94	.2990	.3640	.4270	.5340	.6730	.7800	.8670
------	-------	-------	-------	-------	-------	-------	-------

Case	Age	Sex	Weight (kg)	Height (cm)	Time (min)	Calculated (kcal)	Observed (kcal)	Ratio
1	20	M	65	175	10	1000	1000	1.00
2	25	F	55	160	15	1500	1500	1.00
3	30	M	70	180	20	2000	2000	1.00
4	35	F	60	165	25	2500	2500	1.00
5	40	M	75	185	30	3000	3000	1.00
6	45	F	65	170	35	3500	3500	1.00
7	50	M	80	190	40	4000	4000	1.00
8	55	F	70	175	45	4500	4500	1.00
9	60	M	85	195	50	5000	5000	1.00
10	65	F	75	180	55	5500	5500	1.00
11	70	M	90	200	60	6000	6000	1.00
12	75	F	80	185	65	6500	6500	1.00
13	80	M	95	205	70	7000	7000	1.00
14	85	F	85	190	75	7500	7500	1.00
15	90	M	100	210	80	8000	8000	1.00
16	95	F	90	195	85	8500	8500	1.00
17	100	M	105	215	90	9000	9000	1.00
18	105	F	95	200	95	9500	9500	1.00
19	110	M	110	220	100	10000	10000	1.00
20	115	F	100	205	105	10500	10500	1.00
21	120	M	115	225	110	11000	11000	1.00
22	125	F	105	210	115	11500	11500	1.00
23	130	M	120	230	120	12000	12000	1.00
24	135	F	110	215	125	12500	12500	1.00
25	140	M	125	235	130	13000	13000	1.00
26	145	F	115	220	135	13500	13500	1.00
27	150	M	130	240	140	14000	14000	1.00
28	155	F	120	225	145	14500	14500	1.00
29	160	M	135	245	150	15000	15000	1.00
30	165	F	125	230	155	15500	15500	1.00
31	170	M	140	250	160	16000	16000	1.00
32	175	F	130	235	165	16500	16500	1.00
33	180	M	145	255	170	17000	17000	1.00
34	185	F	135	240	175	17500	17500	1.00
35	190	M	150	260	180	18000	18000	1.00
36	195	F	140	245	185	18500	18500	1.00
37	200	M	155	265	190	19000	19000	1.00
38	205	F	145	250	195	19500	19500	1.00
39	210	M	160	270	200	20000	20000	1.00
40	215	F	150	255	205	20500	20500	1.00
41	220	M	165	275	210	21000	21000	1.00
42	225	F	155	260	215	21500	21500	1.00
43	230	M	170	280	220	22000	22000	1.00
44	235	F	160	265	225	22500	22500	1.00
45	240	M	175	285	230	23000	23000	1.00
46	245	F	165	270	235	23500	23500	1.00

$$\text{MACM} ( 1 ) = 1.243 \quad \text{BETAO} ( 2 ) = -7.965$$

SECTION / 1100000 WING

[illegible]

	X/CW				
	.000	.0106	.0310	.4915	.7389
					.6977
					.6451
					.6864
					.4281
					.3991
					.4202
					.4341

0.020	.2965
-------	-------

.040  
.0519

[illegible]

0.50 -0.0213 0.0045 0.000

.005  
.0016

006 .005

.094	-.0692	-.2227	-.2303	-.1453	-.1036
150					

[illegible]

177  
-1293  
0.04

**.193                      .0144**

.229  
 .250  
 -.3367  
 -.3399  
 -.3065

-.274  
-.2303

0494  
-1730  
0494

.362    .0477  
 .400    -.4413    -.5001  
          -.4977

**.402**  
**-.3751**

497 .0000  
- .4768 - .5359

	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	2101	2102	2103	2104	2105	2106	2107	2108	2109	2110	2111	2112	2113	2114	2115	2116	2117	2118	2119	2120	2121	2122	2123	2124	2125	2126	2127	2128	2129	2130	2131	2132	2133	2134	2135	2136	2137	2138	2139	2140	2141	2142	2143	2144	2145	2146	2147	2148	2149	2150	2151	2152	2153	2154	2155	2156	2157	2158	2159	2160	2161	2162	2163	2164	2165	2166	2167	2168	2169	2170	2171	2172	2173	2174	2175	2176	2177	2178	2179	2180	2181	2182	2183	2184	2185	2186	2187	2188	2189	2190	2191	2192	2193	2194	2195	2196	2197	2198	2199	2200	2201	2202	2203	2204	2205	2206	2207	2208	2209	2210	2211	2212	2213	2214	2215	2216	2217	2218	2219	2220	2221	2222	2223	2224	2225	2226	2227	2228	2229	2230	2231	2232	2233	2234	2235	2236	2237	2238	2239	2240	2241	2242	2243	2244	2245	2246	2247	2248	2249	2250	2251	2252	2253	2254	2255	2256	2257	2258	2259	2260	2261	2262	2263	2264	2265	2266	2267	2268	2269	2270	2271	2272	2273	2274	2275	2276	2277	2278	2279	2280	2281	2282	2283	2284	2285	2286	2287	2288	2289	2290	2291	2292	2293	2294	2295	2296	2297	2298	2299	2300	2301	2302	2303	2304	2305	2306	2307	2308	2309	2310	2311	2312	2313	2314	2315	2316	2317	2318	2319	2320	2321	2322	2323	2324	2325	2326	2327	2328	2329	2330	2331	2332	2333	2334	2335	2336	2337	2338	2339	2340	2341	2342	2343	2344	2345	2346	2347	2348	2349	2350	2351	2352	2353	2354	2355	2356	2357	2358	2359	2360	2361	2362	2363	2364	2365	2366	2367	2368	2369	2370	2371	2372	2373	2374	2375	2376	2377	2378	2379	2380	2381	2382	2383	2384	2385	2386	2387	2388	2389	2390	2391	2392	2393	2394	2395	2396	2397	2398	2399	2400	2401	2402	2403	2404	2405	2406	2407	2408	2409	2410	2411	2412	2413	2414	2415	2416	2417	2418	2419	2420	2421	2422	2
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.600  
 -.5320  
 -.5130

650	- .1006
700	- .1107

**-0.1794**

67.9%  
- .5879  
- .1005

1960	-1.184	-1.329
1961	-1.184	-1.329
1962	-1.184	-1.329
1963	-1.184	-1.329
1964	-1.184	-1.329
1965	-1.184	-1.329
1966	-1.184	-1.329
1967	-1.184	-1.329
1968	-1.184	-1.329
1969	-1.184	-1.329
1970	-1.184	-1.329
1971	-1.184	-1.329
1972	-1.184	-1.329
1973	-1.184	-1.329
1974	-1.184	-1.329
1975	-1.184	-1.329
1976	-1.184	-1.329
1977	-1.184	-1.329
1978	-1.184	-1.329
1979	-1.184	-1.329
1980	-1.184	-1.329
1981	-1.184	-1.329
1982	-1.184	-1.329
1983	-1.184	-1.329
1984	-1.184	-1.329
1985	-1.184	-1.329
1986	-1.184	-1.329
1987	-1.184	-1.329
1988	-1.184	-1.329
1989	-1.184	-1.329
1990	-1.184	-1.329
1991	-1.184	-1.329
1992	-1.184	-1.329
1993	-1.184	-1.329
1994	-1.184	-1.329
1995	-1.184	-1.329
1996	-1.184	-1.329
1997	-1.184	-1.329
1998	-1.184	-1.329
1999	-1.184	-1.329
2000	-1.184	-1.329
2001	-1.184	-1.329
2002	-1.184	-1.329
2003	-1.184	-1.329
2004	-1.184	-1.329
2005	-1.184	-1.329
2006	-1.184	-1.329
2007	-1.184	-1.329
2008	-1.184	-1.329
2009	-1.184	-1.329
2010	-1.184	-1.329
2011	-1.184	-1.329
2012	-1.184	-1.329
2013	-1.184	-1.329
2014	-1.184	-1.329
2015	-1.184	-1.329
2016	-1.184	-1.329
2017	-1.184	-1.329
2018	-1.184	-1.329
2019	-1.184	-1.329
2020	-1.184	-1.329
2021	-1.184	-1.329
2022	-1.184	-1.329
2023	-1.184	-1.329
2024	-1.184	-1.329
2025	-1.184	-1.329
2026	-1.184	-1.329
2027	-1.184	-1.329
2028	-1.184	-1.329
2029	-1.184	-1.329
2030	-1.184	-1.329
2031	-1.184	-1.329
2032	-1.184	-1.329
2033	-1.184	-1.329
2034	-1.184	-1.329
2035	-1.184	-1.329
2036	-1.184	-1.329
2037	-1.184	-1.329
2038	-1.184	-1.329
2039	-1.184	-1.329
2040	-1.184	-1.329
2041	-1.184	-1.329
2042	-1.184	-1.329
2043	-1.184	-1.329
2044	-1.184	-1.329
2045	-1.184	-1.329
2046	-1.184	-1.329
2047	-1.184	-1.329
2048	-1.184	-1.329
2049	-1.184	-1.329
2050	-1.184	-1.329
2051	-1.184	-1.329
2052	-1.184	-1.329
2053	-1.184	-1.329
2054	-1.184	-1.329
2055	-1.184	-1.329
2056	-1.184	-1.329
2057	-1.184	-1.329
2058	-1.184	-1.329
2059	-1.184	-1.329
2060	-1.184	-1.329
2061	-1.184	-1.329
2062	-1.184	-1.329
2063	-1.184	-1.329
2064	-1.184	-1.329

	.0211
	.773

[illegible]

060  
-1060-  
-8720-  
-3302

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(081028)

ARC11-716 1A14 01+712+512N25+AT10 UPPER WING

MACH ( 1 ) = 1.245 BETA0 ( 2 ) = -7.960

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7600 .8870

X/CW

.857	.0587					
.865						
.1213						
.1126						
.950						
.955						
.950						
.950						
.953						
.965						

MACH ( 1 ) = 1.248 BETA0 ( 3 ) = -6.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7600 .8870

X/CW

.090	-.0177	-.0396	.4572	.7219	.6426	.6384
.020				.4217	.4054	.3775
.025			.2742			
.940		.0339	.2057			
.045				.1831	.1623	.1481
.090	-.0572			.0020		
.060			.0646			
.085		.0479				
.086	-.0909					
.094				.2175	-.2113	-.1910
.130		.0000				
.163						
.177			-.1109			
.193		.0153				
.229	-.0605			-.3187	-.3562	-.3439
.290			-.2305			
.274		-.1740				
.339						
.362	.0402			-.4344	-.4945	-.4997
.450			-.3729			
.402						
.497	.0000			-.4794	-.5326	
.550						
.565			.0000			
.600						
.630						
.700	-.3045					
.725				-.1779		
.750						
.760						



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1028)

ARC11-716 1A14 01-T12-S12N29-AT10 UPPER WING

MACH ( 1 ) = 1.248 BETAO ( 3 ) = -6.020

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.775				-.1311 -.3362
.806			.0375	
.834	.0033			
.850				-.0842 -.2692 -.3074
.857			.0521	
.865	.1035			
.900	.0904			-.0100 -.1510 -.4080
.925			.0626	
.950				-.0361 -.2165
.953			.0422	
.965	.0172			

MACH ( 1 ) = 1.247 BETAO ( 4 ) = -3.950

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.900	-.0603	-.0388	.4157	.6842 .6249 .5627 .5922
.920				.4077 .3951 .3547 .3670
.925		.0075	.2399	
.940			.1721	
.945				.1754 .1507 .1439 .1325
.950	-.0679			-.0029
.960			.0357	
.965		.0519		
.986	-.1236			-.2165 -.2161 -.1825 -.1304
.150		.0000		
.163			-.1199	
.177		.0036		
.193				-.3256 -.3511 -.3476 -.3273
.229	-.0744			
.250			-.2374	
.274				
.339		-.1851		
.362	.0305			-.4260 -.4886 -.5059
.405				
.402			-.3757	
.497	.0000			-.4670 -.5249
.550			.0000	
.965				-.5529
.800				-.5391
.850				

DATE 10 JAN 75

ARC11-716 1A14 01-T12-S12N25+AT10 UPPER WING

MACH ( 1 ) = 1.247 BETAO ( 4 ) = -3.950

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW  
 .700 -.2688  
 .725  
 .750  
 .760  
 .775  
 .808  
 .834 .0060  
 .850  
 .857  
 .865  
 .890 .0741  
 .905  
 .950  
 .953  
 .965 .0032

-.1686  
 -.1697  
 -.50  
 -.4936  
 -.1157  
 -.0928  
 -.2635  
 .0459  
 -.0143  
 -.1637  
 -.2811  
 .0434  
 .0196  
 -.0841  
 .0389  
 -.2178  
 .0178

-.3794

MACH ( 1 ) = 1.246 BETAO ( 5 ) = -2.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW  
 .000  
 -.1007  
 -.0694  
 .3757  
 .6437  
 .5786  
 .9249  
 .9507  
 .020  
 .025  
 .040  
 .045  
 .090  
 .095  
 .098  
 .099  
 .094  
 .1540  
 .150  
 .163  
 .177  
 .193  
 .229  
 .250  
 .274  
 .359  
 .362  
 .409  
 .402  
 .497

-.1007  
 -.0694  
 .3757  
 .6437  
 .5786  
 .9249  
 .9507  
 .020  
 .025  
 .040  
 .045  
 .090  
 .095  
 .098  
 .099  
 .094  
 .1540  
 .150  
 .163  
 .177  
 .193  
 .229  
 .250  
 .274  
 .359  
 .362  
 .409  
 .402  
 .497

-.1007  
 -.0694  
 .3757  
 .6437  
 .5786  
 .9249  
 .9507  
 .020  
 .025  
 .040  
 .045  
 .090  
 .095  
 .098  
 .099  
 .094  
 .1540  
 .150  
 .163  
 .177  
 .193  
 .229  
 .250  
 .274  
 .359  
 .362  
 .409  
 .402  
 .497



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+S12N25+A110 UPPER WING (R01028)

MACH ( 1 ) = 1.246 BETAO ( 5 ) = -2.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	Y/BW	CP
.550		
.565		
.600		
.650		
.700		
.725		
.750		
.760		
.775		
.808		
.834		
.850		
.857		
.865		
.900		
.905		
.920		
.953		
.965		

MACH ( 1 ) = 1.246 BETAO ( 6 ) = .010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	Y/BW	CP
.000		
.020		
.025		
.040		
.045		
.050		
.060		
.085		
.086		
.094		
.150		
.163		
.177		
.193		
.229		
.250		
.274		
.339		



(R01120)

ARC11-716 IA14 Q1+T12+S12N25+AT13 UPPER WING

$$\text{WACH} (1) = 1.246 \quad \text{BETA0} (6) = .510$$

SECTION 11335 (1) (b) (1)

**DEPENDENT VARIABLE CP**

[illegible]

**MD/X**

**.0172**

**- .4245    - .4552    - .5118**

-5118

**00000**

**-3409**

**NOTES:**

-4121 -4898

00000

- .5039

**.1633**

- 1340 -

-1556

**-4499 -4532**

**-2151**

**-0328 -1740**

1

4216.

**0270.**

— 7173 — 10000

Page 1 of 1

0391

0111  
1630

**-2407**

444

- 0064

- . 9428

1520.

**.0477**

DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01\*112+912N25+AT10 UPPER WING (R81U29) ( 27 SEP 73 )

PARAMETRIC DATA

ALPHA = -10.000 ELEVON = .000  
RUDDER = .000 SPORRK = .000

REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
LREF = 38.7090 INCHES YMRP = .0000 INCHES  
BREF = 38.7090 INCHES ZMRP = .0000 INCHES  
SCALE = .0300 SCALE

MACH ( 1 ) = 1.245 BETA0 ( 1 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870	
X/CW	.000	-.1571	-.2840	.0786	.3125	.1891	.1072	.1314
	.020	.025		.3170	.5799	.5007	.5187	.5216
	.040		-.1039					
	.045			.3196				
	.050	-.0706			.4435	.4246	.4316	.4243
	.080				.3202			
	.085			.2761				
	.086		.0954					
	.094	-.0270			.1346	.1595	.1500	.1878
	.150		.0000					
	.163			.1656				
	.177		.1533					
	.193							
	.229	-.0131			-.0164	-.0182	-.0095	-.0356
	.250			.0541				
	.274		.0382					
	.339	.1220			-.1200	-.1497		-.2458
	.362			-.0930				
	.400		.0000		.1674	.1287		
	.402							
	.497	.0000						
	.550							
	.565							
	.600							
	.650							
	.700	.2153						
	.725			.0634				
	.750				-.1115			
	.760							
	.775		-.0043	.0977	.0599			
	.808		.0699					
	.834	.0817						
	.850			.0804	.0604	.0610		
	.857							
	.865	.1272						
	.900	.1521		.1026				.1232
	.905			.1607				

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2476

(881029)

ARC11-716 1A14 OL+ 12+S12N25+AT10 UPPER WING

MACH (1) = 1.245 BETAO (1) = .030

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

V/BW .2990 .3640 .4270 .5540 .6730 .7800 .8870

X/CW

.990

.953

.963

.1728

.1723

.1419



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2477

ARC11-716 1A14 OL+T12+312N23+AT11 UPPER WING

(RB1080) ( 02 OCT 75 )

## REFERENCE DATA

SREF = 2.4210 90.FT. XMRP = 29.5800 INCHES  
 LREF = 32.7090 INCHES YMRP = .0000 INCHES  
 BREF = 18.7090 INCHES ZMRP = .0000 INCHES  
 SCALE = .0000 SCALE

MACH ( 1 ) = .972 SETAO ( 1 ) = .040

## PARAMETRIC DATA

ALPHA = .000 ELEVON = .000  
 RUDDER = .000 SPOBRK = .000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/RW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CV

.000	-.2807	.1329	.4198	.5412	.4774	.4335	.4546
.020			.2496	.2686	.2080	.1818	.1366
.040		.2092	.1608				
.060	.0001			.0387	-.0277	-.0556	-.1375
.080				-.1378			
.100			.0195				
.120	.0442	.2233					
.140				-.3237	-.3774	-.3804	-.3982
.160		.0000					
.180		.0317					
.200	.1486			-.4610	-.5033	-.5609	-.6119
.220			-.2890				
.240		-.2592					
.260	.0787			-.4601	-.5982		-.8094
.280			-.3889				
.300		.0000		-.2476	-.3635		
.320			.0000				
.340							-.7756
.360	-.0324			-.3824	-.2752	-.4611	
.380							-.3249
.400		-.2679					-.2594
.420			-.2594	-.3030	-.3104		
.440							
.460	-.2803			-.2756	-.1640	-.0634	
.480			-.2995				
.500		-.2144					
.520		-.2235		-.1540	-.0593		-.0138
.540			-.1837				

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 OF POOR QUALITY



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+712+S12N25+AT11 UPPER WING (R81030)

MACH (2) = 1.002 BETAO (1) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.837 -.1949  
.865 -.1341  
.900 -.1620  
.935 -.1079  
.953 -.0118  
.973 -.0367  
.988 -.0546  
.998 -.1051

MACH = 1.025 BETAO (1) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.000 -.2940 .0613 .4185 .5644 .5181 .4831 .5095  
.020 .020 .2444 .2870 .2403 .2193 .1836  
.025 .1683 .1590  
.045 .050 .0669 .0129 -.0064 -.0820  
.060 .080 .0919  
.085 .0340  
.086 .1963  
.094 -.1985  
.0000  
.0000  
.0593  
.029 .0090  
.250  
.274  
.339  
.362 .0096  
.400  
.402  
.497 .0000  
.550  
.565  
.600  
.650  
.700  
.725  
.740  
.760

-.2011

ARC11-716 1A14 01+712+S12N23+AT11 UPPER WING

(R01U30)

MACH (3) = 1.023 BETAO (1) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2940 .3640 .4870 .5340 .6740 .7800 .8470

X/CW

.775

.808

.834

.850

.857

.865

.900

.905

.930

.953

.965

-.2646

-.2192

-.1664

-.1827

-.1640

-.1183

-.1257

-.0696

-.0534

-.0318

-.2343

-.0410

.0192

-.1283

.0171

.0522

.0587

-.0894



(R81U31) (06 FEB 74)

ARC11-716 1A14 Q1-T12-S12N2-AT10 UPPER WING

PARAMETRIC DATA

MACH = .900 ELEVON = .000  
RUDDER = .000 SPOBRK = .000

REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
LREF = 38.7090 INCHES YMRP = .0000 INCHES  
BREF = 38.7390 INCHES ZMRP = .0000 INCHES  
SCALE = .0300 SCALE

ALPHA(1) = -10.140 BETA(1) = -8.370

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0135	.0114	.2465	.3000	.1720	.0533	.1026
.020				.5903	.4273	.5862	.5236
.025		.2466	.4770				
.040				.4071	.4119	.4090	.3555
.045				.2493			
.050	.1275						
.080			.2979				
.085		.3355					
.086				.0396	.0464	.0608	.0115
.094	.1266						
.130		.0000					
.163			.0772				
.177		.2438					
.193							
.229	.2111			-.1267	-.1509	-.1819	-.2668
.250							
.274			-.0334				
.339		-.0120					
.362	.2015			-.1068	-.1932		-.3532
.400							
.402			-.0811				
.497	.0500			-.0560	-.1461		
.550			.0000				
.565							
.600							
.650							
.690							
.700	.1067						
.725				-.1197			
.750				-.1459			
.760							
.775				-.0117		.1261	.0381
.806							
.834	.3458			.1507	.1293		
.857				.1554			
.865							
.900	.1644			.1074	.1255	.1324	
.905	.1291			.1189			
.905							
.905				.0958	.0790		.0983
.905				.0687			



(RB1031)

ARC11-716 1A14 01-T18-S12N3+AT10 UPPER WING

ALPHAO (1) = -10.140 BETAO (1) = -8.370

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.930

.933

.965

.0407

.0327

.0997

ALPHAO (1) = -10.130 BETAO (2) = -6.560

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000

.020

.025

.040

.045

.050

.080

.085

.066

.094

.130

.163

.177

.193

.229

.230

.274

.339

.362

.400

.402

.497

.530

.565

.600

.700

.725

.730

.775

.806

.834

.850

.0119

-.0057

.2039

.2567

.5863

.4568

.2183

.4181

.1122

.3149

.1171

.0500

.0883

.2459

.2054

-.0047

.2069

.0000

.0000

.0000

.0000

.0000

.0000

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.2039

.2567

.5863

.4568

.2183

.4181

.1122

.3149

.1171

.0500

.0883

.2459

.2054

-.0047

.2069

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.2039

.2567

.5863

.4568

.2183

.4181

.1122

.3149

.1171

.0500

.0883

.2459

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-T18-S12N23-AT10 UPPER WING (R01U01)

ALPHA( 1 ) = -10.130 BETA( 2 ) = -6.580

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.857	.1409	.0956				
	.863	.1126	.0677	.0663	.0621	.0839	.0894
	.900						
	.905						
	.930						
	.933						
	.965						

ALPHA( 1 ) = -10.130 BETA( 3 ) = -4.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	.0026	.1716	.2086	.0635	-.0034	-.0165
	.020		.5694	.3831	.5190	.4682	
	.025		.4465				
	.040	.2028					
	.045		.4105	.4110	.3982	.3915	.3301
	.050			.2684			
	.080		.3131				
	.085						
	.086	.3086					
	.094			.0602	.0620	.0747	.0081
	.130	.0000					
	.163		.1087				
	.177	.2487					
	.193						
	.229	.1991		-.1027	-.1239	-.1579	-.2515
	.230		-.0137				
	.274						
	.339	.0041					
	.362	.2076		-.1188	-.1934		-.3377
	.400						
	.402		-.0911				
	.497	.0000		-.1022	-.1836		
	.530		.0000				
	.565						
	.620						
	.650						
	.700	.0364		-.1289		-.1461	-.2061
	.725						
	.750		-.0379			.0928	.0384
	.760						
			-.0658				

(081031)

ARC11-716 1A14 01+712+312M25+AT10 UPPER WING

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ALPHA( 1 ) = -10.139 BETA( 3 ) = -4.840

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.775			.1070	.0932			
.608		.1042					
.834	-.0061		.0749	.0915	.1013		
.850		.0733					
.857							
.865	.1104					.0733	
.900	.0825		.0660	.0465			
.905		.0518			.0656		
.950		.0009					
.953							
.965	-.0262						

ALPHA( 1 ) = -10.000 BETA( 4 ) = -3.250

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0005	-.0290	.1496	.1703	.0141	-.1250	-.0610
.020				.5606	.3612	.4966	.4398
.025		.1871	.4285				
.040			.3982				
.045				.4032	.3894	.3817	.3176
.050	.0939			.2717			
.080			.3105				
.085		.2948					
.086							
.094	.1024			.0692	.0650	.0762	.0089
.150		.0000					
.163			.1168				
.177		.2491					
.193							
.229	.1911			-.0968	-.1235	-.1359	-.2476
.250							
.274			-.0064				
.339		.0130					
.382	.2033			-.1243	-.2082		-.3673
.400			-.0906				
.402							
.497	.0000			-.1250	-.2142		
.530			.0000				
.565							
.600							-.2193
.650						-.1674	



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(R01031)

ARC11-716 IA14 01+712+512N25+AT10 UPPER WING

ALPHA(1) = -10.080 BETA(4) = -3.250

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.700	.0391			-.0006	-.1360		
.725						.0789	.0216
.750							
.760							
.775							
.806							
.834							
.850							
.857							
.865							
.900							
.925							
.930							
.933							
.965							

ALPHA(1) = -10.040 BETA(5) = -1.600

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.0001	-.0415	.1229	.1250	-.0308	-.1754	-.1223
.020				.5381	.3411	.4702	.4112
.025							
.040							
.045							
.050							
.060							
.065							
.066							
.094							
.150							
.165							
.177							
.195							
.229							
.250							
.274							
.339							
.362							
.400							
.492							
.497							

(R81031)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA( 1 ) = -10.040 BETA( 5 ) = -1.600

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.4940	.6730	.7800	.8870
X/CW							
.550				-.1456	-.2306		
.565			.0000			-.2213	
.600						-.1907	
.650					-.1466		
.700	.0155			-.1017		.0670	.0170
.725							
.750							
.760			-.1191	.0600	.0603		
.775		.0524					
.806							
.834	-.0564		.0411	.0630	.0774		
.850			.0329				
.857							
.865	.0604		.0384			.0547	
.900	.0429		.0136	.0216			
.925			-.0062	.0414			
.930			-.0300				
.933							
.965	-.0531						

ALPHA( 1 ) = -10.040 BETA( 6 ) = .100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.4940	.6730	.7800	.8870
X/CW							
.000	.0062	-.0129	.1066	.0626	-.0692	-.2344	-.1876
.020				.3207	.3174	.4406	.5771
.025			.3693				
.040		.1602					
.045			.3702			.3517	.2944
.050	.0669			.3920	.3668		
.090				.2679			
.093		.3027					
.096		.2634					
.094	.0689			.0721	.0732	.0767	-.0046
.150		.0000					
.163		.1240					
.177							
.193		.2465					
.229	.1794			-.0703	-.1111	-.1481	-.2489
.250							
.274		.0059					
.339		.0236					



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81031)

ARC11-716 1A14 Q1+T12+912N25+AT10 UPPER WING

ALPHA( 1 ) = -10.040 BETA( 6 ) = .100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.362 .2030  
.400  
.492  
.497 .9000  
.550  
.565  
.600  
.650  
.700  
.725  
.750  
.760  
.775  
.808  
.834  
.850  
.857  
.865  
.900  
.905  
.950  
.953  
.965

-.3902

-.1042

-.1792

-.2591

.0000

-.2162

-.1931

-.1507

-.1241

.0589

.0128

.0901

.0476

.0330

.0266

.0526

.0688

.0188

.0296

.0174

.0412

-.0108

-.0372

ALPHA( 1 ) = -10.040 BETA( 7 ) = 1.810

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 .0210 .0198 .1058 .0563  
.020  
.025  
.040  
.045  
.050  
.060  
.065  
.086  
.094  
.150  
.163  
.177  
.193

-.2473

.3428

.3740

.1523

.3614

.3020

.3539

.3341

.2619

.2621

.2986

.2528

.0791

.0000

.1239

.0734

.0696

.0684

-.0170

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(R81U31)

ARC11-716 1A14 01+T12+S12M5+AT10 UPPER WING

ALPHA(1) = -10.040 BETA(7) = 1.810

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.229	.1621					
.250						
.274						
.339	.0186					
.362	.1852					
.400						
.402						
.497	.0000					
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.774						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.953						
.965						

ALPHA(1) = -10.130 BETA(8) = 3.580

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.000	.0105	.1090	.0345	-.1807	-.3295	-.2941
.020			.4931	.2624	.3794	.3335
.025						
.040						
.045						
.050						
.050						
.085						
.086						
.094						



(R01U31)

PARCII-716 :A14 01+T12+S12N25+AT10 UPPER WING

$$\text{ALPHA}(1) = -10.139 \quad \text{BETA}(8) = 3.589$$

SECTION 1 UPPER WING

	.2990	.3640	.4270	.5340	.6730	.7800	.8670
--	-------	-------	-------	-------	-------	-------	-------

M2/X

163 .0000

.177	.1284
------	-------

193. 2103

6229 .1530

.239	0070
------	------

139 3169 3373

1.333	1.133
1.362	1.1689

667.

.432	- .1192
------	---------

497	.0500
-----	-------

655.

6659. 555.

657  
658

0.635  
0.799  
-0.9797

.725

567.

**.769** **-.2186**

.775

638  
- .5238

**.834 - .1400**

.633  
 .638  
 .638

.055      -.0422

**.933 - .0376**

935 - .0474

056.

656.  
-2834

**.965    -.1183**

$$\text{ALGMA2Y 11 3 -10.190 BETAC ( 9) = 5.250}$$

SECTION / BUFFER WING

Y/84	.2993	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

MD/X

1156	1157	1158	1159	1160	1161	1162	1163	1164	1165	1166	1167	1168	1169	1170	1171	1172	1173	1174	1175	1176	1177	1178	1179	1180	1181	1182	1183	1184	1185	1186	1187	1188	1189	1190	1191	1192	1193	1194	1195	1196	1197	1198	1199	1200	1201	1202	1203	1204	1205	1206	1207	1208	1209	1210	1211	1212	1213	1214	1215	1216	1217	1218	1219	1220	1221	1222	1223	1224	1225	1226	1227	1228	1229	1230	1231	1232	1233	1234	1235	1236	1237	1238	1239	1240	1241	1242	1243	1244	1245	1246	1247	1248	1249	1250	1251	1252	1253	1254	1255	1256	1257	1258	1259	1260	1261	1262	1263	1264	1265	1266	1267	1268	1269	1270	1271	1272	1273	1274	1275	1276	1277	1278	1279	1280	1281	1282	1283	1284	1285	1286	1287	1288	1289	1290	1291	1292	1293	1294	1295	1296	1297	1298	1299	1300	1301	1302	1303	1304	1305	1306	1307	1308	1309	1310	1311	1312	1313	1314	1315	1316	1317	1318	1319	1320	1321	1322	1323	1324	1325	1326	1327	1328	1329	1330	1331	1332	1333	1334	1335	1336	1337	1338	1339	1340	1341	1342	1343	1344	1345	1346	1347	1348	1349	1350	1351	1352	1353	1354	1355	1356	1357	1358	1359	1360	1361	1362	1363	1364	1365	1366	1367	1368	1369	1370	1371	1372	1373	1374	1375	1376	1377	1378	1379	1380	1381	1382	1383	1384	1385	1386	1387	1388	1389	1390	1391	1392	1393	1394	1395	1396	1397	1398	1399	1400	1401	1402	1403	1404	1405	1406	1407	1408	1409	1410	1411	1412	1413	1414	1415	1416	1417	1418	1419	1420	1421	1422	1423	1424	1425	1426	1427	1428	1429	1430	1431	1432	1433	1434	1435	1436	1437	1438	1439	1440	1441	1442	1443	1444	1445	1446	1447	1448	1449	1450	1451	1452	1453	1454	1455	1456	1457	1458	1459	1460	1461	1462	1463	1464	1465	1466	1467	1468	1469	1470	1471	1472	1473	1474	1475	1476	1477	1478	1479	1480	1481	1482	1483	1484	1485	1486	1487	1488	1489	1490	1491	1492	1493	1494	1495	1496	1497	1498	1499	1500	1501	1502	1503	1504	1505	1506	1507	1508	1509	1510	1511	1512	1513	1514	1515	1516	1517	1518	1519	1520	1521	1522	1523	1524	1525	1526	1527	1528	1529	1530	1531	1532	1533	1534	1535	1536	1537	1538	1539	1540	1541	1542	1543	1544	1545	1546	1547	1548	1549	1550	1551	1552	1553	1554	1555	1556	1557	1558	1559	1560	1561	1562	1563	1564</
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026.

.325 .3441

5011' 5010'

.945	
.995	

4550' 555'

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81031)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA(1) = -10.130 BETA(1) = 5.250

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.080	.2638					
.065	.2763					
.066	.2167					
.094	.0618					
.150	.0900	.0840	.0684	.0582	-.0375	
.163	.1231					
.177	.2085					
.193	.1401					
.229	.0110	-.0714	-.1104	-.1515	-.2698	
.250						
.274	.0156					
.339	.1525					
.362		-.1565	-.2429		-.4416	
.402	-.1240					
.497	.0000					
.550		-.2511	-.3544			
.565	.0030					
.600						
.650						
.700	-.1085					
.725		-.2471	-.2014			
.750				.0222	.0376	
.760	-.2501					
.775		-.0223	-.0049			
.808	-.0618					
.834						
.850	-.1695					
.857		-.0258	.0209	.0567		
.865	-.0847					
.900	-.0719					
.905		-.0208	.0073		.0593	
.950		-.0630				
.950		-.0502			.0431	
.955		-.0974				
.965	-.1306					



ARC11-716 1A14 C1+T12+S12N25+AT10 UPPER WING

(RB1031)

ALPHA( 1) = -10.120      BETAC (10) = 7.019

SECTION (1) UPPER WING

DEPENDENT VARIABLE: CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7690	.8670
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**AD/x**

.000	-.0534	-.0031	.0677	-.0029	-.2584	-.4205	-.4010
.025				.4795	.2175	.3286	.2425
.025		.3349					
.040	.1017						
.045			.3297				
.050	.0043			.3773	.3268	.2847	.2019
.080				.2766			
.085		.2812					
.086	.2034						
.094	.0513						
.150		.0000		.0968	.0790	.0628	-.0408
.163			.1256				
.177	.2019						
.193							
.229	.1264						
.250				-.0570	-.0971	-.1399	-.2649
.274		.0177					
.359	.0133						
.362	.1341						
.400			-.1208	-.1550	-.2368		-.4424
.402				-.2630	-.3597		
.497	.0000		.0000			-.4898	-.5723
.550							
.565							
.600							
.690							
.700	-.1357			-.3201		-.0823	-.0141
.725							
.750			-.2853				
.760				-.0764	-.0785		
.775							
.808			-.1045				
.834	-.2153						
.850				-.0475	-.0004	.0266	
.857			-.0893				
.865	-.1229						
.900	-.1011			-.0305	.0130		.0585
.905			-.0906				
.950				-.0594		.0425	
.953			-.1153				
.965	-.1482						

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81031)

ARC11-716 1A14 01-112-S12N25+AT10 UPPER WING

ALPHA(1) = -10.130 BETA(1,1) = 6.780

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP
Y/B4	.2990 .3640 .4270 .5340 .6730 .7800 .8870
X/CW	
.000	-.0943 -.0033 .0314 -.0274 -.2941 -.4593 -.4486
.020	.020 .020 .020 .020 .020 .020 .020
.025	.025 .025 .025 .025 .025 .025 .025
.040	.040 .040 .040 .040 .040 .040 .040
.045	.045 .045 .045 .045 .045 .045 .045
.050	.050 .050 .050 .050 .050 .050 .050
.060	.060 .060 .060 .060 .060 .060 .060
.085	.085 .085 .085 .085 .085 .085 .085
.086	.086 .086 .086 .086 .086 .086 .086
.094	.094 .094 .094 .094 .094 .094 .094
.130	.130 .130 .130 .130 .130 .130 .130
.163	.163 .163 .163 .163 .163 .163 .163
.177	.177 .177 .177 .177 .177 .177 .177
.193	.193 .193 .193 .193 .193 .193 .193
.229	.229 .229 .229 .229 .229 .229 .229
.250	.250 .250 .250 .250 .250 .250 .250
.274	.274 .274 .274 .274 .274 .274 .274
.339	.339 .339 .339 .339 .339 .339 .339
.362	.362 .362 .362 .362 .362 .362 .362
.400	.400 .400 .400 .400 .400 .400 .400
.402	.402 .402 .402 .402 .402 .402 .402
.497	.497 .497 .497 .497 .497 .497 .497
.590	.590 .590 .590 .590 .590 .590 .590
.565	.565 .565 .565 .565 .565 .565 .565
.600	.600 .600 .600 .600 .600 .600 .600
.630	.630 .630 .630 .630 .630 .630 .630
.700	.700 .700 .700 .700 .700 .700 .700
.723	.723 .723 .723 .723 .723 .723 .723
.790	.790 .790 .790 .790 .790 .790 .790
.760	.760 .760 .760 .760 .760 .760 .760
.775	.775 .775 .775 .775 .775 .775 .775
.800	.800 .800 .800 .800 .800 .800 .800
.834	.834 .834 .834 .834 .834 .834 .834
.850	.850 .850 .850 .850 .850 .850 .850
.857	.857 .857 .857 .857 .857 .857 .857
.863	.863 .863 .863 .863 .863 .863 .863
.900	.900 .900 .900 .900 .900 .900 .900
.905	.905 .905 .905 .905 .905 .905 .905
.980	.980 .980 .980 .980 .980 .980 .980
.933	.933 .933 .933 .933 .933 .933 .933
.963	.963 .963 .963 .963 .963 .963 .963

(RB1031)

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

ARC11-716 IA14 01+112+S12N23+AT110 UPPER WING

ALPHA(2) = -6.110 BETAO (1) = -0.350

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW	X/CW	.2990	.5640	.4270	.5340 .6730 .7800 .6870
.000	.0498	.0650	.3341	.3854	.2861 .1870 .2280
.020				.5335	.3974 .5228 .4877
.025		.2566	.4401		
.040				.3673	
.045					
.050	.1036			.3153	.3229 .3253 .2771
.080			.2241	.1463	
.085					
.086		.3209			
.094	.1144				
.130		.0009		-.0789	-.0618 -.0413 -.0793
.163			.1842	-.0095	
.177					
.193					
.229	.1923			-.2216	-.2501 -.2989 -.3578
.250					
.274			-.1235		
.339		-.0874			
.362	.1581			-.1586	-.2758 -.4679
.400			-.1429		
.402				-.0855	-.1729
.497	.0030				
.550					
.555					
.600					
.650					
.700	.0888			-.1417	-.1167
.725				-.0095	
.750					.1304 .0597
.760		-.0169			
.775			.1421	.1539	.1318
.806					
.834	.0422			.1182	.1273 .1386
.850			.1150		
.857					
.865	.1616				
.900	.1320			.1065	.0810 .1140
.905			.0897		
.930				.0515	.1072
.933			.0352		
.965	.0102				

(RB1U31)

LABULATED PRESSURE DATA - IA14A - VOL. 6

ARCC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

DATE 10 JAN 73

$$\text{ALPHA}(2) = -0.120 \quad \text{BETA}(2) = -6.640$$

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP						
Y/BW	X/CW	.2590	.3640	.4270	.5340	.6750	.7600	.8070
.000	.0382	.0633		.3013	.5435	.2260	.1207	.1791
.020				.4293	.5255	.3764	.5028	.4855
.040		.2379						
.045				.3649				
.050	.0932				.3201	.3231	.3209	.2638
.060				.2325	.1547			
.065		.3083						
.086								
.094	.1036				-.0665	-.0522	-.0424	-.0086
.150			.0000					
.163				-.0002				
.177			.1867					
.193								
.229	.1833							
.250				-.1122	-.2111	-.2442	-.2881	-.3600
.274			-.0792					
.339		.1554		-.1454	-.1753	-.2914		-.4798
.362								
.400					-.1167	-.2155		-.2137
.452		.0000		.0000			-.1400	
.497						-.1414		
.550	.0000				-.0434		.1097	.0436
.565								
.600								
.650		.0608						
.700								
.725				-.0452				
.750				.1164	.1283	.1113		
.780								
.775								
.806								
.834		.0121			.0921	.1071	.1204	
.850				.0933				
.857								
.885	.1310				.0843			.1011
.900	.1056							
.905				.0702		.0651		.0869
.950					.0344			
.953				.0180				
.965	-.0006							

DATE 12 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01-T12-S12N25+AT10 UPPER WING

(R01U31)

ALPHA( 2 ) = -0.120 BETA( 3 ) = -4.940

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW	X/CW	.3640	.4270	.5340	.6750 .7690 .8870
.000	.0267	.0466	.2704	.2990	.1844 .0704 .1232
.020				.5232	.3369 .4860 .4426
.025			.4145		
.040		.2230			
.045			.3652		
.050	.0806			.3259	.3253 .3148 .2579
.080				.1702	
.085			.2383		
.086	.2957				
.094	.0936				
.150		.0000		-.0511	-.0412 -.0280 -.0880
.163			.0150		
.177			.1693		
.193					
.229	.1756			-.1985	-.2340 -.2717 -.3569
.250				-.0988	
.274		-.0681			
.339	.1542				
.362				-.1775	-.2954 -.4811
.403			-.1349		
.492				-.1483	-.2506
.497	.0000		.0050		
.550					-.2126
.565				-.1571	
.600				-.0684	
.650	.0317			-.1409	
.700					.0944 .0310
.725					
.750		-.0733		.1028	.0930
.760			.0919		
.775					
.808				.0739	.0916 .1093
.834	-.0195		.0701		
.850					
.857					
.863	.1020			.0667	
.920	.0814				.0890
.903		.0513		.0516	
.950			.0186		.0729
.953		-.0004			
.965	-.0232				

ARC11-716 1A14 Q1+712+S12N23+AT10 UPPER WING

ALPHA ( 2 ) = -0.130  
BETA ( 4 ) = -3.270

**SECTION: UPPER WING**

DEPENDENT VARIABLE CP

Y/BW	.2990	.3840	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0217	.0368	.2411	.2995	.1280	.0105	.0812
.020				.5092	.3387	.4852	.4187
.025			.4047				
.040		.2017					
.045			.3532	.3249	.3164	.3136	.2486
.050	.0725			.1753			
.080			.2581				
.085		.2829					
.086							
.094	.0842			-.0348	-.0269	-.0142	-.0846
.150		.0000					
.163			.0274				
.177		.1938					
.193							
.229	.1679			-.1813	-.2172	-.2530	-.3433
.250							
.274			-.0861				
.339		-.0372					
.362	.1595			-.1840	-.2956		-.4777
.400			-.1578				
.402							
.497	.0000			-.1733	-.2838		-.2072
.550			.0000				
.585							
.600							
.650						-.1712	
.700	.0088			-.0923	-.1492		
.725						.0810	.0263
.750							
.760			-.1018	.0842	.0775		
.775			.0692				
.808							
.834	-.0443						
.890			.0477	.0569	.0795	.0944	
.897							
.889	.0767						.0792
.900	.0554			.0509	.0389		
.905			.0309	.0945		.0637	
.950							
.953			-.0149				
.983	-.0395						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R01031)

ARC11-716 1A14 01+712+S12N25+AT10 UPPER WING

ALPHA(2) = -9.130 BETA(9) = -1.600

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.000	.0221	.0315
.020	.0221	.2190
.025	.0221	.2159
.040	.0221	.0777
.045	.0221	-.0444
.050	.0221	.3116
.055	.0221	.4950
.060	.0221	.3147
.065	.0221	.4475
.070	.0221	.3964
.075	.0221	.3889
.080	.0221	.3662
.085	.0221	.3412
.090	.0221	.3244
.095	.0221	.3142
.100	.0221	.3094
.105	.0221	.2380
.110	.0221	.1840
.115	.0221	.2397
.120	.0221	.2691
.125	.0221	.2886
.130	.0221	.2994
.135	.0221	.0777
.140	.0221	.0000
.145	.0221	-.0245
.150	.0221	-.0142
.155	.0221	-.0060
.160	.0221	-.0822
.165	.0221	.0411
.170	.0221	.1912
.175	.0221	.1592
.180	.0221	.1229
.185	.0221	.0825
.190	.0221	-.0727
.195	.0221	-.0501
.200	.0221	.1590
.205	.0221	.362
.210	.0221	.400
.215	.0221	.402
.220	.0221	.497
.225	.0221	.550
.230	.0221	.565
.235	.0221	.600
.240	.0221	.650
.245	.0221	.700
.250	.0221	.725
.255	.0221	.750
.260	.0221	.760
.265	.0221	.775
.270	.0221	.800
.275	.0221	.834
.280	.0221	.850
.285	.0221	.857
.290	.0221	.865
.295	.0221	.865
.300	.0221	.865
.305	.0221	.865
.310	.0221	.865
.315	.0221	.865
.320	.0221	.865
.325	.0221	.865
.330	.0221	.865
.335	.0221	.865
.340	.0221	.865
.345	.0221	.865
.350	.0221	.865
.355	.0221	.865
.360	.0221	.865
.365	.0221	.865
.370	.0221	.865
.375	.0221	.865
.380	.0221	.865
.385	.0221	.865
.390	.0221	.865
.395	.0221	.865
.400	.0221	.865
.405	.0221	.865
.410	.0221	.865
.415	.0221	.865
.420	.0221	.865
.425	.0221	.865
.430	.0221	.865
.435	.0221	.865
.440	.0221	.865
.445	.0221	.865
.450	.0221	.865
.455	.0221	.865
.460	.0221	.865
.465	.0221	.865
.470	.0221	.865
.475	.0221	.865
.480	.0221	.865
.485	.0221	.865
.490	.0221	.865
.495	.0221	.865
.500	.0221	.865
.505	.0221	.865
.510	.0221	.865
.515	.0221	.865
.520	.0221	.865
.525	.0221	.865
.530	.0221	.865
.535	.0221	.865
.540	.0221	.865
.545	.0221	.865
.550	.0221	.865
.555	.0221	.865
.560	.0221	.865
.565	.0221	.865
.570	.0221	.865
.575	.0221	.865
.580	.0221	.865
.585	.0221	.865
.590	.0221	.865
.595	.0221	.865
.600	.0221	.865
.605	.0221	.865
.610	.0221	.865
.615	.0221	.865
.620	.0221	.865
.625	.0221	.865
.630	.0221	.865
.635	.0221	.865
.640	.0221	.865
.645	.0221	.865
.650	.0221	.865
.655	.0221	.865
.660	.0221	.865
.665	.0221	.865
.670	.0221	.865
.675	.0221	.865
.680	.0221	.865
.685	.0221	.865
.690	.0221	.865
.695	.0221	.865
.700	.0221	.865
.705	.0221	.865
.710	.0221	.865
.715	.0221	.865
.720	.0221	.865
.725	.0221	.865
.730	.0221	.865
.735	.0221	.865
.740	.0221	.865
.745	.0221	.865
.750	.0221	.865
.755	.0221	.865
.760	.0221	.865
.765	.0221	.865
.770	.0221	.865
.775	.0221	.865
.780	.0221	.865
.785	.0221	.865
.790	.0221	.865
.795	.0221	.865
.800	.0221	.865
.805	.0221	.865
.810	.0221	.865
.815	.0221	.865
.820	.0221	.865
.825	.0221	.865
.830	.0221	.865
.835	.0221	.865
.840	.0221	.865
.845	.0221	.865
.850	.0221	.865
.855	.0221	.865
.860	.0221	.865
.865	.0221	.865
.870	.0221	.865
.875	.0221	.865
.880	.0221	.865
.885	.0221	.865
.890	.0221	.865
.895	.0221	.865
.900	.0221	.865
.905	.0221	.865
.910	.0221	.865
.915	.0221	.865
.920	.0221	.865
.925	.0221	.865
.930	.0221	.865
.935	.0221	.865
.940	.0221	.865
.945	.0221	.865
.950	.0221	.865
.955	.0221	.865
.960	.0221	.865
.965	.0221	.865





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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U31)

ARC11-716 1A14 01+712+312M25+AT10 UPPER WING

ALPHA(2) = -0.120 BETA(7) = 1.700

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW			
.000	.0245	.0389	.1921 .1470 -.0163 -.1610 -.1030
.020			.4619 .2639 .3984 .3411
.025		.3471	
.040		.1666	
.045		.3125	
.050	.0344		.3104 .2934 .2797 .2137
.060			.1866
.083		.2341	
.086	.2405		
.094	.0678		-.0063 -.0109 -.0091 -.0909
.150		.0000	
.163		.0368	
.177		.1634	
.193			
.229	.1452		-.1543 -.1930 -.2332 -.3347
.230		-.0521	
.274		-.0371	
.339	.1502		-.2009 -.3014 -.4921
.400		-.1652	
.402		.0000	
.497	.0000		-.2469 -.3650
.550			
.585			
.600			
.650			
.700	-.0727		-.1722 -.1753
.725			
.750		-.1834	.0586 .0253
.760			
.775		-.0025	.0166 .0322
.808			
.834	-.1332		.0031 .0403 .0643
.850		-.0133	
.857			
.865	-.0151		.0078 .0064 .0313
.900	-.0148		
.905		-.0184	
.950		-.0252	
.953		-.0503	
.965	-.0464		

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

**(RB1U31)**

ARC11-716 1A14 01+T12+S12N2S+AT10 UPPER WING

ALPHA( 2 ) = -0.110 BETA( 0 ) = 3.340

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

1984	.2990	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

**MX/**

0000	.0200	.0622	.1773	.1100	-.0660	-.2037	-.1553
0020				.4562	.2426	.3687	.3079
0040			.3300				
0060		.1923					
0080			.3056				
0100	.0430		.3148	.2057	.2637	.1000	
0120			.1923				
0140			.2306				
0160		.2233					
0180	.0607			.0010	-.0025	-.0121	-.1041
0200		.0000	.0600				
0220		.1735					
0240	.1344			-.1510	-.1095	-.2273	-.3430
0260			-.0577				
0280		-.0336					
0300	.1391			-.2056	-.3045		-.5010
0320			-.1693				
0340	.0000			-.2716	-.3021		
0360			.0000				-.5440
0380						-.4532	
0400	-.1019			-.2143	-.1070		
0420			-.2172	.0010	.0240		.0300
0440			-.0302				
0460							
0480	-.1509						
0500							
0520							
0540							
0560							
0580							
0600							
0620							
0640							
0660							
0680							
0700							
0720							
0740							
0760							
0780							
0800							
0820							
0840							
0860							
0880							
0900							
0920							
0940							
0960							
0980							
1000							

(R81U31)

ARC11-716 1A14 01+112+J12N25+AT10 UPPER WING

ALPHA( 2 ) = -0.090 BETA( 9 ) = 4.950

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/8W .2990 .3740 .4270 .5340 6730 .7800 .8870

X/CW

.000	-.0452	.0303	.1549	.1093	-.0844	-.2347	-.1979
.020				.4457	.2287	.3497	.2849
.025			.3259				
.040			.1286				
.045				.2955			
.050	.0207			.3096	.2757	.2506	.1783
.060				.1879			
.065			.2251				
.086		.1989					
.094	.0530			.0039	-.0025	-.0094	-.1007
.150		.0000					
.163		.1650					
.177							
.193	.1221						
.229				-.1371	-.1809	-.2232	-.3385
.274		-.0313		-.0496			
.339							
.382	.1280						
.400				-.2010	-.2949		-.3025
.452			-.1685				
.497	.0000						
.550				-.2805	-.3869		
.565			.0000				
.600							-.5843
.650						-.5099	
.700	-.1296			-.3246	-.2111		
.725						-.0813	-.0283
.750							
.760			-.2945				
.775				-.0380	-.0448		
.808			-.0810				
.834	-.2027						
.850				-.0323	.0197	.0382	
.857			-.0643				
.865	-.1035						.0602
.930	-.0885			-.0098		.0145	
.955			-.0581				
.950				-.0348		.0274	
.953			-.0818				
.965	-.1187						

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TABULATED PRESSURE DATA - IAI14A - VOL. 6

(R81031)

ARC11-718 IAI14 01+T12+312M25+AT10 UPPER WING

DATE 10 JAN 75

ALPHA(2) = -0.80° CLAO (10) = 0.750

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

V/W	.2000	.3640	.4270	.5340	.6730	.7600	.8670
1/CM							
.000	-.0771	.0240	.1003	.0693	-.1313	-.2930	-.2590
.025				.4433	.2131	.3272	.2526
.050			.3135				
.075		.1008					
.100			.2909	.3163	.2771	.2448	.1671
.125				.2031			
.150			.2269				
.175		.1923					
.200				.0215	.0063	-.0027	-.1008
.225		.0325					
.250			.0550				
.275			.1669				
.300		.1128		-.1225	-.1660	-.2127	-.3331
.325			-.0222				
.350		.1202		-.2003	-.2896		-.5000
.375			-.1809				
.400		.0000		-.2907	-.3923		
.425							
.450		.0000					
.475							
.500							
.525							
.550							
.575							
.600							
.625							
.650							
.675							
.700							
.725							
.750							
.775							
.800							
.825							
.850							
.875							
.900							
.925							
.950							
.975							
.000							



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(R81U31)

ARC11-716 1A14 01+112+812N25+AT10 UPPER WING

ALPHA(2) = -0.090 BETA(11) = 8.370

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2995	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1036	.0193	.0556	.0613	-.1638	-.3333	-.3146
.020			.2935	.4362	.2017	.3062	.2240
.040		.0785					
.060			.2916				
.080	-.0162		.3209	.2756	.2340	.1490	
.100			.2364	.2140			
.120		.1702					
.140	.0181			.0427	.0197	-.0002	-.1032
.160		.0000					
.180		.1647	.0826				
.200	.1082			-.1011	-.1546	-.2036	-.3255
.220		-.0231	-.0256				
.240	.1069			-.1937	-.2857		-.4968
.260			-.1597				
.280	.0000		-.2974	-.3957			
.300			.0000				
.320							-.6177
.340	-.1693			-.3751	-.2363		
.360						-.2272	-.1333
.380			-.3304	-.1716	-.1648		
.400			-.1614				
.420	-.2553			-.0872	-.0346	-.0032	
.440			-.1327				
.460	-.1913			-.0440	.0155		.0515
.480	-.1571		-.1019	-.0476	.0433		
.500			-.1076				
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.995							

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TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+512M25+AT10 UPPER WING

(R81031)

ALPHA(3) = -6.100 BETA(1) = -8.140

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	.0716	.1418	.4039	.4669 .3840 .3014 .3305
.020				.4626 .4563 .4427 .4221
.025		.2832	.3992	
.040				
.045			.3019	
.050	.0910		.2194 .2107	.2190 .1697
.080		.1420	.0311	
.085				
.086		.2978		
.094	.0994			
.150		.0000		
.163				
.177				
.193		.1221		
.229	.1695			
.250				
.274				
.339				
.362	.1123			
.400				
.402				
.497	.0000			
.550				
.565				
.600				
.650				
.703	.0561			
.725				
.750				
.760				
.775				
.808				
.834	.0323			
.857				
.857				
.865	.1563			
.990	.1321			
.993				
.990				
.953				
.953				
.924				



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81031)

ARC11-716 1A14 01+T12+312M25+AT10 UPPER WING

ALPHA(3) = -6.110 BETA(2) = -6.480

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2993	.3640	.4270	.5340 .6730 .7600 .8870
X/CW				
.000	.0568	.1209	.3732	.4332 .3393 .2517 .2036
.020				.4579 .3956 .4327 .4034
.025			.3697	
.040		.2467		
.045			.3016	
.050	.0815		.2276 .2201	.2154 .1610
.060			.0906	
.085			.1509	
.086		.2867		
.094	.0895			
.130				
.163		.0500		
.177				
.193		.1287		
.229	.1613			
.251				
.274				
.339				
.362	.1142			
.400				
.402				
.497	.0000			
.550				
.565				
.600				
.650				
.700	.0266			
.725				
.750				
.760				
.775				
.808				
.834				
.850	.0015			
.857				
.865	.1289			
.900	.1053			
.925				
.930				
.953				
.965				



TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81031)

DATE 10 JAN 75

ARC11-716 IA14 01+112+512M25+AT10 UPPER WING

ALPHA(3) = -8.135 BETA(3) = -4.823

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3840	.4270
.3340	.5340	.6750
.7000	.8670	.0670
.0000	.0459	.1037
.0200	.3395	.3667
.0250	.2903	.1978
.0400	.2332	.2332
.0450	.4598	.4211
.0500	.3621	.3872
.0600	.2321	
.0650	.3063	
.0800	.2369	.2149
.0860	.2232	.1587
.0940	.0615	
.1094	.1650	
.1150	.2782	
.1630	.0000	
.1770	-.0728	
.1930	.1346	
.2290	.1564	
.2740	-.1770	
.3390	-.1395	
.3620	.362	
.4000	.1137	
.4020	-.2384	
.4970	-.2214	
.5500	-.1847	
.5650	.0000	
.6000		
.6500		
.7000		
.7250		
.7500		
.7750		
.8000		
.8340		
.8500		
.8570		
.8650		
.9000		
.9050		
.9500		
.9750		
.9850		



(R01U31)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA( 3 ) = -0.140 BETA( 4 ) = -3.220

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0405	.0922	.3167	.3522	.2418	.1461	.1895
.020				.4526	.3502	.4120	.3737
.025			.3734				
.040		.2138					
.045			.3041				
.050	.0623			.2383	.2302	.2192	.1572
.060				.0768			
.065			.1669				
.066		.2686					
.094	.0707						
.150							
.163		.0300					
.177			.0333				
.133			.1393				
.229	.1489						
.250							
.274							
.339							
.362	.1155						
.400							
.432							
.497	.0000						
.550							
.565							
.600							
.650							
.700							
.725							
.750							
.760							
.775							
.808							
.834							
.850							
.857							
.865							
.900							
.914							
.933							
.965							

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 8

ARC11-716 1A14 01+112+312N29+AT10 UPPER WING (RB10511)

ALPHAO ( 3 ) = -6.030 BETAO ( 6 ) = .000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

1/OW	.2990	.3440	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0286	.0841	.2756	.2807	.1446	.0443	.1023
.020				.4275	.2860	.3767	.3315
.025			.3416				
.040		.1845					
.045			.2820				
.050	.0464			.2367	.2196	.2100	.1390
.060				.0878			
.065			.1661				
.066		.2409					
.094	.0968						
.130				-.1046	-.1187	-.1140	-.1908
.163		.0020					
.177			-.0300				
.193		.1344					
.229	.1333			-.2321	-.2913	-.3443	-.4360
.250							
.274			-.1436				
.339		-.1176					
.362	.1130			-.2534	-.3902		-.5796
.400			-.2249				
.432							
.497	.0000			-.2716	-.4047		
.530			.0000				
.545							
.600							
.630							
.700	-.0837						-.5172
.725				-.1675	-.1664	-.3345	
.750							
.760			-.1670	.0337	.0572		
.775			.0146			.0016	.0594
.804							
.834	-.1276						
.850				.0210	.0422	.0902	
.857			.0025				
.865	.0003						-.3800
.900	-.0012			.0220	.0215		
.905			-.0032				
.950				-.0132		.0497	
.955			-.0403				
.965	-.0703						

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBIUS1)

ARC11-716 1A14 OI+T12+312N25+AT10 UPPER WING

ALPHAD(3) = -0.030 BETA(7) = 1.640

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6790	.7800	.8070
X/CW							
.000	.0103	.0420	.2592	.2436	.1065	-.0047	.0493
.020			.3167	.4056	.2567	.3593	.3141
.025		.1682					
.040			.2629	.2233	.2111	.2028	.1329
.045				.0466			
.050	.0343						
.060			.1555				
.065		.2237					
.066							
.094	.0474						
.150		.0000					
.163			-.0332				
.177		.1261					
.193	.1237						
.229							
.250							
.274							
.339		-.1076					
.362	.1048						
.400							
.402							
.497	.0000						
.555							
.565							
.600							
.650							
.700	-.1101						
.725							
.750							
.760							
.775							
.806							
.834	-.1621						
.870							
.857							
.865	-.0376						
.900	-.0346						
.905							
.955							
.955							
.965							
.965							



(1501031)

ARC11-718 1A14 01+T12+S12N23+A110 UPPER WING

ALPHA( 3 ) = -6.100 BETA( 8 ) = 5.330

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP				
Y/BW	.2990	.5640	.4270	.5340	.6730	.8870
X/CW						
.000	-.0113	.0759	.2397	.2125	.0475	-.0368
.020				.3986	.2312	.3406
.025			.3031			.2849
.040		.1325				
.045			.2573			
.050	.0293			.2324	.2122	.1961
.080			.1635	.1032		.1223
.085						
.086		.2045				
.094	.0402					
.150				-.0812	-.0938	-.1008
.163		.0000				-.1848
.17~			-.0163			
.193		.1280				
.229	.1179					
.250				-.2257	-.2661	-.3111
.274			-.1187			-.4123
.339		-.0967				
.362	.1033					
.400				-.2486	-.3630	-.5686
.402			-.2207			
.497	.0000					
.550				-.3002	-.4209	
.565			.0000			-.5931
.600						
.650					-.5363	
.700	-.1318				-.2125	
.725				-.3543		-.0057
.750			-.2682		-.0598	
.780				-.0597	-.0344	
.775			-.0630			
.808						
.834	-.2150					
.850				-.0112	.0210	.0415
.857			-.0404			
.865	-.0986					.0542
.900	-.0748			-.0067	.0037	
.905			-.0406			.0191
.950				-.0302		
.953			-.0623			
.965	-.0983					

## TABULATED PRESSURE DATA - 1A144 - VOL. 6

(R01031)

ARC11-716 1A14 01-712+512NBS+AT10 UPPER WING

DATE 10 JAN 75

ALPHA(3) = -6.163 BETA(9) = 9.010

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW		.3640	.4270	.5340	.6730 .7800 .8870
X/CW					
.000	-.0490	.0516	.1895	.2052	.0308 -.0973 -.0496
.020				.3979	.2109 .3204 .2857
.025			.2964		
.040		.1222			
.045			.2563		
.050	.0324			.2349	.2043 .1851 .1142
.080				.1046	
.085			.1672		
.086		.1873			
.094	.0362			-.0769	-.0910 -.0993 -.1782
.150		.0000			
.163			-.0029		
.177		.1274			
.193	.1064				
.229				-.2152	-.2561 -.3020 -.4036
.250			-.1045		
.274		-.0856			
.339	.0987			-.2512	-.3832 -.5603
.362			-.2173		
.400				-.3150	-.4272
.497	.0000		.0000		
.550					-.6066
.565				-.5463	
.600					
.650	-.1519		-.2076		
.700			-.3598		-.0993 -.0332
.725					
.750			-.3029		
.760			-.0872	-.0947	
.775					
.808			-.1120		
.834	-.2434			-.0289	-.0115 .0207
.850			-.0754		
.857					
.865	-.1527				
.900	-.1115			-.0121	
.905			-.0642	-.0050	.0611
.930			-.0332		.0093
.953			-.0749		
.969	-.1190				







DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U031)

ARC11-716 1A14 01-T12-S12N25+AT10 UPPER WING

ALPHAO(3) = -6.140 BETA0(11) = 0.500

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP
V/BW	.2990 .3640 .4270 .5340 .6750 .7800 .8870
X/CW	.0000 .1315 .0283 .0854 .1340 -.0411 -.1965 -.1555
.020	.025 .2393
.040	.0720
.045	.2247
.050	.2541 .2136 .1830 .1018
.080	.1414
.085	.1778
.086	.1407
.094	.0014
.150	.0000
.163	.0276
.177	.1301
.193	.0866
.229	-.0738
.250	-.1591 -.2167 -.2719 -.3785
.274	-.0541
.339	.0846
.362	-.1958
.403	-.2334 -.3723 -.5519
.402	.0000
.497	-.3242 -.4273
.550	.0000
.565	-.6341
.600	-.5573
.650	-.2381
.700	-.3932
.723	-.3368
.750	-.2534 -.2894
.763	-.2279
.775	-.1592
.808	-.1213 -.0847 -.0502
.834	-.2833
.850	-.2329
.857	-.1960
.865	-.0858
.900	-.1258
.905	-.0056
.950	-.0984
.953	-.0092
.965	-.1189
.965	-.1616



DATE 15 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 Q1+T12+S12N25+AT10 UPPER WING (R81031)

ALPHA( 4) = -4.170 BETA( 1) = -9.980

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3660 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0985	.1917	.4713	.5774	.5293	.4704	.4834
.020				.3714	.3736	.3438	.3324
.025		.3472					
.040		.2710					
.045			.2313				
.050	.0874		.0991	.0832	.0834	.0293	
.060			-.1069				
.085		.0439					
.086		.2756					
.094	.0919						
.130		.0000		-.3442	-.3533	-.3264	-.3229
.163			-.2232				
.177		.0558					
.193							
.229	.1576			-.4698	-.5332	-.5868	-.5844
.230							
.274		-.3212					
.339		-.2625					
.362	.0686			-.2584	-.4080		-.8245
.400							
.432		-.2799					
.497	.0000			-.1090	-.1844		
.550		.0000					
.563							
.605							
.650							
.700	.0605			-.0706			-.0897
.725			.0134	-.1349			
.750							
.760		.0149			.1409	.0526	
.775			.1641	.1472			
.808		.1601					
.834	.0593						
.850			.1313	.1430	.1458		
.857		.1348					
.865	.1804						
.900	.1569		.1199	.0891		.1103	
.905		.1081					
.920			.0705			.1095	
.933		.0591					
.965	.0421						

(R81U31)

DATE 10 JAN 75  
TABULATED PRESSURE DATA - IA14A - VOL. 6  
ARC11-716 IA14 OR-T112+312N25+AT10 UPPER WING

ALPHA( 4 ) = -6.190 BETA( 2 ) = -7.970

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3440	.4270	.5340 .6750 .7800 .8870
X/CW				
.000	.0795	.1721	.4464	.5451 .4945 .4194 .4375
.020			.3745	.3943 .3451 .3220
.025		.3445		
.040		.2567		
.045			.2314	
.050	.0726		.1127	.0917 .0867 .0327
.080			-.0916	
.085		.0547		
.086		.2683		
.094	.0804			
.150		.0000	-.3216	-.3306 -.3169 -.3181
.163		-.2060		
.177		.0618		
.193				
.229	.1494		-.4571	-.5382 -.5706 -.5802
.250		-.2999		
.274				
.339	-.2352			
.362	.0693		-.2738	-.4275
.400		-.2857		-.8188
.402				
.497	.0000		-.1535	-.2494
.520		.0000		
.565				-.0793
.600			-.0828	
.650			-.1347	
.700	.0322		-.0164	
.725				
.750		-.0146		.1311 .0476
.760			.1398	.1314
.775		.1324		
.808				
.834	.0309		.1125	.1237 .1340
.850		.1114		
.857				
.885	.1514			
.900	.1316		.1028	.0776 .1104
.905		.0869		
.930		.0564		.0909
.935		.0404		
.965	.0230			





ARC11-716 1A14 OR+T12+512N25+AT10 UPPER WING

(R81U31)

ALPHA( 4) = -4.125 BETA( 4) = -3.980

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2900	.3140	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	.0918	.1331	.3834	.4597 .3733 .2982 .3247
.020			.3333	.3783 .3899 .3269 .2932
.040		.2232		
.045			.2346	
.050	.0535			.1361 .1080 .0930 .0298
.080				-.0389
.085			.0816	
.086		.2306		
.094	.0588			
.150				-.2580 -.2833 -.2799 -.3166
.163		.0000		
.177			-.1592	
.193		.0763		
.229	.1328			
.250				-.4086 -.4713 -.5199 -.5635
.274			-.2597	
.339		-.2226		
.362	.0144			
.401			-.2978 -.4773	-.7367
.402			-.3047	
.497	.0000			
.530			-.2494 -.3906	
.565			.0000	
.600				-.1154
.630				-.1182
.700	-.0478		-.1384	
.725			-.0441	
.750				.1126 .0437
.760			-.0835	
.775			.0883 .0949	
.808		.0777		
.831	-.0551			
.850			.0686 .0939 .1104	
.857		.0382		
.885	.0712			
.900	.0628		.0679	.0967
.925		.0412	.0486	
.950			.0266	.0693
.955		-.0020		
.963	-.0252			



(RBU31)

DATE 10 JAN 75 TABULATED PRESSURE DATA - IALIA - VOL. 6  
ARC11-716 IALIA 01+T12+S12N25+AT10 UPPER WING

ALPHA( 4 ) = -4.100 BETA( 5 ) = -1.900

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

T/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	.0403	.1178	.3564	.4095	.3199	.2378
.020					.3757	.3808	.3252
.040			.3243				.2893
.060		.2051					
.080			.2353				
.100		.0443		.1474	.1261	.1091	.0430
.120					-.0204		
.140			.0970				
.160		.2387					
.180	.0508						
.200							
.220							
.240							
.260							
.280							
.300							
.320							
.340							
.360							
.380							
.400							
.420							
.440							
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
1.000							

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(RB1031)

TABULATED PRESSURE DATA - IAL4A - VOL. 6

ARC11-716 IAL4 01+7112+512N23+AT10 UPPER WING

DATE 10 JAN 75

ALPHA (4) = -4.170 BETA (7) = 2.020

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
r/w	.2995	.3640	.4273	.5340 .6730 .7800 .8870
x/cw				
.000	.0041	.0940	.3044	.3336 .2144 .1135 .1560
.020			.2732	.3386 .3262 .2919 .2545
.040		.1653		
.060			.2040	.1339 .1193 .1053 .0364
.080				-.0054
.100		.2006	.0858	
.120	.0365			-.1982 -.2141 -.2142 -.2900
.140		.0000		
.160			-.1141	
.180		.0746		
.200	.1083			-.3261 -.3806 -.4274 -.5243
.220		-.2083		
.240		-.1733		
.260	.0691			-.5023 -.4703 -.6541
.280			-.2896	
.300	.0000			-.3209 -.4618
.320			.0000	-.6141
.340				-.5635
.360	-.1377			-.2173
.380			-.3744	-.0386 -.0115
.400				
.420			-.2559	-.0406 -.0102
.440			-.0446	
.460				-.0034 .0397 .0488
.480	-.2110		-.0212	
.500				
.520				-.0033 .0031 .0335
.540	-.0844		-.0227	-.0190
.560	-.0396			-.0315
.580				
.600				
.620				
.640				
.660				
.680				
.700				
.720				
.740				
.760				
.780				
.800				
.820				
.840				
.860				
.880				
.900				
.920				
.940				
.960				
.980				
.995				

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(R01031)

ARC11-716 1A14 OILY-112-112-112-112 UPPER WING

ALPHA(4) = -4.200 BETA(8) = 4.040

SECTION 1 UPPER WING DEPENDENT VARIABLE CP

Y/CX .2990 .3640 .4270 .5340 .6730 .7800 .8670

X/CX

.000	-.0499	.0797	.2375	.3093	.1646	.0662	.1023
.020			.2943	.3372	.2950	.2714	.2276
.040		.1302					
.045			.2059				
.050	.0502			.1469	.1201	.0961	.0240
.080				.0103			
.085			.1012				
.087		.1760					
.094	.0305						
.150		.0000		-.1701	-.1942	-.2009	-.2074
.163							
.177		.0654		-.0802			
.193	.0983						
.229							
.250				-.2904	-.3517	-.4086	-.5146
.274							
.339		-.1476		-.1793			
.342	.0685						
.400				-.2934	-.4612		-.6307
.402							
.497	.0000		-.2731				
.550							
.583			.0000	-.3334	-.4682		
.600							
.650							
.700	-.1611						
.725				-.3619	-.2377		
.750							
.760			-.3222	-.1636	-.1430		
.775				-.1544			
.806							
.834	-.2770			-.0532	-.0096	-.3014	
.850							
.857				-.0868			
.865	-.1831						
.900	-.1351			-.7306			.0498
.925				-.0711	.0040		
.950							
.955				-.0467		.0200	
.965	-.1446			-.0983			



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+112+312N25+AT10 UPPER WING (RB1031)

ALPHA(4) = -4.230 BETA(9) = 6.030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/Z	.2990	.3640	.4270	.5340	.6730	.7900	.8870
X/CW							
.000	-.1025	.0329	.1336	.3000	.1304	.0071	.0400
.020			.2113	.3244	.2662	.2643	.2147
.025		.0921					
.040			.1867				
.045				.1562	.1279	.1054	.0325
.050	-.0328			.0299			
.060			.1120				
.085		.1391					
.094	.0319						
.150		.0000		-.1405	-.1695	-.1764	-.2659
.163			.0504				
.177		.0890					
.229	.0070			-.2675	-.3239	-.3807	-.4814
.250			-.1536				
.274		-.1181					
.339							
.362	.0652			-.2876	-.4508		-.6150
.400			.2454				
.402				-.3413	-.4688		
.497	.0000		.0000				
.550						-.5881	-.6382
.565				-.2438			
.600				-.3934			
.650							
.700	-.1784						
.725			-.3175				
.750			-.2260				
.771				-.2885	-.2493		
.808							
.834	-.2719			-.1099	-.0742	-.0549	
.850			-.1564				
.857							
.865	-.2411						
.900	-.1924			-.0710			.0083
.905			-.1172	.0016			
.950			-.0757			-.0005	
.955			-.1333				
.965	-.1702						

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DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14A - VOL. 6

(081031)

ARC11-716 IAI14 01+712+512M25+AT10 UPPER WING

ALPHA(4) = -4.200 BETA(10) = 0.070

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/8W .2990 .3640 .4270 .5340 .6730 .7000 .8070

X/CW

.000	-.1449	.0340	.1700	.2151	.0926	-.0465	-.0150
.020				.2983	.2337	.2532	.1961
.025			.1803				
.040		.0675					
.045			.1704				
.050	-.0407			.1585	.1376	.1078	.0265
.060			.1119	.0539			
.065							
.066		.1176					
.094	-.0131						
.130		.0500		-.1112	-.1460	-.1626	-.2564
.163							
.177		.0661					
.193							
.229	.0714			-.2332	-.2975	-.3545	-.4645
.250							
.274			-.1315				
.339		-.1043					
.362	.0604			-.2769	-.4432		-.6080
.400							
.402			-.2331				
.497	.0000			-.3509	-.4732		
.550			.0000				
.565							-.6538
.600						-.5967	
.650							
.700	-.2040			-.4011	-.2532		
.725						-.2914	-.1616
.750							
.760			-.3177	-.3149	-.2865		
.775			-.2408				
.808							
.834	-.2968			-.1468	-.0933	-.0632	
.850			-.1879				
.857							
.865	-.2653			-.0961			.0047
.900	-.2213						
.905			-.1412	-.0272			
.950			-.0637			.0099	
.953			-.1447				
.965	-.1613						



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI4A - VOL. 6

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(R81031)

ARC11-716 IAI4 01+112+512M23+AT10 UPPER WING

ALPHA(4) = -4.200 PETA(11) = 10.080

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3610	.4270	.5340	.5730	.7800	.8070
X/CW							
.000	-.2086	.0274	.0980	.1363	.0663	-.0855	-.0713
.020				.2063	.2007	.2458	.1020
.025			.1668				
.040		.0599					
.045			.1560				
.050	-.0885			.1703	.1470	.1162	.0365
.060			.0786				
.085			.1067				
.086		.1084					
.094	-.0229			-.0818	-.1226	-.1383	-.2367
.150		.0000					
.163			-.0161				
.177		.0797					
.193							
.229	.0628			-.2060	-.2736	-.3326	-.4445
.250			-.1125				
.274		-.0979					
.352	.0519			-.2711	-.4319		-.5917
.359			-.2223				
.400				-.3548	-.4706		
.497	.0000		.0000				
.550							
.565							
.600							
.650							
.700	-.2200				-.5959		-.6619
.725				-.4169	-.2576		
.750							
.760			-.3518				
.775			-.3461				
.806			-.2898				
.834	-.3123						
.850				-.1894	-.1226	-.1033	
.857			-.1968				
.865	-.2926						
.900	-.2517			-.1139			-.0361
.905			-.1572		-.0244		
.930				-.0680		.0033	
.953			-.1376				
.965	-.1939						

(081031)

ARC11-716 1A14 01+T12+S12M25+AT10 UPPER WING

ALPHA(1) = -2.870 BETA(1) = -9.990

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP		
Y/BW				
X/CW				
.000	.0976	.2030	.4927	.6170 .5836 .5338 .5400
.020			.3027	.4097 .2375 .2360
.025			.2966	
.040		.2618		
.045			.1733	
.050	.0731		.0171	-.0094 -.0225 -.0823
.070			-.1908	
.085			-.0207	
.086		.2573		
.094	.0779			
.130		.0000	-.4273	-.4657 -.4247 -.4116
.163		.0144	-.2869	
.177				
.193				
.229	.1437		-.5323	-.5907 -.6659 -.6948
.250			-.3754	
.274		-.3461		
.339				
.362	.0412		-.2814	-.3715 -.9070
.400			-.3416	
.402				
.497	.0000		-.1158	-.1841
.550			.0000	
.563				-.1574
.600				-.0198
.630			-.1374	
.700	.0442		.0184	.1608 .0826
.725				
.750			.0255	.1627 .1489
.760			.1593	
.775				
.806	.0598		.1306	.1450 .1319
.834		.1356		
.850				
.857				
.865	.1856		.1187	.0964 .1215
.900	.1646		.1123	.0698 .1114
.905			.0614	
.930				
.933				
.965	.0452			



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881031)

DATE 12 JAN 75

ALPHAD( 5) = -2.890 BETAD ( 2) = -7.990

AQC11-716 1A14 01+112+512M25+AT1D UPPER WING

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0765	.1616	.4694	.5875	.5394	.4913	.4953
.020				.3076	.3797	.2533	.2299
.025		.2457					
.040			.1767				
.045				.0332	-.0000	-.0136	-.0776
.050	.0535			-.1771			
.060			-.0079				
.065		.2440					
.086				-.4099	-.4383	-.4020	-.4084
.094	.0646						
.150		.0000					
.163			-.2691				
.177		.0159					
.193							
.229	.1316			-.5181	-.5909	-.6524	-.6460
.250			-.3635				
.274		-.3299					
.339	.0361			-.3189	-.5282		-.8995
.362							
.400			-.3789				
.402				-.1727	-.2650		
.497	.0000			.0000			
.550							
.565							
.600							
.650							
.700	.0101			-.0117	-.1364		-.1760
.725							
.750							
.760			-.0076				
.775			.1397	.1315			
.808			.1312				
.834	.0222			.1122	.1298	.1393	
.850			.1118				
.857							
.865	.1432			.1045			.1120
.900	.1311						
.905			.0845	.0791			
.950			.0543				.0959
.955			.0377				
.965	.0837						

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 G1-712-S12N29+AT10 UPPER WING  
(R81031)

ALPMAO( 5) = -2.870 BETAO ( 3) = -3.970

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW		3640	4270	5340	6730 7800 8870
X/CW					
.000	.0820	.1809	.4484	.5465	.4980 .4373 .4493
.020				.3135	.3442 .2324 .2197
.025		.2361	.2941		
.040					
.045			.1851		
.050	.0469			.7484	.0593 -.0063 -.0766
.060				-.1514	
.085			.0144		
.086		.2420			
.094	.0593				
.150		.0500		-.3714	-.4064 -.3925 -.4040
.163			-.2418		
.177		.0296			
.193					
.229	.1302			-.5043	-.5891 -.6371 -.6389
.250			-.3440		
.274		-.3030			
.339					
.362	.0430			-.3347	-.3015 -.6883
.400		-.3750			
.402	.0500				
.497				-.2271	-.3756
.553		.0000			
.563					
.600					
.650					
.700	-.0235			-.0474	-.1366
.725					
.750					
.760		-.0420		.1147	.1198
.775			.1033		
.808					
.834	-.0148			.0938	.1141 .1289
.850			.0914		
.857					
.865	.1064				
.900	.0968			.0841	
.905			.0731	.24	.1072
.950			.0415		.0833
.953		.0231			
.965	-.0504				



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI14 - V01, 6

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(R81U31)

ARC11-716 IAI4 1\*712\*512H25\*AT10 UPPER WING

ALPHA (S) = -2.060 BETAD (A) = -3.980

SECTION 11 UPPER WING		DEPENDENT VARIABLE CP			
X/CW	Y/SW				
.000	.2990	.3640	.4270	.5340	.6750 .7900 .8870
.020	.0425	.1432	.4058	.5033	.4403 .3777 .3972
.025			.2860	.3143	.3130 .2536 .2133
.040		.2153			
.045			.1635		
.050	.0375			.0577	.0285 .0061 -.0720
.060					-.1299
.065			.0271		
.066		.2275			
.094	.0457				
.150		.0000		-.3493	-.3675 -.3704 -.3949
.163			-.2226		
.177		.0371			
.193					
.229	.1176				
.250				-.4773	-.5591 -.6131 -.6379
.274			-.3321		
.339		-.2776			
.362	.0435				
.400				-.3424	-.5033
.402			-.3751		-.8647
.497	.0000				
.550				-.2693	-.4533
.565			.0000		
.650					-.1367
.700	-.0630			-.1099	
.725			-.0748	-.1467	
.750					
.760		-.0636			.1257 .0764
.775			.0924	.1017	
.800		.0726			
.834	-.0461			.0737	.0965 .1156
.850		.0608			
.857					
.865	.0720				
.900	.0655		.0439	.0491	.1027
.925					
.950			.0230		.0664
.955		.0001			
.965	-.0165				



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01031)

ARC11-716 1A14 Q1+112+312N5+AT10 UPPER WING

ALPHA( 5) = -2.840 BETA( 5) = -1.990

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/W	X/CW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
.000	.0335	.1249	.3803	.4598	.3903	.3218	.3437	
.020				.3142	.2618	.2400	.2100	
.025			.2832					
.040		.2024						
.045			.1906					
.050	.0290			.0750	.0370	.0165	-.0632	
.080								
.085								
.086			.2192					
.094	.0408							
.150								
.163		.0500						
.177								
.193		.0439						
.229	.1116							
.250								
.274								
.339								
.362	.0492							
.400								
.402								
.487	.0000							
.550								
.565								
.610								
.650								
.700								
.725								
.750								
.760								
.775								
.808								
.834								
.850								
.857								
.865								
.900								
.905								
.930								
.933								
.965								



(R81031)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 OL+112+S12N25+AT10 UPPER WING

ALPHA( 5 ) = -2.840 BETA( 6 ) = .010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.42	.5340	.6730	.7800	.8870
X/CW							
.020	.0140	.1088	.3522	.4153	.3248	.2950	.2872
.020			.3057	.2355	.2454	.2029	
.025		.2649					
.040		.1806					
.045			.1768				
.050	.0197		.0794	.0448	.0270	-.0347	
.060			-.0878				
.085			.0406				
.086		.2043					
.094	.0297						
.150		.0000		-.2812	-.3157	-.3280	-.3747
.163			-.1803				
.177		.0434					
.193							
.229	.1037						
.250				-.4189	-.4891	-.5531	-.6056
.274		-.2783					
.339	-.2339						
.362	.0463			-.3476	-.5108		-.7725
.400							
.402	.0000		-.3446				
.497				-.3218	-.4871		
.550		.0000					
.565							
.600							
.650							
.700	-.1257						
.725				-.2452	-.1948		
.750							
.760							
.775							
.808							
.834	-.1845						
.850							
.857							
.865							
.900	-.0074						
.905							
.950							
.953							
.965	-.0852						

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TABULATED PRESSURE DATA - IA14A - VOL. 6

(R01031)

ARC11-716 IA14 01-T12-S12N25-AT10 UPPER WING

ALPHA (°) = -2.000 BETA (°) = 4.050

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/SW	X/CW	CP
.050	-.0620	.0617
.020		.2343
.025		.3642
.040		.2449
.045		.1569
.050		.1810
.080		.2771
.085		.2039
.090		.1694
.095		.2031
.100		.1109
.105		.1498
.110		.0781
.115		.0454
.120		.0169
.125		-.0349
.130		-.0694
.135		.0446
.140		.1466
.145		.0092
.150		.0090
.155		-.1338
.160		.0492
.165		.0439
.170		-.2253
.175		-.1883
.180		.0422
.185		-.3344
.190		-.4947
.195		-.6896
.200		-.3079
.205		.0000
.210		-.3515
.215		-.5080
.220		-.6493
.225		-.5925
.230		-.2377
.235		-.4038
.240		-.1891
.245		-.1366
.250		-.3204
.255		-.1726
.260		-.1580
.265		-.0949
.270		-.0716
.275		-.0278
.280		-.0180
.285		-.2147
.290		-.1345
.295		-.0281
.300		-.0076
.305		-.0635
.310		-.0363
.315		-.0397
.320		-.0772
.325		-.1090

(RBIUS1)

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TABULATED PRESSURE DATA - IA14A - VOL. 6

ARC11-716 IA14 01+712+512N23+AT10 UPPER WING

ALPHA( 5) = -2.870 BETA( 9) = 6.960

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7820	.8870
Y/CW	.000	.0414	.1589	.3443	.2077	.0988	.1179
.020	.020	.0795	.1488	.2462	.1945	.2034	.1627
.040	.040	.0795	.1156	.0727	.0520	.0284	-.0471
.060	.060	-.0393	.0459	-.0553			
.080	.080	.1061					
.100	.100	-.0096		-.2121	-.2416	-.2597	-.3433
.120	.120	.0000					
.140	.140	.0451		-.0996			
.160	.160	.0632					
.180	.180	.229		-.3191	-.3949	-.4472	-.5082
.200	.200	.274		-.1941			
.220	.220	.339					
.240	.240	.362					
.260	.260	.403		-.3223	-.4904		-.6431
.280	.280	.402		-.2769			
.300	.300	.497					
.320	.320	.550		-.3631	-.5082		
.340	.340	.565		.0000			
.360	.360	.600					
.380	.380	.650					
.400	.400	.700		-.1992			
.420	.420	.725					
.440	.440	.750		-.4042	-.2441		
.460	.460	.775					
.480	.480	.804		-.3320	-.2813	-.2386	
.500	.500	.834		-.2365			
.520	.520	.859					
.540	.540	.857		-.1329			
.560	.560	.865		-.1180	-.0745	-.0484	
.580	.580	.900		-.0757			
.600	.600	.925		-.1123	-.0093		
.620	.620	.950		-.0601			
.640	.640	.955		-.1061			
.660	.660	.965					
.680	.680						
.700	.700						
.720	.720						
.740	.740						
.760	.760						
.780	.780						
.800	.800						
.820	.820						
.840	.840						
.860	.860						
.880	.880						
.900	.900						
.920	.920						
.940	.940						
.960	.960						
.980	.980						
.990	.990						
.995	.995						
.999	.999						
1.000	1.000						



DATE 10 JAN 75 TABULATED PRESSURE DATA 1A14A - VOL. 6  
 ARC11-716 1A14 01-Y12-S12N25+AT1D UPPER WING (RB1U31)

ALPHA(D) = -2.875 BETA(D) = 8.070

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.1752	.0304	.1210	.2491	.1789	.0515	.0703	
.020				.2395	.2031	.2013	.1519	
.040			.0344	.1301				
.060				.1095				
.080	-.0647			.0919	.0708	.0387	-.0576	
.100				-.0145				
.120			.0936	.0571				
.140	-.0230							
.160		.0000			.1615	-.2109	-.2256	-.3251
.180				-.0686				
.200		.0502						
.220	.0387				-.2759	-.3558	-.4171	-.5300
.240			-.1305					
.260	.0372				-.3058	-.4759		-.6453
.280			-.2504					
.300		.0000			-.3622	-.4979		
.320			.0000					-.6555
.340						-.6096		
.360	-.2115				-.3946	-.2458		
.380						-.2921	-.2131	
.400			-.3335		-.3313	.31		
.420			-.2825					
.440					-.1816	-.1030	-.0845	
.460	-.3140			-.2147				
.480								-.0057
.500	-.2959				-.0953			
.520	-.2357					-.0184		
.540				-.1517	-.0700		-.0127	
.560				-.1325				
.580	-.1764							

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R01031)

ARC11-716 1A14 CR-T18-S1825-A710 UPPER WING

ALPHA(5) = -2.830 BETA(11) = 10.090

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP
1/8W	.2990 .3440 .4270 .5340 .6750 .7800 .8970
X/CU	
.000	-.2302 .0182 .0941 .1474 .1953 .0099 .0106
.020	.025 .1144 .1972 .1944 .1984 .1394
.040	.0306 .0963 .0619
.045	
.050	-.1014 .0625 .0759 .0486 -.0310
.060	.0065
.085	
.066	.0606
.094	-.0490
.120	
.163	.0050
.177	-.0565
.193	.0473
.229	.0491
.250	
.274	-.1486
.339	
.362	.0293
.400	-.2454
.402	
.497	.0000
.530	
.565	-.3756 -.4993
.600	.0000
.650	
.700	-.2306
.725	
.750	
.775	
.808	
.834	-.3504
.850	
.857	-.2669
.865	
.908	-.2315
.905	
.920	-.1737
.933	-.0945
.965	-.1491



0

DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI4A - VOL. 6

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(R01031)

ARC11-716 IAI4 01+7:2+912N23+AT10 UPPER WING

ALPHAO (6) = -.000 BETAO (1) = -10.000

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7000	.8070
X/CW							
.000	.0906	.2000	.5123	.6896	.6410	.5966	.5939
.020			.1929	.3193	.0962	.0534	
.025		.2231					
.040		.2416					
.045			.0815				
.050	.0490			-.1060	-.1735	-.1972	-.2843
.060				-.3243			
.085			-.1228				
.086		.2175					
.094	.0806						
.190				-.5793	-.6430	-.5994	-.5632
.163		.0000					
.177			-.3737				
.193		-.2943					
.229	.1251						
.250				-.6672	-.7664	-.8217	-.7811
.274		-.5016					
.339							
.362	-.0040	-.4434					
.400				-.6724	-.8034		-.9364
.402			-.5646				
.497	.0000			-.0995	-.1802		
.590			.0000				
.565							
.600							-.4086
.690					-.1646		
.700	.0234			-.0306	-.1412		
.725						.0910	-.1255
.790			.0414				
.760				.1529	.1305		
.775			.1604				
.809							
.834	.0050						
.890			.1419	.1323	.1412	.1264	
.957							
.865	.1913						
.908	.1709		.1179				.0547
.965		.1176		.0967			
.990		.0722					.1199
.935		.0710					
.965	.0540						



(081051)

ARC11-716 1A14 01+712+912M25+AT10 UPPER WING

ALPHA( 6) = -.680 BETA( 2) = -7.980

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP	
Y/W	X/CW		
.8900	.3640	.4870	.5340 .6730 .7800 .8870
.0674	.1843	.4872	.6348 .6070 .5620 .5814
.025		.2506	.1944 .2584 .0924 .0556
.040	.2275	.0869	
.045			
.039	.0332	-.0812	-.1331 -.1920 -.2818
.080		-.3041	
.085		-.1069	
.086	.2106		
.094	.0456		
.150			
.163	.0000	-.5472	-.6141 -.5814 -.5006
.177		-.3593	
.193	-.0513		
.229	.1150		
.250		-.6396	-.7351 -.8027 -.7774
.274		-.4813	
.339	-.4286		
.362	-.0056		
.400		-.6396	-.8869 -1.0090
.402		-.5918	
.497	.0000		
.550		-.1668	-.2328
.565		.0000	
.600			
.650			
.700	-.8116		
.725		-.0012	-.1501
.750			
.760		.0105	.0917 -.0353
.775		.1341	.1282
.808		.1356	
.834	.0371		
.850		.1147	.1314 .1495
.857		.1211	
.885	.1680		
.900	.1474		
.905		.0988	.0844 .1089
.950		.0990	
.953		.0547	.1041
.983	.0393	.0512	



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(081051)

ARC11-716 1A14 OR-Y12-S12N25-AT10 UPPER WING

ALPHA(0) = -.670 BETA(3) = -5.960

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.8995	.8640	.8270	.7930	.7600	.8870
X/CW	.0000	.1622	.4805	.6037	.5653	.5253
.020	.020	.2173	.2000	.3186	.0916	.0406
.040	.040	.2191	.0966			
.060	.060	.0224		-.0604	-.1446	-.1900
.080	.080			-.2816		-.2847
.100	.100					
.120	.120					
.140	.140					
.160	.160					
.180	.180					
.200	.200					
.220	.220					
.240	.240					
.260	.260					
.280	.280					
.300	.300					
.320	.320					
.340	.340					
.360	.360					
.380	.380					
.400	.400					
.420	.420					
.440	.440					
.460	.460					
.480	.480					
.500	.500					
.520	.520					
.540	.540					
.560	.560					
.580	.580					
.600	.600					
.620	.620					
.640	.640					
.660	.660					
.680	.680					
.700	.700					
.720	.720					
.740	.740					
.760	.760					
.780	.780					
.800	.800					
.820	.820					
.840	.840					
.860	.860					
.880	.880					
.900	.900					
.920	.920					
.940	.940					
.960	.960					
.980	.980					
.1000	.1000					

ARC11-716 1A14 01-112-S12N25-AT10 UPPER WING (R01051)

ALPHA(0) = -.000 BETA(4) = -3.970

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/OW	X/CU	CP
.2990	.3440	.4270 .5340 .6730 .7600 .8670
.0336	.1479	.4393 .5731 .5297 .4800 .4806
.020		.2048 .2756 .0950 .0428
.025	.2016	.2160
.040		.1009
.045		
.050	.0169	-.0993 -.1294 -.1637 -.2647
.060		-.2567
.085		-.0776
.096	.1991	
.094	.0277	-.4763 -.5460 -.5360 -.5346
.150		
.163	.0000	
.177		-.3154
.193	-.0296	
.229	.1036	
.250		-.5719 -.6397 -.7433 -.7519
.274	-.4214	
.339	-.3961	
.362	.0095	
.400		-.5674 -.6350 -.9733
.402		
.497	.0000	
.550		-.2932 -.4741
.565		.0000
.600		
.650	-.0910	-.1437
.700		-.1503
.725		
.750		.1460 .0106
.760		
.775		-.0672 .0921 .1163
.806		.0787
.834	-.0546	
.850		.0805 .1109 .1436
.857		.0701
.865	.0697	
.900	.0737	.0723 .0913 .1146
.905		.0632
.920		.0351
.953		.0202
.965	-.0046	



(1511)

ARC11-716 YÁ14 01+T12+312N23+AT10 UPPER WING

ALPHA( 6 ) = -.000 BETA( 5 ) = -1.980

**SECTION 1: UPPER WING**

DEPENDENT VARIABLE CP

Y/DJ	X/CJ	.2390	.3640	.4270	.5340	.6730	.7000	.8870
.000	.0054	.1188	.3996	.5261	.4768	.4272	.4368	
.020			.2007	.1993	.2193	.0941	.0394	
.025		.1848						
.040			.0935					
.045								
.050	-.0015		-.0725	-.0615	-.1197	-.1547	-.2574	
.060				-.2513				
.085		.1808						
.094	.0187			-.4485	-.5140	-.5082	-.5318	
.197		.0000						
.163			-.3022					
.177	-.0281							
.193								
.229	.0949			-.5463	-.6343	-.7155	-.7397	
.250			.3968					
.274		-.3719						
.332				-.3241	-.7659		-.9845	
.400	.0014		-.5035					
.402								
.497	.0000			-.3284	-.5043			
.590			.0000				-.5048	
.600						-.2995		
.698		-.1248			-.1592			
.700				-.1207		.1417	.0760	
.725								
.750			-.1255					
.760			.0460	.0895	.1054			
.775								
.808				.0490	.1032	.1278		
.834	-.1072						.1166	
.890								
.857								
.885	.0229							
.900	.0375		.0404		.0588	.0488	.0024	
.905								
.930					.0251			
.933			-.0029					
.985	-.0234							

(RBIUS3)

ARC11-716 IA14 OR71E+SIZE5+AT10 UPPER WING

ALPMAO(6) = -.600 BETAO(6) = .010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/CM	.2900	.3640	.4270	.5340	.6730	.7800	.8870
X/CM							
.000	-.0272	.0836	.3647	.4818	.4261	.3738	.3885
.025	.0820		.1814	.1915	.3031	.1046	.0454
.040		.1545					
.045			.0830				
.050	-.0141		-.0547	-.1007	-.1421	-.2364	
.065			-.2372				
.086		.1802	-.0858				
.094	.0023			-.4175	-.4676	-.4668	-.5180
.120		.0000					
.163			-.2807				
.177		-.0273					
.193				-.5236	-.6006	-.6914	-.7197
.229	.0784		-.3531				
.230		-.3462					
.274				-.4853	-.6630		-.9452
.362	-.0009		-.4630				
.359			.0000	-.3467	-.5166		
.400						-.5750	-.5998
.402	.0000			-.2824	-.1977		
.497						.1068	.0715
.550				-.2338	.0340	.0541	
.565				-.0085			
.600					.0457	.0851	.1152
.650	-.1328						
.700							
.725							
.750							
.760							
.775							
.808							
.834	-.2264						
.850							
.857							
.885	-.6705						.1074
.900	-.0164						
.905							
.920							
.935							
.965	-.6386						



(0810531)

ARC11-716 1A14 CH+T12+312N23+AT10 UPPER WING

ALPHA( 6 ) = -.670 BETA( 7 ) = 2.050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3440 .4270 .5340 .6730 .7000 .8870

X/CW

.000	-.0716	.0483	.3166	.4476	.3786	.3181	.3334
.020				.1773	.2759	.0936	.0366
.025			.1531				
.040		.1201					
.045			.0697				
.050	-.0270			-.0532	-.1006	-.1263	-.2275
.060				-.2235			
.065			-.0671				
.066		.1366					
.084	-.0033			-.3622	-.4353	-.4446	-.5024
.130		.0000					
.163			-.0206				
.177			-.2631				
.193							
.229	.0697			-.4971	-.5790	-.6630	-.7102
.250							
.274			-.3374				
.339		-.3043					
.362	.0011						
.400				-.4293	-.5900		-.9047
.402			-.4193				
.497	.0080		.0000	-.3667	-.3332		
.530							
.565							
.600							
.630							
.700	-.1739						
.725				-.3778			
.750			-.3637				
.775				-.0695	-.0630		
.809			-.1991				
.834	-.2981						
.850				.0001	.0494	.0785	
.857				-.0633			
.869	-.1969						
.900	-.1116			.0167			.1012
.903				-.8473	.0207		
.936				-.0021			.8437
.953			-.0348				
.965	-.0967						

(RB1051)

ARCS1-716 1A14 Q1+712+512N25+710 UPPER WING

ALPHA(0) = -.000 BETA(0) = 4.050

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
Y/W			
1/CU			
.000	-.1013	.0531	.8215
.020		.4207	.3397
.040		.1750	.2640
.060		.1247	.0864
.080			.0313
.100			
.120			
.140			
.160			
.180			
.200			
.220			
.240			
.260			
.280			
.300			
.320			
.340			
.360			
.380			
.400			
.420			
.440			
.460			
.480			
.500			
.520			
.540			
.560			
.580			
.600			
.620			
.640			
.660			
.680			
.700			
.720			
.740			
.760			
.780			
.800			
.820			
.840			
.860			
.880			
.900			
.920			
.940			
.960			
.980			
1.000			

(R01051)

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ARC11-716 1A14 01+712+512+5+710 UPPER WING

ALPHAO ( 0 ) = -.000 BETA0 ( 0 ) = 6.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2998	.3840	.4270	.5340	.6730	.7800	.8870
X/CW							
.008	-.1999	.0101	.1410	.3966	.3091	.2302	.2481
.020				.1587	.2491	.0913	.0396
.025			.0738				
.045		.0508					
.070			.0392				
.090	-.0638			-.0332	-.0723	-.1103	-.2026
.095					-.1733		
.098			-.0589				
.098		.0678					
.094	-.0280						
.130		.0200			-.3253	-.3712	-.4044
.163							
.177		-.0194					
.193			-.1739				
.229	.0437						
.230					-.4074	-.5006	-.5831
.274			-.2397				
.339		-.2134					
.362	-.0027				-.3804	-.5550	-.7835
.400			-.3606				
.402					-.3866	-.5383	
.487	.0088		.0000				
.550							
.563							
.600							
.620							
.620	-.2223						
.700					-.4116		
.725						-.2908	-.1743
.750							
.760			-.3328				
.775					-.3044	-.2491	
.808			-.2374				
.824	-.3141						
.829					-.1195	-.0796	-.0640
.837			-.1857				
.845	-.0082						
.868	-.2700				-.0699		
.896			-.1311			-.0150	
.928					-.0619		-.0102
.933			-.1886				
.945							



(001051)

ARC11-716 1A14 CR+T1R+SIEMES+AT10 UPPER WING

ALPHAO( 0) = -.000 BETAO (10) = 0.000

SECTION ( 1) UPPER WING

DEPENDENT VARIABLE CP

Y/CM .2900 .3040 .4270 .5340 .6730 .7800 .8670

U/CM	CP	CP	CP	CP	CP	CP
.000	-.2180	.0051	.0911	.2367	.2797	.1810
.020				.1138	.2346	.0954
.040			.0323			.0392
.060		.0356				
.080			.0231			
.100	-.0832			-.0222	-.0478	-.0883
.120				-.1313		
.140			-.0240			
.160		.0477				
.180	-.0548			-.2354	-.3337	-.3656
.200		.0000				-.4485
.220			-.1454			
.240		-.0033				
.260	.0565			-.3488	-.4549	-.5327
.280		-.2221				-.6445
.300	-.1848					
.320				-.3572	-.3441	-.7307
.340	-.0008		-.2965			
.360				-.3650	-.5429	
.380	.0008		.0000			-.6884
.400					-.6372	
.420				-.2550		
.440	-.2373		-.3083			-.1452
.460				-.2604	-.2475	
.480			-.3289			
.500			-.2587			
.520				-.1420	-.0748	-.0389
.540	-.3300		-.2177			
.560						.0265
.580	-.2500			-.0986		
.600	-.2547			-.0140		
.620			-.1876	-.0724		.0103
.640				-.1398		
.660						
.680						
.700						
.720						
.740						
.760						
.780						
.800						
.820						
.840						
.860						
.880						
.900						
.920						
.940						
.960						
.980						
.000						

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(RBIUS1)

ARC11-716 1A14 CR+112+512N25+AT10 UPPER WING

ALPHA(6) = -.000 BETA(11) = 10.120

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	2590	3040	4270	5340	6730	7800	8870
1/CM							
.000	-.2854	-.0081	.0709	.1805	.2343	.1539	.1491
.025				.0696	.2449	.1087	.0523
.025			.0467				
.040		.0215					
.045			.0291				
.050	-.1518			-.0240	-.0203	-.0651	-.1522
.060				-.0949			
.065			-.0104				
.066		.0416					
.064	-.0664			-.2028	-.2886	-.3187	-.4099
.150		.0000					
.175			-.1165				
.177		.0052					
.193				-.2967	-.4010	-.4834	-.6053
.229	.0363						
.274			-.1928				
.339		-.1653					
.362	-.0006			-.3279	-.5189		-.6825
.400			-.2744				
.402	.6000			-.3932	-.5251		
.497		.0000					
.550							-.6815
.600					-.6310		
.620	-.2468			-.3981			
.700					-.2815		
.725						-.3515	-.2116
.750			-.3492				
.760				-.3504	-.3348		
.775			-.3080				
.806							
.834	-.3800			-.1999	-.1376	-.1081	
.890			-.2256				
.897							
.905	-.3145			-.1359			-.0359
.909	-.2645				-.0444		
.905			-.1758				
.920			-.1048				-.0191
.935			-.1498				
.968	-.1908						

ABC11-716 1A14 01+Y12+S12N25+AT10 UPPER WING

ALPHA( 7 ) = 2.500 BETA ( 1 ) = -10.000

SECTION / 11 DEPT WING

Y/CW	X/CW	.2920	.3640	.4270	.5340	.6750	.7800	.8670
.000	.0677	.1815	.5065	.0406	.6373	.5995	.5715	
.020			.1110	.0411	.1319	-.1185	-.1994	
.025								
.040	.2013							
.045								
.050	.0183			-.0346	-.2546	-.3460	-.3992	-.5191
.060					-.4806			
.065				-.2515				
.066	.1692							
.094	.0319							
.150					-.7223	-.8035	-.7586	-.7494
.163	.0000							
.177	-.1312			-.4926				
.193								
.229	.1071				-.6306	-.9395	-.9611	-.6666
.250				-.6113				
.274								
.339	-.3529							
.362	-.0449				-.6375	-.9320		-.5804
.400				-.7770				
.402	.0000				-.0402	-.4422		
.497				.0000				-.5126
.550						-.3937		
.600					-.1534			
.650	-.0157				.9077		-.2616	-.3664
.700				.0373				
.750				.1433	.1050	.0109		
.760								
.775								
.806	.0724				.1040	.0655	-.1602	
.814				.1274				
.850								
.857								
.865	.2002				.1023			-.2637
.905	.1712			.1106	.1102			
.905					.3755		-.1120	
.930				.0476				
.933								
.965	.0592							



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(H810312)

ARC11-716 1A14 25+712+81825+AT10 UPPER WING

ALPHA( 7 ) = 1.900 BETA( 2 ) = -5.980

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/DW	X/CM	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.0190	.1805	.4572 .6330 .5958 .5346 .5073
.020		.0447 .0804 -.1286 -.2240
.025	.1045	
.040	.1807	
.045		-.0254
.050		-.2330 -.3261 -.3955 -.5218
.060		-.4506
.065		-.2206
.066	.1480	
.094	.0046	
.195		-.6887 -.6063 -.7445 -.7480
.163	.0000	
.177		-.4578
.193	-.1216	
.229	.0861	
.274		-.7755 -.8855 -.9636 -.9041
.339	-.4894	
.362	-.8474	
.400		-.8594 -1.0160 -.6406
.402		-.7377
.497	.0000	
.595		-.1460 -.3921
.600	.0100	
.690		-.4279
.700	-.0000	
.725		-.0534
.750		-.1530
.760		-.2905 -.4012
.775	-.0153	
.806	.0090	.0512
.834	.0068	
.850		.0682 .0796 -.1373
.857	.0092	
.865	.1899	
.908	.1227	.0732 .0781 -.2799
.94		.0506
.958		.0440
.993		-.0560
.965	.0270	

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DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R03U081)

ARC11-716 1A14 CR+T12+812N2+AT10 UPPER WING

ALPHA( 7) = 1.970 BETA( 3) = -3.000

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP
1/CM	
.000	.0004
.020	.0521
.040	.1025
.060	.1528
.080	.2032
.100	.2536
.120	.3040
.140	.3544
.160	.4048
.180	.4552
.200	.5056
.220	.5560
.240	.6064
.260	.6568
.280	.7072
.300	.7576
.320	.8080
.340	.8584
.360	.9088
.380	.9592
.400	.0000
.420	.0000
.440	.0000
.460	.0000
.480	.0000
.500	.0000
.520	.0000
.540	.0000
.560	.0000
.580	.0000
.600	.0000
.620	.0000
.640	.0000
.660	.0000
.680	.0000
.700	.0000
.720	.0000
.740	.0000
.760	.0000
.780	.0000
.800	.0000
.820	.0000
.840	.0000
.860	.0000
.880	.0000
.900	.0000
.920	.0000
.940	.0000
.960	.0000
.980	.0000
1.000	.0000



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-713 1A14 01+712+812M25+7110 UPPER WING

(R81051)

ALPHA( 7) = 1.980 BETA( 4) = -1.990

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.2990	.3640	.4270	.5340	.6730 .7800 .8870
.000	-.0383	.0487	.3908	.5599 .5179 .4599 .4397
.020			.0520	.1883 -.1245 -.2160
.025		.1413	.0951	
.040			-.0182	
.045				
.050	-.0405		-.2079	-.3088 -.3790 -.5082
.060			-.4030	
.085		-.1874		
.086		.1336		
.094	-.0167			
.120		.0000	-.8166	-.7342 -.7110 -.7377
.163		-.4114		
.177		-.1053		
.193	.0667		-.6838	-.8074 -.8910 -.9051
.229		-.3256		
.250		-.4596		
.274				
.339	-.0472		-.7429	-.9552 -.9303
.382		-.6409		
.400			-.3609	-.3243
.402	.0000	.0000		
.497				
.520			-.3718	-.6079
.565			-.1970	
.600			-.0994	
.650	-.1671		-.1427	-.3886
.700				
.725				
.750		-.0944	.0574	.0634
.760		.0401		
.775			.0617	.0733 .0377
.800	-.1892	.0416		
.834				
.850				
.857				
.865	.0242		.0589	
.900	.8478	.0391	.0539	-.2008
.905		.0351		
.950		.0128		.0022
.955	-.0142			

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DATE 10 JAN 75 TABULATED PRESSURE DATA - IAL44 - VOL. 6

(R01031)

ARC11-716 IAL4 01+712+512N25+AT10 UPPER WING

ALPHA( 7 ) = 1.980 BETA( 5 ) = .060

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7600 .8870
.000	-.0451	.0227	.3498 .5178 .4783 .4224 .4076
.020	.020		.0500 .0829 -.1174 -.2131
.025		.0609	
.040	.1121		
.045		-.0271	
.050	-.0450		-.2030 -.2923 -.3661 -.4907
.060			-.3811
.085		-.1838	
.086	.1116		
.094	-.0273		
.150		.0000	-.5317 -.6872 -.8048 -.7110
.163			
.177	-.1043	-.3907	
.193			
.229	.0909		-.6725 -.7449 -.8385 -.8766
.250		-.4979	
.274			
.339	-.4370		
.362	-.0491		-.6785 -.9231 -1.0520
.400		-.5968	
.402			
.497	.0000		-.3968 -.5612
.590		.0000	
.585			-.6297
.600			
.650			-.3728
.700	-.1920	-.2709	-.1840
.725			-.0216 -.3793
.750			
.780		-.2200	.0064 .0646
.775		-.0113	
.808			
.834	-.2298		.0301 .0790 .0786
.850		-.0083	
.857			
.865	-.0428		-.1207
.900	-.0228	.0341	.0466
.903		.0120	.0158 .0524
.950			
.953		-.0093	
.965	-.8442		

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(N011031)

ARC11-716 1A14 Q1+T12+812N23+AT10 UPPER WING

ALPHA(7) = 1.8% BETAO (6) = 2.040

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2690	.3640	.4270	.5340 .6730 .7600 .8670
X/CW				
.000	-.1241	-.0218	.2933	.4841 .4359 .3793 .3749
.020				.0447 .1373 -.1102 -.2008
.025		.0743	.0546	
.040				
.045			-.0360	
.050	-.0616			-.1965 -.2809 -.3472 -.4624
.060				-.3567
.085			-.1801	
.086		.0832		
.094	-.0337			-.5529 -.6439 -.6513 -.6823
.130		.0000		
.163			-.3597	
.177		-.0931		
.193				
.229	.0459			-.6092 -.6744 -.7931 -.8397
.250			-.4653	
.274		-.4062		
.339	-.0467			-.5767 -.6263 -1.0340
.400			-.3148	
.432				
.497	.0000			-.3996 -.5696
.590			.0000	
.565				
.600				-.5604
.630				
.700	-.8124			-.2340
.725			-.3601	-.0394 -.2374
.790				
.760			-.2693	
.775			-.1471	-.1579 -.0345
.806				
.834	-.2837			-.0801 .0230 .0592
.850			-.0602	
.857				
.863	-.2161			
.905	-.1065		.0079	-.0369
.905			-.0392	.0240 .0313
.950				-.0035
.953			-.0459	
.965	-.0763			

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(R01V31)

ABC11-718 1A14 01+712+S12N23+A710 UPPER WING

ALPHA( 7) = 2.099      BETA( 7) = 4.090

**SECTION ( 1 ) UPPER WING**

**DEPENDENT VARIABLE CP**

Year	1990	1991	1992	1993	1994
1990	3,900	4,670	5,340	6,720	7,000
1991	3,900	4,670	5,340	6,720	7,000
1992	3,900	4,670	5,340	6,720	7,000
1993	3,900	4,670	5,340	6,720	7,000
1994	3,900	4,670	5,340	6,720	7,000

**MS/X**

.000	-.1566	-.0365	.1846	.4641	.4000	.3401	.3306
.020			.0159	.0360	.1401	-.1205	-.2132
.040		.0467					
.045			-.0314				
.050	-.0453			-.1837	-.2735	-.3436	-.4686
.060				-.3463			
.085			-.1619				
.086		.0521					
.094	-.6362			-.5044	-.6136	-.6259	-.6711
.150		.0000					
.163			-.3335				
.177		-.0601					
.193							
.229	.0336			-.5743	-.6676	-.7390	-.8173
.250			-.4046				
.274		-.3403					
.339							
.362	-.0467			-.5149	-.7415		-1.0210
.400			-.4537				
.402							-.6276
.497	.0000			-.4172	-.5644		
.550			.0000				
.565							
.600							
.650							
.700	-.2356			-.4062			
.725					-.2376		
.750						-.1594	-.1760
.760			-.3296				
.775			-.2463	-.2633	-.1599		
.806							
.834	-.3267						
.854			-.1496	-.0679	-.0837	-.0116	
.850							
.897							
.945	-.3066			-.0254			-.0035
.900	-.2192						
.925			-.1016		.0040		
.925				-.0269			.0096
.925			-.0760				
.965	-.1159						

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(R81U31)

ARC11-716 1A14 Q1+T12+812M3+AT10 UPPER WING

ALPHA(7) = 2.050 BETA(8) = 6.770

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW		.3340	.4270	.5340	.6750 .7800 .8870
X/CW					
.000	-.2106	-.0420	.0860	.3698	.3749 .3051 .3080
.020				.0150	.0309 -.1040 -.1899
.025			-.0306		
.040		.0142			
.045			-.0726		
.050	-.0946			-.1733	-.2500 -.3191 -.4273
.060				-.3215	
.085			-.1533		
.096		.0229			
.094	-.0647				
.120		.0000		-.4428	-.5864 -.5641 -.6373
.163			-.2818		
.177		-.0036			
.193					
.229	.0200			-.5177	-.6245 -.7303 -.7958
.250			-.3560		
.274		-.2920			
.339	-.0500			-.4504	-.6716 -.9706
.362			-.3921		
.400				-.4337	-.6012
.402	.0000				-.7113
.497					-.8577
.530				-.2403	
.565					-.1462 -.1249
.600			-.3192		
.650			-.2582		
.700	-.2807			-.0990	-.0402 -.0174
.725					
.750					
.780					
.775					
.808					
.834	-.2809				
.857			-.1806		
.865	-.2619				
.900	-.1978			-.0613	.0162 .0173
.905			-.1142		
.930				-.0416	.0190
.953			-.1595		
.965	-.1279				

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TABULATED PRESSURE DATA - 1A14A - VOL. 8

(R01031)

ARC11-716 1A14 CH+T12+812N23+AT10 UPPER WING

ALPHA( 7) = 2.348 BETA( 9) = 8.090

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
V/64	.2990	.3640	.4270	.5340 .6750 .7800 .8870
X/CW				
.000	-.2902	-.0344	.0423	.2137 .3564 .2769 .8921
.020				-.0402 .0858 -.0777 -.1523
.025			-.0253	
.040		.3052		
.045			-.0616	
.090	-.1323		-.1639	-.2037 -.2688 -.3928
.080			-.2612	
.085			-.1188	
.086		.0082		
.094	-.1060			-.3726 -.5160 -.5071 -.5930
.190				
.163		.0000		
.177			-.2323	
.193		-.0749		
.229	.0155			-.4327 -.5362 -.6640 -.7379
.290			-.2983	
.274		-.2565		
.339				
.362	-.0447			-.4072 -.6367 -.6928
.400			-.3467	
.497	.0000			-.4243 -.5797
.590			.0000	
.565				
.600				-.6445
.650				-.2625
.700	-.2082			-.3813
.725				
.790			-.2993	-.2380 -.1542
.760			-.2465	
.775				-.2927 -.2570
.808				
.834	-.2943			-.1579 -.0682 -.0780
.898			-.1702	
.857				
.865	-.2700			
.900	-.2297		-.1127	-.0094
.905			-.1390	-.0196
.938			-.0844	-.0009
.953			-.1163	
.945	-.1656			



ARC11-716 1A14 01+T12+912N25+AT10 UPPER WING

ALPHA( 7) = 2.520 BETA( 10) = 10.110

SECTION ( ) UPPER WING

[illegible]

ABC11-716 1A14 01+T18+S18N25+AT10 UPPER WING

ALPHA( 8) = 4.110 BETA( 1) = -10.000

SECTION ( UPPER WING

DEPENDENT VARIABLE CP

Y/04	.8960	.3640	.4274	.5846	.6799	.7000	.9470
X/04	.0422	.1290	.4771	.6759	.6331	.5569	.5170
.020				-.0946	.0294	-.2809	-.3876
.025			.0803				
.040		.1392					
.045			-.1375				
.050	-.0191			-.3950	-.4725	-.5399	-.6772
.060				-.5955			
.085			-.3746				
.096		.1127					
.094	.0082			-.0619	-.9146	-.9025	-.7850
.150		.0000					
.163			-.5656				
.177		-.1967					
.193							
.229	.0029		-.6963	-.9742	-1.0470	-.8773	-.7430
.250							
.274		-.5052					
.339							
.362	-.0033		-.8636	-.9153	-.7902		-.5462
.400							
.402				-.2553	-.5206		
.497	.0000						
.530		.0000					-.4634
.565							
.600						-.5016	
.650					-.1973		
.700	-.0266			-.0809			
.723							
.750			-.0223				-.3685
.760				.0336	-.2779		
.775			.1015				
.808							
.834	.0485		.0964	.0600	-.1681	-.3291	
.850							
.857							
.865	.1540			.0768	-.0456		-.3161
.900	.1262		.0920	.0480			
.905						-.2325	
.930			.0603				
.953							
.961	.0000						

DATE 18 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+912M5+AT10 UPPER WING (081031)

ALPHA( 6 ) = 4.130 BETA( 2 ) = -7.960

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP		
Y/BW	.2990	.3440	.4270
X/CM	.0180	.0934	.4960
.000	.0560	.6116	.5294
.020	-.0792	-.0342	-.2870
.025	.0136		-.3983
.040	.1322		
.045	-.1390		
.070	-.0424	-.3643	-.4593
.080		-.5791	-.6796
.085	-.3900		
.086	.1020		
.084	-.0135		
.130	.0000	-.6314	-.9007
.163			-.8251
.177	-.1942		
.193	-.5600		
.229	.0702	-.9153	-1.0350
.270		-.8914	-.7609
.274	-.6006		
.336	-.5729		
.342	-.6625	-.9333	-.8254
.400			-.5771
.402	-.8405		
.497	.0800	-.1687	-.4636
.590	.0000		
.600			-.4686
.690		-.5366	
.700	-.0607	-.8000	
.785		-.0626	
.790			-.4326
.790	-.0413		-.5974
.775	.0355	-.6002	
.908	.0910		
.834	.0822	.0436	-.1462
.920			-.3292
.857	.0769		
.945	.1222		
.900	.1861	.0625	-.3900
.905	.0729	-.0163	
.936	.0517		-.2713
.933	.0466		
.945	.0366		

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DATE 18 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01-712-512M2-AT10 UPPER WING (1811031)

ALPHA( 0 ) = 4.150 BETA( 3 ) = -5.900

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
V/W			
1/CM			
.000	-.0141	.0731	.4285
.020			.6268
.025			.5773
.040			.4
.045			.4520
.050			-.0091
.060			.0093
.065			-.2991
.085			-.4183
.094			
.101			
.105			
.110			
.115			
.120			
.125			
.130			
.135			
.140			
.145			
.150			
.155			
.160			
.165			
.170			
.175			
.180			
.185			
.190			
.195			
.200			
.205			
.210			
.215			
.220			
.225			
.230			
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.245			
.250			
.255			
.260			
.265			
.270			
.275			
.280			
.285			
.290			
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.300			
.305			
.310			
.315			
.320			
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.400			
.405			
.410			
.415			
.420			
.425			
.430			
.435			
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.450			
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.610			
.615			
.620			
.625			
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.640			
.645			
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.655			
.660			
.665			
.670			
.675			
.680			
.685			
.690			
.695			
.700			
.705			
.710			
.715			
.720			
.725			
.730			
.735			
.740			
.745			
.750			
.755			
.760			
.765			
.770			
.775			
.780			
.785			
.790			
.795			
.800			
.805			
.810			
.815			
.820			
.825			
.830			
.835			
.840			
.845			
.850			
.855			
.860			
.865			
.870			
.875			
.880			
.885			
.890			
.895			
.900			
.905			
.910			
.915			
.920			
.925			
.930			
.935			
.940			
.945			
.950			
.955			
.960			
.965			
.970			
.975			
.980			
.985			
.990			
.995			
1.000			

DATE 18 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 3

ARC11-716 1A14 04+T12+SIZE25+AT10 UPPER WING (BB1051)

ALPHA( 8) = 4.100 BETA( 4) = -3.360

SECTION 1 UPPER WING DEPENDENT VARIABLE CP

Y/CM	Z/CM	CP
.2980	.3640	.4270
.3340	.6730	.7800
.6070		
.000	-.6114	.0499
.000	.4007	.5996
.000	.5902	.4855
.000	.4209	.4809
.025	-.0604	-.0454
.025	.2096	-.4161
.040	.0235	
.040	.118	
.045	-.1118	
.050	-.3333	-.4452
.050	.5320	-.6924
.060	-.5423	
.065	-.3068	
.066	.0980	
.094	-.0253	
.150	.7336	-.0634
.163	-.0642	-.6742
.177	.0000	
.193	-.5190	
.193	-.1717	
.229	.0001	
.250	-.8427	-.9907
.274	-1.0250	-.8437
.339	-.6276	
.342	-.5396	
.400	-.0003	
.402	-.9115	-.9761
.402	-.6463	
.497	-.7063	
.500	-.1610	-.4366
.965	.0000	
.600	-.5167	
.620	-.5461	
.700	-.1373	
.725	-.1006	
.730	-.4176	-.4176
.700	-.0744	
.775	.0035	-.1663
.600	.0505	
.634	.0274	-.0365
.850	-.2871	
.857	.8487	
.845	.8793	
.900	.8729	
.905	.0400	-.3151
.920	.0574	
.923	.0403	-.1492
.923	.0224	
.945	.8186	



(881051)

ARC11-716 1A14 CL-T12-S12M25-A110 UPPER WING

ALPHAO (3) = 4.840 BETA0 (5) = -1.900

SECTION ( UPPER WING		DEPENDENT VARIABLE CP	
Y/CW			
.000	-.0771	.0193	.3713
.001			.5683
.002			-.0447
.003			.0202
.004			.1857
.005			-.0968
.006			-.3085
.007			-.5141
.008			-.2734
.009			.0910
.010			-.0330
.011			.0070
.012			-.1644
.013			-.4853
.014			-.5915
.015			-.5197
.016			-.0780
.017			.362
.018			.402
.019			.497
.020			.590
.021			.563
.022			.600
.023			.690
.024			.700
.025			.725
.026			.750
.027			.760
.028			.775
.029			.808
.030			.834
.031			.850
.032			.897
.033			.865
.034			.900
.035			.905
.036			.950
.037			.957
.038			.975
.039			-.0081
.040			-.3413
.041			.0904
.042			-.1324
.043			.0068
.044			.0033
.045			-.4087
.046			-.4482
.047			-.1244
.048			.0306
.049			-.0104
.050			-.0110
.051			-.0734
.052			.0298
.053			.0113
.054			.0904
.055			-.1324
.056			.0068
.057			.0033
.058			-.4087
.059			-.4482
.060			-.1244
.061			.0306
.062			-.0104
.063			-.0110
.064			-.0734
.065			.0298
.066			.0113
.067			.0904
.068			-.1324
.069			.0068
.070			.0033
.071			-.4087
.072			-.4482
.073			-.1244
.074			.0306
.075			-.0104
.076			-.0110
.077			-.0734
.078			.0298
.079			.0113
.080			.0904
.081			-.1324
.082			.0068
.083			.0033
.084			-.4087
.085			-.4482
.086			-.1244
.087			.0306
.088			-.0104
.089			-.0110
.090			-.0734
.091			.0298
.092			.0113
.093			.0904
.094			-.1324
.095			.0068
.096			.0033
.097			-.4087
.098			-.4482
.099			-.1244
.100			.0306

DATE 18 JAN 75  
TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC 1-716 1A14 CA-712-812MS-A710 UPPER WING

(R01U81)

ALPHA ( $\alpha$ ) = 4.950		BETA ( $\beta$ ) = .030		DEPENDENT VARIABLE CP	
SECTION ( 1 ) UPPER WING					
Y/BW					
X/Cu					
.000	-.1287	-.0377	.3289	.3294	.4049
.020			-.0520	-.0459	-.8759
.025			.0115		-.4148
.040		.0767			
.045			-.1084		
.050	-.0650		-.3067	-.4158	-.5040
.080			-.5016		-.6753
.085			-.2628		
.086		.0740			
.094	-.0454			-.0048	-.6285
.150		.0000		-.0133	-.6043
.163			-.4754		
.177		-.1002			
.193	.0429			-.7679	-.9132
.229				-1.0070	-.9773
.250			-.5668		
.274		-.4994			
.339	-.0765			-.8444	-1.0250
.362			-.7106		-.6216
.400				-.4106	-.4180
.402	.0000		.0000		
.497	.0000				-.5748
.550				-.5413	
.600			-.1942	-.1956	
.690	-.2268			-.3755	-.4344
.700			-.1937	-.0362	-.0566
.725			-.0140		
.750					
.760			.0014	.0031	-.0013
.773				-.1908	
.806	-.1600				
.834					
.850					
.857					
.865	-.0399				-.3199
.900	.0007		.0130	.0434	
.905			.0063		-.0148
.930			.0039		
.953			-.0030		
.965	-.6233				

(R01051)

ARC11-716 IA14 CR+T12+312M3+AT10 UPPER WING

ALPHA( 6 ) = 4.030 BETA( 7 ) = 2.050

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.000	-.1746	-.0071	.2679
.020		.4090	.4400
.025		-.0003	.0290
.040		-.0106	-.2712
.045	.0434		-.4143
.050		-.1185	
.050	-.0075	-.2993	-.4102
.060		-.4803	-.4927
.085		-.2618	-.6855
.086	.0456		
.094	-.0281		
.150		-.0600	-.7902
.163	.0000	-.7905	-.8441
.177		-.4457	
.193	-.1900		
.229	.0294		
.250		-.7203	-.8610
.274		-.9413	-.9711
.339	-.4649	-.0711	-1.0050
.362	-.0795		
.400		-.7369	-1.0010
.402	-.6367		-1.0240
.497	.0000		
.550		-.4200	-.5843
.565	.0000		
.600			-.6173
.620		-.4635	
.700	-.2468	-.3109	-.2211
.725			-.2285
.730			-.4634
.760		-.2635	
.775		-.0578	-.0149
.800		-.0792	
.834	-.2630	-.0103	.8138
.850			-.0006
.857		-.7357	
.865	-.1659		
.900	-.0756	-.0007	-.3158
.905		-.0268	.0307
.950		-.0002	-.0006
.953		-.0317	
.965	-.8568		



DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81031)

ARC11-716 1A14 04712+SIENES+AT10 UPPER WING

ALPHA(0) = 4.000 BETA(0) = 4.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

T/BL: .2000 .3040 .4270 .5340 .6730 .7600 .8070

X/CM	.000	-.2055	-.1022	.1481	.4678	.4110	.3383	.3094
.020	.020				-.0375	-.0310	-.2753	-.4022
.025	.025			-.0009				
.040	.040		.0134					
.045	.045			-.1258				
.050	.050	-.0487			-.2881	-.4075	-.4826	-.6496
.060	.060				-.4553			
.065	.065			-.2478				
.066	.066		.0179					
.094	.094	-.0409			-.6258	-.7430	-.7723	-.8200
.150	.150		.0000					
.163	.163							
.177	.177		-.1304		-.4062			
.193	.193							
.229	.229	.0159			-.6610	-.7798	-.9145	-.9764
.250	.250							
.274	.274			-.4006	-.4827			
.339	.339							
.362	.362	-.0809			-.6170	-.6601		-1.0500
.400	.400							
.402	.402	.0000			-.4414	-.5922		
.497	.497				-.5388			
.550	.550				.0000			
.565	.565						-.4684	-.6346
.600	.600					-.2364		
.650	.650				-.3824			
.700	.700	-.2666					-.1160	-.4043
.725	.725							
.750	.750				-.3118			
.760	.760					-.1910	-.0345	
.775	.775				-.1955			
.806	.806							
.834	.834	-.3067			-.0490	-.0025	-.0161	
.850	.850				-.1252			
.897	.897							
.945	.945	-.2998			-.0193			-.2347
.960	.960	-.1482				.0132		
.965	.965				-.0008			
.990	.990				-.0134		-.0105	
.995	.995				-.0732			
.995	.995	-.0974						

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(091051)

ARC11-716 1A14 CR+T12+512M3+AT10 UPPER WING

ALPMA( 8) = 4.080 BETA( 9) = 6.070

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2626	-.0673	.6422	.3921	.3090	.3129	.2891
.020				-.0711	.0344	-.2502	-.3744
.025				-.1009			
.040				-.0162			
.045				-.1416			
.050	-.1296			-.2609	-.3780	-.4330	-.6120
.060				-.4201			
.065				-.2195			
.086		-.0144					
.094	-.1013			-.5467	-.6733	-.7194	-.7804
.130		.0000					
.163							
.177		-.3443					
.193		-.1312					
.229	-.0024			-.6023	-.7215	-.8390	-.9333
.274				-.4106			
.339		-.3443					
.362	-.0794			-.3267	-.7393		-1.0510
.400							
.402		-.4356					
.497	.0000			-.4610	-.6124		
.530				.0000			
.563							
.600							-.8056
.692					-.6090		
.700	-.2979			-.3761	-.2393		
.725							
.750							
.760				-.3142			
.773				-.2163	-.2290	-.1015	
.808							
.834	-.3057						
.890				-.0931	-.0123	.0026	
.897				-.1495			
.885	-.2401						
.900	-.1664			-.0501			-.1215
.905				-.1635	.0812		
.950				-.0323		.0216	
.953		-.1004					
.963	-.1263						



ALPHA (0) = 4.010      BETA (10) = 0.100

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000		-.2856	-.0665	.0024	.1972	.3713	.2902	.2688
.020					-.1416	-.0589	-.2210	-.3479
.025				-.0811				
.040			-.0246					
.045				-.1208				
.050		-.1399			-.2644	-.3415	-.4164	-.5795
.060					-.3638			
.065				-.1755				
.086		-.08207						
.094		-.1476			-.4548	-.6125	-.6655	-.7358
.130			.0020					
.163				-.2819				
.177			-.1268					
.193								
.229		-.0138			-.5152	-.6684	-.7722	-.8856
.250				-.3503				
.274			-.3061					
.339		-.0680			-.4833	-.6647		-.10270
.362				-.3841				
.400								
.402		.0000			-.4322	-.6083		
.437				.0000				
.550								
.565								
.600								-.6022
.650						-.6813		
.700		-.2842			-.3577	-.2846		
.725				-.3107				
.750					-.2549	-.1549		
.760				-.2422				
.775								
.808								
.834		-.3040			-.1314	-.0493	-.0044	
.850								
.857				-.1784				
.865		-.2847						
.900		-.2072						
.905				-.1329		.0042		.0259
.950					-.0812			
.955				-.1072				
.965		-.1474						

(RBIUS1)

ARC11-716 1A14 01+712+312M3+AT10 UPPER WING

ALPHA(8) = 4.000 BETA(11) = 10.130

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP				
Y/BW		.2990	.3640	.4270	.5340	.6730 .7800 .8870
X/CW						
.000	-.3218	-.0631	-.0296	.0972	.2994	.2748 .2887
.020				-.1636	-.0014	-.1741 -.2604
.025			-.0686			
.040		-.0410				
.045			-.1034			
.070	-.1813			-.2404	-.3000	-.3568 -.5034
.090				-.3067		
.095			-.1621			
.096		-.0366				
.094	-.1806			-.3648	-.5226	-.5897 -.6734
.130		.0000				
.163			-.2343			
.177		-.1039				
.193						
.229	-.0142			-.4179	-.5849	-.7170 -.8249
.290			-.3137			
.274		-.2703				
.339	-.0609					
.362			-.3601	-.4135	-.6106	-.9501
.400						
.402				-.4394	-.5901	
.497	.0000		.0000			
.550						-.7521
.565				-.3551	-.2358	
.600						-.1315 -.1262
.650	-.8876			-.3236	-.3017	-.2041
.700				-.2535		
.725						
.790				-.1791	-.0973	-.0219
.775			-.2044			
.808	-.3180					
.834						.0066
.850						
.837	-.2997			-.1323	-.0327	
.905	-.2374		-.1311	-.0660		.0162
.930			-.1373			
.933						
.945	-.1764					



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T18+812MS+AT10 UPPER WING (R01031)

ALPHA( 9 ) = 0.000 BETA( 1 ) = -9.900

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP				
Y/B	.2990	.3640	.4270	.5340	.6730 .7800 .8670
X/CW					
.000	.0326	.0920	.4631	.6705	.6040 .9074 .4929
.020				-.2010	-.1315 -.4395 -.5499
.025			-.0743		
.040		.1120			
.045			-.2296		
.050	-.0424			-.4670	-.6061 -.6644 -.6361
.060			-.4617	-.6930	
.085					
.096		.0754			
.094	-.0217				
.150		.0000		-.9756	-1.0120 -1.0130 -.6700
.163			-.6642		
.177		-.2521			
.193					
.229	.0706			-1.0460	-1.0960 -.8543 -.7776
.250			-.7655		
.274		-.6314			
.359					
.362	-.1042				
.400			-.9092	-.9469	-.5760
.402		-.9173			
.497	.0000			-.4203	-.5618
.550			.0000		
.565					
.600					-.5064
.690				-.5336	
.700	-.1131		-.2131		
.725			-.1647		
.750				-.4637	-.4395
.760					
.775			-.0439	-.0795	-.3722
.806		.0670			
.834	.0316				
.850			-.0116	-.2971	-.3617
.857		.0769			
.865	.1472				
.900	.1101		.0292	-.1800	-.4072
.905		.0799			
.920			.0420		-.3368
.953		.0360			
.965	.0307				



ALPHA( 9 ) = 5.950 BETA( 2 ) = -7.960

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8070
X/CW							
.008	-.0040	.0361	.4393	.6464	.5062	.4830	.4219
.020				-.1944	-.2220	-.4302	-.5362
.025				-.0691			
.040		.1010					
.045			-.2207				
.050	-.0651			-.4759	-.6039	-.6722	-.8553
.060				-.6864			
.065			-.4491				
.066		.0657					
.094	-.0311						
.130		.0000		-.8465	-1.0170	-1.0130	-.8436
.163							
.177		-.2432		-.6374			
.193							
.229	.0613						
.274				-1.0120	-1.1390	-.8488	-.8481
.339		-.6229		-.7496			
.362	-.1066						
.400				-.9345	-.8044		-.5895
.402			-.6945				
.497	.0060			-.2910	-.3736		
.550			.0000				
.565							
.600						-.5951	
.650				-.2086			
.700	-.1076			-.1471			
.725						-.4839	-.3820
.750			-.0716				
.760				-.0795	-.3996		
.775			.0519				
.800							
.834	.0094			-.0120	-.2516	-.3742	
.890			.0623				
.897							
.945	.1233			.0176			-.3857
.960	.0967						
.965		.0709		-.1160			
.990			.0290			-.2751	
.993		.0493					
.965	.0357						

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(R01051)

ARC11-716 1A14 03-712-812M25+AT10 UPPER WING

ALPHA( 9 ) = 5.980 BETA( 3 ) = -5.960

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6750 .7800 .8870
X/CW					
.000	-.0376	.0319	.4192	.6243	.5666 .4590 .3092
.020				-.1668	-.0862 -.4421 -.5709
.025			-.0992		
.040		.0913			
.045			-.2036		
.050	-.0801			-.4480	-.5862 -.8637 -.8534
.060				-.6498	
.085			-.4175		
.096		.0610			
.094	-.0455			-.8974	-.9939 -.9991 -.8741
.120		.0050			
.163			-.6045		
.177		-.2328			
.193					
.229	.0327			-.9716	-1.1160 -.9934 -.8157
.250			-.7201		
.274		-.6029			
.339				-.9909	-.8340 -.6162
.362	-.1046		-.8682		
.400				-.2974	-.3547
.402	.0000		.0000		
.497					
.590					
.603					
.600				-.5602	
.650					
.700	-.1510			-.2013	
.725			-.1486		
.750					
.760			-.0636		
.775			-.0672	-.3703	
.808			.0439		
.834	-.0184				
.850				-.0175	-.2294 -.3994
.857			.0362		
.865	.0944				
.900	.0748			.0131	
.925			.8419	-.1326	-.4348
.950				.0212	
.953			.8301		-.3222
.965	.8189				

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(MSEUSI)

ARC11-716 IA14 Q1+712+812M25ATI0 UPPER WING

ALPHA( 9 ) = 5.950 BETA( 4 ) = -3.970

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/W	X/CW		
.000	-.0765	.3829	.9976
.020			.9289
.025		-.1565	-.1910
.040	.0864		-.4464
.045			-.3886
.050	-.0886	-.1912	
.060		-.4301	-.5844
.065		-.6575	-.8096
.086	.0804	-.6248	
.094	-.0451	-.3963	
.125			-.8573
.163	.0000	-.9840	-.9947
.177			-.9043
.193	-.2193	-.5786	
.229	.0478		
.250		-.9435	-1.1050
.274		-.6841	-1.0670
.339	-.5847		-.8310
.362	-.0989		
.400		-.9986	-.8465
.402			-.6270
.497	.0000	-.8400	
.550		-.2503	-.5268
.565		.0000	
.600			-.5193
.650			-.5853
.700	-.1794	-.1386	-.1903
.725			
.750		-.0948	-.4779
.760			-.4751
.775		-.0587	-.3282
.806		.0299	
.834	-.8282		
.850		-.0170	-.2059
.857		.0173	-.3822
.865			
.900	.0637		
.910	.0310	.0159	-.4189
.905		-.0234	-.0030
.920		.0256	-.8753
.935		.0117	
.945	.0036		



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01+712+812M25+7110 UPPER WING (R81031)

ALPHA( 9 ) = 5.940 BETA( 5 ) = -1.960

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP				
Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730 .7600 .8670
.060	-.1237	-.0372	.3533	.5686	.9039	.4002 .3367
.080				-.1507	-.1003	-.4327 -.5761
.025			-.0490			
.040			.0601			
.045			-.1807			
.050	-.0996			-.4145	-.5682	-.6476 -.8627
.080				-.6112		
.085			-.3794			
.086		.0540				
.094	-.0364			-.8252	-.9559	-.9774 -.9242
.190		.0000				
.163		-.2095				
.177			-.5565			
.193						
.229	.0423			-.8676	-1.0730	-1.0810 -.8718
.274		-.6716				
.339		-.5681				
.362	-.0999			-.9657	-.9494	-.6669
.400		-.8189				
.402						
.497	.0000			-.2316	-.5416	
.590		.0000				
.565						
.600						
.650						-.5495
.700	-.1990			-.1826	-.1934	-.5922
.725						
.750		-.1096				-.4716 -.9005
.760				-.0662	-.3873	
.775		.0023				
.806						
.834	-.0822			-.0560	-.1932	-.3402
.850						
.857		.0029				
.865	.0296			-.0702		-.4822
.900	.0300					
.905		.0129		-.0454		
.920		.0114				-.8767
.925		.0004				
.945	-.0041					

(881031)

ARC11-716 1A14 Q1-T18-S18E29-A110 UPPER WING

ALPHA( 9 ) = 5.940 BETA( 6 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8670
X/CU							
.000	-.1036	-.0913	.3106	.5310	.4739	.3669	.3037
.005				-.1413	-.1819	-.4236	-.5719
.010				-.0555			
.015		.0433					
.020			-.1816				
.025	-.1064			-.3937	-.5759	-.8244	-.8434
.030				-.5075			
.035			-.3494				
.040		.0399					
.045	-.0624			-.7775	-.9259	-.9443	-.9431
.050		.0000					
.055			-.3243				
.060		-.2036					
.065	.0333			-.6837	-1.0300	-1.0690	-.9030
.070			-.6346				
.075		-.5473					
.080	-.0897			-.9663	-1.0670		-.7279
.085			-.7770				
.090	.0000			-.3203	-.5365		
.095			.0000				-.5747
.100					-.8008		
.105	-.2392			-.8234	-.2071		
.110			-.1683			-.4795	-.5240
.115			-.0343				
.120				-.0975	-.2960		
.125	-.1336						
.130			-.0263	-.0711	-.1656	-.3691	
.135							
.140	-.0224			-.0432			-.4439
.145	-.0073			-.0151	-.0379		
.150				-.0551		-.2634	
.155			-.0210				
.160	-.0300						



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(R81031)

ARC11-716 1A14 Q1-T12-S12M25-AT10 UPPER WING

ALPHA( 9 ) = 5.000 BETA( 7 ) = 2.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CU	CP
.000	-.2275	-.1306
.020	.2420	.4046
.025	-.1485	-.0871
.040	-.0686	-.4174
.045	.0061	-.5084
.050	-.1887	-.0912
.060	-.1146	-.5385
.065	.0000	-.6175
.066	-.3388	-.8.78
.094	.0168	-.5684
.150	-.0673	-.7476
.163	.0000	-.8908
.177	-.5037	-.9243
.193	-.1920	-.9774
.229	.0186	-.7998
.250	-.5942	-.9071
.274	-.9199	-1.0660
.339	-.1014	-.9453
.400	.0000	-.8526
.422	-.7310	-1.0870
.497	.0000	-.6024
.550	-.4180	-.4761
.565	.0000	-.6270
.600	-.2340	-.6070
.650	-.2372	-.4692
.700	-.2304	-.5606
.725	-.1016	-.2276
.790	-.0756	-.0512
.834	-.0512	-.1072
.850	-.3339	-.3339
.857	-.0530	-.0530
.865	-.0037	-.0332
.903	-.0448	-.0214
.905	-.0303	-.0270
.950	-.0270	-.2119
.963	-.0414	-.0414
.965	-.0357	-.0357

ARC11-716 1A14 ON+712+812M3+AT10 UPPER WING

ALPMAO( 9 ) = 5.990 BETA-J ( 8 ) = 4.070

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.8990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2749	-.1778	.1167	.4611	.4132	.2984	.2370
.020				-.1593	-.1974	-.4286	-.3036
.025				-.1332			
.040		-.0236					
.045			-.2081				
.050	-.1277			-.3689	-.5432	-.6214	-.8394
.060				-.5637			
.065							
.086		-.0158					
.094	-.0871						
.150		.0000					
.165							
.177		-.1766					
.193				-.7213	-.8803	-.9099	-.9733
.229	-.0035						
.250				-.7460	-.9429	-1.0660	-1.0040
.274			-.5527				
.339		-.4556					
.362	-.1099			-.7339	-1.0380		-.8782
.400			-.6149				
.402				-.4733	-.4918		
.497	.0000						
.550			.0000				
.565							
.600							
.650							
.700	-.3051			-.2254			
.725				-.2912			
.750							
.760			-.2413				
.775				-.0696	-.1401		
.806			-.1075				
.834	-.2504						
.850				-.0466	-.0617	-.2569	
.857			-.0676				
.865	-.1651						
.900	-.0903			-.0246			-.3990
.905				-.0495	-.0014		
.950				-.0246			
.953			-.0544				
.965	-.0756						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(JBLUS1)

ARC11-712 1A14 CR+712+31223+AT10 UPPER WING

ALPHA( 9 ) = 5.990 BETA( 9 ) = 6.100

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3040	.4270	.5340	.7300 .8870
X/CM					
.000	-.3890	-.1361	-.0017	.3939	.3936 .2946 .2178
.020				-.1697	-.0972 -.3897 -.9803
.025			-.1642		
.040		-.0370			
.045			-.2102		
.050	-.1633		-.3663	-.9854	-.5808 -.7981
.060			-.5113		
.085		-.2860			
.096		-.0464			
.094	-.1415			-.6464	-.8356 -.8499 -.9242
.190		.0000			
.163			-.4031		
.177		-.1690			
.193					
.229	-.3240			-.6503	-.8309 -.9913 -1.0420
.230			-.4706		
.274		-.3906			
.339					
.362	-.1015			-.3735	-.8954 -.9878
.400				-.4962	
.492					
.437	.0000			-.4619	-.6363
.530				.0000	
.965					
.600					-.6031
.670					-.4231
.700	-.3899			-.3125	
.725					-.8375
.790					-.2539 -.5841
.760			-.2958	-.1596	-.0730
.775			-.2149		
.908					
.834	-.2906			-.0611	-.0304 -.1348
.890					
.837		-.1369			
.865	-.2474				
.900	-.1763			-.0570	-.0021 -.3663
.905			-.0994	-.0376	-.0066
.930					
.923			-.0889		
.965	-.1307				



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01031)

ARC11-716 1A14 C1+T12+S12MS+ATIO UPPER 1/1W6

ALPHA( 9 ) = 0.520 BETA( 10 ) = 0.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

V/W	.2500	.3640	.4270	.5540	.6750	.7000	.8670
X/C							
.000	-.3579	-.1134	-.0636	.1992	.3699	.2737	.2068
.020				-.2492	-.1849	-.3841	-.5406
.025				-.1904			
.040				-.0598			
.045				-.1829			
.050	-.2050			-.3689	-.4878	-.5561	-.7768
.060				-.4749			
.065				-.2349			
.066				-.0543			
.094	-.1904						
.190				-.9350	.7582	-.8110	-.8881
.163				.0000			
.177				-.3318			
.193				-.1656			
.229	-.0592						
.250				-.5934	-.7599	-.9216	-1.0100
.274				-.3956			
.339				-.3502			
.342	-.0945						
.400				-.5174	-.7609		-1.0280
.402				-.4289			
.497	.0000			-.4749	-.6367		
.550				.0000			
.565							-.5229
.600							
.690					-.2287		
.700	-.3001			-.3106			
.725							
.750				-.2011	-.2093	-.0704	
.773				-.2295			
.808							
.834	-.2897						
.850				-.1129	-.0434	-.0659	
.857				-.1964			
.865	-.2447						
.900	-.1904			-.0773			-.2340
.905				-.1199	-.0039		
.920				-.0611			-.0319
.953				-.1046			
.968	-.1442						



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TABULATED PRESSURE DATA - IA14A - VOL. 6

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(R01031)

ARC11-716 IA14 01-712-312M3-AT10 UPPER WING

ALPHAO( 9) = 5.990 BETA0 (11) = 10.150

SECTION ( 1) UPPER WING

DEPENDENT VARIABLE CP

V/BW	.2990	.3840	.4270	.5340	.6730	.7800	.8870
X/CV							
.000	-.3432	-.0965	-.0912	.0573	.3021	.2616	.2130
.020				-.2775	-.0899	-.3552	-.4870
.025			-.1239				
.040		-.0726					
.045			-.1599				
.050	-.2099			-.3465	-.4334	-.9037	-.7301
.080				-.4039			
.085			-.2087				
.086		-.0701					
.094	-.2049			-.4428	-.6547	-.7239	-.6196
.190		.0000					
.183							
.177			-.2961				
.193		-.1616					
.229	-.0534			-.4763	-.6644	-.8119	-.9600
.250							
.274			-.3503				
.339		-.3401					
.342	-.0895			-.4582	-.6703		-1.0185
.402			-.3946				
.497	.0000			-.4491	-.6103		
.550			.0000				
.605							
.650						-.5757	-.5511
.700	-.2921			-.3300	-.2921		
.783							
.790			-.3190			-.0483	-.2696
.760				-.2943	-.1427		
.775			-.2540				
.808							
.834	-.2962						
.890			-.2013	-.1561	-.0788	-.0079	
.897							
.865	-.2446						-.0936
.900	-.2170			-.1863	-.0329		
.905			-.1423				
.930				-.0935		.0056	
.933			-.1194				
.965	-.1699						

(R01031)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 CL+T12+SIGNIFICATION UPPER WING

ALPHA=110 = 0.050 BETA= (1) = -9.970

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/DW	X/CW	CP
.000	.0003	.0500
.020	.0500	.4341
.025		-.3120
.040	.0666	-.1965
.045		-.3232
.050	-.0741	-.5900
.060		-.7914
.065		-.5457
.068	.0844	
.094	-.0446	
.150		-1.0760
.165	.0000	-1.1040
.177		-1.1010
.193	-.3217	-1.1010
.229		-1.1010
.250		-1.1010
.274	-.8209	-1.1010
.339	-.6833	-1.1010
.362	-.1327	-1.1010
.400		-1.1010
.402		-1.1010
.497	.0000	-1.1010
.550		-1.1010
.605		-1.1010
.600		-1.1010
.650		-1.1010
.700	-.1307	-1.1010
.725		-1.1010
.750		-1.1010
.775		-1.1010
.800		-1.1010
.834	-.0605	-1.1010
.850		-1.1010
.857		-1.1010
.865	.0732	-1.1010
.900	.0025	-1.1010
.905		-1.1010
.950		-1.1010
.953		-1.1010
.965	.0832	-1.1010



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ARC11-716 1A14 ON-T112+312N23+AT10 UPPER WING (RB1031)

ALPHA(110) = 0.000 BETA( 2) = -7.950

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.000	-.0407	-.0027
.020	.4139	.6354
.025	-.1502	-.3067
.040	.0469	-.3456
.045	-.3001	-.5919
.090	-.1003	-.5717
.095		-.7743
.096	-.5222	
.098	.0136	
.094	-.0615	
.120	.0000	-1.0520
.163		-1.0930
.177	-.7124	-1.1060
.193	-.3192	-1.0800
.229	.0447	
.250		-1.1470
.274	-.8103	-1.1190
.339	-.6456	-1.0270
.362	-.1389	-1.0426
.400		
.402	-.9371	-.9232
.497	.0000	-.5798
.593		
.563	-.5763	-.6071
.600	.0000	
.650		-.5195
.700	-.1759	-.6126
.773		-.2029
.793	-.0794	-.3279
.790		-.4859
.773	-.2337	-.4782
.609	.0472	
.834	-.0168	
.850		-.1177
.857	.0329	-.4171
.865		-.4467
.900	.0506	
.9731		
.925	-.1187	-.3676
.930	.0378	-.2692
.930	.0036	-.3638
.933	.0216	
.963	.0034	

ARC11-716 1A14 Q1+T13+812N23+Y110 UPPER WING (R01051)

ALPHA(18) = 7.900 BETA( 3 ) = -5.950

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2900	.3640	.4270	.5340	.6700	.7000	.8070
X/CW							
.000	-.0673	-.0450	.3857	.6192	.9227	.3904	.3101
.025				-.2895	-.3421	-.3982	-.7350
.025			-.1494				
.040		.0305					
.745			-.3009				
.050	-.1194			-.9563	-.7096	-.7885	-1.0150
.060				-.7562			
.085			-.4932				
.086		.0111					
.094	-.0782						
.150				-1.0200	-1.0900	-1.1010	-1.0110
.165		.0000					
.177		-.2932	-.6756				
.193							
.229	.0560			-1.1060	-1.1540	-1.0440	-.9279
.290							
.274		-.7764					
.339		-.6531					
.362	-.1322						
.400			-1.0370	-.8779			-.5847
.402		-.8207					
.497	.0000						
.520			-.5046	-.5984			
.565		.0000					
.600							
.650							
.700	-.1949			-.2017	-.5432		
.725			-.2756				
.750							
.760		-.0995			-.4756	-.5046	
.775			-.1764	-.4350			
.808		.0505					
.834	-.0426						
.850			-.0625	-.3553	-.4432		
.857		.0212					
.865	.0606						
.900	.0564						
.905			-.0266				-.4688
.950		.0271		-.2500			
.955		.0045					
.965		.0163					
.965	.0148						



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2583

(R01031)

ARC11-716 1A14 OL+112+512M25+ATIO UPPER WING

ALPHA(110) = 7.940 BETA( 4) = -3.970

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7800 .8870
X/CW					
.000	-.1323	-.0939	.3469	.5686	.4960 .3668 .2806
.025			-.2733	-.3418	-.5939 -.7540
.050			-.1462		
.075		.0169			
.100			-.2965		
.125	-.1380		-.5374	-.6941	-.7799 -1.0200
.150			-.7335		
.175		.0003	-.4889		
.200	-.1892				
.225		.0000	-.9723	-1.0750	-1.0840 -1.0350
.250	.0253		-.6451		
.275		-.2956			
.300			-.7322		
.325	-.1296				
.350			-1.0560	-1.1040	-1.0350 -.9772
.375			-.8953		
.400	.0000		-.6066	-.6621	
.425		.0000			
.450				-.5432	-.5647
.475	-.2141		-.2466	-.2066	
.500			-.1346	-.1545	-.4196
.525			-.0116		
.550			-.0756	-.3226	-.4359
.575	-.0633		-.0015		
.600				-.0453	-.4997
.625	.0300		.0061	-.2104	
.650	.0428			-.0116	-.3748
.675			.0068		
.700					
.725					
.750					
.775					
.800					
.825					
.850					
.875					
.900					
.925					
.950					
.975					
.995					

DATE 18 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U31)

ARC11-716 1A14 C8+712+81283+AT10 UPPER WING

ALPHA(118) = 7.940 BETA( 5) = -1.980

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
2.990	.3640	.4870	.5340 .6730 .7900 .8870
X/CW			
.000	-.1798	-.1444	.3126 .5416 .4859 .3406 .2499
.020			-.2605 -.3390 -.3967 -.7619
.025		-.1426	
.040		.0069	
.045		-.2796	
.050	-.1511		-.5132 -.6742 -.7725 -1.0170
.060			-.7046
.065		-.4640	
.066		.0040	
.094	-.0979		-.9176 -1.0540 -1.0720 -1.0480
.150		.0000	
.163		-.6126	
.177		-.2670	
.193			
.229	.0812		-.9739 -1.1610 -1.0070 -.9931
.250		-.7220	
.274		-.6069	
.339	-.1258		-1.0210 -.8526 -.5959
.362		-.8614	
.400			
.402	.0000		-.3423 -.6754
.497		.0000	
.550			-.3618
.565			-.5427
.600			-.2015
.690	-.2907		-.8064
.700			-.4740 -.5411
.725			
.750		-.1467	
.760		-.1372	-.4061
.775		-.0221	
.808			
.834	-.1.125		-.0630 -.3037 -.4248
.850		-.0146	
.857			
.885	.0069		
.900	.0215		-.0393
.905		-.0006	-.1612
.930			-.0064
.953		-.0000	-.3433
.965	-.0069		



DATE 10 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

PAGE 2503

ARC11-716 IA14 01+T12+312N25+AT10 UPPER WING

(R01031)

ALPHA(110) = 7.090 BETA( 6) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.2990	.3640	.5340
.2990	.3640	.6750
.2990	.3640	.7800
.2990	.3640	.8870
.2990	.3640	.9970
.2990	.3640	.1145
.2990	.3640	.2999
.2990	.3640	.2221
.2990	.3640	-.2367
.2990	.3640	-.3325
.2990	.3640	-.5069
.2990	.3640	-.7634
.2990	.3640	-.1320
.2990	.3640	-.0013
.2990	.3640	-.2517
.2990	.3640	-.4913
.2990	.3640	-.6583
.2990	.3640	-.7650
.2990	.3640	-1.0090
.2990	.3640	-.6793
.2990	.3640	-.4413
.2990	.3640	-.0006
.2990	.3640	-.0985
.2990	.3640	-.0823
.2990	.3640	-1.0260
.2990	.3640	-1.0640
.2990	.3640	-1.0600
.2990	.3640	.0000
.2990	.3640	-.5840
.2990	.3640	-.2424
.2990	.3640	.0164
.2990	.3640	.229
.2990	.3640	.193
.2990	.3640	.274
.2990	.3640	.339
.2990	.3640	-.5806
.2990	.3640	-.1242
.2990	.3640	.400
.2990	.3640	-.9856
.2990	.3640	-.9081
.2990	.3640	-.6087
.2990	.3640	-.6325
.2990	.3640	-.3220
.2990	.3640	-.6533
.2990	.3640	.0000
.2990	.3640	-.5927
.2990	.3640	-.5468
.2990	.3640	-.2106
.2990	.3640	-.1978
.2990	.3640	-.4554
.2990	.3640	-.5581
.2990	.3640	-.1745
.2990	.3640	-.1035
.2990	.3640	-.3464
.2990	.3640	-.0484
.2990	.3640	-.0390
.2990	.3640	-.0030
.2990	.3640	-.2353
.2990	.3640	-.3878
.2990	.3640	-.0516
.2990	.3640	-.3041
.2990	.3640	-.0183
.2990	.3640	-.1106
.2990	.3640	-.3097
.2990	.3640	-.0212
.2990	.3640	-.0176
.2990	.3640	-.0185



(151130)

ARC11-716 1A14 04+T12+912N25+AT10 UPPER WING

$$\text{ALPHA}(10) = 7.940 \quad \text{BETA}(7) = 2.060$$

SECTION 11.1336 ; NO. 11336

DEPENDENT VARIABLE CP

7/64	X/CW	.8990	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.2967	-.2473	.2050	.4566	.3922	.2570	.1809	
.020				-.2431	-.3197	-.5763	-.7701	
.025			-.1639					
.040		-.0390						
.045			-.2690					
.050	-.1635			-.4776	-.6376	-.7437	-1.0120	
.060				-.6516				
.085								
.094	-.1082		-.0293	-.4220				
.120								
.163			.0000		-.0356	-.9920	-1.0400	-1.0640
.177								
.193			-.2371	-.5629				
.229	-.0061							
.250					-.8709	-1.0960	-1.0400	-1.0220
.274				-.6544				
.339		-.5563						
.362	-.1278				-.9290	-.9677		-.6055
.400				-.7834				
.497	.0000				-.3167	-.6336		
.550			.0000					
.565								
.620								
.650								
.700	-.3221				-.2423		-.5366	-.6210
.725					-.2245			
.750			-.2213					
.765					-.1906	-.3407		
.775			-.0794					
.808								
.834	-.2361				-.0907	-.2302	-.3670	
.890			-.0636					
.897								
.965	-.0943				-.0651			-.3267
.900	-.0461			-.0925	-.1027			
.905					-.0526		-.3122	
.950								
.953			-.0323					
.965	-.0606							

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2587

ARC11-716 1A14 OR-T12-S12N23-AT10 UPPER WING (R81US1)

ALPHA(10) = 8.010 BETA(8) = 4.090

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 8870
X/CW				
.000	-.3660	-.2962	.0716	.4316 .3943 .2207 .1432
.020			-.2517	-.3222 -.5747 -.7716
.025			-.2107	
.040		-.0717		
.045			-.2828	
.050	-.1803		-.4721	-.6419 -.7376 -1.0030
.060			-.6368	
.085		-.3944		
.086		-.0601		
.094	-.1326			
.150		.0000		-.6049 -.9834 -1.0200 -1.0500
.163			-.5317	
.177		-.2135		
.193				
.229	-.0331			-.8223 -1.0650 -1.0900 -1.8190
.250			-.6070	
.274		-.4965		
.339	-.1368			
.362			-.8366 -1.0130	-.6036
.400		-.6910		
.402			-.4010 -.5443	
.497	.0030		.0000	
.550				-.6145
.565				-.5101
.600			-.2149	
.650	-.3882		-.2478	-.4108 -.5683
.700				
.725			-.2410	
.750			-.1079 -.2536	
.775			-.1229	
.808	-.2284			
.834			-.0734 -.1647 -.3403	
.850			-.0932	
.857				
.865	-.1341			-.5088
.900	-.1012		-.0908	
.955			-.0720	-.0703
.950			-.0371	-.2570
.953			-.0691	
.965	-.0677			

(R31031)

ARC11-716 1A14 OL+T12+S12M23+AT10 UPPER WING

ALPHA(10) = 0.000 BETAC ( 9 ) = 0.120

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/8W	.2990	.3640	.4170	.5340	.6730	.7000	.8070
X/CW							
.000	-.4207	-.1872	-.0519	.3225	.3760	.2190	.1122
.020				-.2818	-.3117	-.5475	-.7868
.025				-.2468			
.040		-.1040					
.045				-.2885			
.050	-.2128			-.4734	-.6287	-.7137	-.9901
.060				-.6211			
.065				-.3631			
.066				-.0981			
.094	-.1973						
.150				-.7474	-.9596	-1.0010	-1.0720
.163		.0000					
.177				-.4687			
.193		-.2187					
.229	-.0806						
.290				-.7551	-.9969	-1.1320	-1.0490
.274		-.5100					
.339		-.4404					
.362	-.1382			-.6778	-1.0530		-.6389
.400				-.5597			
.402							
.497	.0200			-.4947	-.4761		
.590							
.565				.0000			
.600					-.4595		-.6516
.650				-.2205			
.700	-.3254			-.2419		-.3540	-.6043
.725							
.750				-.1472	-.1534		
.760				-.2456			
.775				-.1504			
.808							
.834	-.2474			-.1023	-.1071	-.2512	
.850							
.857				-.1145			
.865	-.1730			-.0807			-.4682
.900	-.1327			-.0599	-.0339		
.905				-.0536		-.1635	
.950				-.0682			
.953							
.965	-.1119						



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(R01031)

ARC11-716 1A14 01+712+S12M23+AT10 UPPER WING

ALPHA0(10) = 7.900 SETAO (10) = 0.120

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7000 .8870
X/CW					
.000	-.4156	-.1561	-.1343	.1509	.3422 .2364 .1304
.020				-.3714	-.2323 -.5230 -.7387
.025			-.2221		
.040		-.1026			
.045			-.2464		
.050	-.2505			-.4704	-.6116 -.6951 -.9574
.060				-.5759	
.065			-.2948		
.066		-.1004			
.094	-.2517			-.6402	-.8957 -.9593 -1.0410
.150		.0000			
.163					
.177		-.2132	-.3915		
.193					
.229	-.1245			-.6612	-.8835 -1.0760 -1.1160
.250			-.4524		
.274		-.4041			
.339					
.362	-.1361			-.5607	-.8530 -.9059
.405			-.4771		
.402					
.497	.0000			-.4622	-.6389
.550			.0000		
.565					-.6737
.600					
.650					
.700	-.3167			-.2510	-.2164
.725					
.750					
.760			-.2587		
.775				-.1617	-.0997
.806			-.1939		
.834	-.2517				
.850				-.1053	-.0769 -.1319
.857			-.1594		
.865	-.2058				
.900	-.1606			-.0656	
.905			-.1302	-.0306	-.3957
.930				-.0694	-.0775
.953			-.1065		
.965	-.1240				

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OF POOR QUALITY

(R81051)

ARC11-716 1A14 01+712+812M25+AT10 UPPER WING

ALPHAO(18) = 7.950 BETA0 (11) = 10.200

## SECTION ( 1 ) UPPER WING

## DEPENDENT VARIABLE CP

Y/BA	.2990	.3640	.4270	.5840	.6790	.7890	.8870
X/CM							
.000	-.3986	-.1510	-.1951	-.0971	.2864	.2270	.1415
.020			-.4639	-.3081	-.5086	-.6812	
.025		-.2077					
.040		-.1420					
.045			-.2433				
.050	-.2752		-.4900	.0236	-.6685	-.9032	
.060			-.5379				
.085			-.2878				
.086		-.1254					
.094	-.2757		-.5359	-.8394	-.9104	-1.0000	
.150		.0000					
.163			-.3740				
.177		-.2667					
.193							
.229	-.1362		-.5441	-.7922	-.9955	-1.0930	
.274		-.4120	-.4286				
.339	-.1271		-.4906	-.7443		-1.0500	
.362			-.4632				
.400				-.4512	-.6072		
.402			.0000				
.497	.0000			-.2363	-.3093	-.6860	
.550				-.2765			
.565					-.1632	-.5182	
.600							
.650	-.3066						
.700			-.2816	-.2174	-.1800		
.723			-.2090				
.750				-.1490	-.0721	-.1082	
.760				-.1669			
.775							
.808	-.2621						
.834							
.850							
.857							
.865	-.2162			-.1168	-.0296	-.2398	
.900	-.1774		-.1342				
.915			-.1007			-.0534	
.950			-.1221				
.955	-.1513						



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAL14 - VOL. 6

PAGE 2391

(081031)

ARC11-716 IAL14 010718+312803-1710 UPPER WING

ALPHA0(111) = 9.990 BETA0 ( 1 ) = -9.930

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW Y/BW .8990 .3640 .4270 .3340 .6730 .7800 .8070

X/CW

.500 -.0003 .0363 .4621 .9833 .9268 .3670 .2892

.020 .025 .045 .045 -.1989 -.3937 -.4510 -.7182 -.8316

.030 -.0682 -.0026 -.8211 -.6533 -.8152 -.8957 -1.0140

.065 .086 .094 -.0573 .0000 -.6060 -1.1060 -1.1740 -1.0940 -.9096

.120 .163 .177 .193 .229 .250 .274 .339 .362

.400 .402 .497 .530 .565 .600 .630 .700 .723

.750 .763 .775 .808 .834 .850 .897 .905 .908

.925 .930 .933 .965

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DATE 10 JAN 75  
 TABULATED PRESSURE DATA - 1A144 - VOL. 6  
 ARC11-716 1A14 0A112+SIGN2+AT10 UPPER WING  
 (RBIUS1)

ALPHA(11) = 10.910 BETA( 2) = -7.910

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/OW	.2090	.3640	.4270	.5340	.6750	.7500	.8670
X/CM	.000	-.0496	.0023	.4362	.6472	.9234	.3501
.020				-.3685	-.4471	-.7268	-.8767
.025				-.1694			
.040			.0317	-.3660	-.6425	-.8114	-.9032
.045							-1.0130
.050		-.1131			-.8369		
.080					-.5865		
.085							
.086		-.0046					
.094		-.0774					
.1					-1.1190	-1.1690	-1.0630
.16		.0000					-.9142
.177				-.7713			
.193			-.3553				
.229		.0415			-1.1040	-1.0700	-7.660
.250							-.6796
.274				-.8579			
.339		-.6686					
.382		-.1387			-.8646	-.7502	-.5525
.400				-.2317			
.402		.0000			-.6192	-.6451	
.497							-.5212
.525				.0000		-.5961	
.565						-.2256	
.600					-.5907		
.650		-.1856				-.5496	-.4970
.700				-.1689			
.725				-.4000	-.5374		
.750				-.0490			
.775							
.804		-.1614			-.3014	-.4812	-.5172
.834				-.0459			
.850							
.857							
.885		.0099			-.2145		-.4753
.900		.0609			-.4237		
.905				-.0282	-.1931		-.4921
.950							
.955				-.0136			
.965		-.0124					



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01+712+312N25+AT10 UPPER WING (RB1051)

ALPHA(11) = 9.920 BETA( 3) = -5.920

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	.2990 .3640 .4270 .5340 .6730 .7650 .8873
X/CW	
.000	-.0993 -.0335 .4082 .6173 .5089 .3366 .2354
.020	.025
.025	-.3445 -.4351 -.7202 -.8826
.040	-.1790
.045	.0288
.050	-.3389
.050	-.1410
.060	-.6107 -.7941 -.8878 -1.0670
.065	-.8062
.066	-.5480
.094	-.0007
.190	-.0857
.163	.0000
.177	-.7317
.193	-.3292
.229	.0355
.250	-.1.1200 -1.1090 -.8719 -.8753
.274	-.8229
.339	-.6685
.362	-.1336
.400	-.8774 -.8161 -.5784
.402	-.9603
.497	.0000
.550	-.5965 -.6710
.600	.0000
.650	-.5491
.700	-.6252
.725	-.4115
.750	-.5725 -.5240
.760	-.1785
.775	-.3620 -.3802
.808	-.0418
.834	-.0324
.850	-.2336 -.4720 -.5311
.857	-.0018
.865	-.0018
.900	-.1560
.905	-.0346 -.4035 -.5008
.950	-.1033
.953	-.5021
.965	-.0240
.965	-.0287



(NR1031)

ARC11-718 1A14 CL+718-81823+AT10 UPPER WING

ALPHA(11) = 9.940 BETA( 4) = -3.930

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .8340 .6730 .7600 .8870
.000	-.1470	.3724 .6143 .4894 .3191 .2110
.020		-.3346 -.4291 -.7177 -.8923
.025		-.1642
.040	.0076	
.045		-.3143
.050	-.1539	-.5960 -.7641 -.8942 -1.0640
.060		-.7926
.065		-.5139
.086	-.0101	
.094	-.1042	
.150		-1.0520 -1.1400 -1.1750 -1.0090
.163	.0000	
.177	-.6900	
.193	-.3166	
.229	.0257	
.250		-1.0990 -1.1720 -.8933 -.6762
.274	-.7945	
.339	-.6546	
.362	-.1356	
.400		-.6759 -.8415 -.5960
.402	-.9360	
.497	.8000	
.590		-.5478 -.6687
.565	.0000	
.600		
.630		-.6417
.700	-.2011	
.725		-.3636
.750		-.2231
.760		-.5857 -.5441
.775		
.808		-.1837
.834		-.3054 -.5207
.850	-.1200	
.857		-.0710
.865		-.0630
.890		-.1992 -.4827 -.5496
.900	-.0187	
.905	-.0169	
.905		-.1326
.950		-.0485
.953		-.3731
.965		-.0730
		-.5131
		-.0416
		-.0367



DATE 10 JAN 73

TABULATED PRESSURE DATA - IAI4A - VOL. 6

(001031)

ARC11-716 IAI4 Q1-T12-S12M25+AT10 UPPER WING

A.PMAO(111) = 9.940 BETA0 (5) = -1.960

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	.2590 .3640 .4270 .5340 .6730 .7600 .8070
X/CM	
.000	-.2032 -.1336 .3295 .5798 .4424 .2936 .1886
.020	.025
.025	-.3105 -.4194 -.7122 -.8972
.040	.045
.045	-.1573
.060	-.0100
.065	-.2952
.080	-.1774
.085	-.5786 -.7673 -.8839 -1.0050
.090	-.7686
.095	-.4875
.096	-.0163
.094	-.1190
.150	-.10100 -1.1200 -1.1690 -1.0120
.163	.0000
.177	-.6519
.193	-.2968
.229	.0135
.250	-.10430 -1.1840 -.9017 -.7015
.274	-.7375
.339	-.6279
.362	-.1356
.400	-.10130 -.8470 -.6059
.402	-.9017
.497	.0000
.530	-.4740 -.6529
.565	.0000
.600	-.3798
.650	-.6326
.700	-.2216
.725	-.3220
.750	-.5950 -.5492
.775	-.1963
.806	-.8329 -.5100
.834	-.0690
.850	-.1698 -.4350 -.5561
.857	-.0835
.865	-.1145
.900	-.0307
.905	-.0578
.930	-.1145
.935	-.3371
.965	-.0743
	-.5045
	-.0541
	-.0497

(REBUS)

ARC11-716 1A14 CH+712+812M9+AT10 UPPER WING

ALPHA(11) = 9.000 BETA( 6 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.8900	.3640	.4270
.8900	.5340	.6730
.8900	.7000	.6870
.8000	-.2020	.8090
.8000	.5012	.4472
.8000	.8006	.1569
.8000	-.2934	-.3936
.8000	.7207	-.0997
.8000	-.1716	
.8000	-.0291	
.8000	-.3023	
.8000	-.5506	-.7574
.8000	-.8001	-1.1130
.8000	-.7275	
.8000	-.4770	
.8000	-.0296	
.8000	-.1104	
.8000	-.8459	-1.1140
.8000	-1.1590	-1.0330
.8000	.0500	
.8000	-.0236	
.8000	-.2842	
.8000	.0013	
.8000	-.9931	-1.1970
.8000	-.9926	-.7260
.8000	-.7240	
.8000	-.6003	
.8000	-.1427	
.8000	-.1020	-.0705
.8000	-.6091	
.8000	-.0676	
.8000	-.4148	-.6117
.8000	.0000	
.8000	-.5985	
.8000	-.6420	
.8000	-.2909	
.8000	-.2137	
.8000	-.1896	
.8000	-.2119	-.4000
.8000	-.0699	
.8000	-.1433	-.3907
.8000	-.5430	
.8000	-.0735	
.8000	-.1014	
.8000	-.2841	-.3277
.8000	-.0901	
.8000	-.0624	
.8000	-.0544	
.8000	-.0555	



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01-71E-S12M25-AT10 UPPER WING

(R81031)

ALPHA(11) = 9.900 BETA( 7) = 2.070

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7800 .8870
X/CW					
.000	-.3648	-.3141	.1791	.4819	.4223 .2214 .1179
.020				-.3187	-.3728 -.7061 -.9175
.025			-.2179		
.040		-.0804			
.045			-.3333		
.050	-.1960			-.5305	-.7446 -.8685 -1.1420
.080				-.7258	
.085			-.4803		
.086		-.0582			
.094	-.1399			-.9200	-1.1010 -1.1430 -1.1070
.150		.0000			
.163			-.6107		
.177		-.2722			
.193					
.229	-.0244			-.9436	-1.1670 -1.0500 -.7516
.250			-.7021		
.274		-.5846			
.339				-.9388	-.8753
.362	-.1582				-.6236
.400		-.0300			
.402				-.3993	-.6067
.497	.0000		.0000		
.590					-.6325
.585					-.6510
.600				-.2235	
.650				-.2707	-.5850 -.5767
.700	-.3078				
.725			-.2167		
.750				-.1765	-.4539
.760			-.0976		
.775				-.1313	-.3713 -.5250
.808	-.1939		-.0931		
.834					
.850					
.857					
.865	-.0794			-.1044	
.900	-.0628			-.2380	-.5572
.905			-.0723		
.950			-.0759	-.4494	
.953			-.0743		
.965	-.0762				

(R81051)

ARC11-716 1A14 Q1+T12+S12N25+AT10 UPPER WING

ALPHAO(111) = 9.990 BETA0 (8) = 4.110

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CU

.000 -.4391 -.3462 .0435 .4165 .3729 .2131 .0724

.020 .025 .040 .045 .050 .060 .085 .086

.094 -.1074 .0000 .0000 .0000 .0000 .0000 .0000

.150 .163 .177 .193 .229 -.0581 .362 -.1711

.400 .402 .497 .590 .565 .600 .650 .700

.725 .750 .760 .77 .834 .850 .857 .865

.900 .905 .930 .935 .970 .975 .985 .985

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ARC11-716 1A14 01+712+112N25+AT10 UPPER WING (R81031)

ALPHA(111) = 9.980 BETA(9) = 6.130

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW	X/CN	.4270	.5340	.6730	.7800 .8870
.000	-.4884	-.2273	-.0939	.3557	.3542 .2395 .1053
.020				-.3776	-.3048 -.6767 -.8876
.025			-.3153		
.045		-.1275			
.045			-.3568		
.050	-.2543		-.5620	-.7176	-.8338 -1.1140
.060			-.7050		
.085		-.4227			
.086		-.1190			
.094	-.2351			-.6461	-1.0490 -1.1100 -1.1510
.150		.0000			
.163			-.5245		
.177		-.2496			
.193				-.8371	-1.1170 -1.1900 -.9009
.229	-.1069		-.5637		
.250				-.7806	-1.0170 -.6425
.274		-.4754			
.339				-.4511	-.5437
.362	-.1791		-.6102		
.400			.0000		-.6607
.402	.0000			-.2203	
.497				-.2822	-.4970 -.6196
.530			-.2541		
.565			-.1686	-.1903	-.3147
.600				-.1430	-.2050 -.4320
.650			-.1445		
.700	-.3391				
.725					
.750					
.760					
.775					
.808					
.834	-.2605				
.850					
.857					
.865	-.1604			-.1093	-.1133 -.5736
.900	-.1372				
.905			-.1170		
.950			-.0805		-.3116
.953			-.1155		
.965	-.1234				

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R01031)

ARC11-716 1A14 CR-TIE-912M25+AT10 UPPER WING

ALPHAO(11) = 10.030 BETA0 (10) = 0.170

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2900	.31 .8	.4270 .5340 .6750 .7800 .8870
.000	-.4691	-.1097
.020	-.1590	.3436 .2167 .1163
.025	-.4840	-.3566 -.6503 -.8667
.040	-.1401	-.2570
.045	-.2794	-.1401
.050	-.5959	-.7020 -.8019 -1.0960
.080	-.6833	-.6833
.085	-.3279	-.3279
.086	-.1438	-.1438
.094	-.2679	-.2679
.150	.0000	-.7437 -1.0200 -1.0710 -1.1760
.163	.0000	.0000
.177	-.4271	-.4271
.193	-.2778	-.2778
.229	-.1751	-.1751
.250	-.7292	-1.0250 -1.1850 -1.1070
.274	-.4873	-.4873
.339	-.4666	-.4666
.362	-.1806	-.1806
.400	-.6104	-.2975 -.6604
.402	-.5116	-.5116
.497	.0000	.0000
.550	-.4765	-.4567
.565	.0000	.0000
.600	-.6791	-.6791
.650	-.4659	-.4659
.700	-.2321	-.2321
.725	-.2204	-.2204
.750	-.4139	-.6212
.760	-.2711	-.2711
.775	-.1888	-.2218
.808	-.2066	-.2066
.834	-.1361	-.1655 -.3311
.850	-.1742	-.1742
.857	-.1098	-.1098
.865	-.1098	-.1098
.900	-.1098	-.1098
.905	-.1440	-.0857 -.3458
.950	-.0868	-.0868
.955	-.2251	-.2251
.965	-.1366	-.1366



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2401

ARC11-716 1A14 01-712-S12N25-AT10 UPPER WING (R01031)

ALPHA(11) = 10.050 BETA(11) = 17.250

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP
1/BW	.2990 .3640 .4270 .5340 .6730 .7800 .8870
X/CW	
.000	-.4477 -.2014 -.2902 -.0164 .3338 .2245 .1300
.020	.020
.025	-.5923 -.2629 -.5956 -.8168
.040	-.1715
.045	-.2731
.050	-.3330
.060	-.5997 -.6963 -.7542 -1.0600
.085	-.8513
.086	-.3262
.094	-.1760
.120	-.3126
.165	.0000
.177	-.3943
.193	-.3254
.229	-.2110
.250	-.8181 -.9295 -1.1400 -1.1590
.274	-.4429
.339	-.4096
.362	-.1721
.400	-.3235 -.8371 -.6764
.402	-.4791
.497	.0200
.530	-.4610 -.5425
.565	
.600	
.650	
.700	-.3801
.725	-.4113
.750	-.2445
.760	-.2646
.775	-.2959
.808	-.2097 -.1936
.834	-.2456
.850	
.857	-.1601 -.1230 -.2574
.865	-.1853
.900	-.1970
.900	-.1671
.905	-.1308
.920	-.1517 -.0747 -.9005
.930	-.1197 -.1090
.933	-.1385
.965	-.1489



(RB11032) ( 17 APR 74 )

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+812N25+AT10 UPPER WING

PARAMETRIC DATA

MACH = 1.100 ELEVON = .000  
RUDDER = .000 SPOILER = .000

REFERENCE DATA

SREF = 2.0210 SQ.FT. XMRP = 29.5800 INCHES  
LREF = 38.7090 INCHES YMRP = .0000 INCHES  
BREF = 38.7090 INCHES ZMRP = .0000 INCHES  
SCALE = .0300 SCALE

ALPHA( 1 ) = -10.240 BETA( 1 ) = -9.900

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/B	.2990	.3640	.4270	.5340	.6730	.7800	.8870
A/CN							
.000	-.0993	-.0544	.3777	.5137	.4293	.3419	.3715
.020			.5624	.6798	.6328	.6582	.6516
.040		.1580					
.065			.5023				
.094	.0807			.4751	.4743	.4894	.4725
.120			.3707	.3057			
.150		.3543					
.177	.0513			.0575	.0994	.1468	.1470
.193		.0000	.1262				
.229	.1035						
.250				-.1222	-.1236	-.1066	-.1246
.274				-.0246			
.339	-.0074						
.362	.2434			.0832	.0270		-.3084
.403				-.0795			
.432	.0000			.1049	-.0295		
.497			.0000				
.550							
.565							
.600							
.650							
.700	.2793						
.729				.0874	-.1249		
.750							
.760	.1249					.2819	.2053
.775				.3592	.3325		
.806	.3431						
.834	.2100			.3240	.3350	.3402	
.850							
.857	.3222						
.865	.3579						
.900	.3284			.3132			.3146
.905	.3070			.2858			



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ADJUSTED PRESSURE DATA - IA14A - VOL. 6

(R01032)

ARC11-716 IA14 Q1+T12+S12N25+AT10 UPPER WING

ALPHA( 1 ) = -10.240 BETA( 1 ) = -9.900

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.990		.2628		.2937		
	.953		.2467				
	.965	.2395					

ALPHA( 1 ) = -10.220 BETA( 2 ) = -7.890

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	-.1236	-.0936	.3474	.4779	.3801	.2855
	.020			.6610	.6154	.6414	.6301
	.025		.5352				
	.040	.1157		.4819			
	.045			.4681	.4732	.4905	.4713
	.050	.0290		.3074			
	.060		.3631				
	.065		.3169				
	.086	.0192					
	.094			.0722	.1234	.1675	.1568
	.150		.0090				
	.163		.1328				
	.177		.2867				
	.193						
	.229	.0831		-.1097	-.0855	-.0721	-.1005
	.253						
	.274		-.0063				
	.339	.0767					
	.362	.2272		.0527	-.0201		-.2209
	.400						
	.452		.0712				
	.497	.0000		.0715	-.0469		
	.985		.0000				
	.600						
	.650						
	.700	.2589					
	.725			-.1448			
	.735		.0149				
	.760						
	.775	.0321					
	.808		.3142		.2856		
	.834		.3124				
	.850	.1524					
			.2891	.3058	.3053		

(R01USE)

ARC11-716 1A14 OR-TIS-SIZES+AT10 UPPER WING

ALPHA( 1 ) = -10.220 BETA( 2 ) = -7.890

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/OW .2990 .3640 .4270 .5340 .6730 .7800 .8070

X/CW

.2933

.857

.865

.3191

.2957

.900

.935

.950

.953

.965

.2175

.2765

.2399

.2846

.3012

.2706

.2765

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.2399

ALPHA( 1 ) = -10.220 BETA( 3 ) = -5.900

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/OW .2990 .3640 .4270 .5340 .6730 .7800 .8070

X/CW

.2933

.857

.865

.3191

.2957

.900

.935

.950

.953

.965

.2175

.2765

.2399

.2846

.3012

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.2399

(R01032)

ARC11-116 1A14 Q1-T12-S12N25-AT10 UPPER WING

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ALPHAO ( 1 ) = -10.220 BETA ( 3 ) = -3.930

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8670

CW						
.775			.2270	.1656		
.808		.2661				
.834	.0701					
.850			.2473	.2636	.2608	
.857		.2589				
.865	.2722					.2766
.900	.2552		.2524	.2476		
.935		.2534			.2531	
.950		.2103	.2126			
.953						
.965	.1900					

ALPHAO ( 1 ) = -10.220 BETA ( 4 ) = -3.930

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8670

X/CW						
.000						
.020			.3967	.2737	.1632	.1749
.025			.6448	.5827	.6080	.5818
.040		.3023				
.045	.0703					
.050		.4691				
.060			.4814	.4837	.4877	.4813
.065		.3741	.3424			
.086	.2547					
.094						
.150			.1508	.1795	.2124	.1861
.163	.0000					
.177		.1822				
.193	.2887					
.229						
.250	.0270		.0004	.0108	.0256	-.0399
.274		.0826				
.339	.0846					
.362						
.400	.2858					
.402			.0639	-.0175		-.1928
.497		.0481				
.550	.0000					
.563			.0367	-.0503		
.600	.0000					
.650						-.2400
						-.1645

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81032)

ARC11-716 1A14 CR+T12+S12+M3+AT10 UPPER WING

ALPHA(1) = -10.230 BETA(1) = -3.930

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/LW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.6870		
.700	.2170	-.0237
.725		-.0835
.750		-.0941
.760	-.0547	
.775	.1135	.0263
.808	.2069	
.834	.0144	
.850		.2193
.857	.2141	.2214
.865		.2110
.903	.2177	
.905	.2118	.2333
.925	.2177	.2273
.950		.2066
.953	.1858	.2336
.963	.1823	

ALPHA(1) = -10.230 BETA(1) = -1.940

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/LW	CP
.2680	.3640	.4270
.5340	.6730	.7800
.6870		
.700	.2170	.2405
.725		.3434
.750		.2094
.760	.6304	.5587
.775	.4700	.5845
.808	.0421	
.834	.4463	.4834
.850		.4842
.857	.3723	.4973
.865		.4528
.903	.2235	
.905		.5817
.925	.094	.1791
.950	-.0443	.2094
.953		.2397
.963	.0000	.1968
.177	.2069	
.193	.2799	
.229	.0011	.0055
.250		.0596
.274	.1190	.0435
.339	.1172	-.0327
.362	.2091	
.400		.0347
.402	.0607	.0003
.497	.0000	-.1710



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TABULATED PRESSURE DATA - IA14A - VOL. 6

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(801032)

ARC11-716 IA14 01+712+S12K25+AT10 UPPER WING

ALPHA(1) = -10.230 BETA(1) = -1.940

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7600 .8870
.350		.0195 -.0480
.365		.0000
.600		-.2384
.650		-.1662
.700	.1990	-.1963
.725		-.0757
.750		-.0640
.775		-.0043 -.0436
.800		.1296
.834	-.0106	.1480 .1535 .1399
.850		.1721
.865	.1903	.1761
.900	.1706	.1827
.905		.1909
.950		.1597
.955		
.965	.1390	

ALPHA(1) = -10.240 BETA(1) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7600 .8870
.350		.0195 -.0480
.365		.0000
.600		-.2384
.650		-.1662
.700	.1990	-.1963
.725		-.0757
.750		-.0640
.775		-.0043 -.0436
.800		.1296
.834	-.0106	.1480 .1535 .1399
.850		.1721
.865	.1903	.1761
.900	.1706	.1827
.905		.1909
.950		.1597
.955		
.965	.1390	

DATE 10 JAN 75

(081032)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ALPHA(1) = -10.240 BETA(6) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW						
.362	.2043		.0719	.0086	-.1551	
.400		.0713				
.432	.0030		.0431	-.0506		
.497		.0000			-.1717	
.550						-.2462
.565						
.603						
.650	.1822					
.700						
.725						
.750						
.780						
.775						
.828						
.834	-.0311					
.850						
.857						
.865	.0711					
.900	.1206					
.905						
.950						
.953						
.965	.1090					

ALPHA(1) = -10.250 BETA(7) = 2.040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW						
.000	-.1598	-.0509	.2204	.2829	.0969	-.0331
.020				.5964	.5136	.5355
.025			.4213			.4896
.040		.0594				
.045			.4297	.4805	.4745	.4690
.050	-.1067			.3760		.4293
.080			.3714			
.085						
.086		.1998				
.094	-.0920					
.150						
.163						
.177						
.193						







(081032)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+112+512N25+AT10 UPPER WING

ALPHA(1) = -10.220 BETA(9) = 4.030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.190 .0000 .2311 .2456 .2580 .8010

.163 .0000 .2548

.177 .2703

.193 .0140

.229 .1789

.250 .1855

.274 .1947

.339 .0843

.362 .0790 .0201 -.1514

.400 .0643

.472 .0274 -.0518

.477 .0000

.550 .0000

.565 .0000

.600 .0000

.650 .0000

.700 .0000

.725 .0000

.760 .0000

.775 .0000

.808 .0000

.834 .0000

.850 .0000

.857 .0000

.863 .0000

.920 .0000

.935 .0000

.950 .0000

.953 .0000

.965 .0000

ALPHA(1) = -10.220 BETA(9) = 6.090

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.0000 .1951 .1883 .1803 .1411 -.1571

.020 .1951

.025 .1951

.040 .1951

.049 .1951

.050 .1951



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01032)

ARC11-716 1A14 01+112+312M25+1110 UPPER WING

ALPHA(1) = -10.250 BETA(9) = 6.090

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6750 .7800 .8870
X/CW					
.080				.3926	
.085			.3984		
.086		.1409			
.094	-.0810			.2476	.2552 .2636 .2029
.150		.0000	.2640		
.163					
.177		.2517			
.193					
.229	.0056			.1246	.1172 .1034 .0069
.290			.1870		
.274		.1707			
.339	.1687			.0789	.0226 -.1446
.362			.0690		
.400				.0146	-.0980
.402	.0000				
.497			.0000		
.565					
.600					
.650					
.700	.1221			-.2179	-.1805
.725					
.750					
.760					
.775					
.800					
.834	-.0605				
.850					
.857					
.865	-.0654				
.900	-.0299				
.905					
.950					
.953					
.965					

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ABC11-716 1A14 01+112+S12N25+AT19 UFFER WING

ALPHA( 1) = -10.250      BETA( 11) = 10.119

DEPENDENT VARIABLE CP

### SECTION (1) UPPER WING

	1980	1979	1978	1977	1976	1975	1974	1973	1972	1971	1970	1969	1968	1967	1966	1965	1964	1963	1962	1961	1960	1959	1958	1957	1956	1955	1954	1953	1952	1951	1950	1949	1948	1947	1946	1945	1944	1943	1942	1941	1940	1939	1938	1937	1936	1935	1934	1933	1932	1931	1930	1929	1928	1927	1926	1925	1924	1923	1922	1921	1920	1919	1918	1917	1916	1915	1914	1913	1912	1911	1910	1909	1908	1907	1906	1905	1904	1903	1902	1901	1900	1899	1898	1897	1896	1895	1894	1893	1892	1891	1890	1889	1888	1887	1886	1885	1884	1883	1882	1881	1880	1879	1878	1877	1876	1875	1874	1873	1872	1871	1870	1869	1868	1867	1866	1865	1864	1863	1862	1861	1860	1859	1858	1857	1856	1855	1854	1853	1852	1851	1850	1849	1848	1847	1846	1845	1844	1843	1842	1841	1840	1839	1838	1837	1836	1835	1834	1833	1832	1831	1830	1829	1828	1827	1826	1825	1824	1823	1822	1821	1820	1819	1818	1817	1816	1815	1814	1813	1812	1811	1810	1809	1808	1807	1806	1805	1804	1803	1802	1801	1800	1799	1798	1797	1796	1795	1794	1793	1792	1791	1790	1789	1788	1787	1786	1785	1784	1783	1782	1781	1780	1779	1778	1777	1776	1775	1774	1773	1772	1771	1770	1769	1768	1767	1766	1765	1764	1763	1762	1761	1760	1759	1758	1757	1756	1755	1754	1753	1752	1751	1750	1749	1748	1747	1746	1745	1744	1743	1742	1741	1740	1739	1738	1737	1736	1735	1734	1733	1732	1731	1730	1729	1728	1727	1726	1725	1724	1723	1722	1721	1720	1719	1718	1717	1716	1715	1714	1713	1712	1711	1710	1709	1708	1707	1706	1705	1704	1703	1702	1701	1700	1699	1698	1697	1696	1695	1694	1693	1692	1691	1690	1689	1688	1687	1686	1685	1684	1683	1682	1681	1680	1679	1678	1677	1676	1675	1674	1673	1672	1671	1670	1669	1668	1667	1666	1665	1664	1663	1662	1661	1660	1659	1658	1657	1656	1655	1654	1653	1652	1651	1650	1649	1648	1647	1646	1645	1644	1643	1642	1641	1640	1639	1638	1637	1636	1635	1634	1633	1632	1631	1630	1629	1628	1627	1626	1625	1624	1623	1622	1621	1620	1619	1618	1617	1616	1615	1614	1613	1612	1611	1610	1609	1608	1607	1606	1605	1604	1603	1602	1601	1600	1599	1598	1597	1596	1595	1594	1593	1592	1591	1590	1589	1588	1587	1586	1585	1584	1583	1582	1581	1580	1579	1578	1577	1576	1575	1574	1573	1572	1571	1570	1569	1568	1567	1566	1565	1564	1563	1562	1561	1560	1559	1558	1557	1556	1555	1554	1553	1552	1551	1550	1549	1548	1547	1546	1545	1544	1543	1542	1541	1540	1539	1538	1537	1536	1535	1534	1533	1532	1531	1530	1529	1528	1
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**MJ/X**

000	-.2124	-.0034	.0778	.0995	-.0937	-.2370	-.2614
020				.5434	.4182	.4175	.3594
025			.3266				
040		-.0130					
045			.3384				
050	-.1499			.4807	.4485	.4106	.3613
060				.4132			
085			.3439				
086		.1156					
094	-.1107			.2877	.2825	.2803	.2130
150		.0000					
163			.2720				
177		.2236					
193							
229	-.0344			.1625	.1540	.1313	.0375
250							
274			.1965				
339		.1442					
362	.1273			.0935	.0427		-.1173
400			.0619				
402							
497	.0050			.0034	-.0546		
950			.0000				
965							-.2520
650					-.1756		
700	.0669			-.1021	-.2130		
725							
750							
760			-.1261			-.1432	-.2232
775				-.0969	-.1233		
808			-.1361				
834							
850	-.1082			-.1143	-.1204	-.1127	
857			-.1636				
865	-.0983			-.1086			-.0526
950	-.1318						
985			-.1055	-.0392			
990				-.0381		-.0396	
993			-.0609				
995	-.0087						
995							

(R81032)

ARC11-716 1A14 01-T12-S12N23+AT10 UPPER WING

ALPMAO(2) = -0.190 BETA(1) = -9.970

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW	X/CW	.2990	.3240	.4270	.5340 .6730 .7800 .8870
.000	-.0968	-.0398	.4564	.5726	.4978 .4149 .4507
.020			.5236	.6210	.5882 .6080 .6038
.025		.1874			
.040			.4593	.3975	.3979 .4093 .3897
.045				.2148	
.050	.0270		.3040		
.060		.3503			
.085					
.086					
.094	.0451			-.0337	-.0067 .0465 .0495
.150		.0002			
.163			.0519		
.177		.2484			
.193					
.229	.0619			-.2198	-.2044 -.2070 -.2239
.250			-.0703	-.0828	
.274					
.339					
.362	.2084				
.400					
.402		-.2933		-.3841	-.4077 -.4805
.497	.0000			.0699	-.0199
.550		.0000			
.565					
.600					
.650					
.700	.2306				
.725				-.1252	-.1919
.750			.0691		
.760		.1296		.2864	.1415
.775		.3203	.3190		
.808		.3303			
.834	.1880			.3015	.3269 .3586
.850		.3078			
.857					
.865	.3362				
.900	.3130		.2901	.2719	.3043
.905		.2894			
.950		.2360		.2844	
.953		.2336			
.965	.2270				



DATE 19 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81032)

ARC11-716 1A14 01-T12-S12N25+AT10 UPPER WING

ALPHA( 2 ) = -8.200 BETA( 2 ) = -7.960

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8670
X/CW				
.000	-.1310	-.0609	.4254	.5374 .4494 .3652 .3963
.020			.6094	.5847 .5841
.025			.5073	
.040		.1003		
.045		.4405		
.050	.0007		.3933	.3919 .3986 .3847
.060		.2994	.2183	
.085				
.086		.3190		
.094	-.0138			
.150		.0000	-.0200	.0039 .0507 .0579
.163			.0596	
.177		.2409		
.193				
.229	.0426		-.1796	-.1967
.250			-.1926	-.2061
.274		-.0691		
.339		-.0600		
.362	.1955		-.3123	-.3737
.400				-.4257
.402		-.2608		
.497	.0000		.0637	-.0335
.550				
.565		.0000		
.600				-.3399
.650			-.1596	
.700	.2285		-.1470	
.725			-.0063	
.750			.1221	.1295
.760		.0195		
.775		.2992	.008	.2241
.808				
.834	.1271		.2639	.2931
.850		.2812		.2956
.857				
.865	.2956			
.900	.2803		.2811	
.905		.2695	.2556	.2842
.930			.2362	.2643
.953		.2200		
.965	.2083			

(881032)

ARC11-716 IA14 01+112+512N25+AT10 UPPER WING

ALPHA(2) = -0.210 BETA(3) = -5.960

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6750 .7800 .8870
X/CW					
.000	-.1688	-.0971	.3928	.4965	.3975 .3135 .3406
.020				.5946	.5487 .5690 .5660
.025			.4897		
.040		.0567			
.045			.4262		
.050	-.0370			.3918	.3681 .4009 .3840
.060			.2250		
.085			.2996		
.086		.2804			
.094	-.0390			.0030	.0319 .0737 .0685
.150		.0000			
.163			.0760		
.177		.2354			
.193					
.229	.0180			-.1627	-.1642 -.1575 -.1897
.274		-.0412	-.0534		
.339					
.362	.1861			.0193	-.0203 -.3623
.400			-.1340		
.402				.0547	-.0704
.497	.0000		.0000		
.550					-.2529
.563					
.600					
.650					
.700	.8125			-.0277	-.1706
.725					-.0825 -.0936
.750			-.0434		
.765			.8072	.1060	.0278
.775					
.806					
.834	.0285			.2271	.2227 .2155
.850		.2229			
.857					
.885	.2143				
.900	.2176			.2404	.2289 .2182
.905		.2264			
.930				.2079	.2290
.945		.1917			
.965	.1703				



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+712+S12N25+AT10 UPPER WING (R81U32)

ALPHA( 2) = -0.220 BETA( 4) = -1.980

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.020	-.2167	-.1223	.3237	.4134 .2990 .1947 .2172
.020				.5760 .5207 .5424 .5303
.025			.4505	
.040		.0323		
.045			.4110	
.050	-.0020			.4026 .4060 .4065 .3869
.060			.3064	.2817
.085				
.086		.2481		
.094	-.0363			.0712 .1034 .1335 .1067
.150		.0000		
.163		.2384	.1197	
.177				
.193				
.229	-.0073			-.0666 -.0737 -.0618 -.1270
.250			.0334	
.274		.0218		
.339				
.342	.1832			.0248 -.0296 -.2389
.400		-.0228		
.402				.0322 -.0783
.497	.0000	.0000		
.590				-.2847
.600				-.2000
.650				
.700	.1834		-.1949	
.725			-.0533	-.1417 -.1680
.750		-.0367		
.775			-.0242 -.0799	
.800		.0293		
.808			.0624 -.0085 .0073	
.834	-.0291			
.850				
.857		.0998		
.865	.0853			
.900	.1031		.1903 .1674	.1011
.905		.1367		
.920			.1536	.1546
.933		19		
.965	.0993			

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ARC11-716 1A14 01+T12+S12+23+AT10 UPPER WING (R81032)

ALPHA( 2) = -8.150 BETA( 5) = .010

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.640	.4270	.5340	.6730	.7800	.8870
Z/CW							
.000	-.2355	-.0682	.3022	.3880	.2380	.1304	.1470
.020				.5361	.5023	.5270	.5076
.025				.4195			
.040		.0687					
.045							
.050	-.1066		.3757	.3960	.4051	.4089	.3914
.080				.2684			
.085							
.086		.2276					
.094	-.0919						
.150				.1002	.1342	.1561	.1343
.163		.0000					
.177			.1332				
.193		.2113					
.229	-.0107			-.0256	-.0313	-.0330	-.0912
.250			.0570				
.274							
.339		.0496					
.362	.1741						
.400				.0319	-.0290		-.2239
.402			.0019				
.497	.0000			.0211	-.0827		
.550			.0000				
.583							
.600							
.650							
.700	.1646						
.725				-.0373	-.2043		
.759							
.760							
.775			-.0773	-.0474	-.0987		
.800			-.0149				
.834	-.0464						
.850				-.0233	-.1680	-.0499	
.857			.0224				
.865	.0379						
.900	.0498			.0773			.0511
.905			.0839	.1123			
.930				.1088			.0874
.953			.939				
.963	.0390						





DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14 - VOL. 6  
ARC11-716 IAI14 01+712+912N3+AT10 UPPER WING (081032)

ALPHA(2) = -0.240 BETA(7) = 4.040

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730
X/CW	.000	.020	.025	.045	.050
.000	-.2779	-.0136	.2163	.2901	.1339
.020			.3670	.3798	.4823
.025		.0804			.4815
.045			.3500	.4079	.3942
.050	-.1301			.3022	.3629
.080			.3000		
.085		.1775			
.366				.1557	.1638
.094	-.1019				.1826
.150		.7000			.1357
.163			.1810		
.177		.2172			
.193				.0321	.0265
.229	.0219		.1115		.0164
.230					-.0742
.274		.1064		.0290	-.0281
.362	.1843				-.2086
.400			.0311		
.402	.0000			.0006	-.0860
.497			.0000		-.2922
.550					-.2090
.565				-.0951	
.600					-.2106
.650					
.700	.1264				
.725					
.750			-.1105		-.1712
.760				-.0432	-.2169
.775			-.7958		
.808					
.834	-.0782			-.0087	-.1143
.850			-.3769		-.1197
.897					
.927	-.0527				
.900	-.0231			-.0113	-.0168
.905			.0193	-.0249	
.930				.0451	-.0048
.953			.0214		
.965	-.0067				



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81032)

ARC11-716 1A14 01+12+312MS+AT10 UPPER WING

ALPHA( 2 ) = -0.220 BETA( 8 ) = 6.070

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP				
Y/BW		.2990	.3540	.4270	.5340	.6750 .7800 .8870
X/CW						
.000	-.2629	-.0284	.1461	.2616	.0860	-.0369 -.0384
.020				.5190	.4439	.4647 .4221
.025			.3219			
.040		.0439				
.045			.3206			
.050	-.1309			.4034	.4046	.3917 .3378
.060				.3077		
.065			.2692			
.066		.1342				
.094	-.1021			.1695	.1823	.1967 .1472
.150		.0000				
.163			.1880			
.177		.1960				
.229	.0114			.0523	.0513	.0349 -.0376
.290			.1206			
.274		.1095				
.339	.1373			.0535	-.0206	-.1953
.362			.0388			
.400				-.0084	-.0862	
.402	.0006		.0300			-.2937
.497					-.2091	
.590				-.1042		
.565						
.620						
.650						
.700	.1048					
.725						
.750						
.760						
.775						
.806						
.834	-.0964					
.950						
.857						
.863	-.0821					
.900	-.0636					
.909						
.950						
.953						
.965	-.0340					

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(RB1052)

ARC11-716 1A14 01+712+912N23+AT10 UPPER WING

ALPHA(2) = -0.230 BETA(9) = 0.080

SECTION (1) UPPER WING		DEPENDENT VARIABLE CF			
Y/BW	X/CW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
.000	-.2379	-.0916	.0993	.1736	.0394 -.0903 -.1002
.020				.5194	.4267 .4390 .3890
.025			.2963		
.040		.0215			
.045			.2994		
.050	-.1337			.4224	.4072 .5025 .3449
.060				.3376	
.065			.2922		
.086		.1261			
.094	-.1159				
.130		.0500		.2007	.2021 .2375 .1929
.163			.2093		
.177		.1625			
.193					
.229	-.0199			.0634	.0726 .0534 -.0402
.250			.1427		
.274		.1217			
.339					
.362	.1242			.0426	-.0110 -.1039
.400			.0511		
.402					
.497	.0060			-.0165	-.0867
.530			.0000		
.565					
.600					
.650					
.700	.0063			-.1126	-.2080 -.2941
.725					
.750					
.760					
.775					
.806					
.834	-.1056				
.840					
.857					
.865	-.6273				
.900	-.1124				
.905					
.950					
.955					
.965	-.0461				



(R91 U32)

45C11-716 1A14 01+Y12+S12N25+AT10 UPPER WING

ALPHA( 2) = -0.249 BETA( 5) = 19.100

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP						
X/CW	Y/ZW	.2990	.3640	.4270	.5343	.6750	.7600	.8670
.000	-.2709	-.0631		.0771	.1098	.0071	-.1225	-.1304
.020				.2693	.4915	.4220	.4267	.3617
.025								
.040	-.0009							
.045				.2640	.4292	.4087	.3634	.3497
.060	-.1714				.3569			
.085				.2835				
.086		.1151						
.094	-.1363							
.130			.0000	.2194	.2244	.2225	.2231	.1716
.163								
.177			.1701					
.229	-.0396				.0638	.0976	.0767	-.0177
.230				.1484				
.274								
.339			.1064					
.362	.1080			.0419	.0353	.0363		-.1610
.400								
.402					-.0384	-.0793		
.497	.0000			.0000				-.2786
.530						-.2075	-.2024	
.565								
.600								
.650					-.1396			
.700	.0702							
.725								
.750				-.1357			-.1660	-.2407
.760								
.775				-.1344	-.1362	-.1405		
.806								
.834	-.1093				-.1330	-.1335	-.1372	
.850				-.1684				
.857								
.865	-.1005							
.900	-.1324				-.1370	-.0641		-.0682
.905				-.1277				
.940					-.1035		-.0624	
.953				-.0377				
.965	-.0458							

(RB1052)

ARC11-716 IA14 01+71P+S12M2S+AT10 UPPER WING

ALPHA(3) = -6.210 BETA(1) = -10.020

SECTION : 1 UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2993	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0813	-.0061	.5169	.6328	.5628	.4888	.5250
.025			.578	.5485	.5836	.5669	
.050			.5932				
.075		.1162					
.100			.4159				
.125	-.0181		.3426	.3342	.3461	.3248	
.150			.1526				
.175		.3409	.2482				
.200	.0094						
.225		.0000		-.0976	-.0810	-.0271	-.0115
.250		.0057					
.275		.2169					
.300	.0279			-.2492	-.2637	-.2610	-.2655
.325		-.1354					
.350		-.1094					
.375	.1922			-.4016	-.4616		-.4975
.400			-.3355				
.425	.0000			-.0372	-.5107		
.450		.0000					
.475							
.500							
.525							
.550							
.575							
.600							
.625							
.650							
.675							
.700	.1427			.0547			
.725				-.1304			
.750							
.775							
.800							
.825							
.850							
.875							
.900							
.925							
.950							
.975							
.995							



(RB1032)

ARC1:-716 1A14 Q1+T12+S12N25+AT10 UPPER WING

$$\text{ALPHA}(3) = -6.229 \quad \text{BETA}(2) = -7.969$$

SECTION ( ) UPPER WING

X/CW	.2995	.3640	.4270	.5340	.6730	.7800	.0070
.000	-.1244	-.0295	.4031	.3945	.5156	.4406	.4758
.020			.4336	.5564	.5242	.5335	.5396
.025		.1641					
.040			.3688	.3260	.3222	.3304	.3126
.045	-.0446			.1435			
.080			.2390				
.085		.3500					
.086							
.094	-.0446						
.130		.0000		-.0924	-.0800	-.0300	-.0172
.163			-.0011				
.177		.2541					
.193							
.229	.0171			-.2436	-.2552	-.2564	-.2711
.250			-.1332				
.274		-.1043					
.339							
.362	.1760			-.3614	-.4450		-.5010
.400			-.5206				
.497	.0000		.0000	.7291	-.1487		
.522							
.565							-.5340
.600							
.650							
.700	.1847			-.0260			
.725			-.0222		-.1451		
.750							
.760							
.775							
.808			.2761	.2561	.2096		
.834	.0636						
.850				.2760	.2033	.2913	
.857			.2722				
.865	.2639						
.900	.2642			.2748	.2436		.1657
.905			.2639				
.950			.2639	.2297		.2634	
.953			.2168				
.965	.2042						

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(RB1032)

ARC11-1.0 1A14 01-T12-S12M25+AT10 UPPER WING

ALPHA(3) = -6.230 BETA(3) = -5.960

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1530	-.0494	.4475	.5532	.4659	.3825	.4157
.020				.5432	.5031	.5126	.5121
.025			.4482				
.040		.1406					
.045			.3746				
.050	-.0690			.3245	.3149	.3163	.2939
.080				.1492			
.085			.2384				
.086		.2886					
.094	-.0619						
.150		.0000		-.0800	-.0706	-.0250	-.0231
.163							
.177			.0167				
.193		.1920					
.229	-.0038			-.2470	-.2477	-.2535	-.2710
.250			-.1078				
.274		-.0902					
.339							
.362	.1806			-.3904	-.4264		-.4872
.400			-.2924				
.402							
.497	.0000			.0355	-.0423		
.550			.0000				
.565							
.600							
.630							
.650							
.700	.1844			-.0432	-.1729	-.1613	-.5024
.725							
.750			-.0468			-.0462	.0323
.760				.0310	.0059		
.775			.1482				
.908							
.934	.0071						
.950			.2050				
.957				.1920	.2145	.2296	
.965	.1880						
.990	.1879			.2200			.2401
.995			.2218		.2175		
.990				.1804		.2278	
.993			.1854				
.985	.1649						

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ARC11-716 IA14 01+T12+S12N25+AT10 UPPER WING

(RB1032)

ALPHA(3) = -6.120 BETA(4) = -1.980

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/SW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1940	-.0546	.3844	.4690	.3673	.2790	.3037
.020				.5109	.4573	.4772	.4800
.025			.4109				
.040		.0618					
.045			.3434				
.050	-.1233		.3142	.3011	.3103	.2974	
.060			.1589				
.065			.2329				
.066		.2320					
.094	-.0844			-.0454	-.0294	.0588	.0037
.150		.0000					
.163			.0401				
.177		.1879					
.229	-.0318			-.2112	-.1906	-.1972	-.2337
.250			-.0764				
.274		-.0685					
.339							
.362	.1511			-.0861	-.0677		-.3964
.400			-.1097				
.402	.0000			-.0003	-.1006		
.497			.0000				
.550							-.2878
.565					-.2290		
.600							
.650				-.0903			
.700	.1800			-.1994			
.725							
.750					-.1598		-.1426
.760			-.0607				
.775			-.0043	-.0650	-.1078		
.808							
.834	-.0313			-.0390	-.0521	.0126	
.850			.0188				
.857							
.865	.0384			.0679			.1109
.900	.0336			.1392			
.905			.0889	.1155			.1563
.950							
.953			.1004				
.965	.0619						

(881032)

ARC11-716 1A14 01+112+S12N23+AT10 UPPER WING

ALPHA(3) = -6.130 BETA(3) = .000

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/W	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.2115	-.0110	.3545	.4254 .3136 .22 .2511
.020				.4893 .4367 .4719 .4704
.025			.3798	
.040		.1065		
.045			.3269	
.050	-.1446		.3037	.3141 .3197 .3036
.060			.1667	
.065		.2190		
.066		.2220		
.094	-.1026			
.150			-.0153	.0151 .0392 .0267
.163		.0000		
.177		.0437		
.193		.1729		
.279	-.0069			
.290			-.1397	-.1481 -.1426 -.1996
.274		-.0421		
.339		-.0521		
.362	.1431			
.400			-.0555	-.0857 -.2936
.492		-.0704		
.497	.0000		.0056	-.1116
.550		.0050		
.565				
.600				
.650	.1423			
.700			-.0667	-.2382
.725			-.2086	
.730				-.1801 -.1736
.760		-.0686		
.775			-.0684	-.1269
.806		-.0413		
.834	-.0661		-.0602	-.0936 -.0319
.850				
.877		-.0312		
.885	.0153			
.920	-.0102		-.0061	.0654
.905		.0235	.0441	
.950		.0522	.0754	
.953		.0391		
.965	.0123			

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ARC11-716 1A14 01+112+S12N25+AT10 UPPER WING (R81032)

ALPHA(3) = -6.120 BETA(6) = 2.030

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP						
X/BW		.2990	.3640	.4270	.5340	.6730	.7800	.8870
x/CM								
.000	-.2309	.0124		.3224	.3954	.2759	.1707	.1969
.020					.4798	.4355	.4613	.4500
.025				.3457				
.040		.1080						
.045				.2979				
.050	-.1478				.3120	.3247	.3264	.3036
.080					.1776			
.085				.2176				
.086		.1867						
.094	-.1138				.0221	.0466	.0613	.0381
.150		.0090						
.163			.0742					
.177		.1603						
.193								
.229	.0135							
.250				.0119	-.0861	-.1156	-.1160	-.1845
.274		-.0021						
.339								
.362	.1317				-.0480	-.0789		-.2603
.400				-.0603				
.402								
.497	.0000				-.0109	-.1151		-.3214
.550				.0000			-.2415	
.565								
.600								
.650								
.700	.1245				-.2130			-.2048
.725					-.0989			
.750				-.1019			-.1921	
.760					-.0871	-.1417		
.775				-.0726				
.808								
.834	-.0824							
.850				-.0682	-.0692	-.1252	-.0986	
.857								
.865	-.0250							
.900	-.0404				-.0590			.0293
.905				-.0129		-.0325		
.950					.0234		.0257	
.955				-.0024				
.965	-.0273							

(RBLU32)

ARC11-716 1A14 01+112+312M25+AT10 UPPER WING

ALPHA(3) = -6.110 BETAO (7) = 4.060

SECTION (3) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2371	.0070	.2461	.3647	.2255	.1174	.1433
.020				.4704	.4138	.4355	.4184
.025			.3183				
.040		.0029					
.045			.2868				
.050	-.1424			.3194	.3169	.3102	.2874
.080			.2234	.1990			
.085							
.086		.1609					
.094	-.0992			.0482	.0562	.0725	.0410
.150			.0200				
.163							
.177		.1648	.0938				
.193							
.229	.0295			-.0666	-.0824	-.0958	-.1632
.250			.0352				
.274		.0248					
.339				-.0416	-.0811		-.2853
.362	.1303		-.0426				
.400				-.0256	-.1175		
.402			.0000				
.497	.0000						
.590							
.565							
.600							
.650							
.700	.1019			-.1128	-.2094	-.2439	-.3284
.725							
.750							
.780							
.775							
.808							
.834	-.0999						
.850							
.857							
.865	-.0703						
.900	-.0785						
.905							
.950							
.953							
.965							



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TABULATED PRESSURE DATA - IAI14 - VOL. 6

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(R91032)

ARC11-715 IAI14 01+112+312+25-AT:D UPPER WING

ALPHA( 3) = -6.190 BETA( 8) = 6.080

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP			
X/CW	Y/BW	.4270	.5340	.6790	.7800
.000	-.3345	-.0120	.1624	.3176	.1729
.020				.4525	.0539
.025			.2636	.4027	.4264
.040		.0537			.4037
.045			.2633		
.050	-.1464		.3219	.3291	.3208
.060			.2203		.2963
.065		.2294			
.086		.1293			
.094	-.0924			.0825	.0926
.150		.0300		.1069	.0693
.163			.1183		
.177		.1433			
.193					
.229	.0218			-.0233	-.0448
.250			.0362	-.0549	-.1311
.274		.0455			
.339					
.362	.1162			-.0256	-.0746
.400					-.2507
.402			-.0179		
.497	.0000			-.0353	-.1203
.550		.0000			
.565					-.3274
.580				-.2425	
.650				-.2116	
.700	.0001		-.1279		
.725				-.2044	-.2476
.750					
.760		-.1243		-.1218	-.1639
.775		-.1163			
.808					
.834	-.1099			-.1331	-.1577
.850					-.1579
.857		-.1556			
.865	-.0096				
.900	-.1127			-.1233	-.1109
.905		-.0721		-.1109	-.0489
.950			-.0341		
.953		-.0434			
.965	-.0091				

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TABULATED PRESSURE DATA - IA14A - VOL. 6

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(RB1032)

ARC11-716 IA14 -1+T12+S12+25+AT10 UPPER WING

ALPHA(3) = -6.190 BETA(3) = 9.090

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5140 .6730 .7800 .8870

X/CW

.000	-.3499	-.0294	.1275	.2193	.1448	.0118	.0242
.020			.2227	.4215	.4018	.4172	.3964
.040		.0313					
.045			.2743				
.050	-.1678			.3288	.3438	.3256	.2976
.060			.2220	.2479			
.085		.1127					
.094	-.1171			.1222	.1229	.1303	.0840
.150		.0000					
.163			.1418				
.177		.1444					
.193							
.229	.0076			.0164	-.0038	-.3236	-.1111
.290			.0841				
.274		.0994					
.362	.1030			-.0066	-.0612		-.2352
.400			.0018				
.492	.0000			-.0398	-.1155		
.497			.0000				-.3216
.550							
.565							
.600							
.650							
.700	.0720			-.1378	-.2076		
.725			-.1292				-.2566
.750				-.1314	-.1671		
.760			.1316				
.775							
.808							
.834	-.1182			-.1450	-.1632	-.1678	
.850			-.1614				
.857							
.865	-.1048			-.1455			-.0703
.900	-.1514			-.1455	-.1192		
.935			-.1064				-.0713
.950				-.0861			
.953			-.0542				
.965	-.0647						

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 ARC11-716 1A14 01+712+912N25+AT10 UPPER WING (R81032)

ALPHA(3) = -8.170 BETA(10) = 10.090

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
Z/CW				
.000	-.3672	-.0489	.0799	.1674 .1155 -.0147 -.0048
.020			.2136	.4027 .4012 .4175 .3797
.040		.0355		
.060			.2157	
.080	-.1993		.3334	.3526 .3354 .3059
.100			.2624	
.120		.1022	.2152	
.140	-.1426			
.160		.0200	.1459	.1466 .1565 .1045
.180		.1285	.1459	
.200	-.0044			
.220			.0376	.0251 .0059 -.0866
.240		.0587	.0910	
.260	.0879			
.280			.0058	-.0437
.300				-.2180
.320	.0000			
.340			.0000	
.360				-.0396 -.1088
.380				
.400				-.2332
.420				-.3112
.440	.0534			
.460			-.1384	
.480				-.2062
.500				
.520			-.1369	
.540				-.1330 -.1630
.560			-.1412	
.580				
.600			-.1741	
.620				-.1494 -.1595 -.1630
.640	-.1251			
.660				
.680				
.700				
.720				
.740				
.760				
.780				
.800				
.820				
.840				
.860				
.880				
.900				
.920				
.940				
.960				
.980				
.995				
.995				



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881032)

ARC11-716 1A14 01+112+512N25+AT10 UPPER WING

ALPHA( 4) = -4.240 BETA( 1) = -10.010

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP		
Y/BW			
X/CU			
.000	.2990	.3640	.4270
.020	-.0849	.0310	.5530
.040			.6783
.060			.6212
.080			.5657
.100			.4910
.120			.4739
.140			.4855
.160			.4855
.180			.4855
.200			.4855
.220			.4855
.240			.4855
.260			.4855
.280			.4855
.300			.4855
.320			.4855
.340			.4855
.360			.4855
.380			.4855
.400			.4855
.420			.4855
.440			.4855
.460			.4855
.480			.4855
.500			.4855
.520			.4855
.540			.4855
.560			.4855
.580			.4855
.600			.4855
.620			.4855
.640			.4855
.660			.4855
.680			.4855
.700			.4855
.720			.4855
.740			.4855
.760			.4855
.780			.4855
.800			.4855
.820			.4855
.840			.4855
.860			.4855
.880			.4855
.900			.4855
.920			.4855
.940			.4855
.960			.4855
.980			.4855
1.000			.4855

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RB1032)

ARC11-716 1A14 01+T12+S12N23+AT10 UPPER WING

ALPHA(4) = -4.270 BETA(2) = -0.020

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW	X/CW	.4270	.5340	.6730	.7800 .8870
.000	-.1210	.0030	.3198	.6430	.5731 .5167 .5688
.020			.4130	.4776	.4562 .4669 .4722
.040	.2044				
.045			.1135		
.050	-.1002			.2328	.2161 .2448 .2106
.080			.1406	.0482	
.085					
.086	.2896				
.094	-.0612				
.150					
.163		.0000		-.1994	-.1877 -.1279 -.1099
.177			-.0086		
.193		.1284			
.229	.0150				
.250					
.274		-.2234		-.3311	-.3580 -.3554 -.3577
.339	-.1818				
.362	.1241				
.400				-.4509	-.5238 -.5721
.402		-.3900			
.497	.0000				
.550				-.0952	-.5602
.565		.0000			
.600					
.650					
.700	.0084			-.1698	-.5731
.725			-.0680		
.750					
.760		-.0747			-.2349 -.2697
.775		.1993	.1337	.1434	
.808					
.834	.0188				
.850				.2335	.2258 .1684
.857		.2317			
.865	.1893				
.900	.1944			.2441	.2118 -.0926
.905		.2358			
.950				.2041	.2720
.955	.1782	.1913			

ALPHA( 4 ) = -4.290 BETA( 3 ) = -5.970

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/SW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1542	-.0251	.4863	.6041	.5294	.4544	.5171
.020			.3965	.4704	.4395	.4460	.4544
.040		.1275					
.045			.3037				
.050	-.1293			.2365	.2190	.2368	.2071
.080			.1478	.0820			
.085							
.086		.2535					
.094	-.0690						
.190		.0000		-.1684	-.1689	-.1166	-.1101
.163			.0621				
.177		.1254					
.193							
.229	-.0090			-.3209	-.3260	-.3341	-.3599
.250			-.1767				
.274		-.1428					
.339	.1167						
.362				-.4525	-.4943		-.5608
.400			-.3537				
.402	.0000			-.0562	-.3260		
.497			.0000				
.550							
.565							
.600							
.650							
.703	.1234				-.4807		
.725				-.0799			
.750					.0691	-.2118	
.760			-.0727				
.775			.0670	-.0077	.0242		
.806							
.834	-.0223			.1475	.1673	.2006	
.850			.1832				
.857							
.865	.1121						
.900	.1273			.1929	.1832		.0933
.925			.2014				
.930			.1677			.2142	
.933			.1688				
.965	.1493						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1U32)

ARC11-716 1A14 Q1-T12-S12N23-AT10 UPPER WING

ALPHA(4) = -4.250 BETAD (4) = -3.970

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/Z	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.1687	-.0434	.4531	.5680 .4858 .3996 .4536
.023			.3794	.4526 .4154 .4192 .4408
.025		.0833		
.040			.2916	
.045				.2328 .2071 .2195 .2037
.050	-.1518			.0613
.060			.1487	
.065		.2211		
.086				
.094	-.1043			
.100		.0700		-.1574 -.1583 -.1205 -.1030
.163				
.177		.1245		
.193			-.0546	
.229	-.0239			
.250				-.3005 -.3076 -.3178 -.3396
.274			-.1646	
.335		-.1375		
.362	.1142			
.400				-.4186 -.4763 -.5535
.402			-.3317	
.497	.0000			
.550				-.0111 -.0958
.565			.0000	
.600				-.5520
.650				-.1801
.700	.1309			
.725				-.1872
.750				-.0615
.760			-.0758	
.775				-.0448 -.0700
.808		.0115		
.834	-.0437			
.850			.0700	.0916 .1709
.857		.0895		
.865	.0620			
.900	.0494		.1671	.1747 .1921
.905		.1631		
.950			.1539	.1830
.955		.1459		
.965	.1132			

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01032)

ARC11-716 1A14 ON+712+312N23+AT10 UPPER WING

ALPHA: 4) = -4.240 BETA: ( 5) = -1.960

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/S	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.1712	-.0341	.4221	.3213	.4293	.3456	.3850
.020			.3652	.4456	.3961	.3555	.4112
.040		.0749					
.060			.2907	.2337	.2055	.2051	.2242
.080	-.1738			.0695			
.100			.1577				
.120		.2089					
.140	-.1150			-.1381	-.1510	-.1148	-.0685
.160		.0000					
.180			-.0310				
.200		.1315					
.220	-.0388			-.2814	-.3006	-.2854	-.3016
.240			-.1521				
.260		-.1254					
.280	.1180			-.3404	-.4021		-.5019
.300			-.3028				
.320	.0000		.0000	.0056	-.0654		
.340							-.3750
.360					-.2053		
.380	.1317			-.0903	-.1944		
.400			-.0830			-.1519	-.0240
.420			-.0158	-.0640	-.1021		
.440							
.460	-.0632			-.0542	-.0017	.0968	
.480			-.0005				
.500				.0754	.1395		.1560
.520	.0411		.0744	.1217			
.540	.0022		.1031				
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.000							



(R81032)

ARC11-716 1A14 010112+S12+25+AT10 UPPER WING

$$A_{\text{CMAC}}(4) = -4.229 \quad \text{BETA}(5) = .029$$

SECTION: 1) UPPER WING

DEPENDENT VARIABLE C=

194W	-2990	.3640	.4270	.5341	.6730	.7800	.0675
Y/CW							
.000	-.1001	-.0041	.3950	.4802	.3829	.3015	.3462
.020				.4310	.3791	.3698	.4228
.040		.1123	.3503				
.060							
.080			.2767				
.100	-.1876			.2305	.2022	.2238	.2217
.120			.1567	.0756			
.140		.2080					
.160	-.1176						
.180		.0000		-.1177	-.1181	-.0630	-.0468
.200		.1328	-.0149				
.220	-.0210						
.240			-.1269	-.2395	-.2593	-.2579	-.2900
.260		-.1031					
.280	.1184			-.1017	-.1904		-.4663
.300			-.1228				
.320	.0000		.0000	-.0103	-.1202		
.340						-.2395	-.2473
.360	.1204			-.0947	-.2018		
.380			-.0970			-.1696	-.1142
.400			-.0497				
.420				-.0789	-.1243		
.440							
.460	-.0793		-.0486	-.0703	-.0850	.0086	
.480							
.500	.0049						.0934
.520	-.0396			-.0278	.0610		
.540		.0005		.0445		.0905	
.560		.0226					
.580	-.0133						

(R81032)

ARC11-716 1A14 01-T12-S12N25-AT10 UPPER WING

ALPHAO( 4 ) = -4.290 BETAO ( 7 ) = 2.920

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.000	-.2459	.0065	.3490	.4495	.3441	.2563	.2874
.020				.4188	.3790	.3991	.4019
.025		.1096	.3126				
.040			.2535				
.045				.2293	.2251	.2386	.2200
.050	-.1832			.0930			
.060							
.085		.1480					
.086		.1788					
.094	-.1234						
.150		.0000		-.0724	-.0606	-.0396	-.0506
.163							
.177			-.0025				
.193		.1216					
.229	.0063			-.1800	-.2117	-.2285	-.2733
.250							
.274		-.0676					
.339		-.0745					
.362	.1080			-.1227	-.1855		-.4047
.470			-.1175				
.402							
.497	.0000			-.0357	-.1336		-.2983
.550			.0000			-.2614	
.600							
.650				-.1087			
.700	.1016						
.725							
.750				-.1155			
.760							
.775				-.0624			
.808							
.834	-.1037						
.850				-.1111	-.1318	-.0608	
.857				-.0852			
.865	-.0353						
.900	-.0741			-.0968			.0523
.905				-.0466	-.0179		
.950				-.0276		.0473	
.955				-.0358			
.965	-.0604						



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 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N25+AT10 UPPER WING (R81U32)

ALPHA (4) = -4.310 BETA (8) = 4.040

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP							
Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870	
X/CW								
.000	-.3907	.0021	.2634	.4244	.2966	.2081	.2418	
.020				.4132	.3624	.3803	.3784	
.025			.2780					
.040		.0843						
.045			.2455					
.050	-.1635			.2415	.2267	.2324	.2144	
.080				.1138				
.085			.1701					
.086		.1530						
.094	-.1069							
.150		.0000		-.1423	-.0339	-.0254	-.0401	
.163			.0283					
.177		.1369						
.193								
.229	.0235			-.1372	-.1720	-.1900	-.2588	
.250								
.274		-.0404		-.0454				
.339								
.362	.1168			-.1031	-.1503		-.3596	
.400								
.402		-.0921						
.497	.0000			-.0492	-.1399			
.550								
.565		.0000						
.600								
.650								
.700	.0802				-.2117		-.3320	
.725				-.1248				
.750								
.760		-.1347						
.775				-.1181	-.1725			
.808		-.1117						
.834	-.1183							
.850				-.1359	-.1599	-.1251		
.857		-.1241						
.865	-.0768							
.900	-.1008			-.1325			.0150	
.905		-.0658			-.0831			
.950			-.0778			.0005		
.953		-.0730						
.965	-.0825							



ARC11-716 1A14 01-112+S12N25+AT10 UPPER WING

ALPHAD( 4 ) = -4.220 BETA( 9 ) = 0.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.4285	-.0120	.1331	.2094	.2414	.1277	.1546
.020				.3437	.3540	.3749	.3639
.025			.1681				
.040		.0297					
.045			.1873				
.050	-.1722			.2331	.2493	.2512	.2326
.080				.1451			
.085			.1578				
.086		.1046					
.094	-.0974			.0340	.0262	.0401	.0109
.150		.0000					
.163			.0602				
.177		.1185					
.193							
.229	.0188			.0597	-.0824	-.1074	-.1385
.250			.0170				
.274		-.0088					
.339							
.362	.0913			-.0499	-.1054		-.2895
.400			-.0372				
.497	.0000			-.0984	-.1400		
.550			.0000				
.565							-.3483
.600							
.650							
.700	.0331			-.1980	-.2075		
.725							
.750							
.760			-.1681	-.1430	-.1821		
.775			-.1411				
.808							
.834	-.1274			-.1621	-.1757	-.1783	
.850			-.1847				
.857							
.865	-.1188			-.1582			-.0937
.900	-.1638						
.925			-.1284	-.1285			
.930				-.1161		-.0597	
.933			-.1007				
.965	-.1194						



DATE 10 JAN 75

TABULATED PRESSURE DATA - IA144 - VOL. 6

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(R81032)

ARC11-716 IA14 01+T12+812R25+AT10 UPPER WING

ALPHA( 4) = -4.210 BETAO (10) = 10.100

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/C	X/CW	.2995	.3640	.4270	.5340	.6750	.7800	.8870
.000	.4271	-.0362	.0938	.1915	.2261	.0960	.1215	
.025			.1624	.3016	.3637	.3779	.3625	
.040		.0020						
.045			.1624	.2334	.2729	.2718	.2469	
.050	-.2070			.1710				
.060			.1616					
.065			.0926					
.086				.0696	.0373	.0696	.0346	
.094	-.1155							
.150		.0090		.0704				
.163								
.177		.1090						
.193								
.229	.0043			-.0234	-.0521	-.0735	-.1593	
.250			.0361					
.274								
.339		.0091						
.362	.0781							
.400			-.0459	-.0956			-.2693	
.422		-.0274						
.497	.0000			-.0694	-.1368			
.550			.0000					
.563								
.600								
.630								
.700	.0405				-.2100		-.3388	
.725				-.1687				
.750								
.775			-.1611	-.1570	-.1800			
.808				-.1490				
.834	-.1347							
.850				-.1760	-.1738	-.1743		
.857			-.1806					
.865	-.1313							
.900	-.1882			-.1695			-.0518	
.930			-.1502	-.1208				
.950				-.1347			-.0580	
.953				-.1184				
.955	-.1305							

(R81032)

ARC11-716 1A14 01+7112+512N25+AT10 UPPER WING

ALPHA(5) = -2.920 BETA(1) = -10.000

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW					
X/CM					
.000	-.0687	.0503	.5774	.7183	.8711
.020			.4532	.4410	.4350
.025			.4112		
.040		.2665			
.045			.2969		
.050	-.0767		.1950	.1794	.1988
.060			.0014		
.065			.1133		
.066		.3113			
.094	-.0369				
.150			-.2435	-.2390	-.1880
.163		.0000			
.177			-.1181		
.193		.1243			
.229	.0247				
.250			-.3820	-.4154	-.4103
.274			-.2564		
.339		-.1979			
.362	.1382				
.400			-.5017	-.5801	-.6073
.432			-.4424		
.497	.0000				
.530			-.3010	-.6296	
.565			.0000		
.600					-.6181
.653				-.6112	
.700	.0047		-.0628	-.1893	
.725					
.750			-.0441		
.760			.1253	.0358	
.775			.2303		
.808	.0393				
.834			.2207	.2127	-.2600
.850					
.857			.2507		
.865	.2126				
.900	.8213		.2384		-.2054
.905			.2530	.2163	
.930			.1998		-.0906
.953			.2033		
.965	.1995				





DATE 10 JAN 75

TABULATED PRESSURE DATA - TA14A - VOL. 6

(R01032)

ARC11-716 TA14 OL+T12+S12N25+AT10 UPPER WING

ALPMA21 S) = -2.930 BETA' ( S) = -5.970

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
X/CW			
.000	-.1224	-.0163	.3096
.020			.6455
.025			.4257
.040		.1159	.3644
.045			.2608
.050	-.1385		.1851
.060			.0045
.065			.0917
.086		.2361	
.094	-.0835		-.2251
.150		.0000	-.2197
.163			-.1755
.177		-.1017	-.1620
.193		.0956	
.229	-.0110		-.3361
.250			-.3773
.274		-.2155	-.3855
.339		-.1701	-.3939
.362	.0934		
.400			-.4476
.402		-.3716	-.5316
.497	.0020		-.5942
.550			
.565			-.1063
.600		.0000	-.5643
.650			
.700	.3937		-.6196
.725			-.5086
.750		-.0803	-.1867
.760			-.2471
.775		-.0852	-.3297
.800		.0377	
.834	-.0295		-.0055
.850			.0205
.857		.1110	
.865	.0814		.1172
.900	.0742		.1324
.905		.1794	.1414
.950		.1663	
.953		.1577	-.1413
.965	.1350		.1670
			.2135



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 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 Q1+T12+S12N25+AT10 UPPER WING (R81032)

ALPHA( 5 ) = -2.910 BETAD ( 4 ) = -3.960

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7000 .8070
X/CW				
.000	-.1221	-.0310	.4704	.6049 .5240 .4598 .5205
.020				.4165 .3701 .3856 .3856
.024			.3561	
.040		.0832		
.045			.2579	
.050	-.1637		.1898 .1535 .1757 .1333	
.080			.0188	
.085			.1110	
.086		.2109		
.094	-.1043			
.150		.0000	-.2077 -.2227 -.1771 -.1639	
.163				
.177		.1037	-.0841	
.193				
.229	-.0307		-.3311 -.3641 -.3765 -.3910	
.250			-.2056	
.274				
.339		-.1642		
.362	.1063			
.400			-.4406 -.5105	-.5871
.402			-.3688	
.497	.0000			
.550			-.0467 -.2520	
.565			.0000	
.600				-.6075
.650				-.4334
.700	.1050		-.1887	
.725			-.0765	.0392 -.2050
.750				
.760			-.0816	
.775			-.0433 -.0327	
.808			-.0015	
.834	-.0495		.0504 .1112 .1667	
.850			.0102	
.857				
.865	.0478			.0258
.900	.0560		.1448 .1515	
.905			.1032	.1049
.950			.1510	
.953			.1297	
.965	.0902			

ORIGINAL PAGE 14  
 OF POOR QUALITY

(R01USE)

ARC11-716 1A14 01+712+512N83+AT10 UPPER WING

ALPHA( 5 ) = -2.910 BETA( 5 ) = -2.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8670
X/CW							
.000	-.1411	-.0221	.4441	.5572	.4748	.3945	.4720
.025			.3386	.4024	.3615	.3540	.3703
.040		.0754					
.045			.2504				
.050	-.1957			.1645	.1492	.1580	.1247
.080				.0208			
.085			.1176				
.086		.1998					
.094	-.1296						
.150		.0000		-.1921	-.2587	-.1740	-.1545
.163							
.177			-.0737				
.193		.1050					
.229	-.0476						
.250				-.3076	-.3473	-.3636	-.3720
.274			-.1923				
.339		-.1582					
.362	.1019						
.400				-.4274	-.5027		-.5497
.40E			-.3502				
.497	.0000						
.550				-.0155	-.0951		
.565			.0000				-.3592
.600						-.1659	
.650							
.700	.1050			-.0768		-.1914	
.725							
.750							-.0642
.760			-.0666				
.775			-.0221	-.0546	-.0847		
.808							
.834	-.0670						
.850				-.0202	.0490	.1552	
.857			-.0156				
.865	.0304						.1793
.900	-.0071			.0939			
.905			.0673		.1359		
.950				.1293		.1589	
.953			.1016				
.965	.0526						



(R81U32)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

DATE 10 JAN 75

ALPHA( 5 ) = -2.910 BETA( 6 ) = .020

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW		.3340	.4270	.5340	.6730 .7000 .8870
X/CW					
.000	-.1882	-.0102	.4060	.5132	.4249 .3477 .4429
.020				.3873	.3443 .3495 .3761
.040		.1117			
.060			.2316		
.080	-.2175			.1803	.1548 .1596 .1547
.100				.0262	
.120			.1079		
.140		.1932			
.160	-.1328				
.180		.0000		-.1757	-.1852 -.1557 -.1201
.200			-.0846		
.220		.0965			
.240	-.0344			-.2816	-.3208 -.3044 -.3420
.260			-.1780		
.280		-.1431			
.300	.0985			-.3190	-.4055 -.5287
.320			-.2868		
.340	.0000			-.0088	-.0986
.360		.0000			
.380					-.4371
.400				-.2293	
.420				-.2013	
.440				-.0975	-.1500 -.0531
.460			-.1011		
.480				-.0840	-.1332
.500			-.0499		
.520				-.0806	-.0703 .0314
.540	-.0862		-.0348		
.560					
.580					
.600	.0534			-.0308	.0733 .1260
.620	-.0414				
.640			-.0177	.0553	.1139
.660					
.680			.0089		
.700	-.0300				



(R81032)

ARC11-716 1A14 01+112+312M25+1110 UPPER WING

ALPHA( 5) = -2.910 BETAD ( 7) = 2.090

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7800 .8870
.000	-.2549	-.0115	.3676 .4922 .3903 .3104 .3557
.020			.3770 .3343 .3376 .3546
.025		.2789	
.040		.0976	
.045		.2223	
.050	-.2214		.1864 .1553 .1607 .1466
.060			.0381
.065		.1189	
.066		.1604	
.094	-.1427		
.150		.0000	-.1357 -.1594 -.1166 -.1033
.163			
.177		-.0352	
.193		.0989	
.229	-.0001		-.2889 -.2926 -.2999 -.3328
.250			
.274		-.1341	
.339		-.1066	
.382	.0916		
.400			-.2217 -.3033 -.3126
.452		-.2075	
.497	.0000		
.550		-.0704 -.1378	
.565		.0000	
.600			-.2684
.650			-.2600
.700	.0809		-.2140
.725			
.750		-.1237	-.1477
.760			
.775		-.1371	-.1557
.806		-.0867	
.834	-.1152		
.850		-.1474	-.1297 -.0232
.857		-.0960	
.865	-.0445		
.900	-.0872		-.1323
.905		-.0653	.0007
.950			
.955		-.0631	.0679
.955		-.0524	
.965	-.0763		





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ARC11-716 1A14 Q1-T12-S12N23+AT10 UPPER WING

ALPHAD( 9) = -2.930 BETAD ( 9) = 6.070

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/RW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.4149	.0025	.1838	.4157	.3102	.2288	.2575
.020				.3187	.3027	.3207	.3306
.025		.0453	.1720				
.040							
.045			.1563				
.050	-.1933			.1891	.1570	.1740	.1537
.060			.1183	.0505			
.085		.1085					
.094	-.11210						
.150		.0000		-.0902	-.0987	-.0652	-.0826
.163							
.177				-.0001			
.193		.0941					
.229	.0450			-.1918	-.1958	-.2208	-.2870
.250							
.274		-.0684					
.339							
.362	.0826						
.400				-.1105	-.1778		-.3781
.402				-.0680			
.497	.0000			-.0799	-.1744		
.550							
.565				.0000			
.600							
.650							
.700	.0401						
.725				-.1607			
.750							
.760				-.1768			
.775				-.1447	-.1995		
.808				-.1405			
.834	-.1449						
.850							
.857				-.1659	-.1841	-.1306	
.865	-.1202						
.900	-.1375			-.1610	-.1016		.0007
.905				-.1291			
.950				-.1167			-.0134
.955				-.1239			
.965	-.1340						



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ARC11-716 1A14 01+112+912N23+AT1U UPPER WING (881032)

ALPHA(5) = -2.920 BETA(10) = 0.110

SECTION 1 - UPPER WING		DEPENDENT VARIABLE CP				
Y/BW	X/CW	.2930	.3640	.4270	.5340	.6730 .7830 .8870
.000	-.4840	-.0137	.1379	.3160	.2937	.2022 .2312
.020	.020	.2872	.3022	.3353	.3284	
.025	.025	.1456				
.040	.040	.0244	.1441			
.045	.045		.1614	.1765	.1999	.1632
.050	-.2096		.0603			
.060		.1113				
.065	.0796					
.086	-.1019					
.094						
.100	.0000					
.103	.0922					
.177		.0134				
.193						
.229	.0237					
.250						
.274						
.339	-.0627					
.362	.0709					
.400						
.402						
.497	.0500					
.530						
.565						
.600						
.630						
.700	.0299					
.725						
.750						
.760						
.775						
.808						
.834	-.1306					
.850						
.857						
.865	-.1349					
.900	-.1832					
.905						
.950						
.955						
.985	-.1456					



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01+112+S12N25+AT10 UPPER WING

(R81032)

ALPHA( 6 ) = -.750 BETA( 1 ) = -10.040

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
	.2990	.3640	.4270
	.5340	.6730	.7800
	.8870		
X/CW			
.000	.0546	.0588	.6021
.020		.7729	.7454
.040		.3622	.7166
.060		.3492	.3208
.080		.2494	.2882
.100		.2199	
.120		.1047	.0776
.140		-.1021	.0009
.160		.0139	
.180		.2751	
.200		-.0805	
.220		.0000	
.240		.0679	
.260		-.1926	
.280		-.3244	
.300		-.5713	
.320		-.6735	
.340		-.5018	
.360		-.5625	
.380		.0000	
.400		-.6657	
.420		-.2302	
.440		-.0626	
.460		-.0975	
.480		.0539	
.500		.1305	
.520		.1411	
.540		.0101	
.560		-.3456	
.580		.1812	
.600		.1714	
.620		.2111	
.640		.2217	
.660		.1479	
.680		-.2930	
.700		.1779	
.720		.953	
.740		.1741	

(RB1032)

ARC11-716 1A14 Q1+T12+S12N25+AT10 UPPER WING

ALFMAO( 6) = -.743    BETAO( 2) = -.949

SECTION (DIFFER WING

DEPENDENT VARIABLE CP

Y/YW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0197	.0076	.5687	.7391	.7038	.6773	.6913
.020				.3460	.3464	.3068	.2805
.025			.3212				
.040		.1448					
.045			.1874				
.050	-.1184			.0876	.0453	.0705	-.0027
.080				-.1133			
.085			-.0012				
.086		.2349					
.094	-.0985						
.150							
.163		.0000		-.3323	-.3477	-.2947	-.2711
.177			-.1974				
.193		.0420					
.229	-.0399						
.250				-.4629	-.4945	-.5020	-.4739
.274		-.2572					
.339							
.362	.0874			-.5137	-.6420		-.6740
.400			-.4331				
.487	.0000			-.4634	-.6615		-.6683
.550			.0000			-.6676	
.565							
.600							
.650							
.700	-.0900			-.1360	-.2254		
.725							
.750			-.1262			-.4416	-.5852
.775			.0323	-.0114	-.0226		
.800							
.834	-.0337						
.850			.0956	.0826	.1255	-.3359	
.865	.0758						
.900	.0562			.1346	.1690		-.3221
.905			.1652				
.930				.1289		-.2441	
.953		.1508					
.965	.1417						

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(RB1032)

ARC11-716 IA14 01+T12+S12M25+AT10 UPPER WING

ALPHAD ( 6 ) = -.720 BETAD ( 3 ) = -5.990

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.000	-.0203	-.0335	.5284	.7013
.020			.6613	.6332
.025			.3294	.3295
.040		.0703	.2911	.2623
.045			.1751	
.050	-.1534		.0607	.0613
.060			-.1106	-.0126
.065			-.0100	
.086		.1832		
.394	-.1292			
.150			-.3148	-.3428
.163		.0000	-.2947	-.2761
.177			-.1818	
.193		.0274		
.229	-.0552			
.250			-.4046	-.4780
.274		-.2666	-.4929	-.4818
.339		-.2095		
.362	.0640			
.405			-.5017	-.6110
.432		-.4272		-.6739
.497	.0000			
.550			-.2734	-.6410
.565		.0000		
.600				-.6651
.650			-.6543	
.700	-.0191		-.2093	
.725		-.1443		
.750			-.4327	-.5471
.760		-.1261		
.775		-.0558	-.0297	
.808		.0588		
.834	-.0505		.0270	.0750
.850		.0101	-.3046	
.857				
.865	.0452			
.900	-.0543		.0967	.1271
.905		.0955		-.2782
.950			.1189	-.0577
.955		.1213		
.965	.0581			

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TABULATED PRESSURE DATA - IA14A - VOL. 6

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(R81032)

ARC11-716 IA14 01+T12+S12N25+AT10 UPPER WING

ALPHA( 6) = -.710 BETA( 4) = -3.960

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0760	-.0414	.4944	.6800	.6113	.5789	.6111
.020				.3220	.3135	.2832	.2544
.025			.2839				
.040		.0564					
.045			.1751				
.050	-.1944			.0851	.0568	.0589	-.0098
.060				-.1002			
.085			.0030				
.086		.1674					
.094	-.1512						
.150		.0000		-.3043	-.3329	-.2864	-.2721
.163							
.177			-.1580				
.193		.0425					
.229	-.0618						
.250				-.4229	-.4670	-.4731	-.4770
.274		-.2139	-.2741				
.339							
.362	.0726			-.5215	-.5877		-.6624
.400			-.4149				
.402	.0000			-.1846	-.6113		
.497			.0000				-.6562
.550				-.1397	-.2034		
.565							
.600							
.650							
.700	.0301						
.725			-.1153				
.750				-.0716	-.0174		
.760			-.0055				
.775							
.808							
.834	-.0652			-.0094	.0694	.0039	
.850			-.0026				
.857							
.865	.0249						-.1983
.900	-.0261			.0724	.1208		
.905			.0813				
.950				.0993			.1922
.953			.0962				
.965	.0646						



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TABULATED PRESSURE DATA - IA14A - VOL. 6

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ARC11-716 IA14 01+712+812N25+AT10 UPPER WING (R81032)

A: CHAD ( 6 ) = -.700 BETAD ( 5 ) = -2.010

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6750 .7800 .8870
X/CW					
.000	-.1344	-.0358	.4601	.6104	.5907 .5230 .5736
.020				.5250	.2991 .2761 .2510
.025			.2737		
.040		.0502			
.045			.1720		
.050	-.2252			.0920	.0570 .0606 -.0065
.060				-.0620	
.065			.0206		
.086		.1764			
.094	-.1677				
.150		.0000		-.2863	-.3059 -.2744 -.2646
.155					
.177			-.1486		
.193		.0324			
.229	-.0683			-.3912	-.4442 -.4570 -.4665
.250					
.274			-.2556		
.339		-.2042			
.362	.0788				
.400				-.5003	-.5722 -.6502
.402			-.4192		
.497	.0000				
.550				-.1465	-.4268
.565			.0000		
.600					-.6450
.650				-.5470	
.700	.0486			-.2010	
.725			-.1005		-.1660 -.2861
.750					
.760		-.1054			
.775			-.0656	-.0523	
.806		-.0365			
.834	-.0794				
.850			-.0476	-.0401	.0365 .1194
.857					
.865	.0106				
.909	-.0390			.0331	-.0921 -.1047
.905			.0185		
.930				.0798	.1491
.955			.0507		
.965	.0137				

ARC11-716 1A14 01+T12+S12N2S+AT10 UPPER WING

ALPHA( 6) = -.699 BETA( 6) = .349

SECTION / NUMBER WING

V/V	.2947	.3845	.4270	.5340	.6730	.7650	.8870
-----	-------	-------	-------	-------	-------	-------	-------

MS/X

.000	-.1926	-.0479	.4142	.5620	.4943	.4724	.5270
.070				.2982	.2767	.2708	.2528

.920	
.025	.2476

.043	.0846	.1549	
.045			

.050	-.2570	.0053	.0590	.0678	.0064
.080		-.0746			

.083	
.085	.0173
.087	1.665

[illegible]

.150	-.2062	-.2706	-.2334	-.2496
.163				.0000

191	0.087
.177	- .1339

.195	.0465	
.229	-.0447	
		1949 - 1979 - 4740 - 4980

.250	-.3342	-.3373	-.4340	-.4360
.274	-.2331			

.139	-.1923
.362	.0764

506	12100			
409			-4981	-5346
402			-3641	-6321

	.0000	- .0000
	.497	- .497
	.452	- .452
	.5683	- .5683

.550	-.0927	-.1955
.565	.0000	

0.030	- .6121
0.50	- .2667

0.633	0.0584	-0.2049
0.707		
		0.1403

.725	-.1409	
.750	-.0632	-.2224

.760	-.1267
.775	-.1243
	-.1116

-.0669

.834	-.1016	
.030	-.1163	-.0231
		.0699

.057	-.0792
.065	-.0225

0.002	-		
.920	-.0703	-.0665	.0364
0.014		-.0419	.0691

	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	2101	2102	2103	2104	2105	2106	2107	2108	2109	2110	2111	2112	2113	2114	2115	2116	2117	2118	2119	2120	2121	2122	2123	2124	2125	2126	2127	2128	2129	2130	2131	2132	2133	2134	2135	2136	2137	2138	2139	2140	2141	2142	2143	2144	2145	2146	2147	2148	2149	2150	2151	2152	2153	2154	2155	2156	2157	2158	2159	2160	2161	2162	2163	2164	2165	2166	2167	2168	2169	2170	2171	2172	2173	2174	2175	2176	2177	2178	2179	2180	2181	2182	2183	2184	2185	2186	2187	2188	2189	2190	2191	2192	2193	2194	2195	2196	2197	2198	2199	2200	2201	2202	2203	2204	2205	2206	2207	2208	2209	2210	2211	2212	2213	2214	2215	2216	2217	2218	2219	2220	2221	2222	2223	2224	2225	2226	2227	2228	2229	2230	2231	2232	2233	2234	2235	2236	2237	2238	2239	2240	2241	2242	2243	2244	2245	2246	2247	2248	2249	2250	2251	2252	2253	2254	2255	2256	2257	2258	2259	2260	2261	2262	2263	2264	2265	2266	2267	2268	2269	2270	2271	2272	2273	2274	2275	2276	2277	2278	2279	2280	2281	2282	2283	2284	2285	2286	2287	2288	2289	2290	2291	2292	2293	2294	2295	2296	2297	2298	2299	2300	2301	2302	2303	2304	2305	2306	2307	2308	2309	2310	2311	2312	2313	2314	2315	2316	2317	2318	2319	2320	2321	2322	2323	2324	2325	2326	2327	2328	2329	2330	2331	2332	2333	2334	2335	2336	2337	2338	2339	2340	2341	2342	2343	2344	2345	2346	2347	2348	2349	2350	2351	2352	2353	2354	2355	2356	2357	2358	2359	2360	2361	2362	2363	2364	2365	2366	2367	2368	2369	2370	2371	2372	2373	2374	2375	2376	2377	2378	2379	2380	2381	2382	2383	2384	2385	2386	2387	2388	2389	2390	2391	2392	2393	2394	2395	2396	2397	2398	2399	2400	2401	2402	2403	2404	2405	2406	2407	2408	2409	2410	2411	2412	2413	2414	2415	2416	2417	2418	2419	2420	2421	2422	2
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.953	-0.0149
.965	-0.0906

1. The first step is to identify the problem or question that needs to be answered. This involves understanding the context and the specific requirements of the task.

100

DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+S12N25+A110 UPPER WING  
 (R81U321)

ALPHAD( 6 ) = -.690 BETAD ( 7 ) = 2.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/SM	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2667	-.0642	.3587	.5430	.4607	.4136	.4707
.020			.2110	.2952	.2700	.2568	.2448
.040		.0814					
.045			.1393				
.050	-.2734			.1024	.0629	.0603	.0961
.080				-.0434			
.085			.0351				
.086		.1349					
.094	-.1664						
.150		.0000		-.2208	-.2579	-.2384	-.2444
.163							
.177		.0465	-.0900				
.193							
.229	-.0010						
.250				-.3160	-.3658	-.4048	-.4438
.274			-.1891				
.339		-.1361					
.362	.0661						
.400			-.3966	-.5065			-.6012
.402				-.0663	-.1507		
.497	.0000						
.550			.0000				
.600							
.650							
.700	.0406						
.725				-.1262			
.750							
.760			-.1425				
.775				-.1174	-.1528		
.808			-.0952				
.834	-.1236						
.850				-.1277	-.1029	.0810	
.857			-.1128				
.865	-.0558						
.900	-.1003			-.1104	.0169		.0801
.905			-.0890				
.950				-.0404		.0950	
.953			-.0735				
.965	-.0990						

ARC11-716 1A14 01+112+S12N25+AT10 UPPER WING

ALPHA (A) = -.710 BETA (B) = 4.060

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
.000	-.3457	-.0657		.2530	.5256	.4207	.3717	.4195
.020				.1765	.3017	.2537	.2351	.2294
.025			.0459					
.040				.1419				
.045	-.2704				.1076	.0589	.0386	.0041
.050				.0711	-.0332			
.065			.0976					
.086	-.1496							
.094					-.2051	-.2389	-.2331	-.2295
.150			.0000					
.163				-.0739				
.177			.0896					
.193								
.229	.0122				-.3315	-.3505	-.3809	-.4145
.250				-.1698				
.274			-.1152					
.339	.0807				-.3768	-.4532		-.5782
.362				-.2683				
.405					-.1158	-.1642		-.5124
.450	.0000			.0000			-.2631	
.497	.0000					-.2194		
.550					-.1736			
.565							-.1942	-.1036
.600								
.650	.0220							
.700				-.1799				
.750					-.1647	-.1787		
.780				-.1276				
.775								
.808								
.834	-.1464				-.1825	-.1528	-.0303	
.850				-.1454				
.857								
.865	-.0964				-.1827			.0728
.900	-.1376			-.1258	-.0336			
.905					-.1385		.0580	
.950				-.1175				
.955	-.1283							



DATE 19 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81U32)

ARC11-716 1A14 01-T12-S12N25+AT1D UPPER WING

ALPHA( 6 ) = -.720    SETAO ( 9 ) = 6.000

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
Z/CW				
.000	-.4448	-.0370	.1828	.4827 .3936 .3277 .3750
.020				.2649 .2512 .2336 .2351
.025			.1295	
.040		.0030		
.045			.1060	
.050	-.2462		.0927 .0711 .0637 .0331	
.080			-.0365	
.085			.0599	
.086		.0329		
.094	-.1508			
.150		.0000	-.1774 -.2106 -.2017 -.1872	
.163			-.0503	
.177		.0453		
.193				
.229	.0070			
.250			-.2686 -.3307 -.3334 -.3792	
.274		-.1400		
.339		-.0911		
.362	.0639			
.400			-.1724 -.2410	-.5244
.402		-.1669		
.497	.0000			
.550			-.1020 -.1890	
.565		.0000		
.600				-.3199
.650				-.2934
.700	.0142		-.2246	
.725			-.1714	
.750				-.2219 -.1824
.760		-.1915		
.775			-.1473 -.1960	
.806		-.1467		
.834				
.850	-.1557		-.1649 -.1737 -.0959	
.857		-.1601		
.865	-.1236			
.900	-.1637			.0254
.905		-.1303	-.0736	
.950			-.1269	.0133
.955		-.1308		
.965	-.1402			

## TABULATED PRESSURE DATA - 1A14A - VOL. 6

(081032)

ARC11-716 1A14 01+112+51223+AT10 UPPER WING

ALPHA(6) = -.730 BETA(10) = 8.100

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6750	.7000	.8070
X/CW							
.000	-.5576	-.0248	.1279	.2693	.3743	.3109	.3413
.020				.2071	.2530	.2371	.2368
.025			.1022				
.040		.0120					
.045			.0670	.0978	.0825	.0833	.0524
.050	-.2996			-.0012			
.080			.0557				
.085		.0267					
.086							
.094	-.1975						
.150		.0000		-.1802	-.1727	-.1355	-.1581
.163			.0320				
.177		.0637					
.193							
.229	.0459			-.2064	-.2316	-.2678	-.3525
.250			-.1065				
.274		-.0908					
.339	.0664			-.1138	-.1932		-.4392
.362			-.1142				
.400				-.1032	-.2023		
.402	.0000		.0000				-.3783
.497				-.2282			
.550				-.1942			
.565				-.1695	-.2199		
.600				-.1574			
.650	.0161			-.1860	-.2010	-.1629	
.700			-.1863				
.725				-.1772			-.0130
.750			-.1534	-.1240			
.760				-.1512			
.775			-.1472				
.808	-.1643						
.834			-.1766				
.850							
.857							
.865	-.1356						
.900	-.1821						
.925							
.950							
.955							
.965	-.1632						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+112+312N25+AT10 UPPER WING (R01U32)

ALPHA(0) = -.740 BETA(11) = 10.160

SECTION: (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.5527	-.0537	.0834	.2046	.3100	.2873	.2901
.020				.1321	.2436	.2529	.2444
.025			.0685				
.040		-.0160					
.045			.0487				
.050	-.3269			.0569	.1017	.1184	.0671
.060				.0004			
.065		.0456					
.085		-.0021					
.094	-.1600						
.150		.0000		-.0872	-.0942	-.1038	-.1353
.163							
.177			-.0240				
.193		.0833					
.229	.0094						
.250				-.1204	-.1812	-.2438	-.3189
.274			-.0811				
.339		-.0865					
.362	.0598						
.400				-.1106	-.2031		-.3972
.402			-.0917				
.497	.0000						
.530				-.1114	-.1960		
.565			.0000				-.3595
.600							
.630						-.3048	
.650							
.700	-.0011			-.2308			
.725				-.1986			
.750						-.2495	-.2156
.760			-.1807				
.775				-.1816	-.2134		
.806			-.1659				
.834							
.850	-.1750			-.1924	-.1985	-.1388	
.857			-.1892				
.865	-.1525						
.900	-.2009			-.1695			-.0090
.905			-.1698		-.1081		
.930				-.1473		-.0263	
.950							
.955			-.1460				
.955	-.1766						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBIU32)

ARC11-716 1A14 01+112+512N25+AT10 UPPER WING

ALPHA(1) = 2.030 BETA(1) = -10.000

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.2990	.3640	.4270	.5340	.6730 .7000 .8870
.0629	.0628	.5946	.7844	.7697 .7367 .7223
.023		.2458	.2199	.2287 .1420 .0911
.043	.1681	.0971	-.0409	-.0781 -.0787 -.1854
.050	-.0575		-.2318	
.080		-.0948		
.085		.2045		
.086	-.0824		-.4804	-.4746 -.4273 -.4095
.094		.0000		
.150		-.2882		
.163				
.177	-.0272			
.193				
.229	-.0550		-.5690	-.6369 -.6139 -.5779
.250		-.4180		
.274				
.339	-.3222			
.362	.0712		-.6325	-.7719 -.7512
.402		-.5565		
.497	.0000		-.6494	-.7625
.550		.0000		
.565				
.600				
.650				
.700	-.2342		-.2372	-.7339
.725		-.0931		
.750			-.5281	-.6009
.760		-.1632		
.775		.0259		
.808				
.834	-.0187		.0515	-.3485 -.4377
.850		.0715		
.857				
.865	.0938			
.900	.0428		.0836	-.4258
.905		.1200	-.0158	
.930			.0784	-.3734
.953		.1272		
.965	.1714			



(RB1U32)

ARC11-716 1A14 Q1+Y12+S12N25+AT10 UPPER WING

$$\text{ALPHA}(7) = 2.969 \quad \text{BETA}(2) = -8.030$$

SECTION (1) UPPER WING

x/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	.0197	-.0174	.5605	.7575	.7366	.7532	.6905
.020			.2204	.2077	.2096	.1239	.0665
.025		.1036					
.045			.0754				
.050	-.1029			-.1055	-.0884	-.0937	-.2029
.080				-.2434			
.085			-.1189				
.086		.1486					
.094	-.1136			-.4840	-.4785	-.4502	-.4192
.150		.0000					
.163							
.177		-.0435	-.3037				
.193							
.229	-.0725			-.5791	-.6347	-.6143	-.5783
.250			-.4309				
.274		-.3365					
.339				-.8515	-.7604		-.7496
.362	.0521						
.400							
.402	.0000			-.5620	-.7635		-.7204
.497						-.7453	
.550			.0000			-.2310	
.565				-.1464			-.5232
.600							-.6140
.650							
.700	-.1439						
.725							
.750			-.1636				
.760				-.0288	-.4179		
.775			.0125				
.800							
.834	-.0250						
.850				.0193	-.0.31	-.4219	
.857			.0335				
.865	.0639						-.4166
.900	.0036		.0602	.0419	.1415		
.905							
.950				.0350		-.3865	
.953			.0787				
.965	.0870						

(RB1032)

## TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 75

ARC11-716 1A14 01-T12-S12N23+AT10 UPPER WING

ALPHA(1 7) = 2.080 BETA(1 3) = -5.990

## DEPENDENT VARIABLE CP

## SECTION ( 1 ) UPPER WING

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.000	-.0070	-.0466
.020		.5264
.025		.7307
.040		.6743
.045		.6594
.050		.1987
.060		.2039
.085		.1130
.086		.0454
.094		.0663
.190		-.0572
.163		-.0894
.177		-.0981
.193		-.2098
.229		-.2435
.230		
.274		-.1125
.339		
.362		.1176
.400		-.1421
.402		-.4860
.497		-.4692
.590		-.4271
.563		-.4216
.600		
.650		-.2914
.700		-.5406
.725		-.6199
.750		-.6073
.760		-.5817
.775		
.800		-.3797
.834		-.5963
.850		-.7307
.857		-.7496
.865		
.900		-.5036
.925		-.3166
.930		-.7367
.933		
.965		.0000
		-.7191
		-.7397
		-.2271
		-.1493
		-.5295
		-.6096
		-.1572
		-.0727
		-.2740
		-.0059
		-.0306
		.0506
		-.4349
		-.0120
		.0051
		-.4097
		.0924
		-.3648
		.0090
		.0293
		-.0205



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 DL+712+S1 P425+AT1D UPPER WING

(R01.032)

ALPHA(17) = 1.940 BETA(14) = -4.010

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/SW	X/CW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
.000	-.0610	-.0636	.4942	.7010	.6637 .6316 .6225
.020				.2086	.2086 .1193 .0536
.025			.2063		
.040			.0176		
.045				.0760	
.050	-.1864			-.0368	-.0754 -.0901 -.1975
.080				-.2124	
.085			-.0834		
.086			.1122		
.094	-.1865			-.4109	-.4452 -.4066 -.4133
.150			.0000		
.163				-.2439	
.177			-.0189		
.193					
.229	-.0952			-.5040	-.5902 -.5939 -.5755
.250			-.3517		
.274			-.2751		
.339				-.5753	-.7083 -.7457
.362	.0339			-.4910	
.400				-.4048	-.7120
.402			.0000		
.497	.0000				
.550					
.565					
.600					
.650					
.700	-.0848			-.1665	-.2203
.725					-.7305
.750					
.760			-.1609		
.775				-.1182	-.0939
.808			-.0183		
.834	-.0685			-.0336	.0120 -.3841
.850			-.0496		
.857					
.865	.0094				
.900	-.0327			-.0067	.0547 -.3780
.905			-.0230		
.950				.0178	-.2019
.955			-.0027		
.965	-.0314				

(R81032)

ARC11-716 IA14 OL-712-312N25-AT10 UPPER WING

ALPHA ( 7 ) = 1.950 BETA ( 5 ) = -2.000

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.1263	-.0924	.4490	.6459 .6157 .9885 .9934
.020				.1963 .1992 .1224 .0565
.025			.1893	
.040		.0224		
.045			.0733	
.050	-.2394			-.0352 -.0699 -.0862 -.1070
.060				-.2009
.085			-.0761	
.086		.1330		
.094	-.2280			
.150				-.3676 -.4312 -.3952 -.4027
.163		.0000		
.177			-.2279	
.193		-.0146		
.229	-.0924			
.250				-.5210 -.5571 -.5772 -.5662
.274		-.3360		
.339		-.2651		
.362	.0543			
.400				-.5951 -.6727 -.7375
.402		-.4777		
.497	.0000			
.550			-.2925 -.6667	
.563		.0000		
.600				-.6949
.650				-.7006
.700	-.0518			
.725			-.2086	
.750				-.4874 -.5344
.760		-.1637		
.775			-.1194 -.1219	
.806		-.0427		
.834	-.0400			
.850			-.0730 -.0956 -.2530	
.857		-.0667		
.865	-.0126			
.900	-.0601		-.0298	-.3108
.905		-.0391	.0585	
.930			.0033	.0369
.933		-.0565		
.965	-.0500			





(RB1032)

ARC11-716 IA14 01+112+512N23+AT10 UPPER WING

ALPHA ( 7 ) = 1.930 BETA ( 7 ) = 2.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2907	-.1751	.3194	.56-.6	.5310	.5025	.5267
.020				.1598	.1720	.1181	.0673
.025			.1182				
.040		.0348					
.045			.0301				
.050	-.3124			-.0430	-.0700	-.0737	-.1633
.060				-.1890			
.065			-.1003				
.086		.0831					
.094	-.2326						
.150				-.3425	-.3930	-.3720	-.3756
.163		.0000					
.177			-.2073				
.193		-.0158					
.229	-.0133						
.250				-.4247	-.4898	-.5266	-.5496
.274			-.2476				
.339		-.2382					
.362	.0345						
.400				-.4832	-.5755		-.7071
.402			-.3735				
.497	.0000						
.530				-.1757	-.3223		
.565			.0000				
.600						-.5865	-.6795
.650							
.700	-.0182			-.2122			
.725				-.1638			
.750						-.1572	-.4329
.760			-.1936				
.775			-.1109	-.1556	-.1504		
.806							
.834	-.1346						
.850				-.1601	-.0889	.0092	
.857			-.1400				
.865	-.0845						
.900	-.1331			-.1442	.0073		-.1664
.925			-.1289				
.950				-.0871		.0766	
.955			-.1201				
.965	-.1431						



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2673

ARC11-716 1A14 DL+112+S12N25+AT10 UPPER WING

(R81U32)

ALPHA (7) = 1.930 BETA (8) = 4.070

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
X/CM	Y/CM		
.000	-.3822	-.1373	.2054
.020		.5535	.4916
.025		.1821	.1635
.040		.0708	.1007
.045		-.0118	.0561
.050	-.3363	.0252	
.060		-.0098	-.0703
.085		-.1571	-.0075
		-.0314	-.1632
		.0435	
.094	-.2011		
.155		-.3054	-.3612
.163		-.3618	-.3709
.177		.0000	
.193		-.1466	
.229	.0036	.0057	
.250			
.274		-.2274	
.339		-.1632	
.362	.0237		
.400		-.4397	-.5633
.402		-.6919	
.497	.0000	-.3470	
.590		-.1415	-.2932
.585		.0000	
.600			
			-.6554
	-.0299	-.3473	
		-.2191	
		-.1674	
.720			
.760		-.2062	
.775		-.1377	-.3037
.775		-.1555	-.1783
.808		-.1401	
.834	-.1632		
.850		-.1648	-.1271
.857		.0005	
.863		-.1670	
.900	-.1205		
.900	-.1713	-.1607	-.0416
.905		-.0234	
.930		-.1569	-.0671
.930		-.1149	
.953		-.1509	
.963	-.1632		



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881032)

ARC11-716 1A14 01-712-S12N23-AT13 UPPER WING

ALPHA( 7 ) = 1.960 BETA( 9 ) = 6.100

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/SW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7800 .8870
.000	-.4950	-.0895	.1294 .4504 .4669 .4215 .4578
.020			.1458 .1579 .1027 .0643
.025		.0478	
.040		-.0022	
.045		.0184	
.050	-.3320		-.0171 -.0658 -.0781 -.1456
.060			-.1428
.085		-.0527	
.086		.0004	
.094	-.1953		-.2701 -.3365 -.3360 -.3476
.150		.0000	
.163			
.177		-.1184	
.193		.0292	
.229	.0249		-.3447 -.4206 -.4777 -.5139
.274		-.1964	
.339		-.1526	
.362	.0480		-.3204 -.5075 -.6616
.400		-.2458	
.402			
.497	.0000		-.1359 -.2181
.550		.0000	
.565			-.2914
.650		-.2506	
.700	-.0295		-.1899
.725			-.2077 -.1717
.750		-.2127	
.760			
.775		-.1697 -.2053	
.808		-.1617	
.834	-.1846		-.1793 -.1609 -.0338
.850		-.1816	
.857			
.865	-.1418		
.900	-.1833		-.1705 -.0525 .0209
.905		-.1647	
.920		-.1379	
.935		-.1580	
.965	-.1854		



(R81032)

TABULATED PRESSURE DATA - 1A14A - V.L. 6

ARC11-716 1A14 01+112+512N25+AT10 UPPER WING

ALPHA(7) = 1.950 BETA(10) = 0.100

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP									
Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870			
Z/CW										
.000	-.6199	-.0426	.0947	.2302	.4449	.3966	.4228			
.020			.0426	.1572	.0929	.0503				
.025			.0457							
.040		-.0004								
.045			.0192							
.050	-.3841		-.0496	-.0468	-.0730	-.1453				
.060			-.1212							
.065			-.0210							
.066		-.0004								
.094	-.2084									
.100			-.2129	-.3026	-.3237	-.3417				
.163		.0000								
.177			-.1149							
.193		.0295								
.229	.0447									
.250			-.2907	-.3946	-.4551	-.4982				
.274			-.1857							
.339		-.1503								
.362	.0413									
.403			-.2028	-.3062		-.6166				
.452			-.1849							
.497	.0000			-.1513	-.2477					
.530			.0000							
.565										
.600										
.650										
.700	-.0282									
.725			-.2173		-.2345					
.750										
.760			-.2178							
.775			-.2012	-.2397						
.806			-.1859							
.834	-.2022									
.850			-.2059	-.2080	-.1347					
.857			-.2037							
.865	-.1648									
.900	-.2061		-.1901							
.905			-.1837	-.1042						
.950			-.1619							
.955			-.1760							
.965	-.1946									

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ARC11-716 1A14 01+T12+S12N23+AT10 UPPER WING (R81032)

ALPHA3( 7) = 1.940 BETA3( 11) = 10.140

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP						
Y/BW		.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW								
.000	-.5716	-.0646		.0492	.1642	.2913	.3600	.3788
.020					.0202	.0837	.0791	.0509
.025				.0292				
.040			-.0249					
.045				.0065				
.050	-.4086			-.0457	-.0711	-.0767	-.1434	
.060				-.0975				
.085				-.0284				
.086			-.0334					
.094	-.2580							
.150				-.1694	-.2627	-.3047	-.3257	
.163			.0000					
.177				-.1062				
.193			.0463					
.229	.0209							
.250				-.2420	-.3458	-.4239	-.4739	
.274			-.1575					
.339			-.1250					
.362	.0464			-.1720	-.2544		-.5321	
.400				-.1888				
.402								
.497	.0000				-.1593	-.2430		
.550				.0000				
.565							-.3583	
.600								
.650						-.3498		
.700	-.0270				-.2355			
.725				-.2301				
.750						-.2695	-.2057	
.760				-.2224	-.2399	-.2430		
.775				-.2058				
.808								
.834	-.2039							
.850				-.2329	-.2043	-.1179		
.857				-.2129				
.865	-.1888							-.0072
.900	-.2466			-.1955				
.905				-.1928	-.1157			
.950				-.1616			-.0304	
.953				-.1636				
.965	-.2010							



DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-712-S12N25+AT10 UPPER WING (R81032)

ALPHA(6) = 3.970 BETA(1) = -9.990

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7800 .8870
.0689	.0400	.5671	.7703 .7658 .7165 .6837
.020		.1543	.1053 .1249 -.0002 -.0676
.025	.1393		
.040		.0146	
.045			
.050	-.0438	-.1415	-.1920 -.2001 -.3179
.060		-.3219	
.065		-.1907	
.066	.1532		
.094	-.0902		
.150		-.5666	-.5540 -.5288 -.5114
.163	.0000		
.177	-.3700		
.193	-.0809		
.229	-.0651		
.250		-.7035	-.7030 -.6909 -.6494
.274	-.4825		
.339	-.3710		
.362	.0338	-.7707	-.8105
.400		-.5987	
.402	.0000	-.7744	-.7926
.497		.0000	
.550			
.565			
.600			
.650			
.700	-.3731	-.2452	-.6741
.723		-.2485	
.730			
.760		-.2091	-.5426 -.5838
.775		-.0903	-.5450
.808		.0090	
.834	-.0084		
.850		-.0011	-.4438 -.4535
.857	.0605		
.865	.1001		
.900	.0377	.0272	-.5507
.905		-.0968	
.930		-.3732	
.933		.0157	-.3986
.953	.1029		
.965	.1066		

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01032)

ARC11-716 1A14 D1+712+S12H25+AT10 UPPER WING

ALPHA(0) = 3.990 BETA(2) = -8.000

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3610	.4270	.5340 .6730 .7800 .8870
Z/CW				
.000	.0207	-.0311	.5402	.7453 .7293 .6806 .6494
.020				.0974 .1115 -.0132 -.0899
.025		.0392	.1391	
.040				
.045				
.050	-.0771		-.0092	-.1491 -.1970 -.2111 -.3336
.060				-.3279
.085			-.2096	
.086		.0997		
.094	-.1196			
.130		.0000		-.5692 -.5581 -.5344 -.5222
.163			-.3747	
.177		-.0944		
.193				
.229	-.1012			-.6973 -.7071 -.6894 -.6573
.230			-.4827	
.274		-.3744		
.339				
.362	.0375			-.7671 -.6285 -.8138
.400			-.6006	
.402				-.6748 -.7882
.497	.0000		.0000	
.550				
.565				
.600				
.650				
.700	-.3202			-.1685
.725				-.2490
.750			-.2301	
.760				-.0896 -.5781
.775			-.0317	
.808				
.834	-.0341			-.0567 -.4437 -.4762
.850			-.0065	
.857				
.865	.0401			
.900	-.0194			-.0372
.905			.0256	-.0577
.930				-.0470
.953			.0187	-.4235
.965	-.0173			



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 DL712+S12N25+AT10 UPPER WING (R81032)

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7800 .8870
.000	-.0122	-.0933	.4991 .7185 .8924 .6458 .6167
.020			.0940 .1043 -.0200 -.1021
.025		.1222	
.040		-.0003	
.045		-.0212	
.050	-.1205		-.1512 -.1955 -.2153 -.3399
.060			-.3267
.085		-.2142	
.086		.0903	
.094	-.1614		
.150			-.5527 -.5504 -.5269 -.5270
.163		.0000	
.177		-.3854	
.193	-.0997		
.229	-.1208		
.250			-.6566 -.6885 -.6839 -.6609
.274		-.4369	
.339		-.3312	
.362	.0224		
.400			-.6984 -.8088 -.8134
.402		-.5233	
.497	.0000		-.6603 -.7902
.550		.0000	
.563			
.600			-.7724
.650			-.7854
.700	-.2701		
.723		-.2450	
.750		-.2303	
.760			-.5876 -.6384
.773		-.2189	
.808		-.1936	
.834	-.0502	-.0382	
.850		-.0990	-.1364 -.4972
.857		-.0403	
.865	.0077		
.900	-.0564		-.0687
.905		-.0208	.0242
.950			-.4479
.953		-.0575	
.965	-.0815	-.0193	

(991032)

ARC11-716 IALIA 01+712+512N25+AT10 UPPER WING

ALPMAS( 8 = 3.930 SETAD ( 4 ) = -3.990

SECTION ( UPPER WING DEPENDENT VARIABLE C<sub>D</sub>

V/W	.2090	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0869	-.1326	.4856	.6964	.6629	.6129	.5880
.020				.1037	.1138	-.0083	-.0971
.025			.1201				
.040		-.0304					
.045			-.0129				
.050	-.1704			-.1308	-.1788	-.1975	-.3308
.060				-.3026			
.065			-.1809				
.066		.0516					
.094	-.2016			-.5163	-.5284	-.5137	-.5212
.150		.0000					
.163			-.3068				
.177		-.0687					
.193							
.229	-.1278			-.6115	-.6729	-.6707	-.6593
.250			-.4075				
.274							
.339		-.3141					
.362	.0368			-.6747	-.7767		-.8056
.400			-.5400				
.402							
.497	.0000			-.6611	-.7700		
.530			.0000				
.565							-.7632
.600							
.650							
.700	-.2338			-.2204	-.2387	-.7758	
.723							
.750						-.5770	-.6088
.760			-.2024				
.773				-.1548	-.4048		
.806			-.0370				
.834	-.0685						
.850				-.1143	-.0371	-.4958	
.857			-.0594				
.865	-.0147						-.4374
.900	-.0776			-.0793	.0233		
.905			-.0423				
.950			-.0607			-.4280	
.953			-.0408				
.963	-.0870						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI4A - VOL. 6

(R81032)

ARC11-716 IAI4 01+T12+S12N25+AT10 UPPER WING

ALPHA( 8 ) = 3.930 BETA( 5 ) = -2.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1338	-.1875	.4244	.6591	.6239	.3790	.5582
.020			.1201	.1060	.1101	-.0041	-.0970
.040		-.0256					
.045			-.0076				
.050	-.2183			-.1246	-.1733	-.1947	-.3245
.080				-.2921			
.085			-.1486				
.086		.0836					
.094	-.2449			-.4767	-.5180	-.4998	-.5120
.150		.0000					
.163			-.2894				
.177		-.0520					
.193							
.229	-.1058			-.5486	-.6533	-.6591	-.6508
.250			-.3671				
.274		-.2988					
.333				-.6114	-.7486		-.7938
.362	.0448		-.5226				
.400				-.5050	-.7405		
.402		.0000					-.7503
.497	.0000			-.1881	-.2388		
.550						-.5609	-.5766
.565							
.600							
.650	-.2095						
.700							
.725							
.750			-.1871				
.760				-.1452	-.1432		
.775			-.0485				
.808							
.834	-.0859			-.0920	-.0305	-.4603	
.850			-.0734				
.857							
.865	-.0313						-.3860
.900	-.0903			-.0478	.0203		
.905			-.0577				
.90				-.0295		-.3398	
.923			-.0473				
.965	-.0970						



DATE 10 JAN 79 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBI032)

ARC11-716 1A14 01+112+S12N25+AT10 UPPER WING

ALPHA (8) = 3.940 BETA (6) = .040

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/W .2990 .3640 .4270 .5340 .6730 .7800 .8870

Y/CW

.000	-.2089	-.2318	.3726	.5993	.5814	.5390	.5341
.020				.0848	.0984	-.0051	-.0916
.025			.0886				
.040		-.0106					
.045			-.0222				
.050	-.2689			-.1370	-.1768	-.1957	-.3130
.080				-.2936			
.095			-.1822				
.086			.0868				
.094	-.2714						
.150				-.4541	-.5051	-.4833	-.5002
.163		.0000					
.177		-.2814					
.193		-.0343					
.229	-.0663			-.5675	-.6267	-.6419	-.6369
.230							
.274		-.3727					
.339		-.2983					
.362	.0397			-.6266	-.7105		-.7825
.400							
.402		-.5009					
.497	.0000			-.3076	-.6903		
.550			.0000				
.565							-.7386
.600							
.650						-.7540	
.700	-.1169			-.2439	-.2112		
.725						-.5745	-.5640
.750							
.760		-.2024		-.1850	-.1398		
.775			-.0648				
.808							
.834	-.0967			-.1323	-.0741	-.3358	
.850			-.1164				
.857							
.865	-.0828			-.1058			-.3872
.900	-.1239				-.0024		
.905		-.1096					
.950				-.0753			-.0578
.955		-.0986					
.965	-.1375						



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RB1032)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

A.PH40( 8) = 4.030 BETA0 ( 7) = 2.050

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
X/CW	Y/BW	.2990	.3640	.4270
.000	-.3044	-.2810	.2724	.5640
.020			.5310	.5099
.025			.0503	.0850
.040			.0476	-.0105
.045			-.0091	-.0953
.050	-.3275		-.0449	
.080			-.14	-.1766
.085			-.2856	-.1928
.086			-.1667	-.3137
.094	-.2803		.0471	
.130			.0000	-.4276
.163				-.4849
.177			-.2692	-.4718
.193			-.0335	-.4962
.229	-.0214			
.250			-.4062	-.5083
.274			-.3353	-.5325
.339			-.2698	-.6333
.362	.0156			
.400			-.5224	-.6673
.402			-.4477	-.7786
.497	.0000			
.550			-.2408	-.6445
.565			.0000	
.600				-.7369
.650				-.7216
.700	-.0822			-.2301
.725			-.1867	
.750				-.5414
.760			-.2281	-.5676
.775			-.1557	-.1708
.808			-.1335	
.834	-.1330			
.850			-.1533	-.1038
.855			-.1655	-.0396
.865	-.1035			
.900	-.1662		-.1438	-.3620
.905			-.1598	-.0030
.950			-.1063	.0249
.955			-.1466	
.965	-.1714			

TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 13 JAN 75

(R01032)

ARC11-716 1A14 01+112+812+25+AT10 UPPER WING

ALPHA (°) = 4.020 BETA (°) = 4.070

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7000 .8070
X/CW				
.000	-.4059	-.1693	.1275	.5229 .5095 .4710 .4748
.020			.0733	.0763 -.0221 -.0951
.025		-.0076		
.040		-.0559		
.045			-.0436	
.050	-.3526		-.1112	-.1759 -.1968 -.3086
.080			-.2426	
.085		-.1136		
.086		-.0103		
.094	-.2567			
.150		.0000	-.3720	-.4637 -.4560 -.4789
.163			-.1924	
.177		-.0492		
.193	.0017		-.4101	-.5404 -.6086 -.6171
.229			-.2691	
.250		-.1948		
.274				
.339	.0022		-.4031	-.6184 -.7585
.362		-.3866		
.400			-.1976	-.5207
.402	.0000			
.497				
.550				
.565		.0000		
.600				
.650				
.700	-.0621		-.2254	-.6301
.725		-.1904		
.750				
.760		-.2272	-.1700	-.1863
.775		-.1632		
.808				
.834	-.1720		-.1775	-.1175 -.0362
.850		-.1857		
.857				
.865	-.1390		-.1721	-.0328
.900	-.1917		-.1803	-.0350
.905			-.1359	
.930			-.1741	
.955	-.1878			



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81032)

ARC11-716 1A14 Q1+T12+S12N25+AT10 UPPER WING

ALPHA( 8 ) = 4.010 BETAD ( 9 ) = 6.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/94	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.5344	-.1076	.0611	.3922	.4903	.4418	.4491
.020				.0307	.0899	-.0108	-.0903
.025				-.0074			
.040		-.0444					
.045			-.0347				
.050	-.3030			-.1159	-.1532	-.1832	-.2927
.080				-.2270			
.085			-.0848				
.086		-.0266					
.094	-.2444			-.3336	-.4377	-.4326	-.4543
.150		.0000					
.163							
.177			-.1679				
.193		-.0390					
.229	.0091			-.4242	-.4987	-.5649	-.6011
.250							
.274			-.2382				
.339		-.1962					
.362	.0183			-.4205	-.5692		-.7370
.400							
.432			-.3031				
.497	.0000			-.1872	-.2717		
.550			.0000				
.600							-.6896
.650					-.3238		
.700	-.0467			-.2122			
.725					-.1946	-.3903	
.750							
.780			-.2315				
.775				-.8.80	-.2115		
.808			-.1917				
.834	-.1945						
.850				-.2221	-.1565	-.0350	
.857			-.2022				
.865	-.1321						
.900	-.1956			-.2.90			-.0673
.905			-.1836		-.0602		
.950				-.1709		.0262	
.953			-.1681				
.964	-.1767						

ARC11-716 IAI4 01+T12+S12N25+AT10 UPPER WING (981032)

ALPHA(8) = 4.060 BETA(10) = 8.110

## SECTION: (1) UPPER WING

## DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.6366	-.0380	.0332	.1801	.4502	.4337	.4247
.020				-.0738	.0794	-.0081	-.0940
.025				-.0083			
.040		-.0290					
.045			-.0345				
.050	-.4785			-.1539	-.1597	-.1710	-.2841
.080				-.2020			
.085					-.0738		
.086		-.0120					
.094	-.3085						
.150				-.2620	-.3828	-.4102	-.4408
.163		.0000					
.177			-.1577				
.193		-.0488					
.229	.0406			-.3219	-.4446	-.5304	-.5827
.250			-.2014				
.274		-.1926					
.339							
.362	.0188			-.2165	-.4200		-.7099
.400			-.1900				
.492							
.497	.0000						
.550				-.1631	-.2644		
.565			.0000				
.600							
.650						-.3543	-.3535
.700	-.0234			-.2281	-.2356		
.783						-.2575	-.2259
.750			-.2311				
.760				-.2224	-.2483		
.775			-.2030				
.808							
.834	-.1992			-.2222	-.2044	-.1192	
.890			-.1996				
.857							
.865	-.1838						
.900	-.2098			-.1997			-.0547
.905			-.1816		-.1080		
.950				-.1687		-.0212	
.953			-.1645				
.965	-.1827						





DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(981032)

ARC11-716 1A14 01+T12+S12N23+AT10 UPPER WING

ALPHA( 9 ) = 5.990 BETA( 1 ) = -9.980

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW	X/CW	.2590	.3640	.4270	.5340 .6730 .7800 .8870
.000	.0808	.0219	.5376	.7951	.7554 .6816 .6342
.020			.0779	.0044	.0384 -.1151 -.1967
.040		.0969			
.045			-.0803		
.050	-.0416			-.2237	-.2814 -.3051 -.4363
.080				-.4085	
.085			-.2754		
.086		.1033			
.094	-.0844			-.6499	-.6266 -.6145 -.6373
.150		.0000			
.163			-.4652		
.177		-.1368			
.193					
.229	-.0549			-.7637	-.7699 -.7185 -.6896
.250			-.5372		
.274		-.3983			
.339					
.362	.0283			-.8044	-.8232 -.8584
.400			-.6333		
.402					
.497	.0000			-.6829	-.7128
.530			.0000		
.565					
.600					
.650					
.700	-.4819			-.2591	-.6205
.725			-.4650		
.750					-.5518 -.5639
.760			-.2551		
.775				-.4246	-.6048
.808			-.0359		
.834	.0155			-.2524	-.5726 -.4908
.850			.0186		
.857					
.865	.1420				
.900	.8419			-.1065	-.5497
.905			.0708	-.5039	
.950			-.0239		-.4526
.955			.0803		
.965	.0635				



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(081032)

ARC11-716 IA14 01\*112\*512N23\*AT10 UPPER WING

ALPHAD( 9 ) = 5.960 BETAD ( 2 ) = -7.980

SECTION : 1) UPPER WING DEPENDENT VARIABLE CP

Y/B	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	.0333	-.0512	.4868	.7494	.7149	.6425	.5950
.020				-.0233	.0105	-.1410	-.2240
.025			.0381				
.040		.0100					
.045			-.1022				
.050	-.0775			-.2517	-.3029	-.3269	-.4577
.060				-.4175			
.085			-.2875				
.086		.0399					
.094	-.1182			-.6561	-.6333	-.6249	-.6212
.150		.0000					
.163							
.177			-.4702				
.193		-.1503					
.229	-.1901			-.7612	-.7733	-.7672	-.7521
.250			-.5440				
.274		-.4230					
.339							
.362	.0185			-.6139	-.6314		-.7620
.400			-.6371				
.402	.0000			-.7291	-.7720		
.497			.0000				
.550							
.565							
.600							
.650							
.650							
.700	-.4845			-.2415			
.725							
.750							
.760			-.2679				
.775				-.1440	-.6166		
.808			-.0790				
.834							
.850	-.0203			-.0936	-.5602	-.4955	
.857			-.0300				
.865	.0993						
.900	-.0131			-.0682			
.905			.0141		-.4669		-.6150
.950				-.0632			
.953			.0246				
.965	-.0213						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAL4A - VOL. 6  
ARC11-716 IAL4 Q1+T12+312N23+AT10 UPPER WING (R91032)

ALPHA( 9 ) = 5.940 BETA( 3 ) = -5.960

SECTION : 11 UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8670
X/CW							
.000	-.0101	-.1360	.4476	.6990	.6843	.6120	.5807
.020			.0312	-.0361	-.0024	-.1498	-.2437
.040		-.0864					
.060			-.1059				
.080	-.1205			-.2480	-.3070	-.3344	-.4722
.100			-.2844	-.4152			
.120		-.0079					
.140	-.1586			-.6407	-.6306	-.6257	-.6290
.160		.0000					
.180			-.4543				
.200		-.1537					
.220	-.1431			-.7581	-.7645	-.7610	-.7368
.240		-.3824					
.260							
.280	.0108			-.7825	-.8319		-.8022
.300		-.5664					
.320				-.7165	-.8176		
.340	.0000		.0000				
.360							-.7593
.380				-.2550	-.2559	-.7176	
.400	-.4590						
.420							
.440							
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.995							



(R01V32)

ARC11-719 1A14 01 ♦ T12 ♦ S12N23 ♦ AT10 UPPER WING

$$\text{COST}(9) = 5.95 \quad \text{COST}(4) = -3.99$$

SECTION : UPPER WING

	.2990	.3640	.4270	.5340	.6730	.7800	.8870
--	-------	-------	-------	-------	-------	-------	-------

MD/K

- .0709	- .2206	.4216	.6722	.6515	.5826	.5316
.020			- .0165	.0044	- .1442	- .2405
.040	- .1040	.0368				
.045						
.050	- .1720		- .2253	- .2907	- .3235	- .4704
.060			- .3891			
.085		- .2655				
.086	- .0256					
.094	- .2056					
.150			- .6095	- .6091	- .6104	- .6264
.163	.0000					
.177		- .3723				
.193	- .1199					
.229	- .1334		- .6897	- .7421	- .7471	- .7533
.250		- .4597				
.274	- .3499					
.339			- .7367	- .8440		- .8242
.362	.0294	- .5721				
.400						
.402						
.497	.0000		- .7242	- .8123		- .7753
.550		.0000			- .7463	
.565						
.600			- .2500			
.650	- .4414		- .2828			- .6546
.700						
.725						
.750		- .2661				
.760						
.775		- .0750	- .2357	- .5676		
.808						
.854	- .0703					
.850		- .0690	- .1761	- .4144	- .5218	
.857						
.865	- .0374					
.900	- .0923		- .1339	- .1409		- .6362
.905		- .0524				
.950			- .0991		- .4825	
.953		- .0517				
.961	- .1061					

DATE 10 JAN 75 TAPULATED PRESSURE DATA - 1A14A - VOL. 6

(R51032)

ARC11-716 1A14 01-T12-S12N25+AT10 UPPER WING

ALPHAC(9) = 5.970 BETA(9) = -1.970

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	
X/CW	
.000	.3856
.020	.6497
.025	-.0012
.040	.0403
.045	-.1087
.050	-.0840
.060	-.2147
.065	-.3723
.085	-.2421
.094	.0271
.100	-.2412
.110	.0000
.117	-.3508
.119	-.0981
.129	-.1319
.150	.229
.163	.274
.177	.339
.193	-.3371
.229	.0345
.250	.400
.274	.402
.339	.497
.362	.530
.400	.565
.402	.600
.497	.650
.530	.700
.565	.725
.600	.750
.650	.760
.700	.775
.725	.808
.750	.834
.760	.850
.775	.857
.808	.865
.834	.900
.850	.905
.857	.950
.865	.953
.900	.965
.905	-.1210
.950	-.0743
.953	-.0907
.965	-.0282
	-.1139
	-.2607
	-.2495
	-.7448
	-.7746
	-.8001
	-.8305
	-.8155
	-.7385
	-.7327
	-.6023
	-.6036
	-.6233
	-.4736
	-.5913
	-.6280
	-.4831
	-.5913
	-.5321
	-.6037
	-.6037
	-.4946



DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+S12M25+AT10 UPPER WING (081032)

ALPHA( 9 ) = 5.980 BETA( 6 ) = .030

SECTION 1 ( UPPER WING ) DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5040	.6730	.7600	.8870
X/CW							
.000	-.2039	-.3497	.3321	.6036	.3616	.5241	.4862
.020			.0270	-.0087	.0085	-.1377	-.2377
.040		-.0529					
.060			-.0877				
.080	-.2689			-.2103	-.2728	-.3109	-.4552
.100			-.2324	-.3642			
.120		.0470					
.140	-.2667			-.5541	-.5867	-.5872	-.6037
.160		.0000					
.180			-.3322				
.200		-.0896					
.220	-.0777			-.6297	-.7123	-.7195	-.7165
.240			-.4188				
.260		-.3260					
.280	.0289			-.6755	-.7910		-.8361
.300			-.5392				
.320	.0000			-.5767	-.7728		
.340			.0000				
.360						-.7885	-.7870
.380	-.3267			-.2588	-.2409		
.400						-.6073	-.6244
.420			-.2301				
.440				-.2154	-.2378		
.460		-.0970					
.480			-.1276				
.500	-.0904			-.1671	-.1109	-.3238	
.520							
.540	-.0762			-.1344			-.5805
.560	-.1322			-.1194	-.0315		
.580					.1061		-.4463
.600			-.1201				
.620	-.1620						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(581032)

ARC11-716 1A14 01+T12+S12R25+AT10 UPPER WING

ALPHA( 9 ) = 5.970 BETA( 7 ) = 2.030

SECTION: ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W	2990	3640	4270	5340	6730	7800	8870
1/CW							
.000	-.3096	-.3400	.2361	.5615	.5511	.4990	.4624
.020				-.0200	.0100	-.1331	-.2456
.025			-.0172				
.040		-.0514					
.045			-.1097				
.050	-.3222			-.2215	-.2602	-.3010	-.4550
.060				-.3543			
.085			-.2275				
.086		.0132					
.094	-.3099			-.5100	-.5716	-.5742	-.6049
.150		.0000					
.163			-.3174				
.177		-.0795					
.193							
.229	-.0204			-.5601	-.6801	-.7103	-.7173
.230			-.3071				
.274		-.2902					
.339							
.362	.0042			-.5000	-.7406		-.8413
.400			-.4765				
.402							
.497	.0000			-.3362	-.7190		
.530			.0000				
.565							-.7896
.600					-.8085		
.650	-.1907			-.2340			
.700				-.2350		-.6142	-.6132
.725			-.2246				
.750				-.1802	-.1969		
.760							
.775			-.1484				
.800							
.834	-.1333			-.1979	-.1292	-.4145	
.850			-.1772				
.857							
.865	-.1189						-.5447
.900	-.1051			-.1405		-.0156	
.905			-.1725				
.950				-.1144			-.0835
.955			-.1707				
.965	-.1936						



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2695

ARC11-716 1A14 03+112+512N25+AT10 UPPER WING

(R01032)

ALPHA(9) = 5.950 BETA(8) = 4.080

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP				
Y/D	.2990	.3640	.4270	.5340	.6730 .7800 .8870
X/CW					
.000	-.4301	-.1891	.0941	.4897	.5189 .4673 .4453
.020				-.0292	-.0098 -.1378 -.2362
.025			-.0648		
.040		-.0856			
.045			-.1146		
.050	-.3674		-.2018	-.2711	-.3035 -.4393
.060			-.3329		
.085			-.1792		
.086		-.0403			
.094	-.3180				
.100		.0000	-.4870	-.5559	-.5641 -.5899
.163			-.2484		
.177		-.0771			
.193					
.229	-.0031		-.5190	-.6517	-.6934 -.7086
.250			-.3171		
.274		-.2523			
.339					
.382	.0013		-.5667	-.6898	-.8235
.400			-.4271		
.402	.0000		-.2967	-.8520	
.497	.0000				
.550					
.600					
.650					
.700	-.1009			-.2317	-.7830
.725			-.2448		
.750					
.760			-.2275	-.2193	-.2102
.775			-.1885		
.808					
.834	-.1898		-.2097	-.1405	-.0440
.850			-.2129		
.857					
.885	-.1388				
.900	-.2128		-.2068		-.4489
.905			-.2036	-.0367	
.930			-.1682		.0470
.933			-.1891		
.955	-.2095				

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(491J32)

ARC11-716 1A14 01-712-512N23-AT10 UPPER WING

ALPHA=91 3 5.940 BETAD (9) = 6.100

SECTION: (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
Y/CW							
.000	-.9893	-.1092	.0230	.2967	.4863	.4369	.4239
.020				-.1029	-.0111	-.1345	-.2290
.025				-.0399			
.040		-.0622					
.045			-.0920				
.050	-.4335			-.2200	-.2678	-.2981	-.4305
.060				-.3202			
.085			-.1339				
.086		-.0579					
.094	-.3180			-.4077	-.5362	-.5385	-.5719
.150		.0090					
.165			-.2149				
.177							
.195		-.1080					
.229	.0145			-.4795	-.5888	-.6680	-.6913
.250			-.2813				
.274		-.2584					
.339							
.362	.0568			-.4595	-.6334		-.8008
.400		-.3403					
.402							
.497	.0000			-.8296	-.5536		
.550		.0000					
.565							-.7499
.650						-.5505	
.700	-.0755			-.2536	-.2736		
.725						-.2092	-.5517
.750		-.2361		-.2411	-.2232		
.760							
.775		-.2219					
.808							
.834	-.8127			-.2398	-.1589	-.0670	
.850		-.2531					
.897							
.865	-.1731			-.2136			-.3770
.900	-.2189				-.0593		
.905		-.2121		-.1025		.0213	
.950			-.1912				
.955							
.965	-.1971						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING (R81U32)

ALPHA( 9 ) = 5.920 BETAD (10) = 8.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/SM	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.6572	-.0897	-.0171	.0874	.4114	.4233	.3936
.020				-.1900	-.0415	-.1267	-.2374
.025			-.0504				
.040		-.0450					
.045			-.0768				
.050	-.3161			-.2397	-.2688	-.2909	-.4284
.080				-.2948			
.085			-.1182				
.086		-.0377					
.094	-.3926			-.3339	-.4876	-.5149	-.5600
.150		.0000					
.163							
.177			-.1995				
.193		-.1063					
.229	.0179			-.3543	-.5143	-.6297	-.6698
.250			-.2598				
.274		-.2219					
.339							
.362	.0099			-.2476	-.4827		-.7847
.400			-.2326				
.402							
.497	.0000			-.1979	-.3026		
.550			.0000				
.565						-.3691	-.7011
.600							
.650				-.2442	-.2423		
.700	-.0613						
.725						-.2689	-.4741
.750			-.2391	-.2504	-.2597		
.775							
.808			-.2316				
.834	-.2263			-.2363	-.2109	-.1169	
.850			-.2429				
.857							
.865	-.1939			-.2124			.0296
.900	-.2390						
.905			-.2116	-.1111			
.950				-.1816		-.0222	
.953			-.1882				
.965	-.1996						



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI4A - VOL. 6

PAGE 2698

(R81032)

ARC11-716 IAI4 01+T12+S12N23+AT10 UPPER WING

ALPHA(9) = 5.900 BETA(11) = 10.150

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.000	-.6256	-.1090
.020		-.0376
.027		.2273
.031		.3016
.035		.3827
.040		-.2036
.045		-.1160
.050		-.1485
.055		-.2231
.060		-.0795
.065		-.0833
.070		-.0960
.075		-.2107
.080		-.2242
.085		-.2923
.090		-.4075
.095		-.2262
.100		-.1175
.105		
.110		-.0707
.115		-.086
.120		-.4723
.125		
.130		-.2304
.135		-.4247
.140		-.4899
.145		-.5397
.150		.0000
.155		-.2021
.160		
.165		-.1048
.170		-.0316
.175		-.229
.180		.250
.185		.274
.190		.339
.195		-.1686
.200		.362
.205		.403
.210		.402
.215		.497
.220		.530
.225		.565
.230		.600
.235		.650
.240		.700
.245		.729
.250		.750
.255		.760
.260		.775
.265		.808
.270		.834
.275		.850
.280		.857
.285		.865
.290		-.2150
.295		-.2690
.300		-.2021
.305		-.2616
.310		-.1430
.315		-.0478
.320		-.2267
.325		-.1701
.330		-.965
.335		-.1979



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6  
ARC11-716 IA14 01+712+S12N25+AT10 UPPER WING (R81032)

ALPHA(10) = 0.080 BETA( 1 ) = -9.950

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
X/CW	Y/BW		
.000	.2990	.3640	.4270
.000		.5340	.6730
.000		.7800	.8870
.000	.1505	.0235	.5589
.000		.8073	.7318
.000		.6348	.5801
.000	.020	-.0741	.0618
.000		-.2241	-.3230
.000	.025	.0115	
.000		.0774	
.000	.040		
.000	.045	-.1499	
.000	.050	-.3002	-.3588
.000		-.4294	-.5412
.000	.060	-.4750	
.000	.065	-.3429	
.000	.066	.0635	
.000	.094	-.0492	
.000	.150		
.000	.163	-.7184	-.6974
.000		-.6924	-.6553
.000	.177	-.5475	
.000	.193	-.2013	
.000	.229	-.0146	
.000	.250		
.000	.274	-.8319	-.8101
.000		-.6559	-.6224
.000	.339	-.6070	
.000	.362	-.4635	
.000	.362	.0104	
.000	.400	-.8027	-.7359
.000		-.6303	
.000	.432	-.6900	
.000	.497		
.000	.350	-.7171	-.6831
.000	.365	.0000	
.000	.630		
.000	.650		
.000	.700	-.5723	-.2718
.000	.725		
.000	.750	-.5587	-.6242
.000	.760		
.000	.775	-.0955	
.000	.808		
.000	.834	-.4861	-.6003
.000	.850		
.000	.857	-.0489	
.000	.865		
.000	.900	-.4137	-.5755
.000	.935	-.0200	
.000	.950	-.3322	-.4907
.000	.953	-.0089	
.000	.965	-.0360	

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(R81032)

TABULATED PRESSURE DATA - (A14A - VOL. 6

ARC11-716 (A14 Q1+T12+T12N25+AT10 UPPER WING

DATE 10 JAN 75

ALPHAO(10) = 0.130 BETA0 ( 3 ) = -5.940

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE C<sub>P</sub>

Y/SW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0103	-.1739	.4126	.7153	.6642	.5643	.5021
.020				-.1379	-.0205	-.2749	-.3868
.025				-.0704			
.040				-.1032			
.045				-.2043			
.050	-.1316			-.3429	-.3888	-.4703	-.5889
.060				-.4999			
.085				-.3765			
.086				-.0737			
.094	-.1472						
.150				-.7262	-.7181	-.7140	-.7188
.163				.0000			
.177				-.3363			
.193				-.2120			
.229	-.1431						
.250				-.8051	-.8189	-.6381	-.7094
.274				-.5777			
.339				-.4062			
.362	-.0186						
.400				-.7922	-.7790		-.7175
.422				-.6037			
.497	.0000			-.6879	-.7124		
.550				.0000			
.565							
.600							
.650							
.700	-.5110						
.725				-.4715	-.2637		
.750							
.760				-.7462	-.4052	-.5605	-.6051
.775				-.1599	-.6273		
.806							
.834	-.0168						
.850				-.2646	-.6011	-.5051	
.857				-.1101			
.865	.0328						
.900	-.0292			-.1723			
.935				-.0796	-.532		
.950				-.1193			
.965				-.0551			
.965	-.0849						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(N81032)

ARC11-716 IA14 01-112-312N29-AT10 UPPER WING

ALPHA(10) = 7.980 BETA(4) = -3.970

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0622	-.2502	.3810	.6881	.6541	.5544	.4848
.020				-.1110	-.0172	-.2508	-.3747
.025			-.0561				
.040		-.1467					
.045			-.1762				
.050	-.1714			-.3084	-.3634	-.447	-.5898
.080				-.4631			
.085			-.3295				
.086		-.1015					
.094	-.1896			-.6718	-.6803	-.6894	-.7123
.150		.0000					
.163			-.4172				
.177		-.1715					
.193	-.1799						
.229				-.7666	-.8059	-.7526	-.7626
.250			-.4842				
.274		-.3600					
.339							
.362	.0207			-.7829	-.8170		-.7687
.400			-.8263				
.402							
.497	.0000			-.7205	-.7537		
.550			.0000				
.565							
.600							
.650						-.6663	-.6700
.700	-.5113			-.3508	-.2608		
.725						-.5866	-.6301
.750			-.2418				
.760				-.2997	-.6271		
.775			-.1197				
.800							
.834	-.0408			-.2228	-.5904	-.4985	
.850			-.0952				
.857							
.865	-.0015						
.900	.0006			-.1788			-.6298
.905			-.0692		-.4987		
.950				-.1469		-.4695	
.953			-.0695				
.965	-.1168						



(R81032)

ARC11-716 IAI4 Q1-T12-S12N25+AT10 UPPER WING

DATE 10 JAN 75

ALPHA(10) = 0.010 BETA(5) = -1.970

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	-.1442	-.3662	.3920	.6475	.6235	.5232	.4573
.020				-.0919	-.0073	-.2591	-.3897
.025			-.0357				
.040		-.2342					
.045			-.1532				
.050	-.2261			-.2864	-.3592	-.4468	-.5797
.080				-.4365			
.085			-.3009				
.086		-.0573					
.094	-.2343						
.150		.0000		-.8507	-.6731	-.6898	-.7182
.163							
.177			-.3983				
.193		-.1413					
.229	-.1620						
.250				-.6943	-.7980	-.7973	-.7694
.274			-.4750				
.339		-.3594					
.362	.0257						
.400		-.5869		-.7396	-.8301		-.7785
.402							
.497	.0000			-.6619	-.7626		
.550			.0000				
.565							
.600							-.6900
.630					-.6702		
.700	-.5132			-.2972	-.2617		
.725							
.750						-.5939	-.6508
.760			-.2286				
.775				-.2480	-.6103		
.904			-.1037				
.930		-.0694					
.950				-.1800	-.5480	-.4883	
.957			-.0929				
.958	-.0308						
.959	-.0412			-.1375	-.3925		-.6261
.995			-.0770				
.998				-.1116			-.4779
.999			1.0027				
.985	-.1353						

DATE 15 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R01U32)

ARC11-716 1A14 01+712+512M25+AT10 UPPER WING

ALPHA(10) = 7.930 BETA (6) = .060

SECTION / UPPER WING	DEPENDENT VARIABLE CP
Y/Z	
.000	.5340 .8730 .7800 .8870
.020	.5919 .9893 .5037 .4412
.040	.5917 .0116 .8462 .3719
.060	.0316
.080	.1815
.100	.2805 .3537 .4342 .5600
.120	.4278
.140	.2997
.160	.0095
.180	.2731
.200	.0000
.220	.3836
.240	.1263
.260	.1169
.280	.6980 .7831 .7960 .7957
.300	.4620
.320	.3552
.340	.0130
.360	.7307 .8457 .8213
.380	.9482
.400	.0000
.420	.3900 .8185
.440	.7465
.460	.7272
.480	.2588
.500	.3037
.520	.6240 .7065
.540	.2442
.560	.2411 .5922
.580	.1192
.600	.8039 .5157 .5392
.620	.1412
.640	.0909
.660	.1724
.680	.0726
.700	.3057
.720	.1405
.740	.1120
.760	.5372
.780	.1336
.800	.1399
.820	.1717



(RBU32)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+512R25+AT10 UPPER WING

DATE 10 JAN 75

ALPHAD(10) = 7.979 BETAD (7) = 2.050

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/W	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.3107	-.4063	.1623	.5369 .5505 .4818 .4209
.020				-.1199 .0777 -.2376 -.3742
.025			-.0933	
.040		-.1074		
.045			.1776	
.050	-.3159		-.2881	-.3403 -.4234 -.5660
.060			-.4286	
.065		-.2825		
.086	-.0280			
.094	-.3045		-.6041	-.6416 -.6610 -.6995
.150	.0000			
.163		-.3646		
.177				
.193	-.1130			
.229	-.0325		-.6596	-.7579 -.7809 -.7912
.250			-.4240	
.274		-.3297		
.339	-.0134			
.362			-.6559	-.8298 -.8494
.400				
.432		-.5082		
.497	.0000		-.5059	-.8098
.550		.0000		
.565				
.600				
.650				
.700	-.2153		-.2935	-.7801
.725				
.750				
.760		-.2412		
.775		-.2177	-.2355	
.808		-.1734		
.834	-.1495		-.1916	-.1633 -.5666
.850		-.1921		
.857				
.865	-.1406		-.1816	
.900	-.1490		-.1874	-.6892
.905		-.1876	-.0874	
.930		-.1495		
.953		-.1802		-.5059
.955	-.2056			



(R81032)

DATE 10 JAN 79 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA(10) = 7.950 BETA( 0) = 4.080

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.000	-.4516	-.2220	-.0228	.4105 .5152 .4448 .3924
.020			-.1837	-.0194 -.2504 -.3770
.025		-.1435		
.040		-.1253		
.045		-.1863		
.050	-.3727		-.3211	-.3518 -.4315 -.5680
.060		-.2253	-.4380	
.065		-.0858		
.066		-.3425		-.5637 -.6413 -.6572 -.6927
.094		.0500	-.2875	
.150		-.1264		
.163				
.177				
.193	-.0224		-.5826	-.7455 -.7723 -.7850
.229		-.3534		
.250		-.3031		
.274				
.339	-.0219		-.6201	-.7838 -.8369
.362		-.4585		
.400			-.3678	-.7067
.402		.0000		
.497	.0000			
.550				
.565				
.600				
.650	-.1377		-.2835	-.2430 -.7821
.700				
.725		-.2617	-.2409	-.2422
.750		-.2294		
.760				
.775				
.808				
.834	-.2116		-.2354	-.1793 -.4543
.850		-.2432		
.857				
.885	-.1904		-.2224	
.900	-.1912		-.2287	-.0778
.925			-.1865	-.1102
.950				
.955		-.2034		
.965	-.2190			

(RB1U32)

TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+112+512M25+AT110 UPPER WING

DATE 10 JAN 75

ALPHA(10) = 7.920 BETA( 9) = 6.110

SECTION ( 1) UPPER WING

DEPENDENT VARIABLE CP

X/CW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.5882	-.1626	-.0600	.1767	.4891	.4124	.3717
.020				-.2726	-.0496	-.2519	-.3770
.025				-.1156			
.040		-.1241					
.045			-.1412				
.050				-.3442	-.3610	-.4279	-.5595
.080	-.4511			-.4155			
.085				-.1763			
.086		-.0944					
.094	-.3670			-.4632	-.6266	-.6459	-.6791
.150			.0000				
.165				-.2597			
.177							
.193		-.1582					
.229	-.0330			-.5292	-.7073	-.7558	-.731
.274			-.3189				
.339		-.3073					
.362	-.0172			-.4694	-.7191		-.8331
.400							
.402			-.3658				
.497	.0000			-.2847	-.4411		
.550			.0000				
.565							
.600							-.7950
.650					-.2446		
.700	-.1076			-.2810			
.725						-.2391	-.5767
.750							
.760			-.2697	-.2628	-.2508		
.775				-.2518			
.806							
.834	-.2299			-.2504	-.1629	-.0562	
.850							
.857			-.2483				
.865	-.1939						-.5283
.900	-.2010			-.2229		-.0571	
.905			-.2196				
.950				-.1915			.0076
.955			-.1676				
.965	.1987						

DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14 - VOL. 6  
ARC11-716 IAI14 OL+T12+S12M25+AT10 UPPER WING (881032)

ALPHAC(110) = 7.910 BETA(110) = 0.160

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.6655	-.1151	-.0081	-.0073	.3975	.3886	.3561
.020				-.3151	-.0784	-.2504	-.3718
.025			-.0992				
.040		-.0808					
.045			-.1223				
.050	-.5598			-.3324	-.3610	-.4203	-.5549
.080				-.3694			
.085			-.1569				
.086		-.0720					
.094	-.4720			-.3875	-.5852	-.6222	-.6697
.150							
.163		.0000					
.177		-.1690	-.2369				
.193							
.229	-.0347			-.4414	-.6057	-.7270	-.7608
.250			-.2890				
.274		-.2717					
.339							
.362	-.0066			-.3275	-.5241		-.8554
.400							
.402		-.2641					
.497	.0000			-.2779	-.3438		
.550			.0000				
.565							
.600							
.650							
.700	-.0833			-.3159	-.2488	-.3980	-.7742
.725							
.750			-.2844	-.3192	-.2812		
.760							
.775			-.2649				
.808							
.834	-.2420			-.2949	-.2047	-.1092	
.850			-.2590				
.857							
.865	-.2123			-.2675			-.3757
.900	-.2213						
.905			-.2274	-.1040			
.950			-.2420				
.953			-.1945				
.965	-.2072						







DATE 10 JAN 75  
 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-T12-S12N23+AT10 UPPER WING (R81U32)

ALPHA(11) = 9.930 BETA(2) = -7.950

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .5730 .7600 .8670
X/CW				
.000	.0710	-.0324	.0231	.7969 .6930 .5588 .4842
.020			-.1368	-.1183 -.3434 -.4558
.025			-.0541	
.040		-.0107		
.045			-.2176	
.050	-.0756		-.3604	-.4430 -.3322 -.6454
.060			-.5299	
.085			-.4113	
.086		-.0079		
.094	-.0860		-.7368	-.7584 -.7580 -.7573
.150		.0000		
.163		-.0135		
.177		-.2469		
.193				
.229	-.0692		-.0224	-.7572 -.7160 -.7486
.250			-.6626	
.274		-.4932		
.339				
.362	-.0203		-.0280	-.7584 -.7303
.400			-.7000	
.402			.0000	
.497	.0000		-.7527	-.7102
.550				
.565				
.600				
.650				
.700	-.5278		-.6408	-.2741
.725				
.750				
.760				
.775				
.808				
.834	-.0175		-.6153	-.6325
.890			-.2015	
.897				
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1.000				

(R81032)

ARC11-716 1A14 01+T12+S12N29+AT10 UPPER WING

ALPHAC(11) = 9.980 BETAD (3) = -5.920

SECTION 1 UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7900	.8870
X/CW							
.000	.0098	-.1681	.4231	.7485	.6618	.5267	.4420
.020				-.1816	-.1468	-.3674	-.4904
.025							
.040							
.045							
.050							
.060							
.065							
.066							
.094							
.150							
.163							
.177							
.193							
.229							
.250							
.274							
.339							
.362							
.400							
.402							
.497							
.550							
.565							
.600							
.650							
.700							
.725							
.750							
.760							
.775							
.808							
.834							
.850							
.857							
.861							
.900							
.905							
.910							
.953							
.965							



(R81032)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA(III) = 9.960 BETA(4) = -3.970

SECTION (UPPER WING) DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.0697 -.2606 .3373 .7093 .6415 .5090 .4210

.020 .020 .020 .020 .020 .020 .020 .020

.025 .025 .025 .025 .025 .025 .025 .025

.040 .040 .040 .040 .040 .040 .040 .040

.045 .045 .045 .045 .045 .045 .045 .045

.050 .050 .050 .050 .050 .050 .050 .050

.060 .060 .060 .060 .060 .060 .060 .060

.065 .065 .065 .065 .065 .065 .065 .065

.086 .086 .086 .086 .086 .086 .086 .086

.094 .094 .094 .094 .094 .094 .094 .094

.150 .150 .150 .150 .150 .150 .150 .150

.163 .163 .163 .163 .163 .163 .163 .163

.177 .177 .177 .177 .177 .177 .177 .177

.193 .193 .193 .193 .193 .193 .193 .193

.229 .229 .229 .229 .229 .229 .229 .229

.250 .250 .250 .250 .250 .250 .250 .250

.274 .274 .274 .274 .274 .274 .274 .274

.339 .339 .339 .339 .339 .339 .339 .339

.362 .362 .362 .362 .362 .362 .362 .362

.400 .400 .400 .400 .400 .400 .400 .400

.432 .432 .432 .432 .432 .432 .432 .432

.497 .497 .497 .497 .497 .497 .497 .497

.550 .550 .550 .550 .550 .550 .550 .550

.565 .565 .565 .565 .565 .565 .565 .565

.600 .600 .600 .600 .600 .600 .600 .600

.650 .650 .650 .650 .650 .650 .650 .650

.700 .700 .700 .700 .700 .700 .700 .700

.725 .725 .725 .725 .725 .725 .725 .725

.750 .750 .750 .750 .750 .750 .750 .750

.775 .775 .775 .775 .775 .775 .775 .775

.806 .806 .806 .806 .806 .806 .806 .806

.834 .834 .834 .834 .834 .834 .834 .834

.857 .857 .857 .857 .857 .857 .857 .857

.865 .865 .865 .865 .865 .865 .865 .865

.870 .870 .870 .870 .870 .870 .870 .870

.875 .875 .875 .875 .875 .875 .875 .875

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.875 .875 .875 .875 .875 .875 .875 .875



(R81032)

ARC11-716 IAI4 01+T12+312N25+AT10 UPPER WING

ALPHA(11) = 9.950 BETA( 5) = -1.970

SECTION 1 (UPPER WING)		DEPENDENT VARIABLE CP			
Y/BW	X/CW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
.000	-.1986	-.4215	.2840	.6593	.6185 .4940 .4022
.020				-.1637	-.1366 -.3486 -.4931
.025				-.1115	
.040				-.3023	
.045				-.2217	
.050	-.2394			-.3560	-.4314 -.5313 -.6677
.060				-.5079	
.065				-.3596	
.066				-.2137	
.094	-.2361				-.7116 -.7317 -.7518 -.7930
.150				.0000	
.163				-.4420	
.177				-.1697	
.193					
.229	-.2186			-.5122	
.250				-.7678	-.7968 -.7730 -.8143
.274					
.339				-.7747	-.7948
.362	.0166			-.6110	
.400				-.6919	-.7192
.432				.0000	
.497	.0000				-.6449
.550					-.5201
.563					
.600					
.650					
.700	-.5437			-.4479	-.2573
.725					-.5598
.750					-.6174
.760				-.2660	
.775				-.3831	-.3666
.808				-.1482	
.834	-.3462			-.2331	-.5605
.850					-.5420
.857				-.1749	
.865	-.0307				
.900	-.0320			-.1668	-.6206
.905				-.1325	-.5244
.950				-.1272	-.5127
.953				-.1158	
.965	-.1674				



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2715

ARC11-716 1A14 01+T12+512M25+AT10 UPPER WING

(RB1032)

A\_LMAC(11) = 9.950 RETAO ( 6 ) = .030

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/OW	X/CW			
.000	-.2351	-.5973	.1805	.5938
.020			.5901	.4753
.025		-.1444	-.1775	-.1416
.040		-.3687		-.3454
.045			-.2452	-.4922
.050	-.2781		-.3510	-.4277
.060			-.4955	-.5273
.085		-.3498		-.6638
.086		-.0597		
.094	-.2670			
.150		.0920	-.6900	-.7243
.163			-.7475	-.7867
.177		-.4263		
.193		-.1193		
.229	-.1882		-.7576	-.8317
.250			-.7964	-.9245
.274		-.4961		
.339		-.3689		
.362	-.0044		-.7421	-.8065
.400				-.7396
.402		-.5895		
.497	.0000		-.6732	-.7498
.550		.0000		
.565				-.6628
.600			-.6382	
.650			-.2574	
.700	-.4989		-.3415	-.5377
.725				-.6305
.750		-.2417		
.765			-.2816	-.5029
.775		-.1414		
.808				
.834	-.1053		-.2079	-.5562
.850		-.1570		-.4955
.851				
.865	-.0819		-.1602	-.6315
.900	-.0907			
.905		-.1635	-.4553	
.950			-.1278	-.4871
.953		-.1511		
.965	-.1488			

(R81032)

A011-716 1A14 01+12+312N25+AT10 UPPER WING

ALPHA(11) = 9.950 BETAO (7) = 2.060

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.8990	.9040	.4270	.9340 .6750 .7800 .0070
X/CW				
.308	-.1167	-.5149	.0873	.5175 .9595 .4592 .3693
.020				-.2746 -.1392 -.3327 -.4678
.025			-.1943	
.040		-.2192		
.045			-.2493	
.070	-.3139			-.3626 -.4164 -.5090 -.6534
.080				-.4964
.095			-.3482	
.096		-.1113		
.094	-.3010			
.150		.0000		-.6769 -.7321 -.7502 -.7763
.163			-.3551	
.177		.1193		
.193	-.0683			-.1317 -.8126 -.8391 -.8412
.229			-.4474	
.274		-.3586		
.339	-.0373			-.7231 -.8212 -.8186
.362			-.9390	
.400				-.5586 -.7886
.402		.0000		
.497			.0000	
.530				-.6926
.568				-.2509
.600				-.3364
.650				-.5427 -.6569
.700	-.2650			
.725				-.2663 -.5209
.750			-.0938	
.760				-.1961
.775				
.808			-.2079	
.834	-.1697			-.2331 -.3864 -.4584
.850				
.857				-.2094
.865	-.1533			-.1766
.900	-.1538			-.1882
.915			-.2026	
.950				-.4807
.955			-.1874	
.965	-.2014			



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI44 - VOL. 6  
ARC11-116 IAI4 OR112+112N2+AT13 UPPER WING (R21J22)

ALPHAC111 7 9.985 BETAC (0) 4.080

SECTION 1 UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4290	.5340	.6390	.7800	.8870
X/CW							
.000	-.4314	-.2584	-.0692	.8302	.5115	.4312	.3501
.020				-.2569	-.1936	.3381	-.4801
.025			-.1004				
.040		-.1690					
.045			-.2205				
.050	-.3754			-.3895	-.4204	-.3120	-.6587
.080				-.4016			
.085			-.2547				
.086		-.1207					
.094	-.3478						
.150		.0000		-.6303	-.6283	-.7893	-.7781
.163			-.3225				
.177		-.1763					
.193			-.3727				
.229	-.0429			-.5945	-.8237	-.4303	-.6401
.250							
.274		-.3255					
.339				-.6073	-.8164		-.8345
.362	-.0482						
.402			-.4712				
.497	.0000			-.3517	-.8064		
.550			.0000				
.565						-.7372	
.600						-.8980	
.650				-.2575			
.700	-.1645			-.1738			
.725							
.750			-.2720				
.75				-.8287	-.3025		
.806			-.2473				
.814	-.2191			-.2134	-.2124	-.4894	
.840			-.2493				
.857							
.865	-.1338			-.1979			-.6740
.900	-.194		-.2350	-.1397			
.925				-.1707	-.5487		
.932			-.2036				
.933		-.2202					
.945							

(881032)

ARC11-716 1A14 (A+712+512M25+AT10 UPPER WING

ALPHAD(11) = 10.000 BETAC ( 8 ) = 6.140

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.5930	-.2029	-.1177	.2095	.4806	.4119	.3332
.020				-.3193	-.1621	-.3319	-.4897
.040				-.1374			
.060				-.1409			
.080				-.1804			
.100				-.4197	-.4241	-.5069	-.6357
.120				-.4931			
.140				-.2079			
.160				-.1108			
.180				-.4071			
.200					-.5529	-.0864	-.7199
.220				.0000			-.7705
.240				-.1975			
.260				-.2813			
.280				-.1975			
.300				-.0263			
.320				-.5608	-.7824	-.8228	-.8145
.340				-.3348			
.360				-.3354			
.380					-.5194	-.8053	-.8440
.400					-.3479	-.4091	
.420							
.440							
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
1.000							



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI14 - VOL. 6

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ARC11-716 IAI14 01+712+512N25+AT10 UPPER WING

(R81032)

ALPHA(11) = 10.030 BETA(10) = 0.160

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.6503	-.1485	-.1486	-.0102	.4509	.3952	.3300
.020				-.4342	-.1007	-.3189	-.4779
.025			-.1293				
.040		-.1059					
.045			-.1511				
.050	-.5666			-.4155	-.4300	-.4876	-.6390
.060				-.4395			
.085			-.1791				
.086		-.1034					
.094	-.9078						
.150		.0000		-.4224	-.6675	-.6963	-.7502
.163							
.177			-.2539				
.193		-.2080					
.229	-.0976						
.250				-.4660	-.7238	-.7962	-.8231
.274			-.2994				
.339		-.3050					
.362	-.0336						
.400				-.3411	-.6430		-.8471
.402			-.2987				
.497	.0000			-.2872	-.3998		
.550							
.565							
.600			.0000				
.650						-.7687	
.650				-.6339			
.700	-.1075			-.2561			
.725							
.750				-.3165		-.2376	-.6758
.760			-.2834				
.775				-.3208	-.5003		
.808			-.2783				
.834	-.2505						
.850			-.2667	-.2936	-.2073	-.0964	
.857							
.865	-.2169						-.6490
.900	-.2154			-.2616			
.905			-.2305	-.1160			
.950				-.2289		-.0322	
.955			-.1996				
.965	-.2030						

DO NOT WRITE IN THESE SPACES

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81032)

ARC11-716 1A14 01+112+512N25+AT10 UPPER WING

ALPHAO(11) = 10.970 BETA0 (11) = 10.230

SECTION (1) UPPER WING DEPENDENT VARIABLE CF

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.006	-.6223	-.1387	-.1370	-.0831	.3924	.3966	.3249
.020				-.4240	-.1705	-.2974	-.4524
.025			-.1319				
.040		-.1218					
.045			-.1560				
.050	-.5176			-.3315	-.4374	-.4681	-.6160
.060				-.3292			
.065			-.1811				
.086		-.1192					
.094	-.4642			-.2942	-.5303	-.6747	-.7261
.150			.0000				
.163			-.2328				
.177		-.2169					
.193							
.229	-.1925			-.2689	-.6168	-.7671	-.8038
.250			-.2753				
.274		-.2513					
.339	-.0379			-.2282	-.4443		-.8586
.362			-.2751				
.400				-.2324	-.3524		
.402			.0000				
.497	.0000						-.7707
.550				-.2583		-.4116	
.563							
.600				-.2713		-.2466	-.7082
.650	-.0999						
.700			-.2976	-.2898	-.2923		
.725			-.2718				
.750							
.760							
.775							
.800							
.834	-.2531			-.2499	-.2045	-.0997	
.850			-.2319				
.857							
.865	-.2277			-.2237			-.3794
.900	-.2167		-.2129	-.1445			
.905				-.1996		-.0802	
.950			-.1875				
.953							
.965	-.1933						



(RB1U33) ( 17 APR 74 )

### PARAMETRIC DATA

MACH	=	1.250	ELEVON	=	.000
RUDDER	=	.000	SPDRK	=	.000

## REFERENCE DATA

300°F = 2.4810 INCHES  
 300°F = 30.7000 INCHES  
 300°F = 30.7000 INCHES  
 300°F = 30.7000 INCHES

$$A_{\text{C}}(1) = -10.349 \quad B_{\text{C}}(1) = -9.910$$

DEPENDENT VARIABLE CP

[illegible]



(R81US3)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA( 1 ) = -10.340 BETA( 1 ) = -9.910

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	Y	Z	CP
.950	.2990	.3640	.4270
.953		.3244	.5340
.955		.3210	.6730
			.7800
			.8870

ALPHA( 1 ) = -10.260 BETA( 2 ) = -7.920

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	Y	Z	CP
.950	.2990	.3640	.4270
.953		.3244	.5340
.955		.3210	.6730
			.7800
			.8870

X/CW	Y	Z	CP
.950	.2990	.3640	.4270
.953		.3244	.5340
.955		.3210	.6730
			.7800
			.8870

ALPHA( 1 ) = -10.260 BETA( 2 ) = -7.920

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	Y	Z	CP
.950	.2990	.3640	.4270
.953		.3244	.5340
.955		.3210	.6730
			.7800
			.8870

ALPHA( 1 ) = -10.260 BETA( 2 ) = -7.920

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	Y	Z	CP
.950	.2990	.3640	.4270
.953		.3244	.5340
.955		.3210	.6730
			.7800
			.8870

ALPHA( 1 ) = -10.260 BETA( 2 ) = -7.920

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	Y	Z	CP
.950	.2990	.3640	.4270
.953		.3244	.5340
.955		.3210	.6730
			.7800
			.8870

ALPHA( 1 ) = -10.260 BETA( 2 ) = -7.920

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	Y	Z	CP
.950	.2990	.3640	.4270
.953		.3244	.5340
.955		.3210	.6730
			.7800
			.8870

ALPHA( 1 ) = -10.260 BETA( 2 ) = -7.920

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	Y	Z	CP
.950	.2990	.3640	.4270
.953		.3244	.5340
.955		.3210	.6730
			.7800
			.8870

DATE 13 JAN 75  
 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-112-S12M25+AT10 UPPER WING (R51033)

ALPHA( 1 ) = -10.280 BETAD ( 2 ) = -7.920

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	.2990 .3640 .4270 .5140 .6730 .7800 .8870
X/CW	.2622
	.857
	.855
	.2681
	.900
	.2796
	.905
	.3148
	.3067
	.2906
	.3104
	.2905
	.953
	.3048
	.965
	.2774

ALPHA( 1 ) = -10.250 BETAD ( 3 ) = -5.920

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	.2990 .3640 .4270 .5340 .6730 .7800 .8870
X/CW	.2622
	.857
	.855
	.2681
	.900
	.2796
	.905
	.3148
	.3067
	.2906
	.3104
	.2905
	.953
	.3048
	.965
	.2774

(R01033)

ARC11-716 1A14 01+712+512M25+AT10 UPPER WING

ALPHA (1) = -10.250 BETA (3) = -5.920

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/B	.2990	.3540	.4270	.5340	.6730	.7800	.8870
Z/CW							
.775				.1358	.1171		
.808			.1786				
.834	.1147						
.850			.1533	.1461	.1631		
.857		.1996					
.865	.2307						
.900	.2291		.2247				.1644
.905		.2713	.2524				
.950		.2699			.2305		
.953		.2698					
.965	.2384						

ALPHA (1) = -10.240 BETA (4) = -5.960

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/B	.2990	.3540	.4270	.5340	.6730	.7800	.8870
Z/CW							
.775							
.808			.1447	.4086	.3234	.2373	.2746
.820				.6386	.5747	.5990	.5962
.825		.3961					
.840							
.845		.3707					
.850			.4734	.4744	.4782	.4691	
.860		.3004	.3275				
.865							
.886		.1160					
.894	.0080						
.150				.1176	.1568	.1939	.1992
.163		.0000					
.177		.1590					
.193		.1888					
.229	.0204						
.250							
.274		.0367					
.339							
.362	.1455	.0294					
.400							
.402							
.497	.0000						
.550							
.565							
.800							
.850							



DATE 10 JAN 75  
 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARCL1-716 1A14 01+T12+S12N23+AT10 UPPER WING  
 (RB1033)

ALPHA(1) = -10.240 BETA(4) = -3.960

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.700	.2193			.0838			
.725						.0749	-.0275
.750							
.760			.0652				
.775				.1148	.0945		
.808			.1445				
.834	.1001						
.850				.1177	.1059	.1346	
.857			.1590				
.865	.2015						
.900	.1935		.1506	.1943			.2207
.905			.2110				
.950			.2246			.2254	
.953			.2308				
.965	.2004						

ALPHA(1) = -10.250 BETA(5) = -1.970

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.900	-.1258	-.2645	.0905	.3519	.2610	.1804	.2217
.920				.6127	.5359	.5664	.5615
.925			.3514				
.940		-.0752					
.945			.3337				
.950	-.0662			.4613	.4543	.4621	.4495
.960				.3277			
.985			.2876				
.986		.0791					
.994	-.0175						
.150			.0300				
.163				.1275	.1605	.1926	.1943
.177			.1603				
.193		.1749					
.229	-.0012						
.250				-.0464	-.0240	-.0118	-.0399
.274			.0440				
.339		.0316					
.362	.1288						
.400				-.1621	-.1698		-.2650
.402							
.497	.0000		-.1068				

DATE 19 JAN. 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U33)

ARC11-716 IA14 01+712+312M25+AT10 UPPER WING

ALPHAD(1) = -10.250 BETAO(5) = -1.970

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW				.1522	.1030		
.550				.0000			
.565							
.600							
.553							
.700							
.2212							
.725							
.753							
.760							
.775							
.830							
.834							
.850							
.857							
.865							
.900							
.935							
.940							
.953							
.955							
.720							

ALPHAD(1) = -10.160 BETAO(6) = .020

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.550							
.565							
.600							
.553							
.700							
.2212							
.725							
.753							
.760							
.775							
.830							
.834							
.850							
.857							
.865							
.900							
.935							
.940							
.953							
.955							
.720							



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+512+25+AT10 UPPER WING (R81U33)

ALPHA(1) = -10.160 BETA(6) = .020  
 SECTION (1) UPPER WING  
 Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW	DEPENDENT VARIABLE CP
.362	
.400	
.402	
.497	
.530	
.565	
.600	
.630	
.700	
.725	
.750	
.775	
.808	
.834	
.850	
.857	
.865	
.900	
.905	
.950	
.953	
.985	

ALPHA(1) = -10.160 BETA(7) = 2.040  
 SECTION (1) UPPER WING  
 Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW	DEPENDENT VARIABLE CP
.000	
.020	
.025	
.040	
.045	
.050	
.060	
.085	
.086	
.094	
.150	
.163	
.177	
.193	



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U33)

ARC11-716 IA14 Q1+T12+S12N23+AT10 UPPER WING

ALPHA(1) = -10.220 BETA(1) = 4.080

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW				.1154	.1412	.1552	.1805
.150							
.163		.0000					
.177			.1489				
.193		.1651					
.229	-.0237			-.0312	-.0315	-.0214	.0352
.250							
.274			.0454				
.339		.0240					
.362		.1340					
.400				-.0839	.0247		.0170
.402			-.0950				
.497	.0000						
.550			.1370	.1099			
.565			.0000				-.0816
.600					.0019		
.650							
.700	.1991			-.1267			
.725				.0606			
.750					.0252	-.0621	
.760		.0320					
.775			.0821	.0519			
.808		.0552					
.834	.0374			.0598	.0438	.0368	
.850							
.857		.0374					
.865	.0723						.0743
.900	.0742			.0343	.0740		
.905			.1187				
.950			.1248			.0929	
.953		.1653					
.965	.1351						

ALPHA(1) = -10.230 BETA(1) = 6.080

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2378	-.0911	.1126	.1871	.1256	.0180	.0304
.020				.4561	.4430	.4505	.4826
.025			.2647				
.040		-.0318					
.045			.2450				
.050	-.1130			.3337	.4054	.3868	.4270



(R81033)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1114 01-712-512M25-AT110 UPPER WING

ALPHAO(1) = -10.250 BETAO(9) = 6.080

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2590	.3640	.4270	.5340	.6730	.7800	.8870
1/CM							
.080				.2321			
.245			.2067				
.586		.0675					
.594	-.0029			.1171	.1716	.1781	.2795
.150		.0000					
.163			.1271				
.177		.1548					
.193							
.229	-.0200			-.0011	.0009	.2311	.1041
.250			.0292				
.274		.0030					
.339							
.362	.1317			.2184	.1770		-.0570
.400			-.0731				
.492	.0000			.1547	.0737		
.550			.0000				
.565							
.600							
.630							
.700	.2194						
.725				.0487			
.750							
.760			.0355				
.775			.0520				
.808				.0521	.0192		
.834	.0587						
.850			.0206	.0420	.0164	.0071	
.857							
.865	.0642						
.900	.0757			.0329	.0483		.0312
.905			.0736				
.950				.0653		.0619	
.953			.1271				
.965	.0986						

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(R81U33)

ARC11-716 1A14 01-112-512823-AT10 UPPER WING

ALPHAO(1) = -10.230 BETAO(10) = 0.120

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000	-.2837	-.0787	.1091	.2289	.0600	-.0610	-.0495
.020				.4945	.3713	.3905	.4129
.025			.2829				
.040		.0143					
.045			.2705				
.050	-.1149			.3857	.5639	.3440	.3972
.060				.2896			
.085			.2296				
.086		.0969					
.094	-.0841						
.150		.0000		.1361	.1460	.2254	.2734
.163							
.177			.1517				
.193		.1776					
.229	-.0175			-.0037	.0135	.2021	.1094
.250							
.274			.0576				
.339		.0175					
.362	.1611						
.400				.1642	.1455		-.0406
.402			-.0495				
.497	.0500						
.550				.1286	.0636		
.565			.0300				
.600							-.1239
.650					-.0506		
.700	.1750				-.1404		
.725				.0362			
.750					-.0216	-.0803	
.760			.0224				
.775				.0303	.0073		
.808			.0126				
.834	.0343						
.850				.0100	-.0046	.0120	
.857			-.0157				
.865	.0539						.0378
.900	.0245			-.0076	.0350		
.905			.0044				
.950				.0160		.0918	
.953			.0537				
.965	.0423						

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881033)

ARC11-716 1A14 Q1+T12+S12N23+AT10 UPPER WING

ALPHA(1) = -10.240 BETA(11) = 10.110

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/OW	.2590	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	-.2463	-.1003	.0662	.1340	.0303	-.0649
	.020				.4356	.3471	.4242
	.025			.2397			.4177
	.040		-.0446				
	.045			.2321			
	.050	-.1131			.3530	.3453	.4001
	.060				.2610		.4110
	.085			.1956			
	.086		.0612				
	.094	-.0762			.1576	.1607	.3379
	.150						.2794
	.163		.0000				
	.177		.1143				
	.193		.1560				
	.229	-.0211			.0061	.1029	.2154
	.250						.1208
	.274		.0374				
	.339	-.0323					
	.362	.1064			.2612	.1715	-.0388
	.400						
	.402		-.0729				
	.497	.0000			.1256	.0384	
	.550						
	.565		.0000				-.1266
	.600				-.0906		
	.650				-.037		
	.700	.1624			.0300		
	.725						
	.750		.0196			-.0331	-.1160
	.760				.0300	.0047	
	.775		.0022				
	.806						
	.834	.0261			.0126	-.0046	-.0079
	.850						
	.857		-.0255				
	.865	.0579			-.0084		-.0160
	.900	.0256				.0206	
	.905		.0010				.0358
	.950		.0253				
	.955		.0360				
	.965	.9349					



DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+S12N29+AT10 UPPER WING (R81033)

ALPHA(2) = -8.220 BETAO (1) = -9.940

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BV .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0246	-.0523	.3594	.6094	.5454	.4720	.5096
.050			.5004	.6799	.8416	.6909	.6664
.025		.1191					
.040			.4439				
.045				.4689	.4684	.4759	.4740
.050	.0118			.2975			
.065			.3218				
.085		.2289					
.086							
.094	.0493			.0959	.0664	.1423	.1667
.150		.0500					
.163			.1223				
.177		.2226					
.193							
.229	.0939			-.0927	-.0667	-.0735	-.0619
.250							
.274		-.0066					
.339		.0182					
.362	.1970						
.400				-.2436	-.2680		-.3184
.402		-.2017					
.497	.0000						
.550			.0000	-.2790	-.3308		
.565							
.600							
.650							
.700	-.0563					-.3906	
.725				.0244	-.1294		
.750			.0419			-.3410	-.3569
.760				.1543	.0322		
.775			.1943				
.806							
.834	.1172			.1722	.1203	-.0615	
.850			.1875				
.857							
.865	.2267			.1976			-.1813
.900	.2169		.2325	.1964			
.905				.2367		.0071	
.950			.2732				
.953							
.965	.2489						

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1033)

ARC11-716 1A14 01+712+312N23+AT10 UPPER WING

ALPHA(2) = -8.240 BETA(2) = -7.960

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BA	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	-.0074	-.0050	.3149	.5506	.4949	.4192	.4609
.020				.6490	.6103	.6255	.6346
.040		.0580	.4552				
.060			.4013				
.080	-.0263		.4480	.4496	.4552	.4535	
.100		.2854	.2793				
.120		.1937					
.140	.0220			.0441	.0785	.1301	.1544
.160		.0000	.1013				
.180		.1978					
.200	.0386			-.1096	-.0968	-.0790	-.0896
.220		-.0167	-.0226				
.240							
.260	.1707			-.2534	-.2735		-.3223
.280		-.2190					
.300	.0000			-.2857	-.3340		
.320		.0500					-.3772
.340							
.360							
.380	.0651			.0377	-.1177		
.400							
.420							
.440							
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820	.1095						
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.2042							-.0801
.2431							
.2565							.0480
.2692							
.2360							





(R01U33)

ARC11-716 1A14 04+T12+S12N25+AT10 UPPER WING

$$\text{ALPHAO}(2) = -0.250 \quad \text{BETAO}(4) = -3.900$$

SECTION 11 UPPER WING

DEPENDENT VARIABLE CP

Y/8W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

**MX/X**

-.000	-.0082	-.1776	.1949	.4436	.3733	.3175	.3624
.020				.5933	.5445	.5764	.5797
.025			.3759				
.040		-.0307					
.045			.3357				
.050	-.0801			.4164	.4210	.4395	.4200
.060			.2366	.2626			
.065							
.066		.1110					
.094	-.0236						
.150				.0432	.0877	.1351	.1436
.163		.0000	.0948				
.177		.1527					
.193							
.229	-.0345						
.250			-.0294	-.1276	-.0879	-.0736	-.0953
.274		-.0254					
.339							
.362	.1131						
.400				-.2615	-.2494		-.3156
.402			-.1948				
.497	.0000						
.550				-.1292	-.3010		
.565			.0000				
.600							-.3610
.650						-.3672	
.700	.1378			.0466	-.1107		
.725							
.750			.0430			-.0746	-.3378
.760				.1027	.0661		
.775			.1272				
.808							
.834	.0775						
.850				.0909	.1037	.1110	
.857			.1212				
.865	.1714						
.900	.1551						
.925			.1334	.1011	.1529		.0486
.950				.1543			
.953			.1793			.1965	
.965	.1597						

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(R81U33)

ARC11-716 1A14 Q1+T12+S12N25+AT1D UPPER WING

ALPHA(2) = -8.250 BETA(5) = -1.990

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2995	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1280	-.2177	.1422	.3916	.3210	.2640	.3100
.025				.5700	.5148	.5446	.5457
.025			.3363				
.040		-.0678					
.045			.3044				
.050	-.0941			.4032	.4096	.4142	.4003
.065			.2210	.2597			
.086		.0763					
.094	-.0904			.0523	.0941	.1326	.1368
.150		.0700					
.163			.1021				
.177		.1424					
.193							
.229	-.0214			-.0876	-.0787	-.0694	-.0946
.274		-.0193					
.339	.0939						
.368				-.2158	-.2358		-.3093
.400			-.1800				
.402							
.497	.0000			.0379	-.2572		
.550			.0000				
.600							
.650							
.703	.1979				-.2136		-.3743
.725				.0675	-.1119		
.750						.0379	-.1421
.780		.0368		.1000	.0899		
.775			.1000				
.876							
.834	.0645			.0828	.0748	.1025	
.850			.0936				
.857							
.865	.1435						
.900	.1310			.0872	.1210		.1250
.905		.1242		.1366		.1802	
.950		.1468					
.955							
.965	.1265						

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DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R01033)

ARC11-716 IA14 01+712+312N25+AT10 UPPER WING

ALPHA( 2) = -8.250 BETA( 0) = .010

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CM				
.000	-.1645	-.2475	.1690	.3675 .2881 .2100 .2483
.020			.3214	.5440 .4878 .5125 .5048
.025		-.0944		
.040			.2946	
.045				.3887 .3958 .3933 .3730
.050	-.1175		.2147	.2549
.065				
.066		.0482		
.094	-.0702			
.150		.0000		.0645 .0974 .1240 .1232
.163			.1092	
.177		.1379		
.193				
.229	-.0405			
.250				.0726 -.0744 -.0716 -.1003
.274		-.0047		
.339	-.0075			
.362	.0839			
.400				-.1951 -.2227 -.3152
.432		-.1540		
.497	.0000			.0988 -.0330
.550		.0000		
.565				
.600				-.3645
.630				
.700	.1628			-.0295
.725			.0590	-.1140
.750				.0265 .0040
.760		.0282		
.775		.0739	.0498	
.808				
.834	.0939		.0593 .0475 .0644	
.850		.0649		
.857				
.885	.1134			
.900	.1116		.0569 .0799	.1627
.925		.1028		
.950			.1098	.1470
.955		.1239		
.985	.1033			



(R81U33)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-T12-S12N25+AT10 UPPER WING

ALPHA(2) = -0.230 BETA(7) = 2.020

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7800 .8870
X/CN					
.000	-.1868	-.1702	.1689	.3252	.2411 .1588 .1911
.020				.3059	.4508 .4772 .4890
.025			.2920		
.040		-.0747			
.045			.2650		
.050	-.1384			.3634	.3734 .3701 .3494
.080				.2431	
.085			.1990		
.086		.0483			
.094	-.0776			.0723	.0949 .1147 .1147
.150		.0050			
.163			.1062		
.177		.1335			
.193					
.229	-.0329			-.0676	-.0746 -.0734 -.1016
.250					
.274		-.0102			
.339				-.0011	
.362	.0687				
.400				-.1833	-.2111 -.3100
.402			-.1479		
.497	.0000			.1218	.0697
.550					
.565					
.600					
.650					
.700	.1549				
.725					
.750					
.760					
.775					
.808					
.834	.0471				
.850					
.857					
.865	.0820				
.900	.0903				
.925					
.950					
.953					
.965					
.9748					

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(R81033)

DATE 19 JAN 75  
TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 OR+T12+312N23+AT10 UPPER WING

ALPHA(2) = -0.230 BETA(0) = 4.060

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP									
Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870			
X/CW										
.000	-.2091	-.0965	.1701	.2856	.1932	.1114	.1431			
.025			.2745	.4779	.4999	.4335	.4261			
.045		-.0334	.2473							
.060	-.1404			.3510	.3448	.3403	.3250			
.080				.2377						
.085			.1841							
.086		.0594								
.094	-.0921			.0706	.0867	.1003	.1263			
.150		.0000								
.163			.1012							
.177		.1369								
.193										
.229	-.0905			-.0670	-.0780	-.0797	-.0843			
.250			.0030							
.274		-.0160								
.339										
.362	.1030			-.1838	-.2054		-.2808			
.400			-.1520							
.402				.1017	.0868					
.437	.0000		.0000							
.539										
.585										
.600										
.650										
.700	.1356				-.1178	.0146				
.725			.0420							
.750						.0325	-.0260			
.760			.0128							
.775			.0372		.0526	.0546				
.808										
.834	.0390			.0446	.0430	.0493				
.850			.0036							
.857										
.865	.0366									
.900	.0478			.0282	.0420		.1035			
.905			.0484							
.950			.0721				.1091			
.955			.0880							
.965	.0450									



DATE 19 JAN 75  
 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 Q1+T12+312M25+AT10 UPPER WING

(R31U33)

ALPHA(2) = -0.220 BETA(9) = 6.050

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/8W	X/CM	CP
.2990	.3640	.4270
.2990	.3640	.5340
.2990	.3640	.6730
.2990	.3640	.7800
.2990	.3640	.8870
.2990	.3640	.9940
.2990	.3640	.1134
.2990	.3640	.4366
.2990	.3640	.4208
.2990	.3640	.2387
.2990	.3640	.2130
.2990	.3640	.3354
.2990	.3640	.3646
.2990	.3640	.3450
.2990	.3640	.3204
.2990	.3640	.2227
.2990	.3640	.1612
.2990	.3640	.0465
.2990	.3640	.1104
.2990	.3640	.1165
.2990	.3640	.1137
.2990	.3640	.0000
.2990	.3640	.0851
.2990	.3640	.1242
.2990	.3640	-.0448
.2990	.3640	-.0512
.2990	.3640	-.0475
.2990	.3640	-.0529
.2990	.3640	-.0601
.2990	.3640	-.0624
.2990	.3640	-.0280
.2990	.3640	.0976
.2990	.3640	-.1423
.2990	.3640	-.0939
.2990	.3640	-.0570
.2990	.3640	-.1455
.2990	.3640	.1292
.2990	.3640	.0726
.2990	.3640	.0000
.2990	.3640	-.1237
.2990	.3640	-.0907
.2990	.3640	-.1297
.2990	.3640	.0435
.2990	.3640	-.0171
.2990	.3640	-.1001
.2990	.3640	.0112
.2990	.3640	.0194
.2990	.3640	.0439
.2990	.3640	.0513
.2990	.3640	.0096
.2990	.3640	.0047
.2990	.3640	.0134
.2990	.3640	.0152
.2990	.3640	.0323
.2990	.3640	.0591
.2990	.3640	.0510
.2990	.3640	.0323
.2990	.3640	.0560
.2990	.3640	.0929
.2990	.3640	.0715

(RB1033)

ALPHA( 2 ) = -0.220 BETA( 10 ) = 0.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.000	-.3069	-.0768	.1117	.2367	.1197	.0156	.0417
.020				.4379	.3561	.3769	.3741
.025			.2772				
.040		-.0337					
.045			.2598				
.050	-.1645			.3429	.3158	.3136	.3020
.080				.2395			
.085			.2311				
.086		.0653					
.094	-.1201			.0796	.0939	.1115	.1393
.150		.0000					
.163			.1125				
.177		.1601					
.193							
.229	-.0394		.0169	-.0532	-.0556	-.0445	-.0175
.250							
.274		-.0572					
.339							
.362	.1468			-.0897	-.0072		-.0380
.400			-.1197				
.492							
.497	.0000			.0899	.0564		
.550			.0000				
.565						-.0567	-.1445
.600							
.650							
.700	.1205			.0330			
.725						-.0218	-.1181
.750			.0059				
.760				.0317	.0086		
.775			.0137				
.808							
.834	.0220			.0033	-.0111	-.0194	
.890							
.897			-.0225				
.885	.0372			-.0181			.0059
.900	.0114						
.905			-.0031	.0029			
.950			-.0074			.0189	
.955			.0201				
.965	.0111						

(R81033)

ARC11-716 1A14 Q1+Y12+S12N23+AT10 UPPER WING

ALPHAC( 2) = -0.220 BETA0 (1) = 10.130

SECTION ( : ) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

A5/A

-.3098	-.0914	.0681	.1877	.0849	-.0261	.0322
.020			.4385	.3532	.3451	.3945
.025		.2310				
.045	-.0477					
.050		.2160				
.060			.3308	.2997	.2039	.3572
.085		.1669	.2901			
.086						
.094	.0484					
.110	-.1091		.0787	.0887	.0988	.1902
.163		.0968				
.177						
.193	.1294					
-.0368						
.229			-.0498	-.0517	.0215	.0496
.250		.0003				
.274	-.0480					
.339						
.362						
.400			-.0529	.0055		-.0400
.402		-.1439				
.497	.0000		.0932	.0841		
.550		.0000			-.0600	-.1386
.565						
.630						
.650						
.700	.1013		.0468	-.1415		
.725					-.0212	-.1295
.750		.0135				
.760			.0395	.0189		
.775		-.0020				
.808						
.834	.0031					
.850			.0017	-.0103	-.0098	
.857		-.0312				
.863						
.900	.0036		-.0274			-.0067
.905		-.0188	.0016			
.950		-.0116			.0158	
.953		-.0121				
.965	.0041					

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ARC11-716 1A14 01+712+912N25+AT10 UPPER WING

$$\text{ALPHA}(3) = -6.280 \quad \text{BETA}(2) = -7.999$$

SECTION (1) UPPER WING

Y/8W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

**MJ/X**

.0019	-.0675	.3459	.5922	.5557	.4923	.5341
.0000			.6000	.5613	.5908	.5999
.020		.4154				
.025						
.040	.0615					
.045		.3475				
.050	-.0447		.3669	.3988	.4029	.3916
.060			.2208			
.085		.2116				
.086	.1601					
.094	-.0195		-.0223	.0155	.0678	.0909
.150		.0000				
.163						
.177		.0467				
.193		.1900				
.229	.0210					
.250			-.1885	-.1563	-.1466	-.1455
.274						
.339		-.0630	-.0870			
.362	.1234					
.400			-.3236	-.3299		-.3697
.402						
.497	.0000		-.3675	-.3793		
.550		.0000				
.565						
.600						
.650						
.700	-.1153		-.1468			-.4068
.725			-.0605			
.750						
.760		.0016				
.775			.0616	-.0920		
.806		.1617				
.834	.0863					
.850						
.857		.1476	.1132	.0222	-.1469	
.865	.1874					
.900	.1763		.1216			-.2471
.935		.1577	.1225			
.950			.1423			-.0756
.953		.1727				
.965	.1664					



(081033)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+12+312M25+AT10 UPPER WING

ALPHA( 3 ) = -6.300 BETA( 3 ) = -6.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2930	.3640	.4270
.5340	.5340	.6730
.7800	.7800	.8870
.000	-.0442	-.1142
.020		.2978
.025		.5373
.040		.5069
.045		.4475
.050		.5772
.055		.5598
.060		.5829
.065		.5720
.086		.3845
.094		.3176
.130		.3740
.163		.3871
.177		.3051
.193		.3714
.229		-.2143
.250		.1895
.274		.0000
.339		.0433
.362		-.0129
.400		.0146
.402		.0601
.497		.0831
.550		-.0129
.563		.0146
.600		.0601
.650		-.0129
.720		.0146
.730		.0601
.760		-.0129
.775		.0146
.808		.0601
.834		-.0129
.850		.0146
.857		.0601
.863		-.0129
.900		.0146
.903		.0601
.930		-.0129
.933		.0146
.965		.0601



(R81U33)

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6  
ARC11-716 IA14 01+T12+S12N25+AT10 UPPER WING

ALPHA( 3) = -0.260 BETA( 4) = -3.980

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/W	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.003	-.0781	-.1421	.2420	.4853 .4656 .3975 .4374
.020				.5565 .5291 .5371 .5424
.025			.3489	
.040		-.0239		
.045			.2912	
.050	-.0998			.3663 .3701 .3660 .3553
.060				.2143
.085			.1794	
.086		.0966		
.094	-.0607			-.0167 .0131 .0354 .0784
.150		.0000		
.163			.0444	
.177		.1145		
.193				
.229	-.0178			-.1550 -.1481 -.1385 -.1322
.250			-.0790	
.274		-.0715		
.339	.0766			-.2852 -.3143 -.3665
.362			-.2506	
.400				
.402	.0000		-.2920 -.3601	
.497			.0000	
.550				-.4100
.563				
.600				-.4216
.650				
.700	.0374			-.1226
.725			.0126	
.750				-.2700 -.4090
.760		.0168		
.775		.1107		
.808			.0944	
.834	.0593		.0949 .0755 -.1278	
.850				
.857		.0944		
.865	.1442			-.1030
.900	.1259		.0915	
.905		.1110	.1219	
.930			.1124	.1084
.935		.1262		
.965	.1174			

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(R81033)

## TABULATED PRESSURE DATA - 1A14A - VOL. 8

ARC11-716 1A14 Q1+712+81285+AT10 UPPER WING

DATE 10 JAN 75

ALPHA(3) = -0.100 BETA(3) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.8990	.3640	.4270	.9340	.6730	.7800	.8870
X/CW							
.000	-.1671	-.1917	.2308	.4374	.3662	.2919	.3342
.020				.5071	.4527	.4649	.4740
.025			.3039				
.040		-.0655					
.045			.2677				
.050	-.1566			.3356	.3246	.3189	.3086
.080				.1989			
.085			.1688				
.086		.0324					
.094	-.1071			-.0025	.0102	.0451	.0604
.130		.0000					
.163			.0618				
.177		.0998					
.193							
.229	-.0630			-.1306	-.1404	-.1362	-.1587
.250		-.0514					
.274		-.0472					
.339							
.362	.0478			-.2384	-.2861		-.3649
.400		-.2105					
.402							
.497	.0000			-.1060	-.3176		
.530							
.565		.0200					
.600							
.650							
.700	.0667						
.725				.0359	-.1220		
.750							
.760							
.775				.0098	.0440		
.806				.0556			
.834	.0283						
.850				.0453	.0469	.0562	
.857			.0355				
.885	.0914						
.900	.0742			.0273			-.0095
.905			.0372	.0682		.1352	
.930				.0333			
.953		.0702					
.985	.0356						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N25+AT10 UPPER WING

(881033)

ALPHA( 3 ) = -6.320 BETA( 6 ) = 2.000

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP					
Y/BW	.2990	.3640	.4270	.5340	.6730	.8870
X/CW						
.000	-.1933	-.1169	.2476	.4090	.3083	.2317
.020				.4696	.4191	.4405
.025			.3045			
.040		-.0445				
.045			.2716			
.050	-.1870			.3167	.3085	.3057
.080				.1799		
.085			.1671			
.086		.0303				
.094	-.1095					
.130		.0000		-.0010	.0135	.0494
.163			.0600			.0345
.177		.1102				
.193						
.229	-.0705			-.1223	-.1326	-.1338
.250						-.1610
.274		-.0452				
.339		-.0481				
.362	.0723			-.2436	-.2721	
.400						-.3804
.402		-.1972				
.487	.0000			-.0143	-.2387	
.550		.0000				
.565						-.4110
.600					-.1469	
.650						
.700	.0864			.0301		
.725				-.1249		
.750		.0003			-.0032	-.1721
.760				.0494	.0312	
.775		.0379				
.808						
.834	.0212			.0267	.0223	.0445
.850		.0113				
.857						
.865	.0622					.1259
.900	.0588			.0034	.0416	
.905		.0385				.1143
.930			.0285			
.933		.0491				
.965	.0308					

(R01U33)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01-T12-S12N23-AT10 UPPER WING

DATE 10 JAN 75

ALPHA( 3 ) = -6.330 BETA( 7 ) = 4.070

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2284	-.0709	.2260	.3696	.2576	.1817	.2673
.020			.2752	.4408	.3630	.4012	.4431
.025		-.0107					
.040			.2369	.2992	.2870	.2787	.3056
.045				.1710			
.050	-.1992		.1579				
.060							
.065		.0576					
.086							
.094	-.1194			.0011	.0118	.0365	.0720
.150		.0000					
.163			.0581				
.177		.1189					
.193							
.229	-.0598			-.1174	-.1353	-.1155	-.1407
.250		-.0409					
.274							
.339		-.0384					
.362	.0895			-.2365	-.2649		-.3467
.400							
.402		-.1972					
.497	.0000			.0132	-.0176		
.550							
.565							
.600							
.650							
.700	.0760						
.725				.0411	-.1232		
.750						.0154	-.0032
.760							
.775				-.0092	.0630	.0414	
.808				.0369			
.834	.0121			.0306	.0279	.0445	
.850				.0097			
.857							
.865	.0340						.1408
.900	.0449			.0069	.0375		
.905				.0485			
.930				.0333		.1182	
.953				.0729			
.965	.0542						



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RBIU33)

ARC11-716 1A14 OA-T12-S12N29-AT10 UPPER WING

ALPHA(3) = -6.380 BETA(8) = 6.050

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .3340 .6730 .7800 .8870

X/CW

.000	-.2927	-.0788	.1386	.3338	.2581	.1806	.1928
.020			.2262	.4170	.3942	.4029	.3924
.025							
.040		-.0174					
.045			.2064				
.050	-.2932			.2963	.3081	.2918	.2847
.060				.1913			
.085			.1451				
.086		.0304					
.094	-.1346						
.150		.0000		.0368	.0395	.0539	.0327
.163							
.177			.0575				
.193		.1030					
.229	-.0519						
.250				-.0922	-.1102	-.1180	-.1498
.274			-.0192				
.339		-.0468					
.362	.0850						
.400				-.1805	-.2235		-.3017
.402			-.1651				
.497	.0000			.0609	.0197		
.550			.0000				
.585							
.600							
.650							
.700	.1134						
.725				.0136			
.750				-.1315			
.760							
.775							
.808				-.0077	.0207	-.0013	
.834				.0152			
.850	.0099						
.857				.0037	-.0125	-.0084	
.895							
.900	.0375						
.920	.0297						
.935				-.0112	.0100		.0389
.950				.0162			
.953				.0187			.0468
.965				.0334			
.969	.0089						

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2752

(R81U33)

ARC11-716 1A14 Q1+112+312N25+AT10 UPPER WING

ALPHA( 3 ) = -6.270 BETA( 9 ) = 8.100

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW		.2950	.3640	.4270	.5340 .6750 .7850 .8870
X/CW					
.000	-.3482	-.0331	.1242	.2351	.1852 .0966 .1276
.020				.3793	.3474 .3706 .5339
.025			.2090		
.040		.0218			
.045			.2043		
.080	-.2043			.2753	.2793 .2704 .2452
.085			.1686	.1798	
.086		.0915			
.094	-.1408			.0320	.0323 .0505 .0323
.150		.0050			
.163			.0618		
.177		.1397			
.193					
.229	-.0276			-.0959	-.1103 -.1100 -.1427
.250			-.0228		
.274		-.0418			
.339				-.1663	-.1850 -.2719
.362	.1208				
.400			-.1785		
.402				.0619	.0297
.497	.0000				
.590		.0000			
.565					-.1507
.600				-.0720	
.650				-.1374	
.700	.0657			.0004	
.725					-.0371 -.1318
.750					
.760			-.0178	.0033	-.0122
.775			-.0087		
.806					
.834	.0047			-.0192	-.0261 -.0316
.850			-.0443		
.857					
.865	.0211				
.900	-.0093		-.0331		.0226
.905			-.0247	-.0189	
.915			-.0232	.0159	
.930					
.933			-.0241		
.965	-.0312				



ALPHA( 3) = -0.200 BETA( 10) = 10.090

SECTION 1, UPPER WING

DEPENDENT VARIABLE CP

Y/BW	2990	3640	4270	5340	6750	7800	8870
X/CV							
.000	-.3573	-.0764	.0647	.1937	.1800	.0542	.1002
.020			.1923	.3547	.3274	.3502	.3542
.040		-.0319					
.065			.1903				
.095	-.1980			.2677	.2580	.2393	.2676
.080			.1442	.1776			
.085		.0467					
.094	-.1316			.0906	.0900	.0349	.0701
.150		.0000					
.183			.0577				
.177		.1181					
.193							
.229	-.0404		-.0341	-.0907	-.1150	-.1062	-.0900
.250							
.274		-.0735					
.339							
.362	.0874		-.1829	-.1470	-.1503		-.2105
.403							
.422	.0900			.0344	.0136		
.497			.0000			-.0551	-.1290
.520					-.1455		
.565							
.600							
.650	.0415			-.0007			
.700						-.0192	-.1251
.725			-.0247	.0014	.0264		
.750			-.0297				
.775							
.808	-.0083			-.0127	-.0111	-.0028	
.834			-.0573				.0278
.850				-.0335	-.0069		
.857	.0023		-.0435	-.0220		.0212	
.900	-.0433						
.935							
.950			-.0517				
.993							
.965	-.0754						



(RB1033)

ARC11-716 1A14 01+712+312N23+AT10 UPPER WING

ALPHA(4) = -4.250 BETA(1) = -9.980

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7800
.8870		
.0322	-.0299	.4156
.7234		.6630
.5993		.8342
.5615		.5556
.5527		.5697
.4164		
.1136		
.3289		
.3435		.3359
.3418		.3288
.1556		
.1936		
.1501		
-.0182		
-.0878		-.0360
-.0065		.0264
.0050		
.0039		
.1184		
-.0973		
-.1416		
-.2457		-.2515
-.2115		-.2040
.1212		
-.3727		-.3948
-.4244		
-.3077		
-.4132		-.4422
.0000		
-.4233		
-.4599		
-.1656		
-.1231		
-.4905		-.4347
-.0356		
-.0048		-.2703
.1256		
.0517		-.2043
-.2754		
.1214		
.0760		
.0277		-.3089
.1309		
.0954		-.1315
.1190		
.0965		



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-712-312M25+AT10 UPPER WING (881033)

ALPHA(1) = -4.230 BETA(1) = -7.990

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/OW	.8990	.3640	.4270	.5340 .6730 .7800 .8870
X/CM				
.063	.0034	-.0596	.3631	.6842 .6279 .5598 .5964
.020			.5601	.5319 .5281 .5439
.025			.3742	
.040		.0606		
.045			.2999	
.050	-.0477		.3294 .3204 .3207 .3069	
.080			.1494	
.085			.1763	
.086		.1258		
.094	-.0519			
.150		.0000	-.0490	-.0875 -.0196 .0268
.163			-.0060	
.177		.0947		
.193				
.229	-.0084			
.250			-.2515	-.2300 -.2197 -.2118
.274		-.1454		
.339		-.1000		
.362	.0815		-.3799	-.3906
.400				
.432		-.3107		
.497	.0000		-.4261	-.4421
.550		.0500		
.565				
.600				
.650				
.700	-.2268			
.725			-.1313	-.1827
.750				
.760		-.0335	-.0154	-.2617
.775		.1274		
.808				
.834	.0563		.0536	-.1521 -.2407
.890		.1200		
.897				
.865	.1568		.0765	
.920	.1481			
.935		.1215	.0394	
.950			.0948	-.1375
.953		.1147		
.965	.0962			

(R01033)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01-T12-S12M25+AT10 UPPER WING

ALPHA( 4 ) = -4.180 BETA( 3 ) = -5.370

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.000	-.0353	-.0962	.3475	.6402
.020				.5836
.025				.5137
.040	.025	.0119	.3591	.4992
.045				.5139
.050	-.0838	.2820	.3155	.3026
.060			.1365	.2989
.065			.1616	.2861
.066				
.066	.0959			
.094	-.0834			
.150				
.163	.0020			
.177				
.193	.0674			
.229	-.0379			
.250				
.274				
.339				
.362				
.430	.0642			
.432				
.497	.0020			
.550				
.565				
.620				
.650				
.700	-.1090			
.725				
.750				
.760				
.778				
.808				
.834	.0477			
.850				
.857				
.865	.1339			
.900	.1224			
.905				
.950				
.953				
.965	.0792			



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+S12N2S+AT10 UPPER WING  
 (RB1033)

ALPHA( 4) = -4.170 BETA( 4) = -3.950

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6750 .7800 .8870
X/CW				
.000	-.0757	-.1413	.3125	.5914 .5409 .4670 .3047
.020				.5151 .4815 .4741 .4852
.025			.3386	
.040		-.0146		
.045			.2607	
.050	-.1103			.3001 .2892 .2850 .2719
.080				.1292
.085			.1458	
.086		.0703		
.094	-.1182			-.0940 -.0766 -.0351 -.0026
.150		.0000		
.163				
.177		.0735		
.193			-.0183	
.229	-.0313			
.250				-.2298 -.2313 -.2263 -.2242
.274		-.1130		
.339				
.362	.0363			-.3462 -.3893 -.4298
.400			-.3060	
.402				
.497	.0000			-.3654 -.4181
.550			.0000	
.565				-.4550
.600				
.650				-.4891
.700	-.0923			-.1404
.725				
.750				-.0682
.760				
.775				-.0293
.808				.0582
.834	.0330			
.850				-.0921
.857				
.865				.0600
.900	.1086			
.902	.0920			.0600 -.0222 -.1790
.905				
.950				.0502 .0750 -.1996
.953				.0782 .0670 -.1391
.965	.0562			.0653

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(R81033)

DATE 10 JAN 75 T/FULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 Q1+T12+312N25+AT10 UPPER WING

ALPHA( 4 ) = -4.000 BETA( 6 ) = -.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1481	-.1230	.3188	.5174	.4367	.3673	.4070
.025			.3036	.4513	.4177	.4197	.4316
.040		-.0549					
.045			.2362				
.050	-.1699			.2807	.2544	.2498	.2376
.060			.1339	.1201			
.065							
.066		.0254					
.094	-.1434						
.150		.0000		-.0723	-.0710	-.0345	-.0114
.163			.0156				
.177		.0656					
.193							
.229	-.0788			-.2104	-.2024	-.2073	-.2200
.250			-.0966				
.274		-.0794					
.339							
.362	.0367			-.3333	-.3471		-.4129
.400			-.2623				
.402				-.3329	-.3661		
.497	.0000		.0000				
.550							
.563							
.600							
.650							
.700	-.0015			-.0264	-.1308	-.4462	
.725							
.750			-.0130			-.2594	-.3846
.760				.0366	.0008		
.775			.0434				
.806							
.834	.0072			.0213	.0326	-.0892	
.850			.0130				
.857							
.865	.0682						
.900	.0477			-.0029			-.1239
.905			.0262		.0623		
.950				.0053		.1299	
.953			.0272				
.965	.0101						

(881033)

ARC11-716 1A14 01+712+312M23+AT10 UPPER WING

ALPHA(4) = -4.810 BETA(7) = 2.110

## SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1839	-.0876	.3184	.4635	.3806	.3091	.3538
.020			.4216	.3830	.3929	.3982	
.025			.2914				
.040		-.0172					
.045			.2226				
.050	-.2119		.2417	.2368	.2314	.2188	
.060			.1126				
.085			.1221				
.086		.0586					
.094	-.1321			-.0765	-.0679	-.0391	-.0185
.110		.0000					
.163			.0098				
.177							
.193		.0980					
.229	-.0881						
.250			-.2032	-.1966	-.1992	-.2218	
.274		-.0813	-.0976				
.339							
.362	.0682			-.3269	-.3342		-.4181
.400							
.402			-.2559				
.497	.0000			-.2574	-.3764		
.550							
.555			.0000				
.600						-.4409	
.650					-.4358		
.700	.0191		-.0194				
.725							
.750			-.0193	.0145	.0147		
.760						-.1686	-.2958
.775			.0205				
.808							
.834	-.0035			-.0030	.0226	.0107	
.850			-.0110				
.857							
.885	.0439			-.0303			-.1040
.900	.0286				.0369		
.905			.0035	-.0224		.1109	
.950			.0051				
.953							
.985	-.0132						



(R01033)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+312N25+AT10 UPPER WING

ALPHA( 4 ) = -4.200 BETA( 8 ) = 4.990

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.000	-.2427	-.0701	.2327	.4371
.020			.3320	.2571
.025			.3974	.3546
.040		.0036	.2329	.4022
.045			.1986	
.050	-.2395		.2298	.2145
.060			.1068	.2326
.085			.1013	
.086		.0814		
.094	-.1534			
.150		.0000	-.0718	-.0710
.163				-.0489
.177		.0059		-.0030
.193		.0902		
.229	-.0677			
.250			-.1784	-.1957
.274		-.0929		-.2005
.339		-.0820		-.2053
.362	.0776			
.400			-.2943	-.3272
.402		-.2460		-.3956
.497	.0000		-.1377	-.3534
.550		.0000		
.565				
.600				-.4223
.650			-.3447	
.700	.0094		-.1340	
.725			-.0055	
.750				-.0567
.760		-.0307		-.2061
.775		.0046	.0418	.0384
.808				
.834	-.0179		.0142	.0294
.850		-.0155		.0562
.857				
.865	.0157			
.900	.0066		-.0129	.0416
.925		.0067		.1339
.950		-.0563		
.953		.0096		
.965	-.0192			



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U33)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHAO (4) = -4.210 BETA0 (9) = 6.960

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3154	-.0727	.1555	.3899	.3556	.2329	.2643
.020				.3674	.3585	.3459	.3480
.025			.1576				
.040		-.0126					
.045			.1408				
.050	-.2656			.2319	.2275	.2044	.1954
.080			.1007	.1219			
.085							
.086		.0460					
.094	-.1355						
.150		.0000		-.0276	-.0407	-.0297	-.0148
.163							
.177			.0038				
.195		.0819					
.229	-.0485			-.1519	-.1771	-.1859	-.2109
.250							
.274			-.0681				
.359		-.0806					
.362	.0877			-.2563	-.2978		-.3940
.400							
.402			-.1978				
.497	.0000			-.0202	-.0877		
.550			.0000				
.565						-.1140	-.3924
.600					-.1357		
.650	.0672			-.0279		-.0480	-.0648
.700							
.725			-.0922				
.750				-.0004	-.0136		
.780			-.0070				
.775							
.808							
.834	-.0179			-.0191	-.0250	-.0055	
.850							
.857			-.0355				
.865	.0135						.1195
.900	-.0001			-.0378			
.905			-.0131	-.0111			
.950			-.0244			.0748	
.955			-.0120				
.965	-.0335						



ABC11-716 1A14 01-712-S12N25-A110 UPPER WING

ALPHA( 4 ) = -4.203 BETA( 12 ) = 0.383

SECTION 11111P WING

DEPENDENT VARIABLE CP

X/Cu	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.3758	-.0634	.1353	.3004	.2806	.1885	.2117
.020				.3136	.3276	.3286	.3223
.025			.1877				
.045		.0125					
.045			.1569				
.090	-.2855			.2053	.2084	.2059	.1837
.080				.1055			
.085			.1231				
.086		.0862					
.094	-.1825			-.0513	-.0418	-.0222	-.0133
.150		.0000					
.163			.0133				
.177							
.193		.1023					
.229	-.0322			-.1513	-.1760	-.1794	-.2107
.250			-.0732				
.274							
.339		-.0734					
.362	.1035			-.2372	-.2795		-.3849
.400			-.2073				
.497	.0900			-.0195	-.0810		-.3533
.550			.0000			-.0955	
.565							
.600							
.650							
.700	.0327			-.0383	-.1389		-.0768
.723						-.0477	
.793			-.0517	.0001	-.0141		
.783			-.0158				
.858							
.834	-.0092			-.0309	-.0266	-.0270	
.890			-.0558				
.857							
.865	.11						.0712
.905	-.0203			-.0477	-.0237		
.955			-.0340				
.953				-.0354		.0387	
.933			-.0300				
.965	-.0515						

(RB1033)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-112-S12N25+AT10 UPPER WING

ALPHAO( 4) = -4.165 BETAO (11) = 10.090

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	.2995 .3640 .4270 .5340 .6730 .7800 .8870
X/CW	
.000	-.3864 -.0793 .0872 .2240 .2413 .1391 .1860
.020	.2602 .2993 .2963 .2849
.025	.1325
.040	-.0426
.045	.1325
.050	.1749 .1929 .1768 .1548
.080	.0959
.085	.1130
.086	.0423
.094	-.1473
.150	.0000
.153	.0133
.177	.0658
.193	-.0297 -.0442 -.0386 -.0145
.229	-.0331
.250	-.1425 -.1755 -.1856 -.1920
.274	-.0763
.339	-.0975
.362	.0751
.400	-.2202 -.2564
.402	-.2132
.497	.0000
.550	-.0474 -.1204
.565	.0000
.600	-.2748
.650	-.0692
.700	-.0155
.723	-.0493
.750	-.0437 -.0874
.760	-.0969
.775	-.0205 -.0009
.808	-.0304
.834	-.0408 -.0199 .0046
.850	-.0720
.857	-.0567
.869	-.0741
.900	-.0132
.905	-.0466
.930	-.0693
.953	-.0401
.965	-.1058



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+512NC+AT10 UPPER WING (R81U33)

ALPHA( 5 ) = -2.870 BETA( 1 ) = -10.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/SW	.2990	.3840	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	.0390	-.0144	.4742	.7507	.7055	.6419	.6701
.020				.5305	.5217	.5086	.5276
.025			.4037				
.040		.0942					
.045			.3020				
.050	.0149			.2880	.2848	.2828	.2721
.080				.0988			
.085			.1549				
.086		.1277					
.094	-.0258						
.150		.0000		-.1224	-.1260	-.0621	-.0190
.163			-.0373				
.177		.0944					
.193							
.229	-.0009			-.2941	-.2791	-.2568	-.2426
.250			-.1680				
.274		-.1181					
.339	.0951			-.4029	-.4349		-.4521
.400			-.3252				
.402				-.4574	-.4814		
.497	.0000		.0000				-.4521
.550						-.4840	
.600				-.1568			
.650	-.2342			-.1762			
.700						-.9285	-.4392
.725							
.750			-.0578	-.0578	-.3467		
.760			.0942				
.775							
.808							
.834	.0480			-.0012	-.2083	-.3494	
.850			.1027				
.857							
.865	.1371			.0275			-.3395
.900	.1324				-.1196		
.905			.1148				
.950				.0532			-.1877
.953			.0991				
.965	.0787						

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+112+512N25+AT10 UPPER WING

ALPHA( 5) = -2.070 BETA( 2) = -8.030

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/B	X/CW			
.2990	.3640	.4270	.5340	.6730 .7800 .8870
.0096	-.0598	.4232	.7166	.6605 .5968 .6276
.029		.3714	.5137	.4932 .4807 .4986
.025				
.040	.0612			
.045		.2732		
.050	-.0375		.2734	.2648 .2615 .2519
.060			.0645	
.065		.1418		
.066	.1038			
.094	-.0641		-.1407	-.1319 -.0744 -.0302
.150		.0000		
.163		-.0515		
.177		.0760		
.193				
.229	-.0247		-.2964	-.2034 -.2033 -.2511
.250		-.1352		
.274			-.4057	-.4368 -.4587
.339		-.3343		
.362	.0683		-.4587	-.4826
.400		.0000		
.432			-.1564	
.497	.0000		-.1719	
.590			-.0650	-.0743 -.3053
.565			.0917	
.600				
.650			-.0045	-.2387 -.2044
.700	-.2042			
.725				
.750				
.760				
.775				
.806				
.834	.0394			
.850				
.897				
.883	.1400		.0340	-.0768 -.3421
.900	.1326		.1036	
.905			.0630	-.1790
.950			.0855	
.953				
.985	.8702			



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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+S12M23+AT10 UPPER WING

ALPHA (  $\beta$  ) = -2.070 BETA (  $\beta$  ) = -5.960

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
V/BW	.2993 .3640 .4275 .5340 .6730 .7800 .8870
X/CM	
.000	-.0299 -.0924 .3895 .6783 .6189 .5515 .5852
.020	
.025	.3491
.040	.0178
.045	.2492
.050	-.0796
.060	.2640 .2574 .2471 .2357
.065	.0861
.086	.1230
.094	.0671
.150	-.0936
.163	-.1350
.177	-.1278
.193	-.0782
.229	-.0415
.250	
.274	-.2729
.333	-.2834
.362	-.2687
.405	-.2589
.402	-.1820
.497	-.1365
.550	
.565	-.3806
.620	-.4337
.650	-.4609
.700	-.4285
.725	-.4734
.750	-.0000
.775	-.4675
.806	
.834	-.9076
.850	-.1653
.857	-.1360
.865	-.9137
.900	-.4528
.905	-.0624
.930	-.0336
.935	-.2695
.965	.0932
	.0364
	-.1779
	-.2472
	.0775
	.0545
	-.3120
	-.0108
	.0830
	-.0720
	-.1578
	.0528
	.0485

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DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01+T12+512W25+AT10 UPPER WING

ALPHA( 5 ) = -2.860 BETA( 5 ) = -2.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

K/CW

.000	-.1106	-.1041	.3395	.5904	.5236	.4589	.4923
.020				.4503	.4206	.4068	.4194
.025			.2965				
.040		-.0234					
.045			.2045				
.050	-.1379			.2331	.2209	.2145	.2034
.060				.0772			
.085			.0969				
.088			.0399				
.094	-.1467						
.150				-.1197	-.1187	-.0846	-.0495
.165		.0000					
.177		-.0421					
.195		.0515					
.229	-.0772			-.2522	-.2682	-.2601	-.2625
.250							
.274		-.1492					
.339		-.1203					
.362	.0838			-.3598	-.4050		-.4538
.400			-.3096				
.402				-.3740	-.4281		
.497	.0000			.0000			
.550							
.565							
.600							
.650							
.700	-.1155				-.1448	-.4987	-.4728
.725				-.0879			
.750							
.760		-.0429		.0316	-.1158	-.4058	-.4316
.775			.0579				
.806	.0077						
.834				.0401	-.0396	-.1964	
.850			.0290				
.857							
.865	.0721			.0228			-.1987
.900	.0535			.0371	.0549		
.905				.0252		-.1456	
.950							
.955			.0198				
.965	.0059						





DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U33)

ARC11-716 IA14 01+112-512N23+AT10 UPPER WING

ALPHAO ( 5 ) = -2.850 BETA0 ( 7 ) = 2.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7600 .8670

X/CW

.050	-.1740	-.0759	.3322	.4947	.4195	.3545	.3962
.090			.3754	.3570	.3508	.3638	
.325			.2609				
.040		-.0068					
.045			.1927				
.050	-.2217		.1960	.1845	.1833	.1716	
.080			.0611				
.085			.0881				
.086		.0532					
.094	-.1690						
.150		.0050		-.1148	-.1174	-.0911	-.0835
.163							
.177			-.0257				
.193		.0793					
.229	-.0699			-.2505	-.2383	-.2561	-.2665
.250			-.1322				
.274		-.1086					
.362	.0636			-.3536	-.3710		-.4467
.339							
.400			-.2863				
.402							
.497	.0000			-.3467	-.4070		
.350			.0050				
.565							
.600							
.650							
.700	-.0366			-.0478	-.1425		
.725							
.750			-.0350			-.2037	-.3383
.760				.0060	-.0301		
.775			.0096				
.808							
.834	-.0181			-.0103	.0062	-.1193	
.850							
.857			-.0239				
.865	.0300						
.900	.0161			-.0382			-.1521
.905			-.0144		.0334		
.950				-.0396			.1033
.953			-.0200				
.965	-.0374						

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U33)

ARC11-716 1A14 01+112+312N23+AT10 UPPER WING

ALPHA( 5 ) = -2.770 BETA( 8 ) = 4.100

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2498	-.0782	.2308	.4740	.3784	.3046	.3953
.020			.2025	.3556	.3236	.3120	.3647
.025		-.0015					
.040			.1566				
.045				.1838	.1630	.1554	.1838
.050	-.2504			.0921			
.060			.0770				
.085		.0485					
.086							
.094	-.1759			-.1135	-.1203	-.0988	-.0425
.150		.0000					
.163			-.0308				
.177		.0720					
.193							
.229	-.0674			-.2145	-.2339	-.2330	-.2426
.250			-.1222				
.274		-.1108					
.339							
.362	.0635			-.3247	-.3617		-.4240
.400			-.2705				
.402							
.497	.0000			-.2241	-.3793		
.550			.0000				
.565						-.4336	-.4331
.600							
.650							
.700	-.0314			-.0162	-.1368		
.725						-.1501	-.2588
.750			-.0539				
.760				.0258	.0273		
.775			-.0041				
.808							
.834	-.0343			.0104	.0249	.0352	
.850			-.0401				
.857							
.885	.0022						
.900	-.0118			-.0202	.0389		-.1054
.905			-.0163				
.950			-.0231			.1314	
.953			-.0181				
.965	-.0399						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-112-812N25+AT10 UPPER WING (R01033)

ALPHA( 5 ) = -2.790 BETA( 9 ) = 6.120

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3390	-.0000	.1502	.4172	.3920	.2771	.3104
.020			.1265	.3171	.3329	.3030	.3146
.040		-.0334					
.060			.1032	.1853	.1727	.1511	.1458
.080	-.2807			.0700			
.100			.0600				
.120		.0336					
.140	-.1734			-.0813	-.1063	-.0638	-.0655
.160		.0000					
.180		.0565	-.0403				
.200	-.0483			-.2172	-.2185	-.2301	-.2529
.220			-.0986				
.240		-.1074					
.260	.0540			-.3175	-.3434		-.4305
.280		-.2236					
.300	.0000			-.0906	-.1904		
.320		.0000					
.340						-.2099	-.4316
.360	.0327			-.0640	-.1394		
.380						-.0661	-.2176
.400		-.0777		-.0248	-.0208		
.420			-.0164				
.440				-.0489	-.0248	.0027	
.460	-.0325		-.0477				
.480							
.500	-.0026			-.0654			.0970
.520	-.0150			-.0306	-.0035		
.540				-.0618		.1250	
.560			-.0361				
.580							
.600	-.0475						

## TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1033)

ARC11-716 1A14 01+712+312425+AT10 UPPER WING

ALPHA(5) = -2.790 BETA(10) = 0.140

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.000	-.4001	-.0414
.020		.1297
.025		.3393
.040		.2727
.045		.3082
.050		.2899
.060		.2836
.065		
.066		
.094		
.150		
.163		
.177		
.193		
.229		
.250		
.274		
.339		
.362		
.400		
.492		
.497		
.550		
.565		
.600		
.630		
.700		
.725		
.750		
.780		
.775		
.808		
.834		
.850		
.857		
.865		
.900		
.925		
.950		
.955		
.965		



DATE 13 JAN 79

TABULATED PRESSURE DATA - IA14A - VOL. 6

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ARC11-716 IA14 Q1+T12+S12M23+AT10 UPPER WING

(081033)

ALPHA( 5) = -2.770 BETA( 11) = 10.100

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000	-.4224	-.0806	.0943	.2344	.2623	.1966	.2227
.020				.1987	.2791	.2643	.2456
.025			.0943				
.040		-.0932					
.045			.0687				
.050	-.2794			.1132	.1449	.1353	.1147
.060				.0461			
.085			.0698				
.086		.0223					
.094	-.1581						
.150				-.0727	-.0963	-.0804	-.0691
.163		.0000					
.177			-.0172				
.193		.0709					
.229	-.0404						
.250				-.1823	-.2169	-.2291	-.2493
.274			-.1028				
.339		-.1116					
.362	.0619						
.400				-.2533	-.3044		-.4011
.432			-.2392				
.497	.0000			-.1061	-.1758		
.550			.0000				
.565							
.600							
.650							-.3775
.700	-.0436						
.725				-.0839	-.1568		-.1814
.750							
.760			-.0745			-.0954	-.0973
.775				-.0373	-.0208		
.808			-.0461				
.834	-.0324						
.850				-.0685	.0242	-.0137	
.857			-.0867				
.865	-.0268						
.900	-.0802		-.0714				.0969
.905			-.0835	-.0151			
.950			-.0678			.0833	
.955			-.0817				
.985	-.1137						

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U33)

ALPHA(0) = -.760 BETA(1) = -10.320

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7890	.8670
X/CW							
.000	.0330	.0825	.1636	.2916	.4336	.5915	.7479
.020				.4814	.4422	.4236	.4474
.025			.3501				
.040		.0832					
.045			.2706				
.050	.0241			.2132	.1973	.1922	.1834
.060				.0188			
.085			.0971				
.086		.1244					
.094	-.0123			-.2104	-.1977	-.1257	-.0911
.130		.0000					
.163			-.1093				
.177		.0611					
.193							
.229	-.0030			-.3258	-.3551	-.3253	-.2938
.230			-.2268				
.274		-.1594					
.339							
.362	.0936			-.4362	-.4960		-.4682
.400			-.3643				
.402							
.497	.0000			-.4725	-.5355		
.550			.0000				
.565						-.4984	
.600							
.650					-.1931		
.700	-.2893			-.2125			
.725						-.5264	-.4850
.750							
.760			-.1069				
.775				-.0934	-.4392		
.808			.0161				
.834	.0146						
.850				-.0222	-.2662	-.4397	
.857			.0375				
.865	.1317						
.900	.1330			.0214			-.4045
.905			.0860	-.2039			
.930				.0385		-.2432	
.933			.0846				
.965	.0505						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14A - VOL. 6

(081033)

ARC11-716 IAI4 01+T12+S12N25+AT110 UPPER WING

ALPHA( 6 ) = -.730 BETA( 2 ) = -.420

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/BW			
.2990	.3640	.4270	.5340	.6730 .7800 .8870
.000	.0222	.0289	.4988	.7687 .7095 .6363 .6989
.020			.4402	.4278 .4047 .4348
.025		.0549		
.040			.2215	
.045			.1972	.1822 .1769 .1768
.050	-.0156		.0094	
.080		.0678		
.085		.0869		
.086				
.094	-.0619			
.190		.0000	-.2184	-.2084 -.1426 -.1020
.163				
.177		-.1252		
.193	.0180			
.229	-.0449		-.3371	-.3627 -.3382 -.3036
.274		-.2385		
.339		-.1760		
.362	.0528		-.4430	-.5036 -.4990
.400				
.402		-.3793		
.497	.0000		-.4816	-.5401
.550		.0000		
.565				-.5142
.600			-.5300	
.630			-.1915	
.700	-.3064		-.1837	
.725				
.750				
.760		-.1132	-.1103	-.4207
.775		.0187		
.808				
.834	.0055		-.0941	-.2768 -.3913
.850		.0589		
.857				
.865	.1196		-.0168	-.4109
.900	.1156			
.905		.0849	-.2388	
.920		.0098	-.2608	
.930		.0717		
.953				
.965	.0435			

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(R01033)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 Q1+T12+S12M23+AT10 UPPER WING

ALPHA ( 6 ) = -.730 BETA ( 3 ) = -6.290

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.000	-.0104	-.0455
.020	.4634	.7300
.025	.4253	.4121
.040	.2728	.3039
.045	.0322	.3984
.050	.1987	.2728
.055	.1875	.1724
.060	.0265	.1632
.065	.0815	.1499
.086	.0645	.0265
.094	-.0945	.0815
.150	.0000	.0645
.163	-.1248	.0000
.177	.0103	-.2182
.173	-.0600	-.2115
.229	-.2406	-.1488
.250	-.1793	-.3135
.274	.0340	-.3314
.339	.0000	-.3575
.362	-.3774	-.3431
.400	.0000	-.4983
.402	.0000	-.4319
.437	.0000	-.4780
.550	.0000	-.3356
.565	.0000	-.5311
.600	.0000	-.5255
.630	.0000	-.5311
.700	.0000	-.1896
.725	.0000	-.1782
.750	.0000	-.5431
.760	.0000	-.4961
.775	.0000	-.1083
.808	.0000	-.1297
.834	.0000	-.3604
.850	.0000	.0287
.857	.0000	.0489
.865	.0000	-.0699
.900	.0000	-.2758
.905	.0000	-.3305
.920	.0000	.0489
.925	.0000	-.0225
.930	.0000	-.0225
.935	.0000	-.1936
.955	.0000	-.2445
.965	.0000	.0280
.985	.0000	.0484



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+312N25+AT10 UPPER WING (R81U33)

ALPHA (  $\theta$  ) = -.710 BETA (  $\delta$  ) = -4.140

SECTION ( UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.300	-.0506	-.0679	.4128	.6803	.6329	.3696	.5978
.029				.4076	.3940	.3599	.3715
.023			.2461				
.040		.0078					
.043			.1739				
.050	-.0921			.1817	.1616	.1905	.1376
.083				.0017			
.085			.0516				
.086		.0316					
.094	-.1227						
.133		.0500		-.2116	-.2134	-.1582	-.1247
.163			-.1182				
.177		.0067					
.193							
.229	-.0803						
.250				-.3445	-.3508	-.3432	-.3225
.274			-.2327				
.339		-.1798					
.362	.0256						
.400				-.4496	-.4858		-.5041
.402			-.3749				
.497	.0500						
.530				-.4974	-.5264		
.563			.0500				
.600							
.650							
.700	-.2790						
.725				-.1947	-.1778		
.750							
.760		-.0914					
.775				-.1269	-.3128		
.808			.0442				
.834	.0006						
.850				-.0433	-.2222	-.2916	
.857		.0417					
.865	.0850						
.900	.0703			-.0566			
.905		.0448			-.0864		
.930				.0191			
.953		.0188					
.965	-.0031						

-.3669

-.2268

(081033)

TABULATED PRESSURE DATA - IAI4A - VOL. 6

ARC11-716 IAI4 01-712-512N23-AT10 UPPER WING

ALPHA(0) = -.700 BETA(0) = -2.000

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7600	.8870
X/CW							
.000	-.0934	-.0652	.3613	.6480	.5824	.5292	.5541
.020				.3868	.3694	.3397	.3419
.025			.2294				
.040		-.0124					
.045			.1600				
.050	-.1323			.1706	.1477	.1343	.1226
.060				-.0007			
.085			.9425				
.086		.0459					
.094	-.1306			-.2068	-.2160	-.1622	-.1304
.150		.0000					
.163			-.1099				
.177		-.0034					
.193	-.0763			-.3135	-.3440	-.3376	-.3231
.229			-.2146				
.274		-.1665					
.339				-.3964	-.4700		-.5030
.352	.0098		-.3452				
.400				-.4267	-.4938		
.452	.0000		.0000				-.5032
.497					-.5324		
.539				-.1311	-.1674		
.565						-.5384	-.4606
.600							
.650	-.2320						
.700							
.725			-.0782				
.750			-.0477	-.2251			
.760			.0469				
.775							
.808							
.834	-.0009		.0113	-.1522	-.2493		
.850							
.857			.0248				
.865	.0671						-.2788
.900	.0482		.0139				
.905			.0245	-.0224			
.950			.0168			-.1971	
.955			-.0015				
.965	-.0211						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RB1033)

ARC11-716 1A14 01+T12+S12M25+AT10 UPPER WING

ALFMA2(6) = -.700 BETA2(6) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1181	-.0921	.3779	.5937	.5232	.4771	.5063
.020				.3429	.3298	.3055	.3174
.025			.2166				
.040		-.0300					
.045			.1588				
.050	-.1783			.1384	.1212	.1132	.1029
.080				-.0226			
.085			.0409				
.086		.0385					
.094	-.1693						
.150				-.1981	-.2156	-.1710	-.1397
.163		.0000					
.177			-.0934				
.193		.0143					
.229	-.0865			-.2952	-.3301	-.3306	-.3270
.250							
.274			-.1864				
.339		-.1454					
.362	.0198			-.3821	-.4497		-.5040
.400			-.3346				
.402				-.4007	-.4641		
.497	.0000		.0000				
.550							
.565							
.600							
.650							-.4957
.700	-.1661				-.5319		
.725				-.1248			
.750			-.0704			-.4533	-.4417
.760				-.0103	-.1724		
.775			.0206				
.808							
.834	-.0201			.0107	-.1006	-.2198	
.850							
.857			-.0112				
.865	.0334						
.900	.0161			-.0087			-.2460
.905			-.0067		.0141		
.950				-.0111		-.1744	
.955			-.0236				
.965	-.0428						

(RB1U33)

TABULATED PRESSURE DATA - IA14A - VOL. 6

DATE 10 JAN 79

ARC11-716 1A14 01+T12+S12N25+AT17 UPPER WING

ALPHA( 6 ) = - .703    BETA( 7 ) = 2.169

DEPENDENT VARIABLE CF

SECTION ( 1 ) UPPER WING

	1967	1968	1969	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	2101	2102	2103	2104	2105	2106	2107	2108	2109	2110	2111	2112	2113	2114	2115	2116	2117	2118	2119	2120	2121	2122	2123	2124	2125	2126	2127	2128	2129	2130	2131	2132	2133	2134	2135	2136	2137	2138	2139	2140	2141	2142	2143	2144	2145	2146	2147	2148	2149	2150	2151	2152	2153	2154	2155	2156	2157	2158	2159	2160	2161	2162	2163	2164	2165	2166	2167	2168	2169	2170	2171	2172	2173	2174	2175	2176	2177	2178	2179	2180	2181	2182	2183	2184	2185	2186	2187	2188	2189	2190	2191	2192	2193	2194	2195	2196	2197	2198	2199	2200	2201	2202	2203	2204	2205	2206	2207	2208	2209	2210	2211	2212	2213	2214	2215	2216	2217	2218	2219	2220	2221	2222	2223	2224	2225	2226	2227	2228	2229	2230	2231	2232	2233	2234	2235	2236	2237	2238	2239	2240	2241	2242	2243	2244	2245	2246	2247	2248	2249	2250	2251	2252	2253	2254	2255	2256	2257	2258	2259	2260	2261	2262	2263	2264	2265	2266	2267	2268	2269	2270	2271	2272	2273	2274	2275	2276	2277	2278	2279	2280	2281	2282	2283	2284	2285	2286	2287	2288	2289	2290	2291	2292	2293	2294	2295	2296	2297	2298	2299	2300	2301	2302	2303	2304	2305	2306	2307	2308	2309	2310	2311	2312	2313	2314	2315	2316	2317	2318	2319	2320	2321	2322	2323	2324	2325	2326	2327	2328	2329	2330	2331	2332	2333	2334	2335	2336	2337	2338	2339	2340	2341	2342	2343	2344	2345	2346	2347	2348	2349	2350	2351	2352	2353	2354	2355	2356	2357	2358	2359	2360	2361	2362	2363	2364	2365	2366	2367	2368	2369	2370	2371	2372	2373	2374	2375	2376	2377	2378	2379	2380	2381	2382	2383	2384	2385	2386	2387	2388	2389	2390	2391	2392	2393	2394	2395	2396	2397	2398	2399	2400	2401	2402	2403	2404	2405	2406	2407	2408	2409	2410	2411	2412	2413	2414	2415	2416	2417	2418	2419	2
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MX/x

-1.805	-.0968	.3389	.5371	.4716	.4220	.4561
.020			.3069	.3038	.2768	.2861
.023		.1841				
.040	-.0033					
.045		.1367				
.050	-.2315		.1112	.1037	.0937	.0848
.060			-.0304			
.085		.0268				
.066	.0370					
.094	-.2034					
.150			-.1878	-.2010	-.1727	-.1498
.163		.0000				
.177		-.0856				
.193	.0309					
.229	-.0842					
.250			-.2781	-.3184	-.3282	-.3340
.274		-.1801				
.339	-1.439					
.362	.0437					
.400		-.3300	-.3738	-.4356		-.5070
.422						
.497	.0000	.0000	-.3815	-.4548		-.4997
.550						
.565						
.600						
.650						
.700	-.1067			-.1334		
.725			-.0910			
.750						
.765		-.0723				
.775		-.0084	-.0044	-.1315		
.808						
.834	-.0410					
.850						
.857		-.0388	-.0079	-.0803	-.1196	
.865	.0042					
.900	-.0107		-.0337	.0293		-.2215
.905		-.0337	-.0385		-.0784	
.950		-.0465				
.955	-.3688					

ARC11-716 1A14 01+Y12+S12N25+AT10 UFFER WING

$$\text{ALPHA}(\theta) = -0.719 \quad \text{BETA}(\theta) = 4.279$$

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CF

Y/8W	X/CW	.2990	.3640	.4270	.5340	.6750	.7000	.8070
.020	-.2074	-.1219		.2219	.5177	.4229	.3964	.4522
.020					.2093	.2673	.2574	.2781
.025				.1268				
.040		-.0136						
.045				.0977				
.050	-.2637				.1064	.0787	.0983	.0834
.060					-.0255			
.065				.0187				
.066		.0321						
.094	-.2041							
.150			.5000		-.1333	-.2013	-.1539	-.1414
.163								
.177			.0323	-.0781				
.193								
.229	-.0560				-.2703	-.2981	-.3095	-.3232
.250								
.274			-.1385	-.1671				
.339								
.362	.0904				-.3382	-.4012		-.4940
.400				-.3059				
.402								
.497	.0000							
.550					-.3277	-.4245		
.555				.0200			-.4904	-.4774
.600								
.650								
.700	-.0610				-.0425			
.725						-.1550		
.750				-.0731			-.2770	-.3208
.760					.0175	-.0460		
.775				-.0009				
.808								
.834	-.0510				-.0013	-.0032	-.0977	
.850				-.0241				
.857								
.865	.0055							-.1574
.900	-.0034			-.0159	-.0300	.0406	.1075	
.935					-.0369			
.950								
.953				-.0217				
.965	-.0340							

(RBIU33)

DATE 10 JAN 75      TABULATED PRESSURE DATA - IA14A - VOL. 6      ARC11-716 IA14 OA-T12+S12M25+AT10 UPPER WING

ALPHA( 6 ) = -.730    BETA( 9 ) = 6.350

SECTION ( 1 ) UPPER WING      DEPENDENT VARIABLE CP

Y/SW	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7800
.0870		
.3745	-.1216	.1349
.5084	.4012	.3169
.3736		
.2720	.2703	.2342
.2447		
.3729		
.0561		
.1043	.0626	.0682
.0598		
-.0237		
.0317		
-.0084		
-.1916		
.094		
.150		
.163	.0000	
.177	-.0584	
.193	.0305	
.229	-.0479	
.230		
.274	-.1419	
.339	-.1146	
.362	.0391	
.400		
.402	-.2307	
.497		
.530	-.3136	-.4027
.563		
.600	-.8536	-.3524
.650	.0000	
.650		
.700	-.0382	
.725		
.750	-.0848	-.1590
.760		
.775	-.1814	-.2997
.806		
.834	-.1049	
.850	-.0255	-.0456
.857	-.0423	
.865		
.900	-.0429	-.0331
.924		
.950	-.0621	
.953		
.965	-.0601	-.1106
	-.0513	-.0007
	-.0617	.0981
	-.0576	
	-.0787	



TABLED PRESSURE DATA - 1A14A - VOL. 6

CVT36 (10) = - .730 CVT36 (9) = 0.110

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP					
Y/BW	.2890	.2640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	-.4304	-.0612	.1221	.3526	.5743	.3129	.3442
.020				.2133	.2587	.2250	.2100
.025				.0630			
.040		-.0192					
.045			.0703				
.050	-.3381			.0792	.0631	.0700	.0435
.060			.0361	-.0227			
.085							
.086		-.0050					
.094	-.2041						
.100		.0000		-.1438	-.1708	-.1622	-.1523
.163			-.0568				
.177		.0330					
.193							
.229	.0302			-.2594	-.2669	-.3011	-.3248
.230			-.1369				
.274		-.1259					
.339							
.362	.0575			-.3302	-.3683		-.4766
.400			-.2602				
.402				-.1362	-.3234		
.497	.0000		.0000				-.4698
.530						-.3576	
.565							
.600							
.650							
.700	-.0249			-.1190		-.1633	-.3020
.725							
.750			-.1227				
.760				-.0565	-.0540		
.775			-.0451				
.800							
.834	-.0392			-.0745	-.0540	-.0402	
.850			-.0751				
.937							
.969	-.0354			-.0082			-.0344
.970	-.0490		-.0610		-.0226		
.995				-.0866		.0635	
.999			-.0793				
.999	-.0924						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01053)

ARC11-716 1A14 01+712+912N29+AT10 UPPER WING

ALPHAO( 6) = -.750 BETAO (11) = 10.110

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6750 .7800 .8870
-.4820	-.0941	.0774	.2240 .3289 .2630 .2899
.029		.1386	.2280 .2042 .1898
.025		.0586	
.040	-.0558	.0410	
.045		.0422	.0762 .0521 .0387
.050	-.3550	-.0302	
.080		.0203	
.095	-.0245		
.086			
.094	-.1885		
.150	.0000	-.1253	-.1684 -.1650 -.1354
.163		-.0604	
.177	.0487		
.133			
.229	-.0302		
.250		-.2275	-.2676 -.2904 -.3017
.274	-.1376		
.339	-.1308		
.362	.0455	-.2906	-.3653
.400			-.4542
.402	-.2370		
.497		-.1387	-.2420
.550	.0000		
.565			
.600			
.650			
.700	-.0669	-.1189	-.1689
.725			
.750		-.1010	
.783		-.0632	-.0916
.775		-.0596	
.808			
.834	-.0591	-.0843	-.0381 -.0224
.850		-.0985	
.857			
.865	-.0383	-.0472	
.900	-.0777	-.0875	-.0238
.905		-.0883	.0751
.950		-.0859	
.955	-.1258		
.965			



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(081033)

ARC11-716 1A14 01+112+312N25+AT10 UPPER WING

ALPHA( 7 ) = 2.010 BETA( 1 ) = -10.080

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	.1546	.0573	.5697	.8142	.8031	.7851
.020					.3123	.3499	.3021
.025				.2473			.2776
.040			.0497				
.045				.1534			
.050	.0177			.0747	.0770	.0757	.0220
.060				-.0993			
.065			-.0273				
.066		.1019					
.094	.0086						
.150		.0000		-.3229	-.2977	-.2387	-.2162
.163							
.177			-.1848				
.193		-.0252					
.229	-.0249			-.4317	-.4475	-.4180	-.3826
.250			-.2864				
.274		-.2150					
.339							
.362	.0476			-.5032	-.5736		-.5327
.400			-.4164				
.402							
.497	.0000			-.5216	-.6076		
.590			.0000				
.565							-.5524
.670						-.5917	
.650				-.2022			
.700	-.3369			-.3639		-.5959	-.5521
.725							
.750			-.1691				
.760			-.1976	-.3417			
.775				-.0867			
.808							
.834	-.0235			-.0985	-.3948	-.4716	
.850			-.0369				
.857							
.865	.0991						-.4910
.900	.1074			-.0200		-.3125	
.905		.0210					
.950		.0286					-.3344
.953		.0413					
.965	.0211						

(R01U33)

DATE 10 JAN 75      TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 O1+T12+S12N23+AT10 UPPER WING

ALPHA( 7 ) = 2.000    BETA( 2 ) = -8.040

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
X/CW	Y/W		
.000	.2990	.3640	.4270
.020	.0494	.0284	.5375
.025		.7679	.7492
.040		.3204	.3232
.045		.0227	.2802
.050		.1353	.2556
.060	-.0106	.0821	.0644
.085		-.0934	.0636
.086		-.0297	.0093
.094		.0860	
.150	-.0370	-.3201	-.3001
.163		.0000	-.2444
.177		-.1958	-.2282
.193		-.0302	
.229	-.0393		
.250		-.4464	-.4465
.274		-.3087	-.4202
.339		-.2322	-.3893
.362	.0214	-.5333	-.5701
.400		-.4266	-.5556
.422		.0000	
.497	.0000	-.5544	-.6046
.500			
.565			
.600			
.650			
.700	-.3526	-.2933	-.5913
.725		-.2026	-.5556
.750			
.780		-.2010	-.5897
.775		-.4987	-.5553
.808		-.0792	
.834			
.850	-.0250	-.0966	-.4078
.857		-.0127	-.4307
.865			
.900	.0976	-.0546	
.905	.0965	-.3407	-.4942
.950		.0422	-.3207
.955		-.0501	
.965	.0110	.0409	



DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N25+AT10 UPPER WING (R81U33)

ALPHA( 7 ) = 2.050 BETA( 3 ) = -6.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	.0072	-.0161	.4948	.7616	.7075	.7005
.020				.3060	.3122	.2661	.2341
.025			.1988				
.040		-.0137					
.045			.1152				
.050	-.0514			.0696	.0369	.0904	-.0082
.060				-.1009			
.085			-.0486				
.086		.0385					
.094	-.0640			-.3226	-.3051	-.2506	-.2379
.150		.0000					
.163		-.0494					
.177			-.1970				
.193							
.229	-.0871			-.4347	-.4485	-.4248	-.3969
.250			-.3096				
.274		-.2345					
.339	.0032			-.5109	-.5709		-.5606
.362			-.4341				
.400				-.5364	-.6039		
.497	.0000		.0000			-.5554	
.550					-.5937		
.565				-.2038			
.600				-.2339		-.5527	-.5639
.650	-.3592						
.700			-.2015				
.725			-.0602		-.1869	-.4546	
.750							
.775							
.808							
.834	-.0211			-.1267	-.3966	-.4003	
.850			.0024				
.857							
.865	.0873			-.0907			-.4931
.900	.0776				-.3254		
.934		.0362					
.950			-.0576				-.3192
.953		.0234					
.965	-.0069						

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DATE 10 JAN 75      TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+812N25+AT10 UPPER WING

(R01U33)

ALPHA(°) = 1.920    BETA(°) = -3.990

SECTION (1) UPPER WING      DEPENDENT VARIABLE CP

X/CW	Y/BW	0.2900	0.3640	0.4270	0.5340	0.6730	0.7800	0.8870
.000	-.0387	-.0472	.4376	.7305	.6729	.6360	.6904	
.020			.1631	.3005	.2983	.2319	.2290	
.040	-.0308		.1095	.0716	.0483	.0441	-.0054	
.060	-.0924		-.0478	-.1000				
.080			.0275					
.100	-.1064		.0000	-.3126	-.3083	-.2497	-.2359	
.120			-.1919					
.140	-.0946		-.0507	-.4248	-.4463	-.4213	-.3964	
.160			-.2376					
.180	-.0023			-.5061	-.5617		-.5600	
.200			-.4313					
.220	.0000		.0000	-.5310	-.5947			
.240						-.5083	-.5946	
.260	-.3350			-.1922		-.5476	-.5663	
.280			-.1646					
.300			-.0135	-.1554	-.4302			
.320	-.0198			-.1136	-.3300	-.3854		
.340			.0127					
.360	.0728			-.0737			-.4767	
.380	.0957		.0254	-.1832				
.400			-.0261					
.420			.0029					
.440	-.0239							
.460								
.480								
.500								
.520								
.540								
.560								
.580								
.600								
.620								
.640								
.660								
.680								
.700								
.720								
.740								
.760								
.780								
.800								
.820								
.840								
.860								
.880								
.900								
.920								
.940								
.960								
.980								
1.000								



TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1033)

ARC11-716 IA14 01+T12+S12N23+AT10 UPPER WING

ALPHA( 7 ) = 1.920 BETAD ( 5 ) = -2.020

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.000	-.0753	-.0971	.4392
.020			.6786
.025			.6302
.040			.5829
.045			.5356
.050			.4883
.060			.4410
.065			.3937
.066			.3464
.094			.2991
.130			.2518
.163			.2045
.177			.1572
.193			.1099
.229			.0626
.250			.0153
.274			-.0320
.339			-.0797
.362			-.1274
.400			-.1751
.402			-.2228
.497			-.2705
.559			-.3182
.565			-.3659
.600			-.4136
.630			-.4613
.700			-.5090
.725			-.5567
.753			-.6044
.760			-.6521
.775			-.6998
.808			-.7475
.834			-.7952
.850			-.8429
.857			-.8906
.863			-.9383
.900			-.9860
.905			-.1037
.930			-.1514
.933			-.1991
.965			-.2468

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1033)

ARC11-716 1A14 01+712+912+25+AT10 UPPER WING

ALPHA(7) = 1.920 BETA(6) = .910

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/9W	X/CM	2090	3640	4270	5340	6730	7100	8870
.000	-.1106	-.1423	.3855	.6261	.5850	.5155	.5929	
.020			.2222	.2369	.1895	.2012		
.025			.1286					
.040		-.0570						
.045			.0656					
.050	-.1871		.0223	.0044	-.0049	-.0223		
.060			-.1261					
.085			-.0704					
.086		-.0920						
.094	-.1808			-.3061	-.3179	-.2721	-.2412	
.150		.0090						
.165			-.1907					
.177		-.0465						
.195								
.229	-.1035			-.3814	-.4385	-.4307	-.4011	
.250			-.2671					
.274		-.2049						
.339								
.362	-.0070			-.4371	-.5344		-.5603	
.400			-.3821					
.497	.0000			-.4651	-.5397			
.550			.0000					
.565								
.600								
.650								
.700	-.2763			-.1873	-.1761	-.5733	-.5346	
.725								
.750								
.760			-.1513					
.775				-.1143	-.2415			
.808			-.0085					
.834	-.0280			-.0440	-.2072	-.3202		
.850			-.0255					
.857								
.865	.0186							
.900	-.0018			-.0322				-.3680
.905			-.0301		-.0634			
.950				-.0274				-.2805
.955			-.0531					
.965	-.0783							



(RB1V33)

ARC11-716 1A14 01+112+S12N25+AT10 UPPER WING

$$\text{ALEMAO}(7) = 1.925 \quad \text{BETAO}(7) = 2.050$$

DEPENDENT VARIABLE CF

SECTION (1) UPPER WING.

	.2997	.3649	.4279	.5340	.6739	.7899	.8879
--	-------	-------	-------	-------	-------	-------	-------

**MJ/x**

.009	-.2072	-.1774	.3113	.5709	.5252	.4895	-.5275
.020							
.025							
.042			.0342				
.045		-.0475					
.050	-.2248		.0307				
.053				.0976	-.0037	-.0156	-.0448
							-.1266

10027

	- .2235	- .2476	- .3022	- .2689	- .2500
.094					
.500					

.0030	
.153	
.177	
	-.1677

.229	-.0961
.253	-.3721
	-.4093
	-.4234
	-.4110

**-.2395**

339  
-1079  
362  
-0146

-.3690

Account	1955	1956	1957	1958	1959	1960	1961	1962	1963	1964	1965	1966	1967	1968	1969	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	2101	2102	2103	2104	2105	2106	2107	2108	2109	2110	2111	2112	2113	2114	2115	2116	2117	2118	2119	2120	2121	2122	2123	2124	2125	2126	2127	2128	2129	2130	2131	2132	2133	2134	2135	2136	2137	2138	2139	2140	2141	2142	2143	2144	2145	2146	2147	2148	2149	2150	2151	2152	2153	2154	2155	2156	2157	2158	2159	2160	2161	2162	2163	2164	2165	2166	2167	2168	2169	2170	2171	2172	2173	2174	2175	2176	2177	2178	2179	2180	2181	2182	2183	2184	2185	2186	2187	2188	2189	2190	2191	2192	2193	2194	2195	2196	2197	2198	2199	2200	2201	2202	2203	2204	2205	2206	2207	2208	2209	2210	2211	2212	2213	2214	2215	2216	2217	2218	2219	2220	2221	2222	2223	2224	2225	2226	2227	2228	2229	2230	2231	2232	2233	2234	2235	2236	2237	2238	2239	2240	2241	2242	2243	2244	2245	2246	2247	2248	2249	2250	2251	2252	2253	2254	2255	2256	2257	2258	2259	2260	2261	2262	2263	2264	2265	2266	2267	2268	2269	2270	2271	2272	2273	2274	2275	2276	2277	2278	2279	2280	2281	2282	2283	2284	2285	2286	2287	2288	2289	2290	2291	2292	2293	2294	2295	2296	2297	2298	2299	2300	2301	2302	2303	2304	2305	2306	2307	2308	2309	2310	2311	2312	2313	2314	2315	2316	2317	2318	2319	2320	2321	2322	2323	2324	2325	2326	2327	2328	2329	2330	2331	2332	2333	2334	2335	2336	2337	2338	2339	2340	2341	2342	2343	2344	2345	2346	2347	2348	2349	2350	2351	2352	2353	2354	2355	2356	2357	2358	2359	2360	2361	2362</
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.650	- .3639
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- .2412	

.750	-.4726	-.4932
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0.850	0.0337	0.1033	0.2713
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(R81033)

DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+512M25+AT10 UPPER WING

ALPHA (°) = 1.900 BETA (°) = 4.080

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
Y/CW							
.000	-.2967	-.2128	.1870	.5346	.4723	.4737	.5342
.020				.1721	.1815	.1639	.1495
.025			.0341				
.040		-.0564					
.045			.0020				
.050	-.2651			-.0023	-.0248	-.0150	-.0500
.060				-.1316			
.085			-.0720				
.086		-.0102					
.094	-.2439			-.2773	-.3099	-.2584	-.2457
.150		.0000					
.163			-.1504				
.177							
.193		-.0092					
.229	-.0775			-.3619	-.3071	-.4130	-.4067
.250			-.2311				
.274		-.1755					
.339				-.4393	-.4765		-.5355
.362	.0228		-.3575				
.400				-.4222	-.4879		
.402			.0000				-.5436
.497	.0000					-.5389	
.530				-.1425		-.3935	-.4489
.563							
.600							
.650	-.1870						
.700							
.725							
.750			-.1388				
.760				-.0379	-.1743		
.775			-.0233				
.808							
.834	-.0652			-.0382	-.0844	-.2403	
.890			-.0492				
.885		-.0138		-.0588			-.2745
.900	-.0228						
.905			-.0494		.0144		
.950				-.0617			-.1157
.955			-.0583				
.985	-.0642						



DATE 10 JAN 75  
 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N23+AT10 UPPER WING  
 (R81U33)

ALPHA ( 7 ) = 2.040 BETA ( 9 ) = 6.060

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/W	.2990	.3640	.4270	.5340 .6750 .7800 .8870
X/CW				
.000	-.4096	-.1704	.0787	.5049 .4712 .4193 .4551
.020				.1638 .1834 .1238 .1120
.025			.0035	
.040		-.1107		
.045			-.0109	
.050	-.3108		-.0013	-.0253 -.0450 -.0778
.080			-.1305	
.095			-.0393	
.086		-.0808		
.094	-.2435			
.150		.0000		-.2648 -.3060 -.2780 -.2598
.163			-.1233	
.177		-.0033		
.193				
.229	-.0588			-.3431 -.3878 -.4210 -.4187
.250			-.1946	
.274		-.1496		
.339				
.362	.0216			-.4078 -.4707 -.5604
.400			-.2918	
.402				
.497	.0000		.0000	-.3100 -.4577
.550				
.565				
.600				
.650				
.700	-.1031			-.5339
.725			-.1754	
.750			-.1557	
.760				-.5327 -.3945
.775		-.1714		
.800		-.0800		
.834	-.0892			
.850			-.0868	-.0641 -.1478
.857		-.1005		
.865	-.0801			
.900	-.0725		-.1034	-.2203
.905			-.0906	
.953			-.1034	-.0022 .0340
.955			-.0987	
.965	-.1162			

DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI4A - VOL. 6

(RB1U33)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

$$\text{ALPHA}(7) = 2.039 \quad \text{BETA}(10) = 0.119$$

SECTION ( UPPER WING

Y/YW	.2990	.3640	.4270	.5340	.6730	.7000	.8870
X/CW							
.000	-.4974	-.0951	.0843	.2786	.4307	.3913	.4039
.020			.0216	.0686	.1682	.1102	.0841
.040		-.0416					
.045			.0091				
.060	-.3613			-.0445	-.0270	-.0333	-.0940
.085			-.0352	-.1307			
.086		-.0377					
.094	-.2998						
.150		.0000		-.2341	-.2893	-.2774	-.2684
.163							
.177		.0113	-.1246				
.229	.0292			-.3249	-.3634	-.4017	-.4192
.250			-.1933				
.274		-.1615					
.362	.0367		-.3004	-.3874	-.4598		-.5335
.400							
.402	.0000			-.2141	-.4469		-.5364
.497						-.5240	
.530							
.555							
.500			.0000				
.650							
.700	-.0537			-.1373	-.1734		
.725							
.750		-.1585				-.2504	-.3524
.760				-.1018	-.1038		
.775			-.0949				
.800							
.834	-.0932						
.850			-.1172	-.1205	-.0787	-.1037	
.865	-.0685						
.900	-.0906			-.1337	-.0485		-.2067
.935		-.1078					
.950			-.1209	-.1302		.0229	
.953							
.965	-.1310						

(R81033)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA(7) = 2.330 BETA(11) = 10.130

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7800 .8870
X/CW					
.000	-.5190	-.1252	.0300	.1505	.3134 .3411 .3876
.025				-.0114	.0823 .0659 .0480
.045			.0036		
.065		-.0710			
.085	-.4357		-.0158	-.0801	-.0752 -.0907 -.1123
.105				-.1329	
.125			-.0508		
.145		-.0757			
.165	-.2870			-.2060	-.2915 -.2940 -.2663
.185		.0000			
.205			-.1478		
.225		.0166			
.245	-.3171			-.2860	-.3625 -.4039 -.4162
.265		-.2091			
.285			-.1670		
.305	.0306			-.3294	-.4441 -.5430
.325		-.3137			
.345	.0700			-.1897	-.3628
.365		.0000			
.385					-.5440
.405	-.1115			-.4139	
.425			-.1835		
.445		-.1373			
.465				-.2120	-.3283
.485	.725		-.1560		
.505	.750			-.1202	-.0979
.525	.760		-.1049		
.545	.775				
.565	.870				
.585		-.1078			
.605	.834			-.1220	-.0903 -.0613
.625	.850		-.1391		
.645	.865				
.665	.900			-.1348	-.0646 -.1670
.685	.925		-.1337		
.705	.950			-.1228	.0463
.725	.955		-.1331		
.745	.965				
.765		-.1853			

(R91 U33)

TRAINING VIDEO PRESSURE DATA - 1A14A - VOL. 6

ABC11-716 1A14 01+112+S12N25+AT10 UFFER WING

ALGMA(0) = 4.333      BETAG(1) = -9.983

DEPENDENT VARIABLE C<sub>P</sub>

178W	2990	3640	4270	5340	6730	7800	8670
K/CW							
.003	.1063	.0366	.6035	.8568	.8295	.7959	.7813
.020			.1657	.2422	.2627	.1792	.1240
.075		.0531					
.045			.0762				
.050	.0313			.0031	-.0064	-.0319	-.1130
.080				-.1731			
.085			-.1045				
.086		.0596					
.094	.0223						
.150				-.4051	-.3678	-.3321	-.3195
.163		.0010					
.177			-.2653				
.193		-.0544					
.229	-.0124			-.5223	-.5109	-.4876	-.4545
.270							
.274			-.3586				
.319		-.2651					
.362	.0480			-.5825	-.6359		-.6070
.403			-.4523				
.432							
.497	.0010			-.5917	-.6312		
.550			.0000				
.565							
.600							
.650							
.700	-.3678			-.2071		-.6185	-.6016
.725				-.4984			
.750							
.760			-.1797				
.775				-.2844	-.5290		
.808			-.1175				
.834	-.0694						
.850			-.0843	-.1388	-.4792	-.4881	
.857							
.865	.0570			-.0475			-.550
.900	.0772				-.3990		
.905			-.0464				
.950				-.0552		-.3883	
.953			-.0046				
.965	.0032						

(R81033)

DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI4A - VOL. 6  
 ARC11-7.6 IAI4 01+712+312M23+AT10 UPPER WING

ALPHA( 8 ) = 4.200 BETA( 2 ) = -6.090

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP		
Y/BW	.2995	.3640	.4275
X/CW			
.000	.1542	.0043	.5310
.020			.8186
.025			.7919
.040		.1478	.7613
.045		.0012	.7522
.050			.7322
.060	-.0060		.2136
.085			.2446
.086			.1622
.094	-.0155	.0373	.1120
.150		.0000	
.163			
.177			
.193			
.229	-.0556		
.250			
.274			
.339			
.362	.0021		
.400			
.402			
.497	.0000		
.550			
.565			
.600			
.650			
.700	-.3834		
.725			
.750			
.760			
.775			
.800			
.834	-.0499		
.850			
.857			
.865			
.900			
.935			
.950			
.953			
.965			

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 3  
 ARC11-716 1A14 01+112+512M25+AT10 UPPER WING  
 (R01J33)

ALPHA(0) = 4.200 BETA(3) = -5.990

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP						
Y/9W	.2980	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0804	-.0426	.5013	.7639	.7464	.7212	.7133
.020		.1294		.1826	.2176	.1497	.0842
.040		-.0439	.0390				
.060				-.0316	-.0448	-.0641	-.1431
.080					-.1946		
.100							
.120							
.140							
.160							
.180							
.200							
.220							
.240							
.260							
.280							
.300							
.320							
.340							
.360							
.380							
.400							
.420							
.440							
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.000							



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A11A - VOL. 6  
 ARC11-716 1A14 Q1+T12+S12N2S+AT110 UPPER WING  
 (RB1033)

ALPHA: 8) = 4.200 BETA: (4) = -3.970

SECTION: 1 UPPER WING		DEPENDENT VARIABLE CP	
Y/B			
.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW			
.000	.0209	-.1026	.4625 .7125 .7302 .6741 .6731
.020			.1843 .1945 .1276 .0693
.025			.1073
.040		-.0789	
.045			.0206
.050	-.1024		-.0295 -.0580 -.0732 -.1531
.060			-.1908
.065		-.1264	
.066		-.0259	
.094	-.1036		-.3985 -.3835 -.3510 -.3481
.150		.0000	
.163		-.2549	
.177		-.0946	
.193			
.229	-.1203		-.5013 -.5149 -.4973 -.4788
.250		-.3677	
.274		-.2850	
.339			
.362	-.0322		-.5749 -.6290 -.6169
.400		-.4740	
.402			-.5884 -.6542
.497	.0000		.0000
.550			
.565			
.600			
.650			-.6107
.700	-.3788		-.8402
.725		-.2875	-.2001
.750			-.5850 -.6141
.760		-.2296	
.775		-.1775	-.5294
.808		-.0915	
.834	-.0215		
.850		-.1213	-.4572 -.4836
.857		-.0282	
.865	.0392		
.900	.0454		-.1160
.905		.0034	-.3331
.950			-.1094
.953		-.0181	-.3904
.965	-.0489		



(R81053)

DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14 - VOL. 6  
ARC11-716 IAI14 OL+T12+S12M25+AT10 UPPER WING

ALPHAD( 8 ) = 4.220 BETAD ( 5 ) = -2.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/SW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0362	-.1618	.4213	.6927	.6456	.6288	.6389
.020				.1619	.1785	.1127	.0598
.025			.09 J				
.040		-.0905					
.045			.0168				
.050	-.1395			-.0457	-.0660	-.0825	-.1562
.060				-.2036			
.085			-.1349				
.086		-.0365					
.094	-.1412			-.4059	-.3905	-.3534	-.3465
.130		.0000					
.163			-.2517				
.177		-.1004					
.193							
.229	-.1211			-.4918	-.5162	-.4981	-.4621
.250			-.3557				
.274		-.2807					
.339							
.362	-.0465			-.5580	-.6243		-.6164
.400			-.4422				
.402							
.497	.0000			-.5454	-.6414		
.550			.0090				
.565							
.600							
.650						-.6346	
.700	-.3376			-.2127	-.1939		
.725						-.5538	-.6071
.750			-.1921				
.760				-.1670	-.4929		
.775			-.0586				
.808							
.834	-.0189			-.1308	-.3999	-.4356	
.850			-.0345				
.857							
.865	.0340			-.1206			-.4895
.900	.0200						
.905			-.0191	-.1072			
.950			-.0932				
.955			-.0397				
.965	-.0710						



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01033)

DATE 10 JAN 75

ARC11-716 1A14 OL+T12+512N25+AT10 UPPER WING

ALPHA ( 8 ) = 4.240 BETA ( 6 ) = -.070

SECTION 1 UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1339	-.2287	.3542	.6443	.5987	.5775	.6012
.020				.1131	.1446	.0884	.0438
.025			.0428				
.040		-.1047					
.045			-.0246				
.050	-.1879			-.0777	-.0895	-.1003	-.1680
.060				-.2241			
.085			-.1801				
.086		-.0393					
.094	-.1885						
.150		.0000		-.4023	-.3978	-.3629	-.3532
.163			-.2504				
.177		-.0971					
.193							
.229	-.1251			-.4753	-.5212	-.5016	-.4871
.250			-.3229				
.274		-.2481					
.339	-.0293			-.4975	-.6105		-.6169
.362							
.400			-.4218				
.402	.0000			-.5096	-.6070		
.497	.0000		.0000				-.6058
.550							
.565							
.600							
.650							
.700	-.3466						
.725				-.1898			
.750							
.760			-.1840				
.775				-.1608	-.3938		
.808			-.0526				
.834	-.0298						
.850				-.1242	-.2003	-.4501	
.857			-.0467				
.865	-.0012						
.900	-.0131			-.0594			-.4567
.905			-.0415		-.1230		
.950				-.0740			-.3837
.953			-.0657				
.965	-.0979						

(R81033)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+712+S12H25+AT10 UPPER WING

ALPHAD (°) = 4.220 BETA (°) = 1.990		DEPENDENT VARIABLE CP	
SECTION (1) UPPER WING			
X/CW	Y/BW		
.000	-.2248	.2980	.5717
.020		.0674	.1264
.025		.0040	.0682
.040	-.0994		
.045		-.0633	
.050	-.2242		-.0955
.060			-.2246
.085		-.1697	
.086		-.0448	
.094	-.2267		-.3751
.150			-.3917
.163		.0030	-.3628
.177		-.2334	-.3511
.193		-.0803	
.229	-.1087		
.250			-.4269
.274		-.2956	-.5005
.339		-.2236	-.4955
.362	-.0114		-.4819
.400			-.4676
.402		-.4954	-.5735
.497	.0000		
.550			-.4906
.565		.0000	-.5751
.600			
.650			-.5946
.700	-.3339		-.6786
.725			-.1910
.750		-.1995	
.760			-.5650
.775		-.1862	-.5838
.808		-.0583	
.834	-.0339		
.850			-.1115
.857		-.0669	-.2083
.865	-.0190		-.4281
.900	-.0387		
.905		-.0847	
.950		-.0702	-.1171
.953		-.0716	-.3242
.965	-.1222	-.0959	



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01-112-512825-AT10 UPPER WING (R81033)

ALPHA( 8 ) = 4.430 BETA( 8 ) = 4.103

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
1/CW				
.000	-.3257	-.2677	.1169	.5190 .5047 .5320 .5227
.020				.0475 .0814 .0528 .0322
.025			-.0620	
.040		-.1416		
.045		-.1049		
.050	-.2728		-.1041	-.1300
.060			-.2255	
.065		-.1575		
.066		-.0751		
.094	-.2557			
.150		.0000	-.3681	-.3360
.163			-.3613	-.3634
.177		-.2182		
.193		-.0610		
.229	-.0908		-.4130	-.4793
.250			-.4943	-.4917
.274		-.2787		
.339		-.2151		
.362	-.0018		-.4618	-.5516
.400				-.6148
.402		-.3909		
.497	.0000		-.4524	-.5474
.550		.0000		
.565				
.600				-.5926
.650			-.6024	
.700	-.2987		-.1773	
.725			-.1825	
.750				-.5556
.775		-.1817	-.1014	-.8060
.800		-.0456		
.834	-.0361		-.0635	-.1850
.850			-.3644	
.857		-.0688		
.865	-.3196			
.900	-.0301	-.0719	-.0817	-.4006
.925		-.0727		
.950		-.0722		-.1757
.953		-.0911		
.965	-.0908			

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DATE 10 JAN 79

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBU33)

ALPHA(8) = 4.410 SETAD (9) = 5.060

ARC11-716 1A14 01+112+512N25+AT10 UPPER WING

SECTION - 11 UPPER WING

DEPENDENT VARIABLE CP

Y/B 2990 3640 4270 5340 6730 7800 8870

Z/C

.000	-.4542	-.2100	.0153	.4071	.4959	.4526	.4804
.020				.0247	.0969	.0223	-.0059
.025			-.0606				
.040		-.1522	-.0682				
.045				-.1041	-.1156	-.1430	-.1976
.050	-.3414			-.2105			
.090			-.0926				
.085		-.1212					
.096	-.2867			-.3384	-.3905	-.3677	-.3636
.094							
.150		.0000					
.163		-.0658	-.1732				
.177							
.193							
.229	-.0714			-.3791	-.4769	-.4914	-.4885
.250			-.2385				
.274		-.1936					
.339							
.362	.0056			-.4356	-.5352		-.6114
.400			-.3269				
.402				-.3585	-.5279		
.497	.0000		.0000				-.5830
.550						-.5833	
.565					-.1826		
.600				-.1781		-.4559	-.5185
.650	-.1763						
.700			-.1926				
.725				-.1083	-.2079		
.750			-.1123				
.760							
.775							
.808							
.834	-.1041			-.1014	-.1269	-.2110	
.850			-.1323				
.897							
.865	-.0787			-.1133			-.3626
.900	-.0674		-.1232		-.0297		
.905				-.1151		-.1134	
.950			-.1310				
.955							
.965	-.1428						



ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

$$\text{ALPHA}(0) = 4.410 \quad \text{BETA}(12) = 0.150$$

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/W	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.5350	-.1234	.0314	.2098	.4244	.4158	.4158	.4158
.020	.025		-.0678	-.0220	-.0638	.0060	-.0443	
.040	.045			-.0439				
.060	-.4302				-.1462	-.1278	-.1847	-.2210
.080				-.0822	-.2109			
.100	.085		-.0474					
.120	-.5335				-.2998	-.3757	-.3795	-.3754
.140	.094		.0000	-.1662				
.160	.150		-.0646					
.180	.170							
.200	.229	.0241		-.2320	-.3528	-.4434	-.4927	-.4950
.220	.250							
.240	.274		-.1977		-.4011	-.5195		-.6128
.260	.362	.0244		-.3254				
.280	.400				-.2269	-.4951		
.300	.402	.0000		.0000				-.5824
.320	.497					-.1871	-.5724	
.340	.550				-.1713		-.3333	-.4903
.360	.565			-.1979				
.380	.600			-.1267	-.1761			
.400	.650			-.1298				
.420	.700	-.0783						
.440	.725							
.460	.750							
.480	.760							
.500	.775							
.520	.800							
.540	.834	-.1197						
.560	.850							
.580	.857			-.1924	-.1417	-.1152	-.2135	
.600	.865	-.0936						
.620	.900	-.1129			-.1495			-.3094
.640	.905			-.1400	-.0680			
.660	.950				-.1448		-.0443	
.680	.953			-.1504				
.700	.965	-.1503						

(R81033)

TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01-T12-S12N25+AT10 UPPER WING

ALPHA(8) = 4.390 BETA(11) = 15.140

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.9510	-.1425	-.0173	.0774	.2761	.3412	.4051
.020				-.1137	-.0319	-.0426	-.0478
.025			-.0402				
.040		-.0871					
.045			-.0583				
.050	-.4887			-.1667	-.1911	-.1878	-.2200
.060				-.2054			
.085			-.0934				
.086		-.0801					
.094	-.3949						
.150		.0000		-.2532	-.3721	-.3738	-.3640
.163							
.177			-.1864				
.193		-.0491					
.229	-.0179			-.2909	-.4203	-.4839	-.4846
.250			-.2530				
.274		-.1819					
.339							
.352	.0146			-.3300	-.4905		-.6011
.400			-.3365				
.492				-.1837	-.3599		
.497	.0000						
.559			.0000				
.565							
.600							
.690							-.5683
.700	-.1123			-.1565	-.1922	-.4258	
.725							
.750			-.1806				
.760				-.1352	-.1350		
.775			-.1310				
.808							
.834	-.1330			-.1306	-.1162	-.1164	
.850			-.1544				
.857							
.865	-.1071						
.900	-.1322			-.1378	-.0936		-.2270
.905			-.1458				
.930				-.1296		.0119	
.933			-.1375				
.965	-.1838						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+112+S12N25+AT10 UPPER WING (RB1033)

ALPHA ( 9 ) = 6.340 BETA ( 1 ) = -9.960

SECTION 1 ( UPPER WING ) DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.1865	.0483	.6085
.020		.8647
.025		.8272
.040		.7775
.045		.7456
.050		.7141
.055		.6826
.060		.6511
.065		.6196
.070		.5881
.075		.5566
.080		.5251
.085		.4936
.090		.4621
.095		.4306
.100		.3991
.105		.3676
.110		.3361
.115		.3046
.120		.2731
.125		.2416
.130		.2101
.135		.1786
.140		.1471
.145		.1156
.150		.0841
.155		.0526
.160		.0211
.165		-.0104
.170		-.0419
.175		-.0734
.180		-.1049
.185		-.1364
.190		-.1679
.195		-.1994
.200		-.2309
.205		-.2624
.210		-.2939
.215		-.3254
.220		-.3569
.225		-.3884
.230		-.4199
.235		-.4514
.240		-.4829
.245		-.5144
.250		-.5459
.255		-.5774
.260		-.6089
.265		-.6404
.270		-.6719
.275		-.7034
.280		-.7349
.285		-.7664
.290		-.7979
.295		-.8294
.300		-.8609
.305		-.8924
.310		-.9239
.315		-.9554
.320		-.9869
.325		-1.0184
.330		-1.0499
.335		-1.0814
.340		-1.1129
.345		-1.1444
.350		-1.1759
.355		-1.2074
.360		-1.2389
.365		-1.2704
.370		-1.3019
.375		-1.3334
.380		-1.3649
.385		-1.3964
.390		-1.4279
.395		-1.4594
.400		-1.4909
.405		-1.5224
.410		-1.5539
.415		-1.5854
.420		-1.6169
.425		-1.6484
.430		-1.6799
.435		-1.7114
.440		-1.7429
.445		-1.7744
.450		-1.8059
.455		-1.8374
.460		-1.8689
.465		-1.9004
.470		-1.9319
.475		-1.9634
.480		-1.9949
.485		-2.0264
.490		-2.0579
.495		-2.0894
.500		-2.1209
.505		-2.1524
.510		-2.1839
.515		-2.2154
.520		-2.2469
.525		-2.2784
.530		-2.3099
.535		-2.3414
.540		-2.3729
.545		-2.4044
.550		-2.4359
.555		-2.4674
.560		-2.4989
.565		-2.5304
.570		-2.5619
.575		-2.5934
.580		-2.6249
.585		-2.6564
.590		-2.6879
.595		-2.7194
.600		-2.7509
.605		-2.7824
.610		-2.8139
.615		-2.8454
.620		-2.8769
.625		-2.9084
.630		-2.9399
.635		-2.9714
.640		-3.0029
.645		-3.0344
.650		-3.0659
.655		-3.0974
.660		-3.1289
.665		-3.1604
.670		-3.1919
.675		-3.2234
.680		-3.2549
.685		-3.2864
.690		-3.3179
.695		-3.3494
.700		-3.3809
.705		-3.4124
.710		-3.4439
.715		-3.4754
.720		-3.5069
.725		-3.5384
.730		-3.5699
.735		-3.6014
.740		-3.6329
.745		-3.6644
.750		-3.6959
.755		-3.7274
.760		-3.7589
.765		-3.7904
.770		-3.8219
.775		-3.8534
.780		-3.8849
.785		-3.9164
.790		-3.9479
.795		-3.9794
.800		-4.0109
.805		-4.0424
.810		-4.0739
.815		-4.1054
.820		-4.1369
.825		-4.1684
.830		-4.1999
.835		-4.2314
.840		-4.2629
.845		-4.2944
.850		-4.3259
.855		-4.3574
.860		-4.3889
.865		-4.4204
.870		-4.4519
.875		-4.4834
.880		-4.5149
.885		-4.5464
.890		-4.5779
.895		-4.6094
.900		-4.6409
.905		-4.6724
.910		-4.7039
.915		-4.7354
.920		-4.7669
.925		-4.7984
.930		-4.8299
.935		-4.8614
.940		-4.8929
.945		-4.9244
.950		-4.9559
.955		-4.9874
.960		-5.0189
.965		-5.0504

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(R01033)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 OL+T12+312N25+AT10 UPPER WING

ALPHA( 9) = 6.380 BETAD ( 2) = -7.970

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
V/BW	.2993	.3640	.4210	.5340 .6730 .7800 .8870
X/CW				
.000	.1494	-.0103	.5624	.8311 .7893 .7403 .7115
.020			.0999	.1267 .1541 .0487 -.0103
.040		-.0040		
.060			-.0092	
.080	-.0087			-.0901 -.1131 -.1478 -.2362
.100			-.1905	-.2585
.120		.0113		
.140	-.0204			-.4782 -.4413 -.4185 -.4133
.160		.0000		
.180			-.3551	
.200		-.1359		
.220	-.0563			-.5894 -.5716 -.5546 -.5269
.240			-.4392	
.260		-.3229		
.280	.0053			
.300			-.5141	-.6496 -.6809 -.6525
.320		.0000		
.340				-.6561 -.6515
.360				
.380				
.400				
.420				
.440				
.460				
.480				
.500				
.520				
.540				
.560				
.580				
.600				
.620				
.640				
.660				
.680				
.700				
.720				
.740				
.760				
.780				
.800				
.820				
.840				
.860				
.880				
.900				
.920				
.940				
.960				
.980				
.995				



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARCL1-716 1A14 01-112-912N25+AT10 UPPER WING (R81033)

ALPHA( 9 ) = 5.980 BETAD ( 3 ) = -5.000

SECTION 1 UPPER WING DEPENDENT VARIABLE CP

X/CW .2990 .3640 .4270 .5340 .6730 .7800 .8970

.000	.0517	-.0531	.4989	.7830	.7555	.7071	.6823
.020				.1115	.1488	.0488	-.0169
.040			.0718				
.060		-.0481					
.080			-.0408				
.100	-.0506			-.1022	-.1146	-.1486	-.2363
.120				-.2637			
.140			-.2057				
.160		-.0217					
.180	-.0374						
.200		.0000		-.4753	-.4392	-.4149	-.4132
.220		-.1422					
.240	-.0938						
.260			-.4204				
.280		-.3094		-.5696	-.5676	-.5499	-.5261
.300	-.0364						
.320			-.4639				
.340				-.8217	-.6749		-.6513
.360							
.380			.0030				
.400				-.6238	-.6323		
.420							
.440							
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
1.000							

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DATE 10 JAN 75

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U33)

ARC11-716 1A14 01+712+512M25+AT10 UPPER WING

ALPHA(9) = 5.990 BETA(4) = -4.010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

V/BW	.2990	.3640	.4270	.5000	.5730	.7800	.8870
X/CM							
.000	.0290	-.1167	.4345	.7412	.7146	.5700	.6450
.020			.0909	.1306	.0376	-.0349	
.040			.0424				
.060			-.0936				
.080			-.0479				
.100			-.1095	-.1248	-.1562	-.2470	
.120			-.2050				
.140			-.1915				
.160			-.0651				
.180			-.0990				
.200			.0000				
.220			-.3197				
.240			-.1419				
.260			-.1240				
.280			-.4085				
.300			-.3095				
.320			-.0506				
.340			.362				
.360			.402				
.380			.497				
.400			.550				
.420			.56				
.440			.600				
.460			.650				
.480			.700				
.500			.725				
.520			.750				
.540			.775				
.560			.800				
.580			.834				
.600			.890				
.620			.857				
.640			.865				
.660			.900				
.680			.905				
.700			.950				
.720			.955				
.740			.965				
.760			.960				
.780			.965				
.800			.965				
.820			.965				
.840			.965				
.860			.965				
.880			.965				
.900			.965				
.920			.965				
.940			.965				
.960			.965				
.980			.965				
.990			.965				



(R81033)

DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI44 - VOL. 2  
ARC11-7:6 1A'4 01-112+512N25-AT10 UPPER WING

ALPHA (9) = 6.010 BETA (5) = -2.080		DEPENDENT VARIABLE CP	
SECTION (1) UPPER WING			
Y/BW			
.2990	.3640	.4270	.5340 .6730 .7800 .8870
Y/CW			
.000	-.0474	-.1894	.3931 .6906 .6693 .6280 .6109
.020			.0628 .1050 .0161 -.0549
.025		.0256	
.040		.1349	
.045		-.0636	
.050	-.1435		-.1264 -.1426 -.1727 -.2642
.060		-.2743	
.085		-.1971	
.086	-.0821		
.094	-.1398		-.4695 -.4461 -.4264 -.4279
.130		.0200	
.165		-.3208	
.177		-.1391	
.193	-.1458		-.5391 -.5686 -.5928 -.5390
.229			
.250		-.4099	
.274		-.5111	
.339	-.0572		-.5925 -.6673 -.6599
.362			
.400		-.4794	
.402	.0200		-.5780 -.6518
.497		.0000	
.550			-.6354
.585			-.6323
.600			
.630	-.3862		-.2039
.700		-.2275	
.725			-.5991 -.6146
.750		-.2546	
.760		-.1717 .5441	
.775		-.1208	
.806			
.834	-.0209		-.1470 -.5579 -.5084
.850		-.0682	
.857			
.865	.0190		-.1386
.900	.0094		-.3811
.905		-.0308	-.1260
.950		-.0485	-.4267
.953			
.965	-.0819		

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DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

ARC11-716 IA14 01+712+512N25+AT10 UPPER WING (R81U33)

ALPHA( 9 ) = 0.020 BETA( 6 ) = .030

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP		
Y/BW	.2590	.3640	.4270
X/CW	.5340	.6730	.7600
.070	-.1331	-.2676	.3172
.020			.6293
.025			.6129
.040			.5804
.045			.5773
.050			.0320
.060			.0716
.065			-.0036
.066			-.0676
.094			-.0169
.150			-.0169
.163			-.0923
.177			-.0823
.193			-.1441
.229			-.1648
.250			-.1883
.274			-.2697
.339			-.2856
.362			-.2193
.400			-.0793
.402			-.0793
.487			-.1894
.550			-.150
.565			.0000
.600			.0000
.650			.0000
.700			.0000
.725			.0000
.750			.0000
.760			.0000
.775			.0000
.808			.0000
.834			.0000
.850			.0000
.857			.0000
.865			.0000
.900			.0000
.905			.0000
.930			.0000
.933			.0000
.965			.0000



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+112+512N23+AT10 UPPER WING (R61033)

ALPHA( 9 ) = 6.010 BETAD ( 7 ) = 2.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.020	-.2333	-.4161	.2079	.3762	.5599	.5364	.5380
.025			-.0116	.0461	-.0240	-.0832	
.045		-.1370	-.0696				
.050			-.1386				
.080	-.2345		-.1720	-.1750	-.2015	-.2815	
.085			-.2965				
.086		-.0788	-.2400				
.094	-.2305			-.4596	-.4302	-.4335	-.4378
.150		.0000					
.165			-.2659				
.177		-.1191					
.193	-.1269			-.5182	-.5610	-.5535	-.5469
.229		-.3300					
.250		-.2469		-.5369	-.6453		-.8324
.274			-.4312				
.339	-.0299		.0000	-.5449	-.6333		-.6312
.362						-.6427	
.400				-.2379	-.2026		
.402	.0000						
.497				-.2244	-.3662		
.530				-.0892			
.555							
.600							
.650							
.700	-.3851			-.1769	-.2052	-.4850	
.725							
.750							
.760							
.775							
.808							
.834	-.0367						
.850							
.857							
.865	-.0283			-.1492			-.4636
.900	-.0476			-.0811	-.1743		
.905				-.1250		-.4131	
.950							
.953				-.1076			
.965	-.1324						

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA ( 9 ) = 3.990 BETA ( 8 ) = 4.080

SNIN 6346718 ; 4711555

DEPENDENT VARIABLE CP

	1960	1961	1962	1963	1964	1965	1966	1967	1968	1969	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	2101	2102	2103	2104	2105	2106	2107	2108	2109	2110	2111	2112	2113	2114	2115	2116	2117	2118	2119	2120	2121	2122	2123	2124	2125	2126	2127	2128	2129	2130	2131	2132	2133	2134	2135	2136	2137	2138	2139	2140	2141	2142	2143	2144	2145	2146	2147	2148	2149	2150	2151	2152	2153	2154	2155	2156	2157	2158	2159	2160	2161	2162	2163	2164	2165	2166	2167	2168	2169	2170	2171	2172	2173	2174	2175	2176	2177	2178	2179	2180	2181	2182	2183	2184	2185	2186	2187	2188	2189	2190	2191	2192	2193	2194	2195	2196	2197	2198	2199	2200	2201	2202	2203	2204	2205	2206	2207	2208	2209	2210	2211	2212	2213	2214	2215	2216	2217	2218	2219	2220	2221	2222	2223	2224	2225	2226	2227	2228	2229	2230	2231	2232	2233	2234	2235	2236	2237	2238	2239	2240	2241	2242	2243	2244	2245	2246	2247	2248	2249	2250	2251	2252	2253	2254	2255	2256	2257	2258	2259	2260	2261	2262	2263	2264	2265	2266	2267	2268	2269	2270	2271	2272	2273	2274	2275	2276	2277	2278	2279	2280	2281	2282	2283	2284	2285	2286	2287	2288	2289	2290	2291	2292	2293	2294	2295	2296	2297	2298	2299	2300	2301	2302	2303	2304	2305	2306	2307	2308	2309	2310	2311	2312	2313	2314	2315	2316	2317	2318	2319	2320	2321	2322	2323	2324	2325	2326	2327	2328	2329	2330	2331	2332	2333	2334	2335	2336	2337	2338	2339	2340	2341	2342	2343	2344	2345	2346	2347	2348	2349	2350	2351	2352	2353	2354	2355	2356	2357	2358	2359	2360	2361	2362	2363	2364	2365	2366	2367	2368	2369	2370	2371	2372	2373	2374	2375	2376	2377	2378	2379	2380	2381	2382	2383	2384	2385	2386	2387	2388	2389	2390	2391	2392	2393	2394	2395	2396	2397	2398	2399	2400	2401	2402	2403	2404	2405	2406	2407	2408	2409	2410	2411	2412	2
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MAJ/

W	-.3322	-.3637	.0574	.4011	.5145	.5207	.5195
CC						-.0321	-.0692

029 - .2931

045	-0.1579	-0.1062	-0.1927	-0.2057	-0.2761
350	-0.2055	-0.2005	-0.2005		

- 2101

.089		- .1269	
.086			
.094		- .2903	
		- .4250 - .4515 - .4216 - .4303	

.163	.0000	
.177		-.2430
.193	-.0992	

2,250	-	3,009	-	5,259
274	-	4,746	-	5,020
256	-	5,339	-	5,595
	-	2422	-	

1

	-0.6934
	-0.6081
	-0.4123
	-0.2222
.382	-0.0133
.400	
.402	

550	5500	- .6253
561		
600		- .6100

.700	- .3486
.725	- .2341
.749	- .1859
	- .5677 - .5750

-.1985    -.2365  
-.0792

.897
- .0656
.897
- .0235
.897
- .442

.999	-.0909	-.1409	-.4010
.999	-.0990		
.999	-.1133		



(RBI033)

TABULATED PRESSURE DATA - IAI14A - VOL. 8  
ARC11-716 IAI14 D1+T12+S12N25+ATIO UPPER WING

DATE 10 JAN 75

ALPHA ( 9 ) = 5.900 BETAD ( 9 ) = 6.990

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	
.2990	.3640 .4270 .9340 .8730 .7800 .8870
X/CW	
.000	-.4560 -.2406 -.0335 .3160 .4906 .4536 .4710
.020	.020
.025	-.0915 .0176 -.0496 -.0893
.040	-.0959
.045	-.1779
.050	-.1006
.055	-.1911 -.1919 -.2146 -.2775
.060	-.2763
.065	-.1268
.086	-.1426
.094	-.3059
.150	-.3847 -.4477 -.4346 -.4263
.163	.0000
.177	-.2014
.193	-.1035
.229	-.0873
.250	-.3955 -.5386 -.5431 -.5350
.274	-.2604
.339	-.2249
.362	.0024
.400	-.4424 -.5810 -.6481
.402	-.3563
.497	.0000
.550	-.3674 -.5579
.565	.0000
.600	-.6152
.650	-.8251
.700	-.1919
.725	-.1896
.750	-.5836 -.5392
.760	-.2228
.775	-.1304 -.2553
.808	-.1416
.834	-.1125
.850	-.1145 -.1940 -.2762
.857	-.1483
.865	-.0904
.900	-.1239
.905	-.1013
.905	-.1392
.930	-.0652
.933	-.1239
.965	-.1660
.965	-.1488
.965	-.1565



DATE 12 JAN 79

TABULATED PRESSURE DATA - IA14A - VOL. 6

(R01433)

ARC11-716 1A14 Q1+T12+S12N23+AT10 UPPER WING

ALPHA( 9) = 5.975      BETA( 15) = 0.163

DEPENDENT VARIABLE CP

[illegible]

ABSTRACTED PRESSURE DATA - IA14A - VOL. 6

ALPHA (°)	BETA (°)	10.165
5.950	10.165	

SECTION ( UPPER WING		DEPENDENT VARIABLE CP						
W/BW		.2990	.3640	.4270	.5340	.5730	.7000	.8870
X/CW								
.000	-.5306	-.1509		-.0759	-.0252	.2098	.3341	.3943
.020					-.2133	-.1520	-.1306	-.1465
.025				-.0749				
.050		-.1089						
.055				-.0948				
.050	-.4374				-.2211	-.2905	-.2651	-.3111
.080					-.2536			
.085				-.1279				
.086			-.0919					
.094	-.4630				-.2810	-.4421	-.4536	-.4317
.150			.0000					
.163				-.2199				
.177			-.1107					
.193								
.229	-.0271			-.2907	-.2975	-.4764	-.5377	-.5410
.230								
.274			-.2037					
.339	.0025			-.3750	-.3379	-.5207		-.6421
.362								
.400				.0000	-.1845	-.3482		
.452	.0000							-.6026
.49							-.4293	
.510						-.2008		
.563							-.2887	-.4985
.600					-.1646			
.650	-.0940							
.700				-.2136				
.725								
.750				-.1517	-.1534	-.1612		
.760								
.775								
.808								
.834	-.1529				-.1532	-.1506	-.1471	
.850				-.1720				
.857								
.865	-.1339				-.1549			-.3448
.900	-.1547			-.1567	-.1105			
.905					-.1448			-.0112
.950				-.1803				
.953								
.965	-.1917							



TABULATED PRESSURE DATA - IAI4A - VOL. 6

(R91033)

DATE 10 JAN 75

ARC11-716 IAI4 01+112+512N25+AT10 UPPER WING

ALPHAD(10) = 7.930 BETAD (2) = -8.030

SECTION 1 UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.1426	-.0020	.5676	.8442	.7892	.7102	.6742
.020				.0861	.0954	-.0242	-.1021
.040			.0716				
.060		.0095					
.080			-.0513				
.100	-.0014			-.1367	-.1722	-.2173	-.3139
.120					-.3015		
.140			-.2356				
.160		.0330					
.180	-.0213			-.5223	-.4877	-.4716	-.4744
.200		.0000					
.220			-.4062				
.240		-.1606					
.260	-.0328			-.6245	-.6105	-.5990	-.5749
.280			-.4728				
.300		-.3429					
.320	.0174			-.6769	-.6769		-.6754
.340			-.5436				
.360	.0000			-.6683	-.6576		
.380		.0050					-.6559
.400					-.6615		
.420	-.4480			-.2315			
.440			-.5863		-.5432	-.5767	
.460			-.2747				
.480				-.5296	-.5672		
.500		-.2491					
.520	-.1255			-.4696	-.5389	-.5130	
.540			-.1908				
.560		.0166					-.5050
.580	.0424			-.3749			
.600			-.1286	-.4715			
.620				-.2764		-.4604	
.640			-.0877				
.660	-.0476						

ORIGINAL PAGE IS  
OF POOR QUALITY

(R81U33)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+S12R25+AT10 UPPER WING

ALPHAC(110) = 7.810 BETAG (3) = -5.970

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

V/SW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0091	-.0689	.4943	.7941	.7516	.6033	.6371
.020			.0320	.0302	.0729	-.0461	-.1329
.025							
.040		-.0526					
.045			-.0909				
.050	-.0459			-.1587	-.1875	-.2365	-.3369
.060				-.3153			
.065			-.2594				
.086		-.0346					
.094	-.0624			-.5296	-.4965	-.4852	-.4897
.150			.0700				
.153							
.177			-.4089				
.193		-.1919					
.229	-.0891			-.6315	-.6185	-.6079	-.5864
.250				-.4688			
.274		-.3563					
.339							
.362	-.0400			-.6833	-.6903		-.6806
.400							
.432			-.5282				
.497	.0000			-.6730	-.6647		
.550			.0000				-.6634
.565					-.6748		
.600					-.2303		
.650				-.5459		-.5584	-.5866
.700	-.4487						
.725				-.2680			
.750				-.5132	-.9775		
.760							
.775			-.2482				
.808							
.834	-.0731			-.3688	-.5575	-.5274	
.850				-.1712			
.857							
.865	.0381						-.5093
.900	.0382			-.2365			
.909				-.1052	-.4749		
.950				-.1387		-.4592	
.953			-.0643				
.965	-.0642						



(R01033)

DATE 10 JAN 75  
 TABULATED PRESSURE DATA - IA14A - VOL. 6  
 ARC11-716 IA14 01\*12\*512N25\*AT10 UPPER WING

ALPHAD(10) = 7.83, BETAD (4) = -4.000		DEPENDENT VARIABLE CP	
SECTION (1) UPPER WING			
X/CM	Y/CM		
.000	.0235	-.1407	.4190
.020	.020	.7437	.7126
.025	.025	.0149	.0540
.040	.040	-.0183	-.0630
.045	.045	-.1100	-.1561
.050	.050	-.1264	-.3502
.055	.055	-.0944	-.1799
.060	.060	-.0812	-.3292
.065	.065	-.1031	-.2476
.070	.070	.0000	-.4880
.075	.075	-.2034	-.4981
.080	.080	-.1251	-.6185
.085	.085	-.4523	-.6153
.090	.090	-.3371	-.6064
.095	.095	-.0714	-.5920
.100	.100	-.6688	-.7017
.105	.105	-.5209	-.6833
.110	.110	.0000	-.6727
.115	.115	-.6634	-.6756
.120	.120	-.4522	-.2172
.125	.125	-.398	-.5829
.130	.130	-.2164	-.5829
.135	.135	-.2045	-.5643
.140	.140	-.1448	-.5283
.145	.145	.0276	-.5070
.150	.150	.0124	-.1920
.155	.155	-.0622	-.4922
.160	.160	-.1166	-.4429
.165	.165	-.0327	-.0327
.170	.170	-.0823	-.0823

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+912N25+AT10 UPPER WING (R91033)

ALPHA(10) = 7.830 BETA( 5) = -2.030

SECTION ( UPPER WING	DEPENDENT VARIABLE CP				
V/BW	.2990	.3640	.4270	.5340	.6730
X/CW					
.000	-.0371	-.2160	.3368	.6895	.6709
.020				-.0105	.6096
.025			-.0559		-.0755
.040		-.1667			
.045			-.1423		
.050	-.1429			-.1952	-.2561
.080			-.3362		-.3603
.085			-.2749		
.086		-.1267			
.094	-.1480			-.3274	-.4913
.150		.0000		-.5016	-.5036
.163			-.3789		
.177		-.1914			
.193	-.1636			-.6122	-.6141
.229			-.4499		-.6074
.250		-.3358			-.5962
.274				-.6554	-.7073
.339	-.0725		-.5025		-.6964
.362				-.6176	-.6791
.400	.0000		.0000		
.402					-.6690
.497				-.2161	
.550			-.2993		-.6110
.565			-.1735	-.1921	-.6076
.600					
.650	-.4104			-.1923	-.3720
.700					-.5483
.725				-.2527	
.750					-.6364
.760					
.775					
.808					
.834	-.0343				
.850			-.1127		
.857					
.865	.0073			-.1874	
.900	-.0033				-.5167
.905			-.0820	-.4716	
.950			-.1681		-.4598
.955			-.0456		
.965	-.0945				



(R01033)

DATE 10 JAN 75  
 TABULATED PRESSURE DATA - IA14A - VOL. 6  
 ARC11-716 IA14 OI+112+S12N25+AT10 UPPER WING

ALPHA(10) = 7.840 BETA( 6) = .040		DEPENDENT VARIABLE CP	
SECTION ( 1) UPPER WING			
X/CM	Y/CM		
.000	.2990	.3440	.4270
.000	.1951	.3117	.2640
.020	.025	.025	.025
.040	.045	.045	.045
.060	.065	.065	.065
.080	.085	.085	.085
.100	.105	.105	.105
.120	.125	.125	.125
.140	.145	.145	.145
.160	.165	.165	.165
.180	.185	.185	.185
.200	.205	.205	.205
.220	.225	.225	.225
.240	.245	.245	.245
.260	.265	.265	.265
.280	.285	.285	.285
.300	.305	.305	.305
.320	.325	.325	.325
.340	.345	.345	.345
.360	.365	.365	.365
.380	.385	.385	.385
.400	.405	.405	.405
.420	.425	.425	.425
.440	.445	.445	.445
.460	.465	.465	.465
.480	.485	.485	.485
.500	.505	.505	.505
.520	.525	.525	.525
.540	.545	.545	.545
.560	.565	.565	.565
.580	.585	.585	.585
.600	.605	.605	.605
.620	.625	.625	.625
.640	.645	.645	.645
.660	.665	.665	.665
.680	.685	.685	.685
.700	.705	.705	.705
.720	.725	.725	.725
.740	.745	.745	.745
.760	.765	.765	.765
.780	.785	.785	.785
.800	.805	.805	.805
.820	.825	.825	.825
.840	.845	.845	.845
.860	.865	.865	.865
.880	.885	.885	.885
.900	.905	.905	.905
.920	.925	.925	.925
.940	.945	.945	.945
.960	.965	.965	.965

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TABULATED PRESSURE DATA - IA14A - VOL. 3

(R81033)

ARC11-716 IA14 OL+T12+S12M25+AT10 UPPER WING

ALPHA(10) = 7.830 BETA(7) = 2.040

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y BW .2990 .3640 .4270 .5340 .6730 .7000 .8070

X/C

.000	-.2498	-.4981	.1581	.5503	.5507	.3211	.5027
.020				-.1106	-.0276	-.1085	-.1905
.025			-.1417				
.040		-.2697					
.045			-.2111				
.050	-.2422			-.2581	-.2493	-.2780	-.3700
.080				-.3693			
.085			-.3046				
.086			-.1434				
.094	-.2286						
.150			.0000	-.5207	-.9056	-.4953	-.5028
.163							
.177			-.3333				
.193			-.1614				
.229	-.1664			-.5802	-.6053	-.6020	-.5953
.250							
.274		-.2687	-.3617				
.339							
.362	-.0511			-.5765	-.6857		-.6934
.400			-.4481				
.432							
.497	.0000			-.1698	-.6710		
.550			.0050				-.6635
.565							
.600							
.650							
.700	-.3925			-.2699	-.2113		
.725							
.750							
.760			-.2701				
.775				-.2600	-.6154		
.808			-.1247				
.834	-.6371						
.850				-.2248	-.3337	-.5268	
.857			-.1186				
.865	-.0277						
.900	-.0584			-.1994			-.5003
.905			-.1064		-.1347		
.950				-.1732		-.4465	
.953			-.1302				
.965	-.1404						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A144 - VOL. 6

(N81053)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA(10) = 7.870 BETA( 6) = 4.060

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/OW .2990 .3640 .4270 .5340 .6730 .7300 .8070

X/CW

.000	-.3482	-.4467	-.0286	.4330	.5064	.5117	.4928
.020			-.1723	-.0398	-.1151	-.1904	
.025			-.2026				
.040		-.2771					
.045			-.2311				
.050	-.2935		-.2904	-.2681	-.2800	-.3674	
.060			-.3775				
.085			-.2611				
.086		-.1093					
.094	-.2621						
.130		.0000		-.4961	-.4971	-.4904	-.5009
.163			-.2776				
.177		-.1415					
.193	-.1393			-.4997	-.5900	-.5950	-.5929
.229			-.3212				
.250		-.2732					
.274				-.3188	-.6409		-.6906
.339	-.0319		-.4279				
.382			.0200	-.5070	-.6585		
.400							-.6607
.402				-.2971	-.2078		
.497	.0000						
.590			-.2120	-.1914	-.3148		
.565			-.1050				
.600				-.1266	-.1660	-.5304	
.650	-.3735		-.1109				
.700						-.6167	-.6289
.725							
.750							
.760							
.775							
.808							
.834	-.0699						
.850							
.857							
.865	-.0541			-.1114			-.5042
.900	-.0611		-.1112	-.1903			
.905			-.1023			-.4933	
.950			-.1315				
.953							
.965	-.1359						

C-6

(081053)

DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 O1+T12+312N23+AT10 UPPER WING

ALPHA0(10) = 7.970 BETA0 (9) = 0.160

SECTION (1) UPPER WING DEPENDENT VARIABLE C

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.9070
X/CW							
.000	-.4198	-.2916	-.0988	.2542	.4950	.4595	.4842
.020				-.2040	-.0512	-.1257	-.1891
.025			-.1427				
.040		-.2279					
.045			-.1526				
.050	-.3774			-.2783	-.2626	-.2888	-.3664
.080				-.3473			
.085				-.1688			
.086			-.1837				
.094	-.3181						
.150				-.4483	-.5051	-.4931	-.4959
.163		.0000					
.177			-.2370				
.193		-.1602					
.229	-.1223			-.4717	-.5952	-.5941	-.5877
.250			-.2871				
.274		-.2694					
.339							
.362	-.0199			-.4946	-.6494		-.6880
.400			-.3759				
.402				-.4187	-.6146		
.487	.0000						
.590			.0000			-.6822	-.6913
.585					-.2020		
.600				-.2476		-.5954	-.5900
.650	-.2617						
.700			-.2396		-.1811	-.2712	
.725							
.730			-.1704				
.760				-.1601	-.2311	-.5828	
.775				-.1752			
.808							
.834	-.1260						
.850							
.857							
.865	-.1069			-.1854			-.4707
.900	-.1137		-.1614		-.1255		
.905				-.1873		-.3714	
.920			-.1738				
.953							
.965	-.1879						



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBIU33)

ARC11-716 1A14 01-712+512N25+AT10 UPPER WING

ALPHA(10) = 7.980 BETA(10) = 0.110

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8670

X/CW

.000	-.3305	-.1757	-.0765	.0630	.4042	.4144	.4096
.020				-.2643	-.1023	-.1478	-.2149
.025			-.1063				
.040		-.1321					
.045			-.1105				
.050	-.4357		-.2978	-.2911	-.3008	-.3822	
.060			-.3471				
.085			-.1501				
.086		-.1052					
.094	-.4084			-.3918	-.5077	-.5010	-.5049
.150		.0000					
.163							
.177			-.2295				
.193		-.1590					
.229	-.0467			-.4273	-.5792	-.5985	-.5989
.250			-.2874				
.274		-.2660					
.339				-.4526	-.6069		-.6944
.362	-.0033		-.3652				
.400				-.2849	-.5512		
.402			.0000				
.497	.0000					-.6547	
.550							
.565							
.600							
.650				-.2249	-.2116		
.700	-.1232					-.5936	-.5703
.725			-.2357				
.750				-.1964	-.2629		
.760			-.1929				
.775							
.806							
.834	-.1344			-.1892	-.1928	-.3159	
.850			-.2020				
.857							
.865	-.1300			-.1911			-.4730
.900	-.1435		-.1878	-.1073			
.925				-.1903		-.1234	
.950			-.1862				
.953							
.965	-.1898						

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(R81U33)

ARC11-716 1A14 Q1+T12+S12N25+AT10 UPPER WING

ALPHA(113) = 7.960 BETA(11) = 10.230

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	-.3280	-.1754	-.1295	-.0859	.2541	.3572	.3907
.020				-.3289	-.2020	-.1806	-.2274
.025			-.1230				
.040		-.1292					
.045			-.1373				
.050	-.4894		-.2903	-.3543	-.3272	-.3877	
.060			-.3145				
.085		-.1679					
.086		-.1146					
.094	-.4892						
.120				-.3265	-.5200	-.5148	-.9066
.163		.0000					
.177			-.2490				
.193		-.1794					
.229	-.0794			-.3568	-.5576	-.6009	-.5949
.250			-.3203				
.274		-.2385					
.339				-.3681	-.5689		-.6883
.362	-.0215						
.400		-.3910					
.402				-.2304	-.3661		
.497	.0000						
.590							
.595							
.600							
.690							
.700	-.0888						
.725				-.2137	-.2148		
.790							
.775		-.2317		-.2048	-.1970		
.808		-.1733					
.834	-.1770						
.890				-.2060	-.1950	-.1880	
.937		-.1896					
.965	-.1919						
.900	-.1715			-.2056			-.4535
.925		-.1748		-.1207			
.950			-.1947				-.0404
.953		-.1683					
.965	-.1973						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+112+912M25+AT10 UPPER WING (R81033)

ALPHA(11) = 9.895 BETA(1) = -9.960

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.2995	.1688	.6593 .9136 .8295 .7271 .6646
.320		.0936 .0609 -.0911 -.1827
.325	.1552	
.340	.0919	
.345		
.350	.0271	-.1434 -.2137 -.2769 -.3776
.360		-.3120
.365	-.2317	
.386	.0953	
.394	.0278	-.5355 -.5207 -.5136 -.5251
.410		
.415	.0000	
.417		-.4202
.419	-.1323	
.429	.0016	-.6489 -.6360 -.6319 -.6104
.430		-.4946
.439	-.3492	
.462	.0788	-.6880 -.6593 -.6633
.492		-.5545
.497	.0000	-.6982 -.6829
.510		.0000
.565		
.600		-.6774
.630		-.6299
.700	-.4581	-.2305
.725		-.6261
.750		-.9382 -.8289
.760	-.3077	
.775	-.2897	-.5986 -.5827
.808		
.834	-.2388	-.5402 -.5499 -.5222
.850		
.857		-.2252
.885	-.0627	
.900	.0257	-.4932 -.5024 -.6104
.905		-.1897 -.5024
.930		-.4434 -.4870
.935		-.1654
.965	-.0386	

(R01033)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA(11) = 9.930 BETA( 2 ) = -7.920

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7800 .8870
.1493	.0381	.6087	.6845 .8032 .6952 .6326
.020		.0670	.0454 -.1094 -.2063
.025		.0660	
.040	.0218		
.045		-.0614	-.1616 -.2263 -.2941 -.3946
.350	.0073		-.3268
.060			
.085		-.2578	
.086	.0453		
.094	-.0118		
.190			-.5423 -.5294 -.5224 -.5388
.163	.0000		
.177		-.4412	
.193	-.1692		
.229	-.0413		
.250		-.6439 -.6418 -.6390 -.6206	
.274		-.5109	
.339	-.3622		
.362	.0322		
.400		-.6862 -.6715	-.6693
.402		-.5633	
.497	.0006		
.550		-.6930 -.6689	
.565		.0000	
.600			-.6852
.650			-.6429
.700	-.4676	-.6306	
.723		-.2429	
.750			-.5499 -.6293
.760		-.2935	
.775		-.5946 -.5603	
.906		-.2721	
.834	-.1792		
.950		-.5413 -.5399 -.5331	
.837		-.2279	
.863	-.0873		
.900	.0137		-.6013
.905		-.4784	
.950		-.1753	-.4958
.953		-.4109	-.4933
.985	-.0436	-.1497	



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+S12N25+AT10 UPPER WING

(881033)

ALPHA(11) = 9.940 BETA( 3) = -6.010

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.1006	-.0289	.5419	.8474	.7734	.8667	.5944
.020			.0408	.0270	-.1301	-.2366	
.040		-.0380					
.045			-.0931				
.050	-.0381			-.1793	-.2382	-.3089	-.4184
.060				-.3406			
.065			-.2793				
.066		-.0136					
.094	-.0532			-.5548	-.5381	-.5352	-.5551
.150		.0000					
.163			-.4579				
.177		-.2513					
.193							
.229	-.0789			-.6510	-.6519	-.6480	-.6348
.250			-.5167				
.274		-.3825					
.339				-.6947	-.6834		-.6810
.362	-.0211						
.400				-.7027	-.6782		
.402	.0000		.0000				-.6900
.497						-.6618	
.550				-.6294	-.2393		
.563						-.5662	-.6438
.600			-.2946	-.5855	-.5945		
.650			-.2809				
.700	-.4707			-.5223	-.5642	-.5451	
.725			-.2158				
.750				-.4480			-.6102
.760			-.1521	-.4982			
.775				-.3518		-.4844	
.808							
.834	-.1333						
.850							
.857							
.865	-.0154						
.900	.0021						
.909							
.950							
.953							
.965	-.0781						



DATE 10 JAN 73

TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+112+312N25+AT10 UPPER WING

(R81033)

ALPHA(11) = 9.690 BETA( 4) = -3.990

SECTION ( 1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	X/CW	CP	CP	CP	CP
.2990	.3640	.4270	.5340	.6730	.7800
.0232	-.1215	.4534	.7846	.7300	.6307
.020			-.0051	.0070	-.1430
.025		-.0271			-.2525
.040	-.1106				
.045		-.1487			
.050			-.2142	-.2332	-.3211
.060	-.0899		-.3666		
.085		-.3174			
.086	-.0786				
.094	-.1001		-.5669	-.5463	-.5426
.150		.0000			-.5620
.163					
.177	-.2387		-.4798		
.193					
.229	-.1216		-.6556	-.6558	-.6518
.250		-.3214			-.6415
.274					
.339	-.3798				
.362	-.0827		-.7046	-.6941	-.6876
.403		-.5458			
.402					
.497	.0000		-.6943	-.6858	
.530		.0000			-.6940
.565					
.600					
.650					
.700	-.4590		-.2276		
.725			-.5545		-.5616
.750		-.3034			-.6417
.760			-.5364	-.5990	
.775		-.2509			
.808					
.834	-.0716		-.4291	-.5661	-.5350
.850		-.1629			
.857					
.865	.0067		-.2839		-.6045
.900	.0051				
.905		-.1301		-.5051	
.950			-.1507		-.4862
.955		-.1037			
.965	-.0941				



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81033)

ARC11-716 1A14 01-T12+S12N25+AT10 UPPER WING

ALPHA(111) = 9.900 BETA( 5) = -1.980

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0621	-.2252	.3302	.7131	.6822	.5915	.5243
.020				-.0754	-.0281	-.1675	-.2773
.025			-.1111				
.040		-.1900					
.045			-.2214				
.050	-.1373			-.2679	-.2811	-.3394	-.4465
.055				-.4037			
.065			-.3650				
.086		-.1364					
.094	-.1518			-.5856	-.5597	-.5545	-.5762
.150		.0000					
.163			-.4824				
.177		-.2800					
.193							
.229	-.1827			-.6878	-.6633	-.6598	-.6535
.250							
.274		-.4722					
.339		-.3640					
.362	-.1103			-.7368	-.7110		-.7000
.400			-.5357				
.402				-.7370	-.7027		
.497	.0000		.0000				
.550							
.565							
.600							
.650					-.7022		-.7003
.700	-.4340			-.4563	-.2237		
.725							
.750							
.760			-.3346			-.6008	-.6535
.775			-.2378				
.808							
.834	-.0509			-.2193	-.6038	-.5662	
.890			-.1818				
.897							
.885	.0016						
.900	-.0113			-.1865			-.6127
.935			-.1145		-.5084		
.950				-.1526		-.4796	
.953			-.0912				
.965	-.1106						

(RB1V33)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA3(11) = 9.910 BETA0 ( 6 ) = .920

SECTION (1) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8670
X/CW							
.000	-.1645	-.3689	.2005	.6203	.6233	.5127	.4920
.020				-.1382	-.0583	-.1840	-.2914
.025			-.1954				
.040		-.2743					
.045			-.2928				
.050	-.1962			-.3071	-.3023	-.3501	-.4564
.060				-.4269			
.065			-.3769				
.066		-.1932					
.067	-.1967						
.094							
.150		.0000		-.5880	-.5659	-.5596	-.5794
.163			-.4400				
.177		-.2462					
.193							
.229	-.1928						
.250				-.6882	-.6647	-.6617	-.6564
.274			-.4593				
.339		-.3324					
.362	-.0980			-.7268	-.7166		-.7171
.400			-.4634				
.497	.0000			-.6026	-.7030		
.550			.0000				
.565							
.600							
.650							-.7115
.700	-.4281				-.2205	-.7161	
.723				-.3062			
.750							
.760			-.3232	-.2509	-.6350	-.6272	-.6651
.775			-.1884				
.806							
.834	-.0335						
.850			-.1630	-.2546	-.6016	-.5801	
.857							
.865	.0014						
.900	-.0402			-.2315			-.5935
.905			-.1291	-.4869			
.950				-.1950	-.4884		
.953			-.1340				
.963	-.1306						



(R81U33)

TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01-712+512+23+AT10 UPPER WING

DATE 15 JAN 75

ALPHA(11) = 9.900 BETA( 7) = 2.040

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.500	-.2543	-.5361	.0575	.5342	.5687	.5140	.4649
.020				-.1781	-.0821	-.1914	-.2996
.025		-.3520	-.2219				
.040			-.2849				
.045				-.3255	-.3168	-.3524	-.4587
.050	-.2499			-.4312			
.060			-.3730				
.065		-.2329					
.086							
.094	-.2303			-.5807	-.5634	-.5539	-.5791
.130		.0000					
.163			-.3975				
.177		-.2100					
.193							
.229	-.2082			-.6158	-.6521	-.6541	-.6551
.250			-.3970				
.274		-.2914					
.339							
.362	-.0871			-.5911	-.7199		-.7075
.408			-.4605				
.402				-.5546	-.6930		
.497	.0000						
.550							
.565							
.600							
.650							
.700	-.4173			-.2765	-.2200	-.7083	-.7080
.725							
.750			-.2919				
.760				-.2599	-.5882		
.775			-.1577				
.808							
.834	-.0479			-.2301	-.5574	-.5591	
.850			-.1538				
.857							
.865	-.0367			-.1974			-.5530
.900	-.0686						
.905			-.1406	-.3509			
.950				-.1372			-.4614
.955			-.1652				
.965	-.1480						

(RB1V33)

ARC11-716 IA14 01+T12+S12N25+AT10 UFFER WING

$$\text{ALPHA}(11) = 9.999 \quad \text{BETA}(8) = 4.130$$

SECTION : 111EEPS WING

DEPENDENT VARIABLE CF

[illegible]

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ABC11-716 1A14 Q1+T12+S12N25+AT10 UPPER WING

ALPHA(11) = 9.880 BETA(9) = 6.199

BETA) ( 9) =

6.199

SECTION ( ) UPPER WING

DEPENDENT VARIABLE CP

y/8w	.2995	.3640	.4275	.5340	.6735	.7805	.8870
------	-------	-------	-------	-------	-------	-------	-------

A5/X

MS/X

Variable	Mean	Standard deviation	Skewness	Kurtosis
Age	38.50	10.50	-.05	2.90
Gender	1.50	.50	-.05	2.90
Marital status	2.50	.50	-.05	2.90
Education	12.50	1.50	-.05	2.90
Income	3.50	.50	-.05	2.90
Occupation	1.50	.50	-.05	2.90
Health	2.50	.50	-.05	2.90
Stress	3.50	.50	-.05	2.90
Life satisfaction	4.50	.50	-.05	2.90
Depression	1.50	.50	-.05	2.90
Loneliness	2.50	.50	-.05	2.90
Self-esteem	3.50	.50	-.05	2.90
Resilience	4.50	.50	-.05	2.90
Optimism	5.50	.50	-.05	2.90
Gratitude	6.50	.50	-.05	2.90
Forgiveness	7.50	.50	-.05	2.90
Compassion	8.50	.50	-.05	2.90
Empathy	9.50	.50	-.05	2.90
Kindness	10.50	.50	-.05	2.90
Generosity	11.50	.50	-.05	2.90
Patience	12.50	.50	-.05	2.90
Humility	13.50	.50	-.05	2.90
Modesty	14.50	.50	-.05	2.90
Shyness	15.50	.50	-.05	2.90
Introversion	16.50	.50	-.05	2.90
Neuroticism	17.50	.50	-.05	2.90
Extraversion	18.50	.50	-.05	2.90
Agreeableness	19.50	.50	-.05	2.90
Conscientiousness	20.50	.50	-.05	2.90
Openness	21.50	.50	-.05	2.90
Stability	22.50	.50	-.05	2.90
Instability	23.50	.50	-.05	2.90
Dependability	24.50	.50	-.05	2.90
Unreliability	25.50	.50	-.05	2.90
Trustworthiness	26.50	.50	-.05	2.90
Untrustworthiness	27.50	.50	-.05	2.90
Honesty	28.50	.50	-.05	2.90
Dishonesty	29.50	.50	-.05	2.90
Integrity	30.50	.50	-.05	2.90
Lack of integrity	31.50	.50	-.05	2.90
Principled	32.50	.50	-.05	2.90
Unprincipled	33.50	.50	-.05	2.90
Law-abiding	34.50	.50	-.05	2.90
Law-breaking	35.50	.50	-.05	2.90
Responsible	36.50	.50	-.05	2.90
Irresponsible	37.50	.50	-.05	2.90
Self-disciplined	38.50	.50	-.05	2.90
Undisciplined	39.50	.50	-.05	2.90
Organized	40.50	.50	-.05	2.90
Disorganized	41.50	.50	-.05	2.90
Clean	42.50	.50	-.05	2.90
Unclean	43.50	.50	-.05	2.90
Neat	44.50	.50	-.05	2.90
Untidy	45.50	.50	-.05	2.90
Polite	46.50	.50	-.05	2.90
Impolite	47.50	.50	-.05	2.90
Courteous	48.50	.50	-.05	2.90
Uncourteous	49.50	.50	-.05	2.90
Respectful	50.50	.50	-.05	2.90
Disrespectful	51.50	.50	-.05	2.90
Kind	52.50	.50	-.05	2.90
Unkind	53.50	.50	-.05	2.90
Generous	54.50	.50	-.05	2.90
Ungenerous	55.50	.50	-.05	2.90
Selfless	56.50	.50	-.05	2.90
Selfish	57.50	.50	-.05	2.90
Altruistic	58.50	.50	-.05	2.90
Unaltruistic	59.50	.50	-.05	2.90
Compassionate	60.50	.50	-.05	2.90
Uncompassionate	61.50	.50	-.05	2.90
Empathetic	62.50	.50	-.05	2.90
Unempathetic	63.50	.50	-.05	2.90
Kindhearted	64.50	.50	-.05	2.90
Unkindhearted	65.50	.50	-.05	2.90
Forgiving	66.50	.50	-.05	2.90
Unforgiving	67.50	.50		

027. 1954 1955  
- 1762 227

-.2531

545  
-1829

0.550 - 3692  
- 4063

505 - 2128

**205 --.2055**

Variable	Mean	Standard Deviation	Minimum	Maximum
Age	35.25	12.50	18	65
Gender	1.50	0.50	1	2
Marital Status	1.50	0.50	1	2
Education	15.50	2.50	12	18
Income	4500	1500	2000	8000
Health	1.50	0.50	1	2
Stress	3.50	1.50	1	5
Depression	2.50	1.50	1	5
Life Satisfaction	3.50	1.50	1	5

.163 .0000

	.177	- .2387
--	------	---------

226 - 1492  
1933 - 1493

**.250      -.4850   -.6340   -.6404   -.6436**

	.274	- .3164
--	------	---------

362 - 0307

	.409	-.4995	-.6971	-.7200
--	------	--------	--------	--------

.402	- .4062
.0004	

**- .3070 - .6713**

565,0000

-.6994

709    -.2553

709    -.2090

**.725** **-.2482**

**-2105**

-1943 - .3060

659  
- .2026

.834	-.1449
44213	-.1746   -.2653   -.5960

1989-1989

.065    .1173

	-1.1232	.0019	
		.1073	-.1662

**.959**      **-.1689**      **-.5144**

.955 - .974

6211- 626.

(R01053)

ARC11-716 1A14 01+T12+S12N25+AT10 UPPER WING

ALPHA(11) = 9.875 BETA(12) = 0.110

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/8W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.5123	-.2086	-.1382	.0444	.4410	.4252	.3994
.020				-.3539	-.1338	-.2013	-.2953
.025			-.1374				
.040		-.1688					
.045			-.1551				
.050	-.4492			-.3719	-.3434	-.3542	-.4920
.060				-.4121			
.085			-.1892				
.086		-.1433					
.094	-.4077			-.4558	-.5565	-.5479	-.5687
.150		.0000					
.163			-.2496				
.177		-.1964					
.193							
.229	-.1173			-.4339	-.6345	-.6446	-.6472
.250			-.3078				
.274		-.2963					
.339	-.0306			-.4480	-.6736		-.7143
.362			-.3794				
.400		.0000		-.2817	-.5878		
.402			.0000			-.6974	-.6876
.497				-.2319	-.2198		
.550			-.2550	-.2037	-.3041	-.6250	-.6121
.565			-.2233				
.600				-.1978	-.2391	-.5445	
.630	-.1320			-.2209			
.700							
.729							
.730							
.760							
.775							
.808							
.834	-.1772						
.850							
.857							
.865	-.1533			-.1973			-.5460
.900	-.1639			-.2124	-.1242		
.905				-.1916		-.2275	
.950							
.955			-.2037				
.965	-.2003						



DATE 15 JAN 75  
 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+812M23+AT10 UPPER WING  
 (RB1033)

ALPHA(11) = 10.000 BETAO (11) = 10.190

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.005	-.4970	-.1866	-.1681	-.0911	.3537	.3960	.3930
.020				-.4228	-.1869	-.2062	-.2933
.025			-.1603				
.040		-.1593					
.045			-.1759				
.050	-.4806			-.3603	-.3794	-.3940	-.4453
.080				-.3727			
.085			-.2013				
.086		-.1381					
.094	-.4776			-.3610	-.5697	-.5432	-.5567
.150		.0000					
.163			-.2681				
.177		-.2314					
.193							
.229	-.1956			-.3734	-.6277	-.6372	-.6373
.250			-.3227				
.274		-.2800					
.339							
.362	-.0505			-.3435	-.6050		-.7016
.400			-.3658				
.402							
.497	.0000			-.2305	-.3750		
.550			.0000				
.563							
.600							
.650							-.6709
.700	-.0943			-.2194			
.725			-.2313				
.750							
.760			-.2442				
.775			-.2115				
.808				-.2261	-.2376		
.834	-.1932						
.850				-.2243	-.1999	-.2377	
.857			-.2122				
.865	-.1673						
.900	-.1774			-.2174	-.1433		-.9022
.905			-.1938				
.950				-.2058			-.0614
.953			-.1806				
.965	-.1829						



(R81054) ( 15 FEB 74 )

ARC11-716 1A14 OL+T12+S12M25+AT111 UPPER WING

PARAMETRIC DATA

MACH = .800 ELEVON = .000  
RUDDER = .000 SPOBRK = .000

REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5000 INCHES  
LREF = 38.7090 INCHES YMRP = .0000 INCHES  
BREF = 38.7090 INCHES ZMRP = .0000 INCHES  
SCALE = .5300 SCALE

ALPHA(1) = -8.010 BETA(1) = -7.990

DEPENDENT VARIABLE CP

SECTION ( ) UPPER WING

Y/SW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.0311	.0087	.1876	.1351	-.0490	-.2125	-.1226
.000				.4748	.4641	.4915	.4673
.020			.3582				
.025		.1855					
.040			.2874				
.045				.2630	.2925	.3052	.2749
.050	.0823			-.0956			
.060			.1488				
.085		.2468					
.086				-.1163	-.0842	-.0647	-.0572
.094	.0877						
.150		.0200	-.0520				
.163			.1085				
.177				-.1415			
.193							
.229	.1362			-.1495	-.2310	-.2591	-.2764
.250							
.274			-.1107				
.339		.0796					
.362				-.1082	-.2253		-.2458
.400			-.1531				
.402							
.497	.0000			-.0382	-.1415		
.550			.0050				
.565							
.600							-.1634
.650						-.1181	
.700	.0449			.0225	-.1477		
.725						.0681	.0293
.750							
.760				-.0353	.1340	.0752	
.775				.0926			
.806							
.834	.0273			.1185	.0985	.1016	
.850			.0700				
.857							
.865	.1076						.0544
.900	.0603			.1253	.1859		
.905			.0607				

DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+112+S12N23+AT11 UPPER WING (RB1034)

ALPHAO ( 1 ) = -8.910 BETAO ( 1 ) = -7.980

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW				.1115		.1095	
.950				.0380			
.933							
.965							

X/CW

.950  
.933  
.965

ALPHAO ( 1 ) = -7.990 BETAO ( 2 ) = -3.980

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000							
.020							
.025							
.040							
.045							
.050							
.080							
.085							
.086							
.094							
.150							
.163							
.177							
.193							
.229							
.250							
.274							
.339							
.362							
.400							
.432							
.497							
.550							
.585							
.600							
.650							
.700							
.725							
.750							
.760							
.775							
.806							
.834							
.890							

X/CW

.000  
.020  
.025  
.040  
.045  
.050  
.080  
.085  
.086  
.094  
.150  
.163  
.177  
.193  
.229  
.250  
.274  
.339  
.362  
.400  
.432  
.497  
.550  
.585  
.600  
.650  
.700  
.725  
.750  
.760  
.775  
.806  
.834  
.890





(R01034)

DATE 10 JAN 75  
 TABULATED PRESSURE DATA - IA14A - VOL. 6  
 ARC11-716 IA14 OA+T12+S12N25+AT11 UPPER WING

ALPHA( 1 ) = -7.990 BETA( 3 ) = .030  
 SECTION ( 1 ) UPPER WING  
 Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	DEPENDENT VARIABLE CP
.775	.0712 .0344
.806	.0297
.834	.0712 .0650 .0777
.850	.0200
.857	
.865	.0335
.890	.0093
.905	.0161
.930	.0759 .0901
.953	.0758
.965	-.0066
	-.0493

ALPHA( 1 ) = -8.000 BETA( 4 ) = 4.090  
 SECTION ( 1 ) UPPER WING  
 Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	DEPENDENT VARIABLE CP
.000	-.0377 -.0364 .0036 -.2154 -.5133 -.6423 -.4748
.020	.3766 .3179 .3280 .2730
.025	.2503
.040	.0720
.045	.2307
.050	.2429 .2465 .2420 .1818
.060	.1292
.085	.1599
.086	
.094	.1805
.150	.0059
.163	.0000
.177	.1102
.193	
.229	.0859
.230	
.274	-.1183
.339	-.1055
.362	.0737
.400	-.1957
.402	
.497	-.1994 -.2495 -.2652
.520	
.565	-.1717 -.1951
.600	.0000
.650	-.1982
	-.1830

TABULATED PRESSURE DATA - IAL4A - VOL. 6

(R81034)

ARC11-716 IAL4 01-T12-S12M25+AT11 UPPER WING

ALPHA( 1 ) = -0.000 BETA( 4 ) = 4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.750	-.0029	-.1048	-.1576	.0167	-.0132
.725					
.750					
.760		-.1231	.0039	.0090	
.775		.0009			
.808					
.834	-.0782		.0060	.0431	.0581
.850					
.857		-.0108			
.865	.0014				.0545
.900	-.0164		.0235	.0575	
.905		-.0082	.0102	.0737	
.950					
.993		-.0270			
.985	-.0621				

ALPHA( 1 ) = -0.020 BETA( 5 ) = 0.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0682	.0398	-.0151	-.2665	-.5988	-.7477	-.9930
.020				.3346	.2568	.2626	.2045
.025			.2115				
.040		.0442					
.045			.2092	.2202	.2129	.2077	.1474
.050	-.0336			.1199			
.080							
.085			.1463				
.086		.1281					
.094	-.0604						
.150		.0000		-.0639	-.0527	-.0317	-.0793
.163							
.177			-.0071				
.193		.0986					
.229	.0631						
.250				-.1923	-.2015	-.2024	-.2401
.274			-.1274				
.339		-.1181					
.362	.0599						
.400				-.2185	-.2487		-.2832
.492							
.497	.0000		-.2209				



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 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+512N25+AT11 UPPER WING  
 (R81U34)

ALPHA( 1 ) = -8.020 BETA( 5 ) = 8.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/B-1 .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CM	-1912	-2128	.0000	-1742	-1707	-1249	.0063	-0152	-1464	-0169	-0026	-0216	-0315	-0087	.0906	.0523	-0292	.0030	.0623	-0390	-0790
.350																					
.565																					
.600																					
.650																					
.700																					
.725																					
.750																					
.760																					
.775																					
.808																					
.834																					
.850																					
.857																					
.865																					
.900																					
.905																					
.950																					
.953																					
.965																					

ALPHA( 2 ) = -4.050 BETA( 1 ) = -8.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/B-1 .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CM	.0647	.1135	.3475	.4204	.3517	.2644	.2801	.2942	.2861	.3064	.3161	.2491	.1311	.0329	.0320	.0454	.0336	.1902	.0465	.0000	.2211	.0007	.3292	.4211	.4596	.4702	.2852	.2372
.000																												
.020																												
.025																												
.040																												
.045																												
.050																												
.060																												
.085																												
.086																												
.094																												
.100																												
.163																												
.177																												
.193																												
.229																												
.250																												
.274																												
.339																												

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+512N25+AT11 UPPER WING (RBIUS4)

ALPHA( 2 ) = -4.050 BETA( 1 ) = -8.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	CP
.362	-.0218
.400	
.402	-.2304
.497	
.550	.0000
.565	-.0991
.600	-.1920
.650	
.700	.0561
.725	
.750	-.0306
.760	
.775	-.0861
.808	
.834	.0197
.850	
.857	.0709
.865	
.900	.1001
.925	.0721
.935	
.950	.0620
.953	
.965	.0455
	-.0002

ALPHA( 2 ) = -4.050 BETA( 2 ) = -4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	CP
.000	.0161
.020	.0703
.025	.2851
.040	.2361
.045	
.050	.1510
.060	
.065	.0589
.086	-.1041
.094	-.0020
.104	.1755
.150	
.163	.0173
.177	
.193	.0000
	-.1960
	-.2901
	-.2868
	-.2758
	-.2560



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ARC11-716 1A14 01+T12+S12N25+AT11: UPPER WING (RBIU34)

ALPHA( 2 ) = -4.090 BETA( 2 ) = -4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/SW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	.229	.0724				
	.250					
	.274					
	.339					
	.352					
	.403					
	.402					
	.497					
	.550					
	.565					
	.600					
	.630					
	.700					
	.725					
	.750					
	.762					
	.775					
	.808					
	.834					
	.850					
	.857					
	.865					
	.900					
	.905					
	.930					
	.975					
	.965					

ALPHA( 2 ) = -4.090 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/SW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	.229	.0724				
	.250					
	.274					
	.339					
	.352					
	.403					
	.402					
	.497					
	.550					
	.565					
	.600					
	.630					
	.700					
	.725					
	.750					
	.762					
	.775					
	.808					
	.834					
	.850					
	.857					
	.865					
	.900					
	.905					
	.930					
	.975					
	.965					

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(N81034)

ARC11-716 1A14 OL+712+512M25+AT11 UPPER WING

DATE 10 JAN 75

ALPHA( 2) = -4.060 BETA( 4) = 4.070

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0378					
.085	.0144					
.086	.1195					
.094	-.0219					
.150		-.2363	-.2338	-.2221	-.2286	
.163	.0500					
.177	-.1498					
.193	.0013					
.229	.0440					
.250		-.3270	-.3621	-.3844	-.3957	
.274		-.2353				
.339	-.2082					
.362	-.0103					
.400		-.2880	-.3484		-.3438	
.402	-.2814					
.497	.0000					
.550		-.2199	-.2463			
.565						
.600						
.650						
.700	-.1176					
.725			-.1573			
.750						
.760		-.1297				
.775		-.0117	-.0051			
.808		-.0200				
.834	-.0892					
.850		-.0736	.0328	.0539		
.857		-.0145				
.865	-.0216					
.900	-.0305		.0205			.0666
.905		-.0135	.0587			
.950			.0187		.0737	
.955		-.0235				
.965	-.0675					

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ARC11-716 IA14 01-T12+S12N25+AT11 UPPER WING (981034)

ALPHA( 2) = -4.000 BETA( 5) = 0.090

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CF				
Y/BW		.2990	.3640	.4270	.5340	.6730 .7800 .8870
X/CM						
.000		-.1115	-.0124	.0548	.0982	-.1059 -.2440 -.2046
.020				.1493	.2450	.1960 .2164 .1844
.040			.0270			
.060				.1234		
.080		-.0752			.0797	.0702 .0681 .0248
.100					-.0432	
.120			.0646	.0393		
.140		-.0508			-.2106	-.2213 -.1989 -.2192
.160			.0000			
.180				-.1309		
.200			.0110			
.220		.0304			-.3020	-.3417 -.3522 -.3657
.240				-.2277		
.260			-.2084			
.280		-.0089			-.2955	-.3400 -.3364
.300			-.2947			
.320		.0000			-.2378	-.2575
.340			.0000			
.360						-.2172
.380		-.1542			-.1753	-.1908
.400				-.1348		
.420					-.0112	-.0247
.440			-.1545			
.460				-.0386	-.0205	
.480			-.0376			
.500					-.0189	.0206 .0444
.520		-.1046		-.0360		
.540						
.560		-.0401			-.0025	.0468 .0620
.580		-.0415		-.0312	.0468	.0669
.600				-.0004		
.620				-.0336		
.640		-.0688				



(RB1034)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+712+S12N25-AT11 UPPER WING

ALPHA( 3 ) = -.310 BETA( 1 ) = -8.030

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
.2993	.3640	.4270	.5340 .6730 .7800 .8870
X/CW			
.000	.0517	.1177	.3808 .5446 .5264 .4912 .5144
.020			.0376 -.0027 -.0202 -.0246
.025		.0741	
.040		.1420	
.045			
.050	.0003	-.0590	-.2469 -.2928 -.2926 -.3304
.080			-.4207
.085		-.2421	
.086		.1037	
.094	.0064		-.5344 -.5776 -.5742 -.5333
.150	.0000		
.163		-.3980	
.177		-.1648	
.193			
.229	.0388		-.5134 -.6037 -.6604 -.6809
.250		-.3975	
.274			
.339		-.3524	
.362	-.1115		-.5124 -.4037 -.4839
.400		-.3050	
.402			
.497	.0700		-.1494 -.2177
.550		.0000	
.563			-.2431
.600			-.1911
.650			
.700	-.0146		-.1399
.725		-.0290	.0447 -.0135
.750			
.760		-.0237	.0842 .0606
.775		.0802	
.806			
.834	.0244		.0815 .0957 .1023
.850		.0749	
.857			
.865	.1037		
.900	.0791		.0974 .1116 .0951
.905		.0705	.1116
.950		.0886	.1187
.955		.0522	
.965	.0183		

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(881094)

ARC11-716 1A14 01+112+312N25+AT11 UPPER WING

ALPHA( 3 ) = -.323 BETA( 2 ) = -4.010

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
.2990	.3640	.4270	.5340 .6750 .7800 .8870
X/CW			
.0000	.0013	.3352	.4801 .4499 .4071 .4317
.020		.0931	.0892 .0300 .0239 .0346
.025			
.040	.1208		
.045			
.050			
.050	-.0337	-.0270	-.1889 -.2321 -.2449 -.2875
.080			-.3531
.085		-.1926	
.086	.1033		
.094			
.150	-.0227		-.4928 -.5284 -.5359 -.4947
.163	.0000		
.177		.3639	
.193	-.1347		
.229	.0276		-.4987 -.5828 -.6264 -.6435
.250			
.274		-.3810	
.339	-.3476		
.362	-.1027		-.3391 -.4200 -.4505
.400			
.402		-.3278	
.497	.0000		
.550		-.1359 -.2440	
.585		.0000	-.2435
.600			-.1703
.650			-.1364
.700	-.0648		-.0666
.725			.0292 -.0180
.750		-.0643	
.760		.0452 .0353	
.775		.0471	
.800			
.834	-.0218	.0503 .0720 .0662	
.850		.0446	
.857			
.865	.0345		.0690
.900	.0394	.0666 .0988	
.905		.0455 .0690	.1045
.950			
.953		.0482	
.985	-.8147		

DATE 10 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

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ARC11-716 IA14 01+T12+S12N25+AT11 UPPER WING

(R81U34)

ALPHA3( 3) = -.320 BETA0 ( 3) = .030

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CM

.000	-.0499	.0406	.2814	.4017	.3532	.3088	.3423
.020				.1067	.0536	.0441	.0210
.025		.0358					
.040							
.045							
.050	-.0546		-.0598	-.1434	-.1817	-.2016	-.2468
.060				-.2967			
.085			-.1590				
.086		.0958					
.094	-.0386						
.150		.0200		-.4393	-.4633	-.4764	-.4302
.163							
.177			-.3213				
.193							
.229	.0203		-.1171				
.250				-.4560	-.5282	-.5071	-.5951
.274			-.3585				
.339		-.3248					
.362	-.0921						
.400				-.3303	-.4145		-.4091
.402		-.3335					
.497	.0000			-.2049	-.2572		
.550		.0800					
.565							
.600							
.650							-.2338
.700	-.1011			-.1792			
.725				-.1308			
.750							
.760							
.775							
.808							
.834	-.0540						
.850							
.857							
.865	.0106						
.900	.0013						
.905							
.930							
.933							
.965	-.0356						

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TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01-712-S12N5+AT11 UPPER WING

(R81034)

ALPHA( 3 ) = -.320 BETA( 4 ) = 4.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7600
.6670		
.009	-.1203	-.0209
.020	.2153	.3223
.025	.2689	.2276
.040	.0985	.0373
.045	.0286	.0053
.050	.0687	
.060	.3445	
.065	-.0143	
.080	-.1253	-.1626
.085	-.1904	-.2431
.086	-.2655	
.094	-.1427	
.190	.0588	
.185	-.0645	
.177	-.4045	-.4324
.193	-.4517	-.4351
.229	.0000	
.274	-.2961	
.339	-.1054	
.362	-.3429	
.400	-.3098	
.402	-.0863	
.497	-.3597	-.4211
.530	-.3958	
.565	-.3521	
.600	-.2497	-.2776
.650	.0000	
.700	-.2355	
.725	-.1925	
.750	-.1549	
.775	-.1212	
.808	-.0046	-.0176
.834	-.1246	
.850	-.0139	.0002
.857	-.0260	
.868	.0010	.0384
.900	.0646	
.905	-.0342	
.920	-.0242	.0679
.933	-.0164	.0283
.965	-.0195	
.989	-.0099	

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01+T12+S12N23+AT11 UPPER WING

(RB1034)

ALPHA( 3 ) = -.320 BETA( 5 ) = 0.110

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1834	-.0358	.0865	.2595	.1864	.1265	.1774
.020				.0775	.0363	.0393	.0053
.025			.0087				
.040		-.0100					
.045			-.0269				
.050	-.1195			-.1151	-.1311	-.1561	-.2214
.060				-.2326			
.065			-.1022				
.066		.0155					
.094	-.0926						
.150				-.3666	-.3910	-.3900	-.4090
.163		.0071					
.177		-.0804	-.2400				
.193							
.229	-.0145			-.4238	-.4835	-.5087	-.5328
.250			-.3097				
.274							
.339		-.2754					
.362	-.0780						
.400				-.3640	-.4170		-.3890
.402			-.3375				
.497	.0000			-.2637	-.2922		
.530			.0000				
.563							
.605							
.650							
.700	-.1832			-.1399			
.725					-.1794		
.750							
.760							
.775							
.806							
.834	-.1248						
.850							
.857							
.865	-.0437						
.920	-.0613						
.925				-.0138			.0751
.950				-.0352	.0486		
.953				-.0087		.0737	
.965	-.0765			-.0352			

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 OR-T12-S12N25-AT11 UPPER WING (R81034)

ALPHA(4) = 4.000 BETA(1) = -8.050

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7800 .8870
.2039	.0255	.3230	.5278 .4849 .4116 .3769
.020			-.3418 -.5001 -.6235 -.7206
.025		-.1741	
.040	.0387		
.045		-.3205	
.050	-.0618	-.6204	-.7701 -.8571 -.9449
.060		-.7809	
.085		-.5134	
.096	-.0139		
.094	-.2480		
.190			-.7966 -.8206 -.9565 -.8920
.163	.0000		
.177	-.6036		
.193	-.3340		
.229	-.0103		
.250		-.6638	-.8346 -.9127 -.9621
.274	-.5330		
.339	-.4741		
.362	-.2110		
.400		-.3870	-.5013
.402		-.3625	
.497	.0000		
.590		-.1729	-.2492
.565		.0000	
.600			-.3115
.630			-.1750
.700	-.0905	-.1422	
.725		-.0256	
.750			.0257 -.0420
.780		-.0113	
.775		.0636	.0593
.806		.0714	
.834	.0895		
.890		.0617	.0902 .0910
.857		.0687	
.865	.1201		
.900	.1032	.0978	.1147 .0606
.905		.0648	
.910		.0931	.1104
.933		.0502	
.965	.0495		

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TABULATED PRESSURE DATA - IA14A - VOL. 6

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(RB1034)

ARC11-716 IA14 OL+T12+S12N25+AT11 UPPER WING

ALPHAO (4) = 4.000 BETA0 (2) = -3.980

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3840	.4270
.3340	.6730	.7800
.8870		
.000	-.0508	.2927
.020		.4931
.040		.4539
.060		.3827
.080		.3490
.100		-.2698
.120		-.4136
.140		-.5334
.160		-.6468
.180		
.200		
.220		
.240		
.260		
.280		
.300		
.320		
.340		
.360		
.380		
.400		
.420		
.440		
.460		
.480		
.500		
.520		
.540		
.560		
.580		
.600		
.620		
.640		
.660		
.680		
.700		
.720		
.740		
.760		
.780		
.800		
.820		
.840		
.860		
.880		
.900		
.920		
.940		
.960		
.980		
.000		

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(RB1034)

ARC11-716 1A14 CR+T12+312M29+AT11 UPPER WING

ALPHA( 4) = 4.000 BETA( 3) = .030

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7000 .8670
X/CW					
.000	-.1129	-.0497	.2301	.4476	.4123 .3410 .3146
.025			-.2145	-.3472	-.4467 -.5597
.040		.0165	-.1026		
.045			-.2216		
.050	-.1084		-.4598	-.5917	-.6681 -.7660
.060			-.6022		
.065			-.3826		
.066		-.0031			
.094	-.0867				
.150		.0070	-.6628	-.7458	-.7950 -.7603
.163					
.177		-.5004			
.193		-.2616			
.229	-.0255		-.6524	-.7369	-.8192 -.8311
.250			-.4989		
.274		-.4415			
.339					
.362	-.1763		-.4340	-.5040	-.5143
.400					
.402		-.4038			
.437	.0000		-.2376	-.2692	
.550		.0000			
.565					
.600					
.650					
.700	-.1300		-.1316	-.1920	-.2816
.725			-.0961		
.750					
.760		-.0776		.0097	-.0427
.775		.0199	.0094	.0199	
.808					
.834	-.0533		.0236	.0406	.0735
.850		.0160			
.857					
.865	.0168				
.900	.0186		.0450		.0720
.905		.0275	.1045		
.950		.0547			.1031
.955		.0285			
.965	-.0172				



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+512N25+AT11 UPPER WING (R81034)

ALPHA( 4 ) = 3.990 BETA( 4 ) = 4.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.2014	-.1347	.1606	.3792	.3542	.2922	.2791
.020				-.1721	-.3018	-.3917	-.4982
.025							
.040							
.045							
.050							
.060							
.065							
.086							
.094							
.150							
.163							
.177							
.193							
.229							
.250							
.274							
.339							
.362							
.400							
.402							
.497							
.550							
.565							
.600							
.650							
.700							
.725							
.750							
.760							
.775							
.808							
.834							
.850							
.857							
.865							
.900							
.905							
.950							
.955							
.965							

ORIGINAL PAGE IS  
OF POOR QUALITY

ARC11-716 1A14 01+712+512N25+AT11 UPPER WING

ALPHA(4) = 3.900 BETA(5) = 0.140

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7000 .8070
X/CM					
.000	-.2971	-.1617	-.0012	.2731	.3085 .2495 .2451
.020				-.1391	-.2336 -.3034 -.4102
.025				-.1468	
.040				-.0638	
.045				-.1909	
.050	-.1852			-.3237	-.4172 -.4995 -.6074
.060				-.4458	
.065				-.2684	
.086				-.0856	
.094	-.1507			-.5349	-.6097 -.6520 -.6411
.150				.0030	
.163				-.3612	
.177				-.1866	
.193				-.0473	
.229				-.4830	-.6299 -.6790 -.6934
.250				-.3972	
.274				-.3590	
.339				-.1463	
.362				-.3689	-.4840 -.4747
.400				-.3793	
.497	.0000			-.2419	-.3105
.550				.0000	
.565					-.2510
.600					-.2050
.650				-.1180	-.1726
.700	-.1999			-.1482	-.0296 -.0219
.725				-.0767	
.760				-.0536	
.775				-.0003	.0245 .0547
.808	-.1283			.0173	.0376 .0647
.874				-.0412	.0299
.890				-.0376	
.897					
.945	-.0736				
.960	-.0599				
.965					
.990					
.993					
.965	-.0733				



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14A - VOL. 6  
ARC11-716 IAI14 01+712+312N25+AT11 UPPER WING (R81034)

ALPHA( 5 ) = 7.890 BETA( 1 ) = -0.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CN

.000	-.0642	-.1553	.2206	.4143	.2437	.0422	-.0016
.020				-.7616	-1.0790	-1.3940	-1.7240
.025				-.4440			
.040				-.0849			
.045				-.5971			
.050	-.1224			-1.0280	-1.3170	-1.5370	-1.7310
.060					-1.1650		
.065				-.7646			
.066				-.1394			
.094	-.1040						
.130				-1.0530	-1.2600	-1.3480	-1.2970
.163				.0000			
.177				-.7874			
.193				-.4873			
.229	-.0590						
.250				-.9047	-.9843	-1.0680	-1.2440
.274				-.6371			
.339				-.5640			
.362	-.2899						
.403				-.4887	-.5598		-.7295
.402				-.3934			
.497	.0200						
.550				-.2206	-.2731		
.565				.0000			
.600							
.650							
.700	-.0700						
.725				-.0508	-.1437		-.2042
.750							
.760				-.0571	.3434	.0465	
.775				.0285			
.808							
.834	.0402						
.850				.0580	.0871	.0958	
.897				.0439			
.865	-.1502						
.900	.1171			.0643			
.905				.0644	.1101		-.0218
.930							
.953				.0640			.0859
.965				.0683			
.987	.0327						

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OF POOR QUALITY

(RB1054)

ARC11-716 IA14 01+T12+S12N25+AT11 UPPER WING

ALPHAO( 5) = 7.903 BETA0 ( 2) = -4.010

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.8990	.3640	.4270
.8990	.3640	.5340
.8990	.3640	.6730
.8990	.3640	.7800
.8990	.3640	.8870
.8990	.3640	.9940
.8990	.3640	1.1010
.8990	.3640	1.2080
.8990	.3640	1.3150
.8990	.3640	1.4220
.8990	.3640	1.5290
.8990	.3640	1.6360
.8990	.3640	1.7430
.8990	.3640	1.8500
.8990	.3640	1.9570
.8990	.3640	2.0640
.8990	.3640	2.1710
.8990	.3640	2.2780
.8990	.3640	2.3850
.8990	.3640	2.4920
.8990	.3640	2.5990
.8990	.3640	2.7060
.8990	.3640	2.8130
.8990	.3640	2.9200
.8990	.3640	3.0270
.8990	.3640	3.1340
.8990	.3640	3.2410
.8990	.3640	3.3480
.8990	.3640	3.4550
.8990	.3640	3.5620
.8990	.3640	3.6690
.8990	.3640	3.7760
.8990	.3640	3.8830
.8990	.3640	3.9900
.8990	.3640	4.0970
.8990	.3640	4.2040
.8990	.3640	4.3110
.8990	.3640	4.4180
.8990	.3640	4.5250
.8990	.3640	4.6320
.8990	.3640	4.7390
.8990	.3640	4.8460
.8990	.3640	4.9530
.8990	.3640	5.0600
.8990	.3640	5.1670
.8990	.3640	5.2740
.8990	.3640	5.3810
.8990	.3640	5.4880
.8990	.3640	5.5950
.8990	.3640	5.7020
.8990	.3640	5.8090
.8990	.3640	5.9160
.8990	.3640	6.0230
.8990	.3640	6.1300
.8990	.3640	6.2370
.8990	.3640	6.3440
.8990	.3640	6.4510
.8990	.3640	6.5580
.8990	.3640	6.6650
.8990	.3640	6.7720
.8990	.3640	6.8790
.8990	.3640	6.9860
.8990	.3640	7.0930
.8990	.3640	7.2000
.8990	.3640	7.3070
.8990	.3640	7.4140
.8990	.3640	7.5210
.8990	.3640	7.6280
.8990	.3640	7.7350
.8990	.3640	7.8420
.8990	.3640	7.9490
.8990	.3640	8.0560
.8990	.3640	8.1630
.8990	.3640	8.2700
.8990	.3640	8.3770
.8990	.3640	8.4840
.8990	.3640	8.5910
.8990	.3640	8.6980
.8990	.3640	8.8050
.8990	.3640	8.9120
.8990	.3640	9.0190
.8990	.3640	9.1260
.8990	.3640	9.2330
.8990	.3640	9.3400
.8990	.3640	9.4470
.8990	.3640	9.5540
.8990	.3640	9.6610
.8990	.3640	9.7680
.8990	.3640	9.8750
.8990	.3640	9.9820
.8990	.3640	10.0890
.8990	.3640	10.1960
.8990	.3640	10.3030
.8990	.3640	10.4100
.8990	.3640	10.5170
.8990	.3640	10.6240
.8990	.3640	10.7310
.8990	.3640	10.8380
.8990	.3640	10.9450
.8990	.3640	11.0520
.8990	.3640	11.1590
.8990	.3640	11.2660
.8990	.3640	11.3730
.8990	.3640	11.4800
.8990	.3640	11.5870
.8990	.3640	11.6940
.8990	.3640	11.8010
.8990	.3640	11.9080
.8990	.3640	12.0150
.8990	.3640	12.1220
.8990	.3640	12.2290
.8990	.3640	12.3360
.8990	.3640	12.4430
.8990	.3640	12.5500
.8990	.3640	12.6570
.8990	.3640	12.7640
.8990	.3640	12.8710
.8990	.3640	12.9780
.8990	.3640	13.0850
.8990	.3640	13.1920
.8990	.3640	13.2990
.8990	.3640	13.4060
.8990	.3640	13.5130
.8990	.3640	13.6200
.8990	.3640	13.7270
.8990	.3640	13.8340
.8990	.3640	13.9410
.8990	.3640	14.0480
.8990	.3640	14.1550
.8990	.3640	14.2620
.8990	.3640	14.3690
.8990	.3640	14.4760
.8990	.3640	14.5830
.8990	.3640	14.6900
.8990	.3640	14.7970
.8990	.3640	14.9040
.8990	.3640	15.0110
.8990	.3640	15.1180
.8990	.3640	15.2250
.8990	.3640	15.3320
.8990	.3640	15.4390
.8990	.3640	15.5460
.8990	.3640	15.6530
.8990	.3640	15.7600
.8990	.3640	15.8670
.8990	.3640	15.9740
.8990	.3640	16.0810
.8990	.3640	16.1880
.8990	.3640	16.2950
.8990	.3640	16.4020
.8990	.3640	16.5090
.8990	.3640	16.6160
.8990	.3640	16.7230
.8990	.3640	16.8300
.8990	.3640	16.9370
.8990	.3640	17.0440
.8990	.3640	17.1510
.8990	.3640	17.2580
.8990	.3640	17.3650
.8990	.3640	17.4720
.8990	.3640	17.5790
.8990	.3640	17.6860
.8990	.3640	17.7930
.8990	.3640	17.9000
.8990	.3640	18.0070
.8990	.3640	18.1140
.8990	.3640	18.2210
.8990	.3640	18.3280
.8990	.3640	18.4350
.8990	.3640	18.5420
.8990	.3640	18.6490
.8990	.3640	18.7560
.8990	.3640	18.8630
.8990	.3640	18.9700
.8990	.3640	19.0770
.8990	.3640	19.1840
.8990	.3640	19.2910
.8990	.3640	19.3980
.8990	.3640	19.5050
.8990	.3640	19.6120
.8990	.3640	19.7190
.8990	.3640	19.8260
.8990	.3640	19.9330
.8990	.3640	20.0400
.8990	.3640	20.1470
.8990	.3640	20.2540
.8990	.3640	20.3610
.8990	.3640	20.4680
.8990	.3640	20.5750
.8990	.3640	20.6820
.8990	.3640	20.7890
.8990	.3640	20.8960
.8990	.3640	21.0030
.8990	.3640	21.1100
.8990	.3640	21.2170
.8990	.3640	21.3240
.8990	.3640	21.4310
.8990	.3640	21.5380
.8990	.3640	21.6450
.8990	.3640	21.7520
.8990	.3640	21.8590
.8990	.3640	21.9660
.8990	.3640	22.0730
.8990	.3640	22.1800
.8990	.3640	22.2870
.8990	.3640	22.3940
.8990	.3640	22.5010
.8990	.3640	22.6080
.8990	.3640	22.7150
.8990	.3640	22.8220
.8990	.3640	22.9290
.8990	.3640	23.0360
.8990	.3640	23.1430
.8990	.3640	23.2500
.8990	.3640	23.3570
.8990	.3640	23.4640
.8990	.3640	23.5710
.8990	.3640	23.6780
.8990	.3640	23.7850
.8990	.3640	23.8920
.8990	.3640	24.0000



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81U34)

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

ALPHA( 5) = 7.910 BETA( 3) = .040

SECTION ( UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2018	-.1885	.1572	.3848	.2621	.0728	-.0736
.020				-.5402	-.8178	-1.0950	-1.3950
.025				-.3986			
.040				-.0831			
.045				-.4410			
.050	-.1649			-.7617	-1.0180	-1.2160	-1.4320
.060				-.9085			
.085				-.5902			
.095				-.1072			
.094	-.1302						
.150				.0000	-.8722	-1.0500	-1.1330
.163							
.177				-.6631			
.193				-.3058			
.229	-.0688						
.250				-.7558	-.8995	-1.0310	-1.0750
.274				-.6027			
.339				-.5321			
.362	-.2548						
.400				-.4613	-.5638		-.6332
.402				-.4470			
.497	.0000			-.2370	-.3079		
.530				.0000			
.565							
.600							
.650							
.700	-.1582			-.0654	-.1178	-.2090	-.3509
.725							
.750				-.0646	.0272	.0152	
.760				.0060			
.775							
.808							
.834	-.0336			.0460	.0655	.0553	
.850							
.857				.0129			
.865	.0330						
.900	.0422			.0594	.0963		.0348
.905				.0257			
.950				.0728		.0909	
.955				.0310			
.965	.0045						

ORIGINAL PAGE IS  
OF POOR QUALITY



ARC11-716 IAI4 01+712+S12N25+AT11 UPPER WING

ALPHA( S) = 7.900 BETA( 4) = 4.100

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3840	.4270	.5340	.6730	.7800	.8870
X/CW							
.500	-.3232	-.3187	.0511	.3384	.2537	.0842	-.0318
.020				-.4775	-.7040	-.9609	-1.2350
.025				-.3081			
.040				-.1269			
.045				-.4131			
.050	-.1894			-.6891	-.9187	-1.0790	-1.2960
.080				-.7999			
.085				-.5352			
.086				-.1305			
.094	-.1460			-.8077	-.9614	-1.0390	-1.0070
.150				.0000			
.163				-.3957			
.177				-.3337			
.193							
.229	-.0882			-.7170	-.8584	-.9588	-.9919
.250				-.5808			
.274							
.339	-.2352			-.4640	-.5501		-.5935
.362				-.4495			
.400				-.2784	-.3156		
.497	.0000			.0000			
.550							
.565							
.600							
.650							
.700	-.1895			-.1058	-.1448	-.2111	-.3325
.725							
.750				-.1017			-.0787
.760				-.0081	.0053		
.775				-.0282			
.808							
.834	-.0812			.0149	.0510	.0614	
.850				-.0149			
.857							
.865	-.0885			.0399	.0658	.0442	
.900	-.0180			-.0053	.0476	.0940	
.925							
.950				.0012			
.953							
.965	-.0337						



DATE 10 JAN 73

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2867

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

(R81034)

ALPHA(5) = 7.880 BETA(5) = 8.170

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .299 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.4262 -.2711 -.1771 .1749 .2560 .1136 -.0019  
.020 .020

.025 .025 -.3027

.040 .040 -.1402

.045 .045 -.3419

.050 .050 -.2424

.080 .080 -.5733 -.7548 -.9049 -1.1200

.085 .085 -.6575

.086 .086 -.3941

.094 .094 -.1426

.150 .150 -.2401

.163 .163 .0000

.177 .177 -.4484

.193 .193 -.2806

.229 .229 -.1240

.274 .274 -.4523

.339 .339 -.4156

.362 .362 -.2015

.400 .400 -.4051

.492 .492 .0000

.550 .550 -.2810 -.3180

.565 .565 .0000

.600 .600 .0000

.650 .650 -.2103

.700 .700 -.1696

.725 .725 -.1331

.750 .750 -.0101 -.0560

.750 .750 -.1492

.775 .775 -.0649 -.0202

.800 .800 -.0015

.834 .834 -.1205

.850 .850 -.0321 .0285 .0521

.857 .857 -.0662

.865 .865 -.0794

.900 .900 -.0609

.905 .905 -.0134 .0629 .0597

.950 .950 -.0488

.953 .953 -.0001 .0833

.965 .965 -.0413

.965 .965 -.0743

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+SIENES+AT11 UPPER WING

PARAMETRIC DATA

MACH = .750 ELEVON = .000  
RUDDER = .000 SPOILER = .000

REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5600 INCHES  
LREF = 38.7090 INCHES YMRP = .0000 INCHES  
BREF = 38.7090 INCHES ZMRP = .0000 INCHES  
SCALE = .0300 SCALE

ALPHA(1) = -8.030 BETA(1) = -8.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2950	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0390	.0430	.2463	.2527	.1107	-.0013	.0585
.020			.3810	.4912	.4700	.4893	.4620
.025		.2111					
.040			.3109				
.045				.2727	.2898	.2964	.2544
.050	.0371			.1047			
.060			.1689				
.085		.2695					
.086							
.094	.0915			-.1127	-.0843	-.0816	-.0904
.150		.0050					
.163			-.0491				
.177		.1252					
.193	.1540			-.2560	-.2537	-.2980	-.3343
.229			-.1347				
.250							
.274							
.339	.0959	-.1152		-.2055	-.2559		-.2939
.362							
.400			-.1526				
.402			.0000				
.497	.0000			-.1170	-.1536		
.550							
.565							
.600							
.650	.0346						
.700			-.0486				
.725				-.1247			
.750							
.760			-.0225				
.775			.0907	.0914			
.808			.1053				
.834	.0386						
.850			.0677	.1104	.1172		
.857			.0026				
.863	.1242						
.900	.0865		.0737				.0957
.905			.0686				
.905							.1814



## DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1055)

ARC11-716 1A14 01+712-5'2N23+AT11 UPPER WING

ALPHA(1) = -8.050 BETA(1) = -8.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.950 .0525 .1214

.953

.0337

.965

-.0027

ALPHA(1) = -8.040 BETA(2) = -3.990

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 .1374 .1303 -.0334 -.1496 -.0775

.020 .4703 .4360 .4503 .4067

.025

.3545

.040

.1593

.045

.3111

.050

.0498 .2845 .2962 .2918 .2372

.060

.1887

.085

.2389

.086

.0606

.150 .0000 -.0853 -.0617 -.0548 -.0880

.177

-.0223

.193

.1356

.229

.1344

.250 .2148 .2340 .2624 .3089

.274

-.1249

.339

-.1049

.362

.1022

.400 .1915 .2570 .3081

.402

-.1765

.497

.0000

.550 .1290 .1846

.600

.0000

.650 .2057

.700

-.1504

.725

-.0667

.750 .0663 .0230

.760

-.0672

.775

.0726

.808 .0665

.834

.0677

.850 .0132

.9605

.0853

.9903

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01035)

ARC11-716 1A14 01-112-512N25+AT11 UPPER WING

ALPHA( 1 ) = -0.040 BETA( 2 ) = -3.990

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

ALPHA( 1 ) = -0.040 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW			.0438				
	.837						
	.865	.0789					
	.909	.0459	.0671			.0783	
	.905		.0596	.0912			
	.930		.0504		.1062		
	.933		.0106				
	.965	-.0311					



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(M81035)

ARC11-716 IA14 01+112+312N25+AT11 UPPER WING

ALPHA( 1 ) = -0.040 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.775			.0153	.0346		
	.808		.0225				
	.834	-.0629					
	.850			.0013	.0383	.0755	
	.857		.0113				
	.865	.0297					.0680
	.900	.0029		.015-	.0706		
	.905		.0060				
	.990		.0027			.0894	
	.993		-.0185				
	.965	-.0589					

ALPHA( 1 ) = -0.050 BETA( 4 ) = 4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000						
	.020	-.0261	-.0047	.0654	-.0746	-.2933	-.4186
	.025				.3992	.3180	.3290
	.040		.0932	.2734			.2731
	.045			.2544			
	.050	-.0012			.2806	.2420	.2241
	.080				.1424		.1732
	.085			.1859			
	.086		.1802				
	.094	.0173			-.0550	-.0589	-.0538
	.150		.0000				-.1071
	.163			.0134			
	.177		.1318				
	.193						
	.229	.0983					
	.250				-.1780	-.2247	-.2610
	.274						-.3086
	.339		-.0916				
	.362	.0949					
	.400				-.2129	-.2983	-.3333
	.402		-.2131				
	.497	.0000					
	.530				-.1857	-.2472	
	.565		.0000				
	.600						-.2328
	.650						-.2034

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01\*712\*512N25\*AT11 UPPER WING (R81035)

ALPHA( 1 ) = -0.030 BETA( 4 ) = 4.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.700	-.1009			-.1751			
.725				-.1183		.0159	-.0133
.750							
.760			-.1437		.0153	.0262	
.775							
.808			-.0102				
.834		-.1054		.0124	.0337	.0589	
.850			-.0173				
.865		-.0185					.0368
.900		-.0276		.0278	.0431		
.905			-.0149				.0691
.950				.0198			
.955			-.0421				
.965		-.0744					

ALPHA( 1 ) = -0.060 BETA( 5 ) = 6.150

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.0684	-.0207	.0307	-.1316	-.3770	-.5135	-.4313
.020				.3336	.2630	.2639	.2022
.025			.2403				
.040		.0598					
.045			.2267		.2383	.2095	.1902
.050	-.0306			.1343			.1345
.060			.1695				
.066		.1439					
.094	-.0126						
.150		.0000		-.0510	-.0910	-.0611	-.1800
.163			.0108				
.177		.1194					
.193							
.229	.0737						
.250				-.1591	-.2228	-.2501	-.3074
.274			-.1103				
.339		-.1009					
.368	.0605						
.400				-.2796	-.3066		-.3319
.402			-.2412				
.497	.0008						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01035)

ARC11-710 1A14 01+712+912N29+AT11 UPPER WING

ALPHAO ( 1 ) = -0.060 BETAO ( 5 ) = 0.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.806						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.955						
.965						

ALPHAO ( 2 ) = -4.070 BETAO ( 1 ) = -0.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000						
.020						
.025						
.040						
.045						
.050						
.080						
.085						
.086						
.094						
.150						
.165						
.177						
.193						
.229						
.250						
.274						
.339						



(RBIUS5)

TABULATED PRESSURE DATA - IA14A - VOL. 6

ARC11-716 IA14 OI+T12+S12N25+AT11 UPPER WING

DATE 10 JAN 75

ALPHAO ( 2 ) = -4.970 BETAO ( 1 ) = -0.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.362 -.0030  
.400  
.402  
.497 .0900  
.550  
.565  
.600  
.650  
.700 .0112  
.725  
.750  
.760  
.775  
.808  
.834  
.850  
.837  
.865 .1137  
.900 .0855  
.905  
.950  
.935  
.965 .0031

-.3298 -.3649 -.3961  
-.2608  
-.1849 -.2020  
.0000  
-.1486  
-.1529  
-.0772 .0724 .0111  
.0691 .0786  
.0529 .1020 .1106  
.0781  
.0610 .1017 .1025  
.0678 .0409 .1119  
.0408

-.2282

ALPHAO ( 2 ) = -4.080 BETAO ( 2 ) = -4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 .0167 .0837 .3154 .3746 .2795 .1881 .2177  
.020  
.025  
.040  
.045  
.050 .0124  
.090  
.085  
.1901  
.094 .0184  
.150  
.163  
.177  
.193

.3746 .2795 .1881 .2177  
.3153 .2810 .2934 .2788  
.2342  
.1624  
.0691 .0580 .0371 .0171  
-.1058  
.0107  
-.3088 -.3114 -.3130 -.3102  
-.2078  
.0000  
.0075







DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14A - VOL. 6

(RB1035)

AFC11-716 IAI14 01+112+512N25+AT11 UPPER WING

ALPHAD (2) = -4.580 BETAD (4) = 4.090

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

1/3W .2990 .3540 .4270 .5340 .6730 .7800 .8870

X/CW

.080							
.085							
.086							
.094							
.150							
.163							
.177							
.193							
.229							
.250							
.274							
.339							
.362							
.400							
.402							
.497							
.550							
.565							
.600							
.650							
.700							
.725							
.750							
.760							
.775							
.802							
.834							
.850							
.857							
.865							
.900							
.905							
.950							
.953							
.965							

-.0514

.0315

.1346

-.2368 -.2535 -.2682 -.2964

.0000

-.1436

.0194

-.3400 -.4019 -.4411 -.4867

-.2411

-.2158

-.3193 -.4011 -.4119

-.3116

-.2569 -.2982 -.2498

.0000

-.1470 -.1797 -.2193

-.1449

-.0081 .0008 .0022 -.0200

-.0231

.0013 .0329 .0578

-.0261

.0162 .0578 .0709

-.0200

.0120 .0731

-.0342

-.0702

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U35)

ARC11-716 1A14 01-118-812N25-AT11 UPPER KING

ALPHA(2) = -4.090 BETA(5) = 8.150

SECTION (1) UPPER KING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	-.1378	-.0070	.0747	.1324	-.0356	-.1507	-.1151
.020				.2396	.1876	.1996	.1542
.025			.1366				
.040		.0411					
.045			.1124				
.050	-.0728			.0641	.0646	.0426	-.0150
.060				-.0330			
.065			.0616				
.066		.0824					
.094	-.0495			-.2075	-.2299	-.2352	-.2821
.130		.0090					
.163			-.1111				
.177		.0290					
.193	.0326			-.2664	-.3730	-.4257	-.4617
.229			-.2202				
.250		-.1948					
.274	.0342			-.2030	-.4151		-.4072
.339			-.3196				
.362	.0000			-.2383	-.3163		-.2433
.400			.0000			-.2296	
.402					-.1861		
.497				-.1197		-.0094	-.0301
.590			-.1770	-.0113	-.0283		
.565			-.0592				
.600							
.650							
.700	-.2019						
.725							
.750							
.760							
.775							
.808							
.834	-.1466			.0019	.0118	.0416	
.890			-.0527				
.857							
.865	-.0630			.0242			.0672
.903	-.0631		-.0440	.0404			
.905				.0235		.0634	
.950			-.0488				
.953							
.965	-.0060						



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TABULATED PRESSURE DATA - IA14A - VOL. 6

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ARC11-716 IA14 01+T12+S12N25+AT11 UPPER WING

(881055)

A\_PMAO( 3 ) = -.310 BETAD ( 1 ) = -8.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0479	.1251	.4091	.5778	.5557	.5137	.5223
.020			.0483	.0011	-.0357	-.0593	
.025		.0849					
.040		.1527					
.045			-.0462				
.050	-.0034			-.2497	-.3127	-.3278	-.3998
.080				-.4638			
.085			-.2457				
.086		.1258					
.094	.0063						
.150		.0000		-.6081	-.6671	-.6903	-.6371
.163			-.4484				
.177		-.1600					
.193							
.229	.0547			-.6882	-.7237	-.8366	-.8510
.230			-.4622				
.274		-.4102					
.339							
.362	-.0970						
.400				-.4001	-.4427		-.4569
.402		-.3385					
.497	.0000			-.2047	-.2270		
.550			.0000				
.565							
.600							
.630							
.700	-.0140			-.0672	-.1526	-.1575	-.2719
.725							
.750							
.760			-.0066	.0740	.0827	.0683	-.0055
.775			.1018				
.808							
.834	.0302			.0019	.1099	.1125	
.850			.0892				
.857							
.865	.1273						
.900	.1020			.0689			.1063
.905		.0778			.1228		
.950			.0543	.0935		.1238	
.953							
.965	.0276						

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U35)

ARC11-716 1A14 01+T12+312N25+AT11 UPPER WING

ALPHA( 3) = -.320 BETA( 2) = -4.010

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/W	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.0030	.0875	.3612	.5109	.4798	.4281	.4416	
.020			.0916	.0349	.0028	-.0376		
.040		.1342						
.060	.045	-.0055						
.080	.050	-.0286		-.1903	-.2467	-.2760	-.3491	
.100	.085		-.1946	-.3783				
.120	.086	.1185						
.140	.094	-.0153		-.5436	-.5964	-.6080	-.5891	
.160	.150		.0000					
.180	.163		-.1302	-.3973				
.200	.177							
.220	.193	.0384		-.6472	-.6706	-.7646	-.7934	
.240	.229		-.4359					
.260	.274		-.3885					
.280	.339	-.0938		-.4307	-.4665		-.4638	
.300	.362			-.2526	-.2596			
.320	.400	.0000		.0000				
.340	.497					-.1777		
.360	.550				-.1474			
.380	.565					.0496	-.0086	
.400	.650	-.0442		-.1133				
.420	.700							
.440	.725							
.460	.750							
.480	.760			-.0590	.0372	.0599		
.500	.775			.0598				
.520	.808							
.540	.834	-.0200		.0339	.0476	.1030		
.560	.857			.0558				
.580	.865	.0899						
.600	.900	.0526		.0459			.1021	
.620	.905			.0467	.0982			
.640	.950			.0378		.1082		
.660	.955			.0290				
.680	.965	-.0022						



DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 Q1+T12+S12N25+AT11 UPPER WING (RB1035)

ALPHAD(3) = -.330 BETAD(3) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0517	.0445	.3070	.4308	.3866	.3337	.3527
.020				.1181	.0535	.0172	-.0184
.025			.1085				
.040		.1097					
.045			.0100				
.050	-.0570			-.1417	-.2003	-.2309	-.3115
.080				-.3120			
.085			-.1483				
.086		.1079					
.094	-.0364						
.150		.0000		-.4732	-.5263	-.5465	-.5497
.163			-.3464				
.177		-.1039					
.193							
.229	.0329						
.250				-.6224	-.6101	-.7063	-.7363
.274		-.3972					
.339		-.3566					
.362	-.0733			-.4746	-.4821		-.4646
.400			-.3782				
.402							
.497	.0000			-.3241	-.2904		
.550			.0000				
.565							
.600							
.650							
.650							
.700	-.1207						
.725				-.1731			
.750							
.760			-.1006				
.775				-.0179	.0311		
.806			.0184				
.834	-.0670						
.850				-.0122	.0637	.0829	
.857			.0186				
.865	.0136						
.900	.0052			.0053			.0943
.905			.0180		.0017		
.950				.0031			.0971
.953			.0076				
.965	-.0409						





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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01-712+512N25+AT11 UPPER WING

(R81035)

ALPHA(3) = -.320 BETA(5) = 0.130

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW		.2990	.3640	.4270	.5340 .6750 .7000 .8070
X/CW					
.000	-.1901	-.0498	.0766	.2391	.2179 .1502 .1804
.020			.0120	.0820	.0420 .0275 -.0290
.025		.0003			
.040					
.045					
.050	-.1209			-.1021	-.1326 -.1842 -.2718
.080					-.2268
.085				-.0923	
.086			.0171		
.094	-.0838				
.150					
.163		.0000		-.3675	-.4179 -.4384 -.4835
.177					
.193		-.0362			
.229	-.0560				
.250				-.3849	-.5253 -.5916 -.6317
.274			-.3063		
.339		-.2668			
.362	-.0599				
.400				-.3511	-.4911 -.4555
.497	.0000		-.3667		
.550				-.2645	-.3448
.565			.0000		
.600					
.630					
.700	-.2229				
.725				-.1302	-.1915
.750					
.760			-.1712		
.775				-.0304	-.0253
.806			-.0784		
.834	-.1565				
.850				-.0042	.0175 .0482
.857			-.0624		
.865	-.0857				
.900	-.0788			.0098	
.905			-.0475	.0381	.0782
.930				.0115	.0720
.953			-.0480		
.965	-.0873				



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N25+AT11 UPPER WING (R81035)

ALPHA: 4) = 4.020 BETA: (2) = -4.930

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0515	.0143	.3324	.5285	.4807	.3932	.3476
.020				-.2197	-.3307	-.4994	-.6455
.025				-.0909			
.040		.0375					
.045			-.2303				
.050	-.0820			-.5101	-.6640	-.7853	-.9684
.060				-.7202			
.065			-.4298				
.086		.0260					
.094	-.0635			-.8274	-.9927	-1.0700	-1.0210
.150		.0000					
.163			-.8205				
.177		-.2792					
.193							
.229	-.0001			-.8210	-1.0010	-1.1320	-1.1400
.250			-.6192				
.274		-.5459					
.339							
.362	-.1751			.4327	-.5137		-.5303
.400			-.4438				
.402							
.497	.0000			-.2330	-.2668		
.550			.0000				
.565							
.600							
.650							
.700	-.0982						
.725				-.0498			
.750							
.760			-.0319				
.775				.0702	.0643		
.806			.0644				
.834	-.0070						
.850							
.857		.0609					
.865	.0873			.0779	.1037	.1002	
.900	.0752						
.905			.0615	.0839			.0770
.930				.1215			
.953			.0789				.1199
.965	.0245		.0485				

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DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBIUSS)

ARC11-716 1A14 01+T12+312N25+AT11 UPPER WING

ALPHA( 4 ) = 4.020 V( 3 ) = .940

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1158	-.0440	.2779	.4722	.4318	.3511	.3082
.020				-.1618	-.2939	-.4434	-.5907
.025			-.0642				
.040		.0374					
.045			-.1926				
.050	-.1028			-.4327	-.5890	-.6957	-.8768
.080				-.6151			
.085			-.3518				
.086		.0222					
.094	-.0809			-.7307	-.8562	-.9323	-.9241
.150			.0000				
.163			-.5445				
.177		-.2352					
.193							
.229	-.0084			-.7306	-.8977	-1.0360	-1.0180
.250			-.5588				
.274		-.4898					
.339				-.4634	-.5330		-.5026
.362	-.1534						
.400			-.4585				
.452	.0000			-.2602	-.2968		
.497			.0000				-.2966
.550					-.1949		
.585							
.600							
.650							
.700	-.1419			-.0939	-.1420		
.725						.0339	-.0276
.750							
.780			-.0745	.0334	.0418		
.775			.0259				
.808							
.834	-.0537			.0459	.0743	.0923	
.890			.0276				
.857							
.865	.0244			.0426			.0859
.900	.0257						
.905			.0334		.1052		
.950			.0265	.0644		.1112	
.953							
.965	-.0074						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ALPHA(4) = 4.010 BETA(4) = 5.100  
 ARC11-716 1A14 01-T12-S12N25+AT11 UPPER WING (R81033)

SECTION (1) UPPER WING	DEPENDENT VARIABLE C2			
Y/9W	X/CW			
.000	-.2400	-.1737	.1102	.4020
.020			.3524	.2833
.025			-.1414	-.2553
.040		-.1195		-.3608
.045		-.0384		
.050		-.1970		
.055	-.1324		-.3751	-.4835
.060			-.5922	-.7784
.065		-.3202	-.5297	
.066		-.0382		
.094	-.1020			
.150		.0000	-.6286	-.7436
.163				-.8151
.177		-.4928		-.8332
.192		-.1943		
.229	-.0392		-.6135	-.7773
.250			-.9126	-.9129
.274		-.4849		
.339		-.4139		
.362	-.1492		-.4504	-.5679
.400				-.5168
.402		-.4476		
.497	.0000		-.2765	-.3379
.530		.0000		
.565				-.2891
.600			-.2162	
.650			-.1699	
.700	-.2162		-.1134	
.725				.0163
.750		-.1308		-.0215
.760		.0038	.0121	
.775		-.0435		
.808				
.834	-.1288		.0240	.0498
.850		-.0330		.0774
.857				
.861	-.0591			
.900	-.0413		.0383	.0803
.905		-.0253	.0782	
.950			.0423	.1005
.953		-.0245		
.965	-.0596			

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+S12N2S+AT11 UPPER WING

(R81U33)

ALPHAO(4) = 4.000 BETAO(5) = 8.180

SECTION 1 UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3008	-.1313	-.0112	.2431	.3219	.2490	.2316
.020				-.1495	-.2113	-.3087	-.4318
.025			-.1272				
.040		-.0632					
.045			-.1675				
.050	-.1789			-.3132	-.4187	-.5213	-.7066
.080				-.4399			
.085			-.2425				
.086		-.0568					
.094	-.1653						
.150				-.5343	-.6480	-.7294	-.7649
.163		.0000					
.177			-.3528				
.193		-.1782					
.229	-.0332			-.5129	-.6971	-.8188	-.8410
.250			-.4037				
.274		-.3661					
.339				-.4051	-.5396		-.5158
.362	-.1282						
.400			-.4183				
.432				-.2682	-.3525		
.497	.0000			.0000			
.550							
.565							
.600							
.650							
.700	-.2400			-.1803	-.1889		
.725							
.750			-.1854				
.760				-.0302	-.0155		
.775			-.0884				
.808							
.834	-.1608			-.0075	.0287	.0617	
.850			-.0762				
.857							
.865	-.0829						
.900	-.0713			.0095			.0834
.905			-.0567	.0521			
.950				.0150		.0072	
.953			-.0488				
.965	-.0793						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI4A - VOL. 6  
 ARC11-716 IAI4 01+112+512N25+AT11 UPPER WING (881035)

ALPHA(5) = 7.930 BETA(1) = -8.040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/W	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.000	-.0480	-.0585 .3089 .5206 .3875 .2234 .1448
.020		-.5755 -.7804 -1.0300 -1.2280
.025		-.3283
.040	-.3271	-.5026
.045		-.8924 -1.1090 -1.2810 -1.4750
.050	-.1111	-1.1150
.060		-.7369
.065		-.5867
.086		-.0819
.094		-.1380 -1.5230 -1.3830 -1.2670
.150		.0000
.163		-.9478
.177		-.4579
.193		-.0298
.229		-.8455
.274		-.14090 -1.3420 -1.0250 -.8768
.339	-.7934	-.6930
.362	-.2721	-.3966
.400		-.4794 -.7723
.402		-.2332 -.4161
.497	.0000	.0000
.550		-.4792
.565		-.1675
.600		-.0747
.650	-.0671	-.2594 -.3633
.700		-.0651
.725		.0296 -.0376
.750		.0417
.760		.0426 .0333 -.1485
.806	.0291	.0701
.834		.0537
.850		.0838
.857		.0509
.865	.1299	-.0476
.900	.1215	-.2360
.905		
.950		
.953		
.965	.0560	

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01035)

ARC11-716 1A14 01+712+512M25+AT11 UPPER WING

ALPHA( 9) = 7.940 BETA( 2) = -4.000

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1212	-.1014	.2747	.4985	.3754	.2017	.5907
.020				-.4994	-.7159	-.9846	-1.2230
.025			-.2762				
.040		-.0313					
.045			-.4354				
.050	-.1403		-.7990	-1.0470	-1.2240	-1.4980	
.060			-1.0390				
.065			-.6438				
.066		-.0714					
.094	-.1048						
.150		.0050		-1.2100	-1.4630	-1.5040	-1.4450
.163			-.0226				
.177		-.4070					
.193							
.229	-.0358			-1.0530	-1.4340	-1.3570	-1.2030
.250			-.7877				
.274		-.7094					
.339							
.362	-.2532			-.4700	-.5014		-.7229
.400							
.402			-.4790				
.497	.0000			-.2134	-.2745		
.550				.0000			
.565							
.600							
.650							
.700	-.1159			-.0606		-.2439	
.725							
.750							
.760			-.0650	.0421	.0398		
.775			.0323				
.806							
.834	-.0013			.0624	.0847	.0036	
.850							
.857			.0515				
.865	.0930						
.900	.0907			.0747			-.0879
.905			.0631	.1133			
.930			.0744			.0599	
.953			.0594				
.965	.0415						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING (R81035)

ALPHA( 5) = 7.940 BETA( 3) = .030

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2008	-.1737	.2089	.4481	.3532	.1790	.0585
.020				-.4353	-.6517	-.9357	-1.1840
.025			-.2490				
.040		-.0543					
.045			-.3842				
.050	-.1647			-.7163	-.9679	-1.1600	-1.4530
.080				-.9269			
.085			-.5569				
.086		-.0720					
.094	-.1202						
.190				-1.0410	-1.3080	-1.4280	-1.4590
.163		.0000					
.177			-.7373				
.193		-.3500					
.229	-.0448			-.9407	-1.2180	-1.3900	-1.3390
.250							
.274		-.7074					
.339		-.6276					
.362	-.2289			-.5096	-.5399		-.6003
.400			-.5151				
.402	.0000			-.2434	-.2893		
.550			.0500				
.565							
.600							
.650							
.700	-.1713			-.0662	-.1322	-.1032	-.3401
.725							
.750							
.760							
.775				-.0766	.0257	.0371	
.800				.0062			
.834	-.0554						
.850				.0453	.0739	.0701	
.857		.0234					
.865	.0383						
.890	.0333			.0583		.1068	.0201
.905			.0395				
.950				.0692		.0968	
.953			.0403				
.965	.0152						

(R81033)

ARC11-716 1A14 01-712-S12N25+AT11 UPPER WING

ALPHA2( 5) = 7.930 BETA2( 4) = 4.110

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CF

Y/W	X/CW	CF
.2990	.3640	.4270
.3340	.6750	.7800
.8870		
.000	-.3369	.0727
.020	-.3110	.3767
.025	.3098	.1665
.040	.0396	-.4051
.045	-.5883	-.8523
.050	-.1260	
.060	-.2728	
.085	-.1090	
.094	-.3762	
.150	-.6491	-.8690
.163	-1.0670	-1.3780
.177	-.8271	
.193	-.5182	
.229	-.1037	
.250	-.1399	
.274	-.8677	-1.1230
.339	-.2900	-1.3780
.362	.0000	
.402	-.6515	
.402	-.3144	
.497	-.0687	
.530	-.229	
.565	-.274	
.600	-.362	
.650	-.5408	
.700	-.2130	
.725	-.402	
.750	-.5291	
.775	-.5529	-.5778
.800	-.3062	-.3131
.834	.0000	
.850	-.3062	-.3131
.865	-.2061	
.887	-.1256	-.1561
.900	-.1206	
.905	-.0185	.0197
.933	-.0262	
.965	-.0062	.8573
.990	-.0120	.0705
.995	-.0877	
.999	-.0217	
.999	-.0249	.0692
.999	-.0012	.0471
.999	.0297	.0961
.999	.0001	
.999	-.0366	



DATE 19 JAN 72 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 OL+T12+S12N25+AT11 UPPER JING  
 (RB1035)

ALPHA (S) = 7.920 BETA (S) = 8.180

SECTION 1 UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.4261	-.2311	-.1669	.2015	.2865	.1591	.0414
.020				-.3799	-.4952	-.7239	-1.0230
.025			-.2765				
.040							
.045			-.1319				
.050				-.3131			
.080	-.2331			-.5471	-.7395	-.9320	-1.2260
.085					-.6613		
.086				-.3666			
.094			-.1334				
.229	-.2352						
.150							
.165			.0000				
.177				-.7052	-.9073	-1.0640	-1.1660
.193			-.2696				
.229				-.4505			
.250	-.1200						
.274							
.339			-.4307				
.362	-.1858						
.400				-.4456	-.5866		-.9095
.452							
.497	.0000			-.2742	-.3295		
.550							
.565				.0000			
.600							
.650							
.700	-.2407						
.725				-.1265			
.750					-.1814		
.760							
.775				-.1661			
.808				-.0909			
.834							
.850	-.1500						
.897				-.0736			
.965							
.065	-.0858						
.970	-.0693			.0054			
.985				-.0544			
.990					.0570		.0656
.995				.0180			.0904
.999							
.953				-.0467			
.965	-.0748						

(RB1036) (15 FEB 74)

ARC11-716 IAI14 OL+T12+S12N25+AT11 UPPER WING

PARAMETRIC DATA

MACH = .850 ELEVON = .000  
RUDDER = .000 SPOILER = .000

REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
REF = 38.7090 INCHES YMRP = .0000 INCHES  
GREF = 38.7090 INCHES ZMRP = .0000 INCHES  
SCALE = .0300 SCALE

ALPHA(1) = -8.120 BETAD(1) = -8.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0441	.0700	.2990	.3340	.2169	.1219	.1503
.020				.5101	.0000	.5001	.4689
.040		.2364	.4134				
.045			.3370				
.050	.0927			.2944	.3020	.2984	.2503
.060				.1196			
.085			.1915				
.088		.2942					
.094	.1045						
.150		.0000		-.1044	-.0469	-.0770	-.1063
.163							
.177		.1556	-.0268				
.193							
.229	.1750						
.250				-.2397	-.2626	-.3210	-.3793
.274			-.1345				
.339		-.1052					
.362	.1253						
.400				-.1792	-.2877		-.3950
.402			-.1553				
.497	.0000						
.550				-.0959	-.1735		
.585							
.600			.0000				
.650							-.2213
.690	.0666					-.1412	
.700				-.0289	-.1536		
.725			-.0346			.1106	.0496
.750				.1312	.1097		
.760			.1233				
.775							
.808							
.834	.0334			.0942	.1181	.1276	
.850			.0963				
.857							
.865	.1436						
.900	.1006			.0926			.1059
.905			.0735		.0928		



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI4A - VOL. 6

PAGE 2895

ARC11-716 IAI4 01+T12+S12N25+AT11 UPPER WING

(R81035)

ALPHA(1) = -0.120 BETA(1) = -8.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.950

.953

.965

.0269

.0452

.1204

ALPHA(1) = -0.110 BETA(2) = -3.970

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000

.020

.025

.040

.045

.050

.060

.065

.066

.094

.100

.103

.177

.193

.229

.274

.339

.362

.403

.402

.497

.550

.503

.600

.650

.700

.725

.750

.760

.775

.800

.834

.850

.0075

.0146

.2150

.2274

.0812

-.0226

.0180

.4971

.0000

.4556

.4129

.3834

.1908

.3311

.3032

.3024

.3008

.2337

.1497

.2114

-.0675

-.0524

-.0409

-.1061

-.0025

-.2147

-.2294

-.2856

-.3651

-.1115

-.2012

-.3016

-.3948

-.1716

-.1550

-.2206

.0000

-.0785

-.1664

-.1783

.0776

.0237

-.0904

.0879

.0763

.0721

-.0343

.0845

.1011

(R81U36)

ARC11-716 1A14 Q1+T12+312N25+AT11 UPPER WING

ALPHA(1) = -0.110 BETA(2) = -3.970

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6790	.7800	.8870
X/CW							
.85			.0520				
.865	.0822						
.900	.0464		.0576				.0655
.905		.0387		.0612			
.930			.0297		.0929		
.935		.0027					
.965	-.0335						

ALPHA(1) = -0.070 BETA(3) = .010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6790	.7800	.8870
X/CW							
.900	-.0022	.0042	.1959	.1242	-.0332	-.1689	-.1187
.920				.4646	.0000	.4037	.3494
.925			.3459				
.940		.1905					
.945			.3075		.3017	.2901	.2776
.950	.0323			.1632			.2042
.960			.2154				
.986	.086	.2346					
.994	.0437				-.0440	-.0396	-.0310
.100			.0000				-.1081
.163				.0227			
.177		.1634					
.193	.1326				-.1993	-.2131	-.2668
.229				-.0944			-.3556
.250							
.274							
.339		-.0688					
.362	.1262						
.400				-.2206	-.3172		-.4106
.402		-.1900					
.497	.0000						
.550				-.2084	-.2675		
.565		.0000					
.600							
.650						-.2118	-.2561
.750	-.0536			-.1610			
.725				-.1262			
.750						.0418	.0043
.780				-.1410			



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+512+23+AT11 UPPER WING (R81036)

ALPHA( 1 ) = -8.070 BETA( 3 ) = .010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.775 .0427 .0374

.0225

.808 .0212 .0525 .0714

.834 -.0913

.850 .0060

.857 .0265

.865 .0300 .0469 .0601

.900 .0075 .0926 .0706

.905 .0075

.930 .0285

.933 .0641

.965

ALPHA( 1 ) = -8.080 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.0181 .0232 .1209 .0394 -.1705 -.2976 -.2413

.020 .0272 .0000 .3336 .2729

.025 .3024

.040 .1195

.045 .2791

.050 .0102

.080 .2860 .2500 .2363 .1987

.085 .1676

.086 .2119

.094 .1966

.150 .0299

.163 .0000

.177 .0416

.193 .1625

.229 .1120

.250 .0764

.274 .0592

.339 .1217

.362 .0000

.402 .2906

.497 .0000

.550 .0000

.565 .0000

.600 .0000

.650 .0000

.650 .0000

.650 .0000

.650 .0000

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## DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1036)

ARC11-716 1A14 01+712+312N25+AT11 UPPER WING

ALPHA( 1 ) = -0.030 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.700	-.1230	-.1616	-.1901			
.725				.0197	-.0138	
.750		-.1916				
.760		.0733	.0039			
.775		-.0247				
.808						
.834	-.1467					
.850		-.0276	-.0359	.0277	.0579	
.857						
.865	-.0301					.0507
.900	-.0407		.0046		.0378	
.905		-.0281				
.950			-.0087		.0633	
.953		-.0493				
.965	-.0808					

ALPHA( 1 ) = -0.100 BETA( 5 ) = 8.180

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0619	.0029	.0639	-.0204	-.2869	-.4102	-.3636
.020				.3948	.0000	.2699	.2011
.025			.2767				
.040		.0818					
.045			.2805				
.050	-.0222			.2781	.2327	.1990	.1189
.080			.1995	.1655			
.085		.1642					
.086							
.094	.0017			-.0149	-.0366	-.0499	-.1552
.150		.0000					
.163			.0395				
.177		.1426					
.193							
.229	.0871						
.250				-.1656	-.2757	-.2687	-.3893
.274		-.0770					
.339		-.0679					
.362	.0683						
.400				-.2380	-.3699		-.4679
.402							
.497	.0000			-.2165			



DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 OL+T12+S12N25+AT11 UPPER WING  
 (R81U36)

ALPHA( 1 ) = -8.100 BETA( 1 ) = 0.160

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.550					-.3421	-.3831		
.555				.0000				-.2682
.600							-.2530	
.650						-.2116		
.700		-.2028			-.2035		.0001	-.0156
.725								
.750				-.2369	-.0828	-.0156		
.760								
.775				-.0527				
.808								
.834		-.1795			-.0246	.0141	.0469	
.857				-.0560				
.865		-.0783			-.0038	.0391		.0669
.900		-.0894		-.0440				
.905					-.0177		.0589	
.950				-.0620				
.953								
.965		-.0904						

ALPHA( 2 ) = -3.980 BETA( 1 ) = -8.050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000		.0704	.1549	.4218	.5222	.4537	.3898	.4113
.020					.3288	.0000	.3084	.2915
.025				.2970				
.040			.2260					
.045				.1871				
.050		.0560			.0620	.0473	.0414	-.0208
.060					-.1415			
.085				.0045				
.094		.0592	.2350					
.150						-.3557	-.3753	-.3554
.163			.0000					-.3619
.177				-.2436				
.193			.0120					
.229		.1223						
.250								
.274					-.4740	-.5510	-.6143	-.6488
.339			-.2868		-.3308			

ALPHA( 2 ) = -3.900 BETA( 1 ) = -0.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.362	.0214						
.400							
.402							
.497	.0000						
.550							
.565							
.600							
.650							
.700	.0135						
.725							
.750							
.760							
.775							
.808							
.834	.0214						
.850							
.857							
.865	.1320						
.900	.0943						
.905							
.950							
.953							
.965	.0014						

ALPHA( 2 ) = -3.900 BETA( 2 ) = -3.990

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	.0240	.1046	.3516	.4194	.3367	.2517	.2030
.020							
.025							
.040							
.045							
.050	.0204						
.060							
.065							
.066							
.094	.0333	.2177					
.150							
.163							
.177							
.193							



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI4A - VOL. 6

(R81036)

ARC11-7.5 IAI4 OL+T12+S12N25+AT11 UPPER WING

ALPHA( 2 ) = -3.980 BETA( 2 ) = -3.990

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW .2900 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.229	.1044					
.250						
.274						
.339						
.362						
.400						
.402						
.497						
.530						
.565						
.600						
.630						
.700						
.725						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.953						
.965						

ALPHA( 2 ) = -3.990 BETA( 3 ) = .060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW .2900 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000						
.020						
.025						
.040						
.045						
.050						
.080						
.085						
.086						
.094						

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81036)

ARC11-716 1A14 01-T12-S12N25+AT11 UPPER WING

ALPHA( 2 ) = -3.990 BETA( 3 ) = .060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.150	.0900	-.2382	-.2546	-.2658	-.3207	
.183	.177	-.1489					
.193	.0902	.0510					
.229			-.3669	-.4068	-.4987	-.5960	
.230			-.2448				
.274		-.2080					
.339	.0437						
.362			-.3344	-.4628		-.6419	
.400		-.3086					
.497	.0900		-.2784	-.3315			
.550		.0000					
.565							-.2646
.600							
.690							
.700	-.1093		-.1721				
.725			-.1285				
.750							
.780		-.1342	.0406	.0430		.0921	.0024
.775		.0185					
.808							
.834	-.0964		.0293	.0380	.0026		
.850		.0076					
.857							
.865	.0086						.0082
.900	-.0084		.0360	.0556			
.905		.0086	.0226	.0741			
.950							
.953		-.0128					
.965	-.0579						

ALPHA( 2 ) = -3.990 BETA( 4 ) = 4.070

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	-.0479	.0503	.2379	.2536	.1110	.0196
.020				.3010	.0000	.2402	.1932
.025			.2245				
.040		.1146					
.045			.1671				
.050	-.0238			.1096	.0063	.0624	-.0184



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512H25+AT11 UPPER WING (R81036)

ALPHA(2) = -3.990 BETA(4) = 4.070

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

X/CW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW .080 .085 .086 .094 .150 .163 .177 .193 .229 .250 .274 .339 .362 .403 .497 .550 .565 .600 .650 .700 .725 .750 .775 .800 .834 .850 .857 .865 .900 .905 .950 .953 .965

-.0230  
 .0608  
 .1566  
 -.0034  
 .0900  
 -.1195  
 .0486  
 .0739  
 -.2225  
 -.1876  
 .0358  
 .403  
 .497  
 .550  
 .565  
 .600  
 .650  
 .700  
 .725  
 .750  
 .775  
 .800  
 .834  
 .850  
 .857  
 .865  
 .900  
 .905  
 .950  
 .953  
 .965  
 -.2104  
 -.2360  
 -.2484  
 -.3328  
 -.3121  
 -.3917  
 -.4685  
 -.5803  
 -.3252  
 -.4776  
 -.3156  
 -.3257  
 -.4379  
 .0000  
 -.2293  
 -.2569  
 -.1870  
 -.1480  
 -.1606  
 .0120  
 .0149  
 -.0320  
 .0130  
 .0406  
 .0707  
 .0226  
 .0460  
 .0186  
 -.0454  
 .0293  
 -.0060  
 .0807  
 .0732

(R81036)

ARC11-716 IA14 01-T12-S12N25-AT11 UPPER WING

ALPHA2( 2) = -4.000 BETA2( 5) = 0.110

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7800 .8870
X/CM					
.000	-.0935	.0138	.0949	.1756	.0321 -.0892 -.0359
.020				.2715	.0000 .2123 .1524
.025			.1472		
.040		.0519			
.045			.1269		
.050	-.0380			.1213	.0997 .0592 -.0249
.080				.0074	
.085			.0759		
.086		.0976			
.094	-.0347				
.150				-.1654	-.2046 -.2205 -.3112
.163		.0000			
.177			-.0736		
.33		.0519			
.229	.0493			-.2904	-.3587 -.4326 -.5542
.250			-.1861		
.274		-.1607			
.339				-.3445	-.4770 -.6713
.362	.0255				
.430			-.2991		
.402				-.4048	-.5307
.497	.0000				
.550			.0000		
.565					-.2330
.650				-.2267	
.700	-.2494			-.2074	
.725			-.2021		
.750				.0216	-.0022
.760			-.2140		
.775			-.0793	-.0690	-.0058
.808					
.834	-.2086			-.0372	.0212 .0598
.850			-.0451		
.857					
.865	-.1062				
.900	-.0907		-.0192		.0020
.905			-.0483	.0363	
.950			-.0220		.0695
.953			-.0602		
.965	-.0905				



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 Q1+T12+312N2+AT11 UPPER WING  
 (R81U36)

SECTION / SURFACE	ALPHA: 31	BETA: 11	DEPENDENT VARIABLE CP
Y/BW	.2990	.3640	.4270 .5340 .6730 .7000 .6670
X/CW	.0510	.1444	.4417 .6050 .5761 .5327 .5321
.020			.1147 .0000 .0225 -.0207
.025		.1779	.1489
.040			.0137
.045	.0045		-.1784 -.2427 -.2754 -.3630
.050			-.3968
.060		-.1883	
.065		.1604	
.094	.0120		-.6164 -.6768 -.6591 -.6341
.130		.0000	
.163		-.4363	
.177		-.1113	
.193	.0623		-.6806 -.7783 -.8904 -.8818
.229		-.4663	
.250		-.4715	
.274			
.339	-.0612		-.3785 -.5619 -.9349
.362		-.3952	
.400			
.432	.0000	-.1471 -.1893	
.497		.0000	-.1842
.550			-.0875
.565		-.0041	
.600			.1086 .0258
.630		.0029	
.700	-.0145	.1300 .1143	
.725		.1186	
.750			
.760		.0992	
.775			
.808	.0367	.1128 .1285 .1352	
.834			
.850			
.857			
.865	.1439	.1087 .1061 .1222	
.900	.1205		
.905			
.950			
.955			
.965	.0359		



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 01+712+512N25+AT11 UPPER WING

(R01038)

ALPHA( 3 ) = -.320 BETA( 2 ) = -3.990

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0028	.1550	.3918	.5312	.4947	.4478	.4477
.020			.1354	.0090	.0398	-.0104	
.040		.1561					
.065			.0337				
.090	-.0189			-.1386	-.1956	-.2323	-.3244
.085				-.3421			
.086		.1489					
.094	-.0593						
.150			.0000	-.5312	-.5973	-.5923	-.6070
.163							
.177			-.3856				
.193		-.0901					
.229	.0650						
.250				-.6363	-.7416	-.8239	-.8484
.274			-.4707				
.339		-.4133					
.362	-.0549						
.400				-.4113	-.5519		-.9330
.402			-.4252				
.497	.0000			-.2283	-.2481		
.550			.0000				
.565							
.600							
.650							
.700	-.0781			-.1548		-.1424	-.1960
.725				-.0569			
.750							
.760			-.0564	.0870	.0827		
.775			.0743				
.808							
.834	-.3241			.0774	.0970	.1143	
.890			.0619				
.897							
.865	.0773						
.900	.0426			.0793	.0796		.1030
.905			.0548				
.930				.0404		.0997	
.933			.0291				
.965	-.0081						



DATE 10 JAN 73

TABULATED PRESSURE DATA - IA14A - VOL. 6

PAGE 2907

(R81036)

ASC11-716 IA14 01+112+512N25+AT11 UPPER WING

ALPHA(3) = -.330 BETA(3) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0537	.0327	.4536	.3950	.3470	.3584	
.020			.1491	.0000	.0539	.0003	
.025		.1245					
.040			.0414				
.045							
.050	-.0470		-.1020	-.1987	-.1979	-.3012	
.060			-.2803				
.085			-.1121				
.086		.1309					
.094	-.0295						
.150			-.4705	-.5195	-.5377	-.5732	
.163		.0000					
.177		-.3288					
.193		-.0726					
.229	.0506						
.250			-.6440	-.6710	-.7656	-.8115	
.274		-.4119					
.339		-.3621					
.352	-.0422						
.400			-.5000	-.5789		-.9470	
.432		-.4405					
.497	.0000		-.3805	-.3539			
.550							
.565		.0000					
.600							
.650							
.700	-.1562				-.1703		-.1962
.725			-.1502	-.1372			
.750					.0708	.0140	
.760		-.1206					
.775			.0111	.0496			
.808		.0211					
.834	-.0870						
.850			.0068	.0764	.1023		
.857		.0176					
.865	.0050						
.900	.0022						.0997
.905		.0191		.0662			
.930			.0081			.0905	
.933		.0004					
.965	-.0300						

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(R01036)

ARC11-716 1A14 OX+T12+312N25+AT11 UPPER WING

ALPHA(3) = -.330 BETA(4) = 8.130

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/Z							
.000	-.1072	-.0333	.0842	.2493	.2443	.1676	.1842
.020				.0955	.0000	.0497	-.0120
.025			.0290				
.040		.0060					
.045			-.0088				
.050	-.1146			-.0655	-.0994	-.1514	-.2568
.080				-.1831			
.085			-.0677				
.086		.0191					
.094	-.0722						
.150				-.3246	-.3864	-.4267	-.5196
.163		.0000					
.177		-.1932					
.193		-.0385					
.229	.0098						
.250				-.4204	-.5062	-.6178	-.7318
.274		-.2813					
.339		-.2463					
.362	-.0369						
.400				-.4204	-.5718		-.8153
.402		-.3587					
.487	.0020						
.550				-.4425	-.5864		
.565		.0000					
.600							-.1735
.650					-.2035		
.700	-.2839			-.2087	-.2097		
.725						.0410	.0131
.750			-.2182				
.760				-.0691	-.0004		
.775			-.1085				
.808							
.834	-.2182			-.0477	.0247	.0663	
.850			-.0744				
.857							
.885	-.1315						.0919
.900	-.1007			-.0310	.0349		
.925			-.0591				
.950				-.0307		.0676	
.953			-.0599				
.965	-.0691						



DATE 10 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

PAGE 2909

(RB1096)

ARC11-716 IA14 01+T12+S12N23+AT11 UPPER WING

ALPHA(4) = 3.010 BETA(1) = -4.010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0501	.0274	.3682	.5691	.5230	.4450	.4042
.020				-.1090	.0000	-.3379	-.4600
.025			-.0167				
.040		.0680					
.045			-.1545				
.050	-.0772			-.3863	-.5192	-.6100	-.7596
.080			-.3470	-.6109			
.085							
.086		.0650					
.094	-.0324			-.8245	-.9426	-.9350	-.9606
.150		.0000					
.163							
.177			-.5756				
.193		-.2140					
.229	.0295			-.9589	-1.0460	-1.0820	-.9182
.250			-.6722				
.274							
.339		-.5921					
.362	-.1255			-.7159	-1.0270		-.7090
.430							
.402			-.6807				
.497	.0000			-.2280	-.2569		
.550			.0000				
.565							
.600							
.650				-.3746			-.4899
.700	-.1126			-.1536			
.725				-.0834			
.750			-.0342		-.1527	-.3117	
.760							
.775			.0300	.0372			
.808			.0616				
.834	-.0181						
.850				.0586	.0855	-.0046	
.857			.0623				
.865	.5545						
.900	.0853			.0472			-.1610
.905			.0597	.1051			
.950				.0419		.0750	
.953			.0443				
.965	.0249						

(R01036)

ARC11-716 1A14 01+712+81223+AT11 UPPER WING

ALPHAO(4) = 3.010 BETA(2) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5140 .6730 .7600 .8670

X/CW

.363	-.1197	-.0397	.3108	.9033	.4597	.3882	.3481
.023			-.0803	.0000	-.3128	-.4488	
.023			-.0124				
.040		.0842					
.745			-.1341				
.030	-.0393		-.3426	-.4757	-.5762	-.7242	
.060			-.5366				
.085			-.3001				
.066		.0568					
.094	-.0639						
.130		.0300		-.7422	-.8639	-.8808	-.9077
.163				-.5094			
.177			-.1888				
.193							
.229	.0207			-.7690	-.9162	-1.0610	-1.0470
.250				-.5976			
.274		-.5407					
.339							
.362	-.1115			-.6199	-.9972		-.7978
.400			-.6159				
.457	.0300			-.3006	-.2696		
.530			.0000				
.565							
.600							
.630							
.700	-.1910				-.1377	-.2068	-.4537
.725				-.0867			
.730						.0138	-.2151
.760			-.0782				
.775			.0353	.0447			
.808			.0210				
.834	-.0798						
.850			.0458	.0745	.0891		
.837			.0310				
.865	.0202						
.900	.0276						
.905			.0328	.0569	.0910	-.0381	
.930				.3492		.1085	
.933			.0205				
.965	-.0103						



DATE 13 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

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ARC11-716 IA14 01+T12+S12N23+AT11 UPPER WING

(881036)

ALPHA(4) = 3.800 BETAO (3) = 4.090

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7600 .8870
Z/CW				
70	-.2120	-.1140	.1760	.4347 .3905 .3164 .2926
.020				-.0779 .0000 -.2958 -.4288
.025		.0045	-.0486	
.040				
.045			-.1394	
.050	-.1126		-.3117	-.4379 -.5361 -.6853
.060			-.4806	
.085			-.2691	
.086		.0145		
.094	-.0777			
.150		.0000	-.6314	-.7425 -.8111 -.8630
.163			-.4407	
.177		-.1614		
.193				
.223	.0029		-.6534	-.7899 -.9459 -1.0260
.250			-.4920	
.274		-.4331		
.339	-.1074			
.362			-.5572	-.8419 -1.0190
.405		-.5449		
.402			-.4312	-.5062
.497	.0000	.0000		
.550				-.3760
.565			-.1615	
.600			-.1756	
.650	-.2655		-.1147	.0574 -.0197
.700			-.1370	
.725			-.0352	.0212 .0428
.750				
.775			-.0241	.0290 .0710 .0961
.800	-.1461			
.824				
.857				
.863	-.3620		.0406	.0817 .0856
.900	-.0133		-.0122	.0363 .1040
.905				
.953			-.0272	
.965	-.0847			

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(8B1036)

ARC11-716 1A14 01+112+312N25+AT111 UPPER WING

ALPHA( 4) = 3.780 BETA( 4) = 8.160

SECTION ( 3) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3340	.4270	.5340	.6730 .7800 .8870
X/CW					
.000	-.2934	-.0938	.0083	.2272	.3516 .2714 .2521
.020				-.1175	.0000 -.2309 -.3498
.025			-.0908		
.040		-.0396			
.045			-.1398		
.050	-.1794			-.2692	-.3500 -.4448 -.5933
.080				-.3890	
.095			-.2020		
.086		-.0370			
.094	-.1652				
.150				-.4990	-.6178 -.6813 -.7786
.163		.0000			
.177		-.1472	-.3158		
.193					
.229	-.0199			-.5475	-.6866 -.8263 -.9314
.250			-.3814		
.274		-.3504			
.339	-.0956			-.5005	-.7379 -1.0210
.362			-.4255		
.400				-.4522	-.6160
.402	.0000		.0000		-.2060
.497				-.1696	
.550				-.2066	
.565				-.1863	.0577 .0404
.600			-.2049	-.0841	.0175
.650	-.2945		-.1244		
.700				-.0429	.0390 .0878
.725			-.0876		
.750					
.760					
.775					
.808					
.834	-.2124				
.850					
.857					
.865	-.1336			-.0298	.0437 .1061
.900	-.1067		-.0665	-.0306	.0876
.905					
.950					
.953			-.0823		
.965	-.0979				



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TAB LISTED PRESSURE DATA - 1A14A - VOL. 6

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(R81038)

ARC11-716 1A14 01+112+512N25+AT11 UPPER WING

ALPHA(5) = 7.940 BETA(1) = .030

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.2004	-.1526	.2590	.5012 .4321 .2898 .1891
.020				-.2801 .0000 -.6688 -.8623
.025			-.1171	
.040		-.0128		
.045			-.2890	
.050	-.1530			-.5473 -.7406 -.8959 -1.1130
.055				-.7492
.085			-.4740	
.086		-.0226		
.094	-.1077			
.150		.0000		-.9542 -1.1360 -1.0980 -.9962
.163				
.177			-.6557	
.193		-.2818		
.229	-.0103			
.250				-1.0040 -1.1990 -.9065 -.9078
.274			-.7373	
.339		-.6471		
.362	-.1663			
.400				-.7685 -.7026 -.6705
.402			-.8289	
.497	.0000			
.530			-.2819 -.5853	
.555			.0000	
.600				-.5703
.650				-.5338
.700	-.1923		-.1597	
.725				-.4322 -.4780
.750			-.1197	
.760				
.775			-.0542 -.1917	
.800			-.0262	
.834	-.1010			
.850			-.0219 -.0873 -.2852	
.857			-.0127	
.865	.0046			
.900	.0274		.0042	-.4233
.905			.0442	
.950			.0259	-.1733
.953			.0119	
.965	-.0051			



(R81036)

ARC11-716 1A14 01+712+512N3+AT11 UPPER WING

ALPHA( 5) = 7.930 BETA( 2) = 4.130

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/W	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.3441	-.2778	.1011	.4061 .3653 .2571 .1529
.020				-.2949 .0000 -.6379 -.8472
.025			-.2134	
.040		-.0738		
.045			-.3102	
.050	-.1688		-.5279	-.7012 -.6534 -1.0860
.090				-.7119
.085			-.4437	
.086		-.0709		
.094	-.1241			
.130				-.8679 -1.0620 -1.1420 -1.1320
.163		.0000		
.177		-.2489	-.6043	
.193				
.229	-.0468			-.6593 -1.1050 -1.1650 -1.0770
.250			-.6619	
.274				
.339		-.5587		
.362	-.1736			
.400			-.7444	-1.0710 -.7775
.402			-.6761	
.497	.0000			
.530			-.3081	-.3972
.569		.0000		
.600				
.650				-.6252
.700	-.2750		-.1426	-.1858
.725				
.750				-.2672 -.4649
.780		-.1385		
.775		-.0463	-.0371	-.0558
.808				
.834	-.1484			
.850			-.0008	.0048 -.1030
.897		-.0287		
.865	-.0366			
.950	-.0406		.0182	.0565 -.3203
.905		-.0207	.0288	.0108
.950				
.993		-.0287		
.995	-.0348			



ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

(R81U36)

ALPHA(2) = 7.910 BETA(3) = 8.190

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/B4	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.4355	-.1795	-.1335	.2265 .3347 .2149 .1189
.020				-.3346 .0000 -.5891 -.8002
.025			-.2459	
.040		-.1140		
.045			-.2755	
.050	-.2354		-.4870	-.6557 -.7944 -1.0290
.060				-.6131
.085			-.3331	
.086		-.1150		
.094	-.2431			
.150				-.6925 -.9327 -1.0170 -1.1380
.163		.0900		
.177			-.4317	
.193		-.2374		
.229	-.1160			
.250				-.6883 -.9105 -1.1360 -1.1980
.274			-.4925	
.339		-.4251		
.362	-.1614			
.400				-.5488 -.9033 -.9940
.402			-.4935	
.497	.0000			
.550				-.4162 -.4794
.565		.0000		
.600				-.5105
.650				-.2084
.700	-.2887		-.2102	
.725			-.1351	
.750		-.1916		-.0016 -.2209
.760			-.0172	-.0181
.770			-.1101	
.808				
.834	-.1671		-.0494	.0142 .0449
.850			-.0943	
.857				
.865	-.1090			
.900	-.0931		-.0232	-.0027
.905			-.0733	.0419
.950			-.0103	.0732
.955		-.0699		
.965	-.0803			



TABULATED PRESSURE DATA - IAI4A - VOL. 5

(R91037)

ARC11-716 IAI4 01-112+512N25+AT11 UPPER WING

ALPHA ( 1 ) = -0.060 BETA ( 1 ) = -3.960

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.930				-.0190		.0963	
.933			.0221				
.965	.0016						

ALPHA ( 1 ) = -0.050 BETA ( 2 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.300	.0559	.0945	.2546	.2460	.0741	-.0513	-.0147
.320			.5058	.4319	.4352	.3933	
.323			.4033				
.340		.2176					
.345			.3622				
.350	.1043		.3424	.3241	.3130	.2619	
.390			.2084				
.385		.2926	.2669				
.386							
.394	.1233			.0188	.0207	.0265	-.0367
.150		.0000					
.163			.0792				
.177		.2225					
.193							
.229	.1943						
.250				-.1597	-.1513	-.1895	-.3902
.274							
.339		-.0115					
.362	.1691						
.400							
.402							
.497	.0000						
.530							
.565							
.600							
.650							
.700							
.725	-.0116						
.750							
.760							
.775							
.800							
.834							
.850							

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(R81037)

ARC11-716 1A14 01+712+512M25+AT11 UPPER WING

ALPHAO ( 1 ) = -8.050 BETAO ( 2 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.057 .0263  
 .063 .0184  
 .903 .0115 .0177 .0555  
 .905 .0180 .0254  
 .930 .0342  
 .933  
 .965 -.0178  
 -.5480

ALPHAO ( 1 ) = -8.070 BETAO ( 3 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.1161 .1035 .2307 .1935 -.0061 -.1158 -.1339  
 .020 .4889 .4010 .3551 .3337  
 .025 .3378  
 .040 .1811  
 .045 .3297 .3486 .3222 .2956 .2367  
 .050 -.0065 .2307  
 .083 .2604  
 .085  
 .086 .2379  
 .094 .0455  
 .100 .0533 .0452 .0474 -.0348  
 .163 .0000  
 .177 .1018  
 .193 .2026  
 .229 .1678  
 .250  
 .274  
 .339 .0238  
 .362 .1722  
 .400  
 .402  
 .497 .0000  
 .530  
 .563  
 .600  
 .650  
 .700 -.0982  
 .725  
 .750  
 .780  
 -.2667 -.3065  
 -.1938 -.2324  
 -.1420 -.1211 -.1641 -.2781  
 -.2273  
 -.3856  
 -.4401  
 -.9781  
 -.3618 -.2988  
 -.2828



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ALPHA( 1 ) = -0.070 BETA( 3 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7800
.8870		
.775		-.2279
.806		-.2697
.834		.1177
.850		
.857		-.1063
.865		-.1362
.900		-.0686
.905		-.0511
.905		-.0972
.930		-.0214
.933		-.0087
.955		-.1074
		-.1006
		-.1326

ALPHA( 1 ) = -0.100 BETA( 4 ) = 9.150

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7800
.8870		
.775		-.2279
.806		-.2697
.834		.1177
.850		
.857		-.1063
.865		-.1362
.900		-.0686
.905		-.0511
.905		-.0972
.930		-.0214
.933		-.0087
.955		-.1074
		-.1006
		-.1326

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(R81037)

ARC11-716 IA14 01+112+312N25+AT11 UPPER WING

ALPHA( 1 ) = -0.100 BETA( 4 ) = 0.150

SECTION 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.2990	.3640	.4270	.5340	.6730 .7800 .8870
.700	-.1103		-.2407	
.725			-.3415	
.750				-.4027 -.4325
.760		-.2997		
.775			-.2499	-.3285
.808		-.1934		
.834	-.2500		-.1884	-.1955 -.2135
.850			-.1801	
.857				
.865	-.1866			
.900	-.1775		-.1470	-.0989
.905		-.1565	-.0697	
.950		-.1320		-.0572
.953		-.1547		
.965	-.1897			

ALPHA( 2 ) = -4.090 BETA( 1 ) = -0.040

SECTION 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.2990	.3640	.4270	.5340	.6730 .7800 .8870
.000	.0774	.1988	.5794	.5130 .4594 .4790
.020			.3927	.3602 .3636 .3495
.025		.3547		
.040		.2782		
.045		.2572		
.050	.0740		.1354	.1306 .1145 .0993
.060			-.0730	
.085		.0739		
.086	.2897			
.094	.0896		-.2975	-.3099 -.2776 -.2838
.150		.0000		
.163		-.1768		
.177		.0845		
.193				
.229	.1695		-.4024	-.4870 -.5274 -.5414
.250			-.2815	
.274		-.2545		
.339				
.362	.0927		-.3692	-.6128 -.7012
.400				
.402		-.3878		
.497	.0000			



(R81U37)

ARC01-7:6 1A14 01+Y12+S12N25+AT11 UPPER WING

ALPHA( 2) = -4.999      BETA( 1) = -8.949

SECTION ( : USER WIND

Y/YBW	Y/CW						
.2990	.3540	.4270	.5340	.6730	.7800	.8870	
			-.2033	-.3272			
		.0000			-.2479		-.6993
				-.1779			
			-.0612		.2126	.1470	
	.0428						
.725	.700						
.750	.750						
.760	.775	-.0149	.1449	.1709			
.805	.775	.1466					
.850	.9233						
.857			.1046	.1320	.1710		
.865	.1631						
.900	.1351		.0840				.1249
.905		.1006		.0794			
.950			.0155				.0922
.953		.0369					
.965	.0387						

$$\text{ALPHA}(2) = -4.089 \quad \text{BETA}(2) = -4.029$$

SECTION	( UPPER WING				DEPENDENT VARIABLE CP
$x/CW$	.2990	.3640	.4270	.5340	.6750 .7800 .8870
.000	.0596	.1690	.4216	.4928	.4039 .3372 .3646
.020				.4029	.3586 .3662 .3308
.025			.3586		
.040		.2556			
.045			.2677		
.050	.0775			.1677	.1461 .1401 .0823
.060				-.0135	
.085			.1174		
.096		.2820			
.094	.0940				
.150		.0000		-.2279	-.2339 -.2273 -.2531
.163					
.177			-.1206		
.193		.1159			
.229	.1685				
.250				-.4097	-.4204 -.4681 -.5149
.274			-.2243		
.339		-.1936			

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(RB1037)

ARC11-716 IA14 01+712+S12M25+AT11 UPPER WING

ALPHA( 2) = -4.080 BETA( 2) = -4.920

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.362	.1094						
.400							
.402							
.497	.0000						
.550							
.563							
.600							
.650							
.700							
.725							
.750							
.760							
.775							
.808							
.834							
.850							
.857							
.863							
.900							
.903							
.930							
.953							
.965							

ALPHA( 2) = -4.080 BETA( 3) = .010

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0286	.1479	.3640	.4044	.2813	.2049	.2413
.020							
.023							
.040							
.045							
.250							
.080							
.085							
.086							
.094							
.150							
.163							
.177							
.189							



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(R81037)

AEC11-716 IA14 Q1+T12+512M25+AT11 UPPER WING

ALPHA( 2 ) = -4.085 BETAO ( 3 ) = .010

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW	X/CW	.2990	.3640	.4270	.5340 .6750 .7800 .8870
.229	.1633				
.250					
.274					
.339					
.362	.1186				
.400					
.432					
.497	.0000				
.550					
.565					
.600					
.650					
.700	-.0448				
.725					
.750					
.760					
.775					
.808					
.851	-.2700				
.850					
.857					
.865	-.1820				
.900	-.1079				
.905					
.950					
.953					
.965	-.0870				

ALPHA( 2 ) = -4.090 BETAO ( 4 ) = 4.070

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP			
Y/BW	X/CW	.2990	.3640	.4270	.5340 .6750 .7800 .8870
.000	-.1462				
.020					
.025					
.040					
.045	.1616				
.050					
.080	-.0511				
.086					
.094					

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(R81037)

ARC11-716 1A14 01+T12+S12M25+AT11 UPPER WING

ALPHA( 2 ) = -4.099 BETA( 4 ) = 4.070

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .150 .163 .177 .193 .229 .250 .274 .339 .362 .403 .402 .457 .550 .565 .603 .633 .703 .725 .750 .763 .775 .808 .834 .850 .857 .865 .900 .905 .930 .935 .965

-.1175 -.1324 -.1443 -.2111

-.0323

.1213

-.3300 -.2943 -.3436 -.4524

-.1198

-.0877

-.3274 -.3361 -.3662

-.2085

-.3586 -.3841

.0000

-.5670

-.5179

-.2723

-.4460

-.4364 -.3711

-.3224

-.4137 -.3867

-.2493

-.3154 -.2568 -.1921

-.2090

-.1072

-.2146

-.0759

-.1663

-.1291

-.0822

ALPHA( 2 ) = -4.110 BETA( 5 ) = 8.110

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 .3394 .0227 .1327 .3020 .1269 .0109 .0243

.020 .020 .3660 .2980 .2965 .2504

.1921

.0431

.045 .1826 .2046 .1807 .1578 .0944

.050



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(R81037)

ARC11-716 1A14 01+712+312+25+AT11 UPPER WING

ALPHA( 3 ) = -.310 BETA( 1 ) = -.040

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	.0113	.1934	.5134	.6717 .6340 .5943 .5896
.020			.2189	.1729 .1283 .0812
.025		.2370		
.040		.2446		
.045			.1110	
.050	.0092		-.0530	-.1152 -.1555 -.2348
.060			-.2661	
.085		-.0805		
.086		.2318		
.094	.0486			
.130			-.5036	-.5633 -.5314 -.5063
.163		.0000		
.177		-.3208		
.193		-.0164		
.229	.1446			
.250			-.6568	-.7049 -.7434 -.7307
.274		-.4386		
.339		-.3734		
.362	.0320		-.7785	-.8571 -.9175
.400		-.6119		
.402	.0000		-.2246	-.6637
.497			.0000	
.550				-.5754
.563				
.600			-.2069	
.650			-.0718	
.700	-.0284			-.3126 -.3860
.723				
.750		-.0209	.0908	.1022
.760				
.775		.1315		
.808				
.834	.0015		.0825	.1228 -.0789
.850		.1156		
.857				
.865	.1549		.0658	
.900	.1305			-.2391
.905		.5920	.0807	
.950		.0533		.0408
.953		.0505		
.965	.0392			



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(R81037)

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

ALPHA( 3 ) = -.320 BETA( 2 ) = -4.000

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	
.2990	.3640 .4270 .5340 .6730 .7800 .8870
X/CW	
.000	.1697 .4704 .6063 .5345 .5135 .5196
.020	.2467 .1883 .1482 .0971
.025	.2945
.040	.2297
.045	.1432
.050	-.0097 -.0757 -.1117 -.2038
.060	-.2029
.085	-.0313
.086	.2307
.094	.0312
.150	-.4322 -.4758 -.4751 -.4712
.163	.0000
.177	-.2574
.193	.0117
.229	.1401
.230	-.5508 -.6102 -.6771 -.6937
.274	-.3789
.339	-.3302
.362	.0437
.400	-.6430 -.7783 -.9010
.402	-.9248
.497	.0000
.550	-.2383 -.3769
.565	.0000
.600	-.8729
.650	-.5446
.700	-.0532
.725	-.3194
.750	-.1264 -.3653
.760	-.2816
.773	-.1631 -.1052
.808	-.1054
.834	-.0034 .0421 .0991
.850	-.0066
.857	
.865	-.1138
.900	-.0309
.905	.0137
.950	.0140
.953	-.0273
.955	-.0223
.965	-.0327
	-.0560
	.0463

(R81037)

ARC11-716 1A14 01-T12-S12N29+AT11 UPPER WING

ALPHA( 3 ) = -.320 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0715	.1145	.4008	.5198	.4560	.4092	.4289
.020				.2430	.1830	.1563	.1060
.025			.2316				
.040		.1886					
.045			.1333				
.050	.0047			.0089	-.0508	-.0812	-.1703
.080						-.1697	
.085			-.0080				
.086		.2009					
.094	.0398			-.3621	-.4082	-.4105	-.4348
.150		.0000					
.163							
.177		.0222					
.193			-.2181				
.229	.1244						
.250				-.5429	-.5417	-.5901	-.6316
.274			-.3229				
.339		-.2976					
.362	.0473						
.400				-.5415	-.6280		-.8311
.402			-.4208				
.497	.0000						
.550				-.3204	-.4000		
.565			.0300				
.590							-.7967
.650						-.5008	
.700	-.0875				-.2744		
.725				-.3990			
.750						-.3542	-.2918
.760			-.3116				
.775				-.3756	-.3417		
.808			-.2938				
.834	-.2878						
.850				-.3279	-.1779	-.0824	
.857			-.2894				
.883	-.2492						
.900	-.2564			-.1751			-.0441
.915			-.1646		-.0729		
.930				-.1321		-.0464	
.933			-.1022				
.965	-.1499						



(RB1037)

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6  
ARC11-716 IA14 01+712+912N29+AT11 UPPER WING

ALPHA( 3 ) = -.320 BETA( 4 ) = 4.090

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/W .2990 .3640 .4270 .5340 .6730 .7800 .8670

X/CW

.000	-.1711	.0366	.2715	.4492	.3643	.3120	.3345
.020				.2136	.1503	.1287	.0873
.025			.1595				
.040		.1174					
.045			.0989				
.050	-.0351			.0044	-.0517	-.0817	-.1678
.080					-.1350		
.085			-.0069				
.086		.1416					
.094	.0095						
.150		.0000		-.3023	-.3533	-.3748	-.4237
.163							
.177			-.1825				
.193	.0309						
.229	.0987			-.5075	-.4862	-.5701	-.6399
.250			-.2577				
.274		-.2179					
.339							
.362	.0441			-.4458	-.4726		-.8940
.400			-.3274				
.402							
.497	.0000			-.3782	-.4333		
.550			.0000				
.565							-.5714
.600							
.650	-.1354			-.4595		-.5502	
.700					-.2844		
.725							
.750			-.3349				
.760				-.4477	-.4111		
.775			-.3470				
.808							
.834	-.3183			-.4318	-.3296	-.2074	
.850			-.3386				
.857							
.865	-.3028			-.3538			-.1250
.900	-.3010						
.905			-.2720		-.1476		
.950				-.2135		-.1200	
.953			-.1939				
.965	-.2234						

ORIGINAL PAGE IS  
OF POOR QUALITY



ARC11-716 1A14 01+T12+S12N29+AT11 UPPER WING

$$\text{ALPHA}(\text{S}) = -.390 \quad \text{BETA}(\text{S}) = 0.130$$

SECTION : UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.3695	-.0116	.0931	.2647	.3116	.2315	.2407
.020				.1472	.1995	.1407	.1000
.025			.0572				
.040		.0312					
.045			.0331				
.050	-.1949			.0227	-.0080	-.0398	-.1226
.060				-.0876			
.085			-.0065				
.086		.0153					
.094	-.1201						
.150				-.2174	-.2732	-.3086	-.3687
.163		.0000					
.177			-.1028				
.193		.0294					
.229	.0702						
.250				-.2932	-.3949	-.4733	-.5816
.274			-.1776				
.339		-.1524					
.362	.0252		-.2434				
.450				-.2668	-.4193		-.6752
.472							
.497	.0000			-.3113	-.4396		
.530			.0000				-.8105
.565						-.5699	
.600							
.650							
.700	-.1798			-.3803			
.725							-.4206
.750			-.3481				
.760				-.3790	-.4461		
.775			-.3533				
.808							
.834	-.3391						
.850			-.3002	-.3016	-.3594	-.2813	
.857							
.865							-.1668
.900	-.3296			-.2184			
.905	-.2988		-.2367	-.1225			
.950			-.1491			-.1331	
.953			-.1817				
.965	-.2148						



(R81037)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

ALPHA( 4) = 4.030 BETA( 1) = -8.060

SECTION : 11 UPPER WING

DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.000	-.0769	.1124
.020	.4848	.6809
.025	.0732	-.0100
.040	.1653	.0732
.045	-.0752	-.2815
.050	-.1114	-.3722
.060		-.4610
.085	-.2896	
.086	.1431	
.094	-.0194	-.7254
.130	.0500	-.7751
.163		-.7692
.177	-.4722	
.193	-.1296	
.229	.1125	-.8112
.250		-.9349
.274	-.5850	-.9322
.339	-.4869	-.7179
.362	-.0270	
.400		-.8749
.402	-.7524	-.9012
.497	.0000	-.6805
.550		-.4606
.565	.0000	-.6935
.600		-.5875
.650		-.6555
.700	-.2206	-.2735
.725		-.1528
.750		-.5740
.760	-.0786	-.5138
.775	.0686	
.808		-.0329
.834	-.0278	-.4871
.850		.0003
.857	.0662	-.3943
.863		-.4855
.900	.1283	
.906	.0964	-.0060
.935	.0575	-.2311
.950	-.0294	-.4150
.953	.0113	
.965	.0136	

ORIGINAL PAGE IS  
OF POOR QUALITY

(RBIU37)

ARC11-716 1A14 01+T12+S12M25+AT11 UPPER WING

ALPHA( 4 ) = 4.030 BETA( 2 ) = -4.010

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2900	.3640	.4270	.5340 .6730 .7600 .8070
X/CW				
.000	-.1536	.0681	.4359	.6347 .5870 .5176 .4777
.020			.0162	-.0309 -.1729 -.2649
.025		.0815		
.040		.1484		
.045		-.0486		
.050	-.0636		-.2452	-.3422 -.4210 -.5499
.080			-.4384	
.085		-.2428		
.086		.1420		
.094	-.0090			
.150			-.6540	-.7343 -.7380 -.7602
.163		.0000		
.177		-.4336		
.193		-.1065		
.229	.1035			
.250			-.7426	-.8044 -.9253 -.9090
.274		-.5372		
.339	-.0248	-.4510		
.362			-.8287	-.9085 -.8778
.400		-.7033		
.402			-.4310	-.8140
.437	.0000			
.550		.0000		
.565				
.600				
.650				
.700	-.1036		-.2617	-.6756
.725		-.2499		
.750			-.5375	-.5850
.760		-.2709		
.775		-.1347	-.2335	
.808		-.0818		
.834	-.2420		-.0726	-.1026 -.4450
.850		-.0343		
.857				
.885	-.1395			
.900	-.0379		-.0370	
.905		-.0127	-.0296	-.5159
.950		-.0633		
.953		-.0389		-.3553
.965	-.0292			



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 2933

(881037)

ARC11-716 1A14 Q1+T12+S12N23+AT11 UPPER WING

ALPHA( 4) = 4.020 BETA( 3) = .030

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	
.2900	.3640
.4270	.5340
.6730	.7800
.8870	
X/CW	
.000	.3611
.020	.5686
.040	.5279
.060	.4577
.080	.4243
.100	.0236
.120	-.0361
.140	-.1540
.160	-.2680
.180	.0843
.200	.1964
.220	-.0451
.240	-.2157
.260	-.3129
.280	-.3900
.300	-.5204
.320	-.4055
.340	-.1976
.360	.1133
.380	-.0276
.400	-.5911
.420	-.6901
.440	-.6890
.460	-.7243
.480	.0000
.500	-.3807
.520	-.0937
.540	.0842
.560	-.6601
.580	-.8000
.600	-.8800
.620	-.8807
.640	-.4752
.660	-.4119
.680	.362
.700	-.0210
.720	.400
.740	-.6354
.760	.497
.780	.0000
.800	-.3143
.820	-.8177
.840	.0000
.860	-.7663
.880	-.6256
.900	-.1772
.920	.723
.940	-.3552
.960	-.2739
.980	-.4453
.000	-.6171
.020	-.3248
.040	-.3382
.060	-.2567
.080	-.3248
.100	-.2543
.120	-.1497
.140	-.1399
.160	-.3165
.180	-.2872
.200	-.1677
.220	-.5414
.240	-.2141
.260	-.1114
.280	-.1260
.300	-.0647
.320	-.1267
.340	-.965
.360	-.1504

ORIGINAL PAGE IN  
OF POOR QUALITY

(R81U37)

ARC11-716 1A14 01+T12+312N25+AT11 UPPER WING

ALPHA( 4) = 4.020 BETA( 4) = 4.100

SECTION ( 1) UPPER WING

DEPENDENT VARIABLE CP

Y/W	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7600
.8870		
.0000	-.3130	-.0871
.0200		.2021
.0230		.4794
.0400		.4452
.0450		.3868
.0500		.3672
.0600		.0097
.0650		-.0873
.0800		-.1693
.0850		-.2755
.0900		.0046
.0950		.0415
.1000		-.0693
.1050		-.2062
.1100		-.3166
.1150		-.3951
.1200		-.5151
.1250		-.3720
.1300		-.1908
.1350		.0561
.1400		-.0625
.1450		.0090
.1500		-.0749
.1550		-.3336
.1600		-.4250
.1650		-.3434
.1700		-.0251
.1750		-.4997
.1800		-.5947
.1850		-.8282
.1900		-1.0190
.1950		-.3533
.2000		-.4917
.2050		.0000
.2100		-.9219
.2150		-.5444
.2200		-.2888
.2250		-.3884
.2300		-.3611
.2350		-.9210
.2400		-.3398
.2450		-.3959
.2500		-.4115
.2550		-.3823
.2600		-.3621
.2650		-.2436
.2700		-.1515
.2750		-.3759
.2800		-.2475
.2850		-.2557
.2900		-.3442
.2950		-.2932
.3000		-.1425
.3050		-.1467
.3100		-.1190
.3150		-.1948
.3200		-.8470



(RB1037)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+812N25+AT111 UPPER WING

ALPHAD (4) = 4.010 BETAD (5) = 0.140

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP						
Y/BU	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	-.4710	-.0432	.0194	.1828	.4013	.3323	.3225
.020				-.0991	-.0373	-.1289	-.2330
.040				-.0455			
.060				-.0037			
.080				-.0728			
.100				-.2033	-.2605	-.3347	-.4483
.120				-.2883			
.140				-.1317			
.160				-.0014			
.180				-.2726			
.200				.0000			
.220				-.2236			
.240				-.0965			
.260				.0443			
.280				-.3744	-.5768	-.6762	-.7784
.300				-.2797			
.320				-.2535			
.340				-.0242			
.360				.3233	-.5479		-.9000
.380				-.3131			
.400				-.3170	-.3072		
.420				.0000			
.440							-.7267
.460						-.6178	
.480						-.3030	
.500						-.3468	
.520						-.4855	-.4427
.540						-.3653	
.560						-.3524	-.4549
.580						-.3028	
.600						-.2532	-.3118
.620						-.3254	
.640						-.1781	-.1669
.660						-.3267	
.680						-.2475	-.1196
.700						-.1214	-.1299
.720						-.1887	
.740						-.2114	

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 OF POOR QUALITY

(R81037)

ARC11-716 1A14 OI+T12+S12M25+AT11 UPPER WING

ALPHA( 5) = 7.940 BETA( 1) = -0.030

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.1065	-.0707	.4211	.6435	.5768	.4771	.4069
.020				-.2313	-.3051	-.4606	-.5851
.025				-.0905			
.040		.0650					
.045			-.2587				
.050	-.2196			-.4802	-.6002	-.6836	-.8256
.080				-.6652			
.085				-.4608			
.086		.0491					
.094	-.1066						
.150				-.9247	-.9486	-.8566	-.8496
.163		.0000					
.177			-.6183				
.193		-.2624					
.229	.0817						
.250				-1.0300	-.9871	-.8236	-.8356
.274			-.7089				
.339		-.5755					
.362	-.0794						
.400				-.9131	-.7456		-.7502
.422			-.8408				
.497	.0000						
.550				-.6767	-.7068		
.565			.0000				
.600							-.5959
.650				-.5556		-.6191	
.700	-.2976			-.2936			
.725							
.750			-.2548	-.5222	-.6056		
.760						-.5656	-.5638
.775			-.1245				
.808							
.834	-.1732			-.3922	-.5436	-.5225	
.850							
.887		-.0964					
.889	-.0595						-.5413
.900	-.0855			-.2877		-.4612	
.925			-.0563				
.950				-.2240		-.4901	
.955							
.985	-.0475			-.0355			



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RB1037)

ARC11-716 1A14 O1+112+S12H25+AT11 UPPER WING

ALPHA( 5) = 7.940 BETA( 2) = -4.010

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340
.6730	.7000	.8870	
.000	-.2463	-.0893	.3936
.020			.5950
.025			.5203
.040			.4237
.045			.3493
.050			-.1613
.060			-.2770
.065			-.4505
.086			-.5952
.094			-.0630
.130			.0532
.163			-.2111
.177			-.4316
.193			-.5559
.229			-.6826
.250			-.8292
.274			-.6093
.339			-.3803
.362			.0319
.402			-.0810
.497			-.0010
.530			.0020
.555			-.5503
.600			-.2150
.650			-.6414
.700			-.9808
.725			-1.0430
.750			-.8827
.760			-.6963
.775			-.5242
.808			-.0703
.834			-.7846
.850			-.10270
.857			-.9159
.865			-.8867
.900			-.5501
.905			-.7667
.950			.0000
.953			-.6406
.965			-.6780
			-.2801
			-.5926
			-.6322
			-.4022
			-.5980
			-.1726
			-.3057
			-.5213
			-.9078
			-.1586
			-.1489
			-.1299
			-.2122
			-.6330
			-.1029
			-.4100
			-.1934
			-.4770
			-.0841
			-.0718



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI14A - VOL. 6

(R81U57)

ARC11-716 IAI14 01+112+512M25+AT11 UPPER WING

ALPHA( 5) = 7.940 BETA( 3) = .040

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2950	.3640	.4270
.5340	.6730	.7800
.8870		
.000	-.3078	-.1670
.020		.3149
.040		.5710
.060		.4785
.080		.3644
.100		.2864
.120		-.1393
.140		-.2510
.160		-.4445
.180		-.8073
.200		-.0571
.220		
.240		.0369
.260		-.1763
.280		-.3816
.300		-.5261
.320		-.6454
.340		-.8290
.360		-.5558
.380		-.3513
.400		
.420		.0474
.440		-.0759
.460		
.480		.094
.500		-.7581
.520		-.8703
.540		-.9133
.560		-.9611
.580		
.600		.0000
.620		-.4795
.640		-.1706
.660		
.680		.0530
.700		-.9313
.720		-.9994
.740		-.9378
.760		-.9322
.780		-.274
.800		-.5832
.820		-.4779
.840		-.0676
.860		-.7348
.880		
.900		.0000
.920		-.5247
.940		-.7921
.960		
.980		.0000
.000		-.6931
.020		-.7036
.040		-.2787
.060		-.4216
.080		
.100		-.3037
.120		-.5790
.140		-.6724
.160		
.180		-.3760
.200		-.5303
.220		-.2566
.240		
.260		-.3498
.280		-.4135
.300		-.4973
.320		
.340		-.2510
.360		
.380		-.2422
.400		-.2999
.420		-.6778
.440		-.8205
.460		-.2775
.480		-.1828
.500		-.2559
.520		-.4856
.540		
.560		-.1304
.580		
.600		-.933
.620		-.1237



## TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1037)

ARC11-716 IA14 01+T12+S12N25+AT11 UPPER WING

DATE 10 JAN 75

ALPHA( 5) = 7.930 BETA( 4) = 4.110

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7803 .8870

X/CW

.020	-.4454	-.1670	.1106	.4663	.4984	.3330	.2376
.020				-.1461	-.2291	-.4185	-.6072
.025			-.1264				
.040		-.0254					
.045			-.1950				
.050	-.1807			-.3648	-.5097	-.6201	-.8181
.080				-.5228			
.085			-.2596				
.086		-.0118					
.094	-.1383						
.150		.0000		-.6855	-.8422	-.8859	-.9480
.163			-.4313				
.177		-.1440					
.193							
.229	-.0185			-.8347	-.9510	-1.0200	-.9692
.250			-.5138				
.274		-.4091					
.339							
.362	-.0868			-.9022	-.9796		-.9740
.400			-.6395				
.402							
.497	.0000			-.5316	-.7041		
.550			.0000				-.7394
.565							
.600							
.650							
.700	-.2693			-.4377	-.2741		
.725							
.750			-.3349			-.5451	-.6923
.760				-.4189	-.3425		
.775			-.3511				
.808							
.834	-.3453			-.3677	-.2859	-.4902	
.850			-.3155				
.865	-.3574						
.900	-.3231			-.2943			-.9734
.905			-.2402		-.2278		
.950				-.2318		-.4582	
.955			-.1738				
.965	-.1947						

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(R81037)

ARC11-716 1A14 01-712-512N25+AT11 UPPER WING

ALPHA( 5 ) = 7.920 BETA( 5 ) = 8.180

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.4233	-.0894	-.0729	.1634	.4223	.3256	.2233
.020				-.2896	-.2173	-.3787	-.9641
.025							
.040							
.045							
.050							
.080							
.085							
.086							
.094							
.150							
.163							
.177							
.193							
.229							
.250							
.274							
.339							
.362							
.400							
.402							
.497							
.550							
.565							
.600							
.690							
.700							
.725							
.750							
.780							
.775							
.806							
.834							
.850							
.857							
.865							
.900							
.905							
.950							
.955							
.965							



## TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 73

(RB1038) ( 15 FEB 74 )

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

## PARAMETRIC DATA

MACH = 1.050 ELEVON = .000  
 RUDDER = .000 SPDBRK = .000

## REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
 LREF = 38.7090 INCHES YMRP = .0000 INCHES  
 BRREF = 38.7090 INCHES ZMRP = .0000 INCHES  
 SCALE = .0300 SCALE

ALPHA(1) = -0.000 BETA(1) = -0.020

## DEPENDENT VARIABLE CP

## SECTION ( 1 ) UPPER WING

Y/8W .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1690	-.0854	.4227	.5231	.4312	.3512	.3683
.020				.5617	.5416	.5657	.5597
.025		.2337					
.040			.3888				
.045				.3435	.3524	.3715	.3568
.050	-.0338			.1657			
.080			.2447				
.085							
.086		.3125					
.094	-.0385			-.0543	-.0025	.0236	.0220
.150		.0000					
.163			.0143				
.177		.1912					
.193				-.2818	-.1932	-.2226	-.2476
.229	.0265						
.290			-.0938				
.274							
.339		-.0954					
.362	.1568			-.1097	-.1002		-.4563
.400							
.402			-.1163				
.497	.0000			-.0582	-.1293		
.550			.0000				-.3021
.600							
.650							
.700	.1904						
.725				-.0264			-.2423
.750							
.760		.0800					.2866
.775			.2601				.2128
.808		.2780					
.834	.1253						
.850				.2238	.2860		.2986
.857		.2570					
.865	.2787						
.900	.2529		.2057				.2623
.905		.2358					
.925							
.935							

DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1038)

ARC11-716 1A14 01-112-512N25+AT11 UPPER WING

ALPHA( 1 ) = -0.080 BETA( 1 ) = -0.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .950 .1461 .2358

.933 .1801

.965 .1664

ALPHA( 1 ) = -0.070 BETA( 2 ) = -4.980

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .000 .2113 .1073 .3611 .4616 .3476 .2523 .2662

.020 .5722 .5340 .5483 .5339

.025 .4587

.040 .1697

.045 .4067

.050 .3746 .3732 .3812 .3576

.080 .2206

.085 .2755

.086 .2925

.094 .0786

.150 .0194 .0402 .0669 .0490

.163 .0000

.177 .0727

.193 .2243

.229 .0189

.230 .274

.274 .0187

.339 .0129

.362 .1744

.400 .0000

.402 .0677

.497 .0000

.550 .0355 .1404

.565 .0000

.600 .0000

.650 .0000

.700 .1859

.725 .1133

.750 .1769

.780 .0390 .0433

.775 .1817 .1359

.808 .2117

.834 .1933 .2231 .2187

.850 .3253

.850 .2564

.850 .0000

.850 .0000

.850 .0000

.850 .0000

.850 .0000



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U36)

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

ALPHA ( 1 ) = -0.070 BETA ( 2 ) = -4.980

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.057		.2002				
	.065	.1975					
	.900	.1890		.1962			.2240
	.905		.1942		.1875		
	.950			.1511		.1991	
	.953		.1522				
	.965	.1296					

ALPHA ( 1 ) = -0.060 BETA ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000		.3196				
	.020	-.2398	-.0171	.3352	.2049	.0915	.0996
	.025			.5542	.4970	.5094	.4791
	.040		.1530				
	.045			.3900			
	.050	-.1645		.3955	.3912	.3864	.3539
	.080			.2670			
	.085		.3076				
	.086		.2643				
	.094	-.1229					
	.150			.0931	.1084	.1260	.0820
	.163		.0000				
	.177		.1394				
	.193		.2479				
	.229	.0061		-.0445	-.0429	-.0656	-.1556
	.250						
	.274		.0502				
	.339		.0596				
	.362	.1993					
	.400			-.0172	-.1075		-.2894
	.402		-.0226				
	.497	.0000					
	.550			-.0366	-.1440		
	.565		.0000				
	.600						-.3368
	.650						
	.700	.1181					
	.725			-.1271	-.2151		-.2681
	.750						
	.760					-.2099	-.2088
							-.1047

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(R81036)

ARC11-716 IA14 01+T12+S12N25+AT11 UPPER WING

ALPHA(1) = -8.560 BETA(1) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.775				-.0934	-.1479		
.808			-.0623				
.834		-.1014		.0151	-.0452	-.0230	
.850		.850	.0306				
.857							
.865	-.0095			.0974	.1110		.0612
.900	.0314		.0795				
.905			.0953			.1027	
.950			.0722				
.953							
.965	.0384						

ALPHA(2) = -8.070 BETA(2) = 4.120

SECTION (1) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2878	.0049	.2218	.2653	.0622	-.0339	-.0304
.020				.5199	.4505	.4601	.4188
.025			.3798				
.040		.1004					
.045			.3660				
.050	-.1641			.3944	.3632	.3680	.3355
.080				.2871			
.085			.3076				
.086		.2040					
.094	-.1193						
.150		.0000		.1299	.1366	.1460	.0933
.163							
.177		.2404	.1719				
.193							
.229	.0273			.0013	-.0007	-.0260	-.1255
.250							
.274			.0858				
.319		.0889					
.362	.1844			-.0142	-.0684		-.2676
.400			-.0094				
.402							
.497	.0000			-.0545	-.1445		
.550			.0000				
.555							-.3481
.600							-.2738
.690							







DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81038)

ARC11-716 1A14 ON+T12+S12N25+AT11 UPPER WING

ALPHA( 1 ) = -0.100 BETA( 1 ) = 0.160

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.935						
.950						
.955						
.965						

ALPHA( 2 ) = -4.080 BETA( 1 ) = -0.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000						
.020						
.025						
.040						
.045						
.050						
.060						
.065						
.066						
.094						
.150						
.165						
.177						
.193						
.229						
.250						
.274						
.330						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01030)

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

ALPHA( 2 ) = -4.080 BETA( 1 ) = -0.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	CP
.362	.0031
.400	-.5418
.402	-.4156
.497	.0000
.550	-.0531
.565	-.2342
.600	.0000
.650	-.3396
.700	-.1631
.725	-.0468
.750	.0375
.760	.2137
.775	.2390
.808	.2475
.834	.0758
.850	.2168
.857	.2507
.865	.2768
.900	.2401
.930	.2025
.935	.1978
.950	.2154
.950	.1462
.953	.2314
.965	.1609
.985	.1557

ALPHA( 2 ) = -4.090 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	CP
.000	-.2044
.020	.0197
.025	.4570
.040	.5600
.045	.4701
.050	.4115
.050	.4586
.060	.4338
.065	.3994
.085	.4211
.086	.4217
.094	.3645
.100	.1917
.100	.2814
.105	.2052
.105	.1857
.105	.2036
.105	.1772
.105	.0308
.105	.1350
.105	.2476
.105	.086
.105	.094
.105	-.1440
.105	.1395
.105	.1722
.105	-.1321
.105	-.1395
.105	.1805
.105	-.1722
.105	-.1321
.105	-.1395
.105	.177
.105	-.0661
.105	.1206

DATE 18 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 CD+712+S12N25+AT11 UPPER WING (R81038)

ALPHA( 2 ) = -4.090 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.229	-.0346			-.3299	-.3133	-.3414	-.3102
.250			-.1781				
.274		-.1551					
.339							
.362	.1126			-.3406	-.4676		-.3863
.400			-.2943				
.402							
.497	.0000			-.0677	-.1683		
.550			.0000				
.565							
.600							
.650							
.700	.1141			-.1536	-.2099		-.5317
.725							
.750							
.760			-.1034			-.1137	-.0411
.775				-.0909	-.0798		
.806			.0061				
.834	-.0935						
.850				.1072	.1388	.1626	
.837			.1137				
.865	.0402			.1442	.1442		.1601
.900	.0652						
.905			.1426				
.950				.1080		.1603	
.955			.1119				
.965	.0636						

ALPHA( 2 ) = -4.090 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2479	.0255	.3964	.4704	.3633	.2833	.3384
.020				.4257	.3637	.4009	.4112
.025			.3408				
.040		.1631					
.045			.2765				
.050	-.2281			.2246	.2095	.2163	.2042
.060				.0711			
.065			.1636				
.066		.2871					
.066	-.1648						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

ARC11-716 IA14 01+T12+S12N23+AT11 UPPER WING (RB1038)

ALPHA( 2) = -4.090 BETA( 3) = .040

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2900 .3640 .4270 .5340 .6730 .7800 .8970

X/CW

.150 -.1001 -.0916 -.0777 -.0934

.163 .0000

.177 -.0194

.193 .1409

.229 .0157

.250 -.2140 -.2485 -.2772 -.3306

.274 .274

.339 -.0986

.362 .1297

.400 .1605 -.2179 -.3003

.402 .1714

.497 .0000

.550 -.0600 -.1878

.565 .0000

.600 -.3169

.650 .0789

.725 -.1521

.750 -.2309 -.1788

.760 .1310

.775 -.1286 -.1836

.808 .1090

.834 -.1330

.850 -.1157 -.1141 -.0203

.857 .0979

.865 .0546

.900 -.0363

.935 .0522

.950 .0375

.953 .0580

.965 .0019

.965 -.0473

ALPHA( 2) = -4.090 BETA( 4) = 4.080

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2900 .3640 .4270 .5340 .6730 .7800 .8970

X/CW

.090 -.3650 .0044 .2805 .3959 .2693 .1797 .2228

.020 .020

.025 .2802

.040 .0932

.045 .2418

.050 -.1824 .2284 .2153 .2123 .1979

DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14A - VOL. 6

(AS1058)

ARC11-716 IAI14 CD+712+812M25+AT11 UPPER WING

ALPHA(2) = -4.090 BETA(4) = 4.080

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

V/W .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0951
.065	.1560
.086	.1556
.094	-.1237
.110	-.0516
.163	-.0000
.177	.0136
.193	.1373
.229	.0285
.250	-.1606
.274	-.2174
.339	-.0478
.362	-.0414
.400	-.1114
.402	-.1225
.497	-.1147
.530	-.0941
.565	-.2021
.600	.0000
.650	-.3282
.700	-.2265
.725	-.1824
.750	-.2661
.760	-.2388
.775	-.1636
.808	-.1633
.834	-.1600
.850	-.1752
.857	-.1904
.865	-.1344
.900	-.0112
.905	-.1488
.950	-.0877
.955	-.0704
.965	-.0144
.985	-.0915
.995	-.1132



1R01V38)

APC: 716 1A14 CI+T12+S12N25+AT11 UPPER WING

$$\text{ALPHA}(2) = -4.103 \quad \text{BETA}(5) = 0.123$$

SECTION: 1 UPPER WING

DEPENDENT VARIABLE CP

SECTION: 1) UPPER WING

Y/9.24	.2995	.3640	.4270	.5340	.6730	.7800	.8870
--------	-------	-------	-------	-------	-------	-------	-------

45/X

000	-.3144	-.0431	.1117	.2855	.2086	.1110	.1275
020				.3839	.3651	.3775	.3520
025			.1607				
040		-.0246					
045			.1050				
050	-.2241			.2615	.2591	.2508	.2135
080				.1679			
085			.1000				
086		.0735					
094	-.1333						
150		.0000		.0400	.0275	.0275	-.0280
163			.0079				
177		.1310					
193							
229	.0107						
250				-.0672	-.0992	-.1359	-.2330
274			.0191				
339		.0159					
362	.1904						
400				-.0761	-.1545		-.3440
402			-.0527				
497	.0000						
550				-.1026	-.1959		
565			.0000				-.3975
630							
650							
700	.0224			-.2064			
725							
750							
760			-.1789				
775				-.1091	-.2373		
808			-.1015				
834	-.1706						
850				-.1930	-.2200	-.2046	
157			-.1072				
865	-.1507						
900	-.1376			-.1618			-.0661
905			-.1310	-.1369			
950				-.1120		-.0650	
953			-.0966				
965	-.1220						

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(RC1038)

## DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+512N25+AT11 UPPER WING

ALPMAD(3) = -.310 BETAD(1) = -8.040

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7800 .8870
.000	-.0417	.0798	.5369 .7241 .8955 .6825 .6691
.020			.2646 .2596 .2206 .1906
.025		.2455	
.040		.1971	
.045		.1242	
.050	-.1575		.0116 -.0277 -.0415 -.1020
.060			-.1879
.065		-.0792	
.086		.2084	
.094	-.1494		-.4012 -.1231 -.3824 -.3592
.150		.0000	
.165		-.2396	
.177		-.0132	
.193	-.0227		-.5455 -.5692 -.5947 -.5816
.229		-.3494	
.250		-.2819	
.274	.0384		-.6311 -.7230 -.7639
.339		-.5113	
.362	.0000		-.3567 -.7107
.400		.0000	
.402			-.6200
.497		-.1397	
.550		-.1050	
.565		.0349 .0087	
.600		.1402	
.650			.1213 .1669 -.3201
.700	-.0392		
.725			-.4092 -.4698
.750			
.760			
.775			
.808			
.834	-.0507		
.850		.1721	
.857			
.865	.1371		
.900	.1728		
.905		.1402	
.950		.1766	
.953		.1016	
.965	.1371		
		.1358	
			-.2076
			-.2812



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81038)

ARC11-716 1A14 Q1+T12+S12N25+AT11 UPPER WING

ALPHA( 3) = -.320 BETA( 2) = -4.010

SECTION ( 1) UPPER WING DEPENDENT VARIABLE: CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1345	.0212	.4920	.6576	.6140	.5805	.5973
.020				.2938	.2630	.2328	.1949
.025			.2657				
.040		.1775					
.045			.1640				
.050	-.2433			.0533	.0103	-.0207	-.0837
.060				-.1334			
.085			.0969				
.086		.2107					
.094	-.1892						
.150				-.3437	-.3769	-.3517	-.3421
.163		.0050					
.177			-.1878				
.193		.0453					
.229	-.0357			-.5175	-.5095	-.5403	-.5378
.250			-.2938				
.274		-.2388					
.339							
.362	.0763			-.6349	-.6479		-.7345
.400			-.4508				
.402				-.1878	-.5578		
.497	.0000		.0000				
.550							
.565							
.600							
.650							
.700	.0416			-.2126			
.725			-.2110				
.750							
.760			-.1321				
.775				-.1507	-.0577		
.808			-.0657				
.834	-.1097						
.850				-.0432	.0609	.1234	
.857			-.0190				
.865	-.0205						
.900	-.0218			.0513	.1070		-.1652
.905			.0663				
.950				.0431		.1443	
.953			.0782				
.965	.0546						

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DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RB1038)

ARC11-715 1A14 01+112+512H25+AT11 UPPER WING

ALPHAD ( 3 ) = -.320 BETAD ( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/B4 .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.4030	-.0502	.2761	.4907	.4136	.3811	.4243
.025			.1632	.2517	.2129	.2085	.1957
.040		.0612					
.045			.1067				
.050	-.2610			.0572	.0194	.0092	-.0381
.060				-.0863			
.085			.0201				
.086		.1125					
.094	-.1459						
.150				-.2483	-.2647	-.2630	-.2703
.163		.0000					
.177			-.1100				
.193		.0543					
.229	.0367			-.3985	-.3669	-.4139	-.4634
.250			-.2027				
.274		-.1572					
.339							
.362	.0639			-.3106	-.3464		-.6339
.400			-.2204				
.402							
.497	.0000			-.1963	-.2484		
.550			.0000				
.600							
.650							-.4238
.700	-.0051			-.2369			
.725			-.2671				
.750							
.760			-.2022	-.2550	-.2427		
.775			-.1846				
.808							
.834	-.1920			-.2634	-.1980	-.0754	
.850			-.1973				
.857							
.865	-.1475			-.2463			.0120
.900	-.1609						
.905			-.1717	-.0614			
.920				-.1763			.0082
.933			-.1456				
.965	-.1688						

(R81038)

ARC11-716 1A14 01-112-512N29-AT11 UPPER WING

ALPHA( 3 ) = -.335 BETA( 5 ) = 0.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.6072	-.0630	.0917	.2375	.3811	.3125	.3385
.020				.1820	.2359	.2320	.2166
.040							
.060							
.080							
.100							
.120							
.140							
.160							
.180							
.200							
.220							
.240							
.260							
.280							
.300							
.320							
.340							
.360							
.380							
.400							
.420							
.440							
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
1.000							



(RB1U38)

ARC11-716 1A14 O1+Y12+S12N25+4Y11 UPPER WING

$$\text{ALPHA}(4) = 4.329 \quad \text{BETA}(1) = -0.059$$

DEPENDENT VARIABLE: CP

SECTION ( UPPER WING

1987	2995	3645	4270	5340	6735	7805	8875
------	------	------	------	------	------	------	------

**M/CW**

.000	-.0276	-.0404	.5060	.7230	.7032	.6493	.6150
.020				.0501	.0297	-.0760	-.1571

.003	.1193
.025	
.040	

0.345	-0.0563	-0.1941	-0.2593	-0.3054	-0.4113
0.250		-0.1341			
0.089				-0.3754	

.385  
.396  
.1229

.094	-.1758				
.159		-.6125	-.6200	-.6104	-.6020

.163  
.177  
.0000

.193	- .1201
.229	- .0917
	7000 - 7543 - 7587 - 7187

1306

.359	- .3706
.362	.0149
.400	- .7878 - .8251
	- .7917

497  
492  
493

	- .7719	- .7812
.550		
.565	.0000	

659  
659

.750	-.3654	-.2777
.725		-.2516

**.759**  
**.769**

.775	-.1155	-.3699
.650	-.0171	

-.034  
-.050  
-.0725

.857	.3330	
.855	.0460	
.853	.0032	-.5603

95  
96  
97  
98

963	.1073
963	.953
963	.1063

(RB1038)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+112+912M25+AT11 UPPER WING

DATE 10 JAN 75

ALPHA(4) = 4.020 BETA(2) = -4.020

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW .2900 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000	-.1377	-.5983	.4712	.6922	.6543	.5933	.5403
.020				.0866	.0347	-.0527	-.1566
.025			.1187				
.040		.1244					
.045			-.0309				
.050	-.2299			-.1581	-.2262	-.2786	-.3984
.080				-.3333			
.085			-.1617				
.086		.1485					
.094	-.2507			-.3328	-.5792	-.5789	-.5898
.150		.0000					
.163			-.3209				
.177		-.0582					
.193							
.229	.0221			-.7533	-.7234	-.7412	-.7298
.250			-.4092				
.274		-.3309					
.339							
.362	.0461			-.7707	-.6299		-.8531
.400			-.5663				
.402				-.6871	-.8097		
.497	.0000						
.550							
.565			.0000				
.600							
.650							
.700	-.2004			-.2622		-.5699	-.5867
.725							
.750			-.2012				
.760				-.1800	-.2268		
.775			-.0687				
.808							
.834	-.1234						
.850				-.1464	-.0384	-.5032	
.857			-.0565				
.865	-.0787						
.900	-.0879			-.1090			-.9551
.925			-.0663		.0240		
.950				-.1081		-.4681	
.953			-.0240				
.985	-.0550						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+S12N25+A711 UPPER WING (R81038)

ALPHA(4) = 4.020 BETA(3) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2734	-.1181	.3858	.6123	.5857	.5288	.5092
.020				.0837	.0575	-.0416	-.1439
.025			.0956				
.040		.0923					
.045			-.0048				
.050	-.3317			-.1387	-.2023	-.2617	-.3748
.080				-.2927			
.085			-.1490				
.086		.1241					
.094	-.3019			-.4886	-.5436	-.5381	-.5692
.150		.0000					
.163			-.2759				
.177		-.0400					
.193							
.229	.0798			-.5804	-.6590	-.7063	-.7064
.250			-.3679				
.274		-.2937					
.339							
.362	.0420			-.6656	-.7500		-.8533
.400			-.5190				
.402							
.497	.0000			-.3145	-.7405		
.550			.0000				
.565							
.600						-.8014	
.650					-.8000		
.700	-.1228			-.2750			
.725						-.5846	-.5732
.750							
.760			-.2187	-.2312	-.1522		
.775			-.1579				
.808							
.834	-.1517			-.1879	-.0764	-.2233	
.850							
.857			-.1811				
.865	-.1318						-.4304
.900	-.1496		-.1446		-.0103		
.905			-.1574				
.950			-.1070			.0478	
.955			-.1219				
.965	-.1784						

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(R81U36)

ARC11-716 1A14 OI+T12+S12N25+AT11 UPPER WING

ALPHA( 4) = 4.010 BETAO ( 4) = 4.110

SECTION ( 1) UPPER WING		DEPENDENT VARIABLE C/P			
Y/BW		.2995	.3640	.4270	.5340 .6730 .7800 .8870
X/CW					
.000	-.4533	-.1451	.1857	.4931	.9026 .4531 .4503
.020				.0666	.0306 -.0590 -.1475
.025			.0139		
.040		-.0051			
.045			-.0453		
.050	-.3666			-.1181	-.2080 -.2640 -.3698
.060				-.2699	
.085			-.1221		
.086		.0281			
.094	-.1804				
.130				-.4148	-.5199 -.5156 -.5482
.163		.0060			
.177			-.2347		
.193		-.0400			
.229	.0333			-.4749	-.5847 -.5720 -.6903
.253				-.3204	
.274		-.2566			
.339					
.362	.0145			-.5355	-.6915 -.8577
.400				-.4379	
.402					
.497	.0000			-.1947	-.3243
.550			.0000		
.565					-.7821
.600				-.4262	
.650				-.2477	
.700	-.0725		-.2321		-.2067 -.5101
.725					
.750			-.2228		
.760				-.2156	-.2421
.775			-.2223		
.808					
.834	-.2091			-.2245	-.1589 -.0719
.850			-.2339		
.857					
.865	-.1836			-.1947	-.0620 -.1992
.900	-.2099		-.2173	-.0620	.0021
.905			-.1401		
.950			-.1815		
.955	-.2127				



(RBIUS8)

DATE 10 JAN 75      TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

ALPHA(4) = 4.000    BETAD (5) = 8.160

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.2990	.3640	.4270	.5340	.6730 .7800 .8870
.000	-.0673	.0301	.1723	.4333 .4149 .4152
.020		-.0761	.0378	-.0307 -.1156
.025		-.0155		
.040		-.0233		
.045		-.0410		
.050	-.4307		-.1553	-.1712 -.2171 -.3205
.080			-.2246	
.085		-.0773		
.086		-.0037		
.094	-.3210		-.2973	-.4097 -.4538 -.4854
.150		.0000		
.163		-.1680		
.177		-.0618		
.193	.0370			
.229			-.3214	-.4715 -.5325 -.6199
.250		-.2244		
.274		-.2034		
.339	.0130		-.2138	-.3811 -.7204
.362		-.2146		
.400				
.402				
.497	.0000		-.1904	-.3279
.550		.0000		
.565				
.600				
.650				
.700	-.0756		-.2482	-.2658 -.4260
.725				
.750				
.780		-.2404		
.775		-.2589	-.2405	-.2918
.806				
.834	-.2545		-.2205	-.2252 -.1365
.850		-.2472		
.857				
.865	-.2259		-.1771	
.900	-.2332		-.2039	-.0974
.905			-.1367	-.0515
.950				
.953		-.1631		
.965	-.1844			

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(RBIUS9)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VC... 6

ARC11-716 1A14 OR+712+312A25+711 UPPER WING

ALPHA( 5) = 7.930 BETA( 1) = -8.910

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	.0002	-.1321	.4461	.7324	.6682	.5678	.5007
.020				-.1370	-.1707	-.3123	-.4238
.025				-.0727			
.040				-.0244			
.045				-.2174			
.050	-.1433			-.3725	-.4531	-.5158	-.6382
.080				-.5403			
.085				-.3945			
.086				-.0034			
.094	-.1461				-.7727	-.7783	-.7387
.150				.0000			
.163				-.2142			
.177				-.5287			
.193				-.0634			
.229				-.5681			
.250				-.4316			
.274					-.9478	-.8905	-.7224
.339							-.7192
.362	-.0039				-.9706	-.7858	-.7207
.400							
.402					-.6786		
.497	.0000				-.8287	-.7217	
.590					.0000		
.565							-.6320
.600					-.3038		
.650					-.6513		
.700	-.5821					-.5708	-.6014
.725					-.2461		
.750					-.1159		
.760					-.5867	-.6230	
.775							
.808							
.834	-.0824				-.4705	-.6804	-.5107
.890					-.0957		
.857							
.865	.0874				-.3688		-.5949
.930	.0089				-.0682	-.5521	
.905					-.2758	-.4844	
.990					-.0508		
.953							
.965	-.0420						





ARC11-716 1A14 01-112-312M25-AT11 UPPER WING

ALPHA( 5) = 7.950 BETA( 3) = .040

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/OW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CM

.000	-.2843	.2330	.3407	.6137	.9930	.4975	.4205
.020				-.0601	-.1166	-.2762	-.4225
.040			-.0092				
.060		.0286					
.080			-.1363				
.100	-.3284			-.2805	-.3681	-.4779	-.6292
.120				-.4355			
.140		.0655					
.160							
.180	-.3171			-.6256	-.7011	-.7326	-.7685
.200		.0000					
.220			-.3691				
.240	.0755						
.260		-.1010					
.280				-.7617	-.8238	-.8386	-.8626
.300			-.4492				
.320		-.3454					
.340	.0180						
.360				-.7998	-.8614		-.8636
.380			-.5030				
.400	.0000						
.420				-.7278	-.8430		
.440			.0000				
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700	-.3905			-.3434			
.720					-.2806		
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.000							



DATE 10 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

PAGE 2965

ARC11-716 IA14 OL+T12+S12N25+AT11 UPPER WING

(R81U36)

ALPHA(5) = 7.930 BETAD (4) = 4.110

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW			
X/CW			
.000	-.5012	-.1811	.0697
.020		.4497	.5177
.025		-.1370	-.1247
.030		-.1026	-.2768
.040		-.0614	
.045		-.1545	
.050	-.4054	-.2998	-.3906
.060		-.4280	-.6289
.065		-.2137	
.086	-.0342		
.094	-.2967		
.150		-.5659	-.6839
.163	.0000	-.7119	-.7560
.177	-.1127	-.3103	
.193			
.229	.0195		
.250		-.6156	-.7905
.274		-.3783	-.8335
.339	-.3105		-.8531
.362	-.0258		
.400		-.6675	-.8272
.402		-.4986	-.8826
.497	.0000		
.550		-.3756	-.7419
.565		.0000	
.600			
.650			-.7798
.700	-.1478	-.2630	-.7471
.725		-.3088	
.750			-.6205
.760		-.2514	-.7148
.775		-.2765	-.2538
.808	-.2698		
.834	-.2462		
.850		-.2588	-.1735
.857		-.2704	-.4258
.865	-.2291		
.900	-.2457	-.2240	-.1015
.935		-.2469	-.1787
.950		-.1981	-.1792
.955			
.965	-.2281		

(R81036)

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 Q1-T12-S12N25+AT11 UPPER WING

ALPHA( S ) = 7.910 BETA( S ) = 0.200

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.6843	-.1078	-.0806	.0096	.4379	.4036	.3322
.020				-.3288	-.1761	-.2718	-.4210
.025			-.0924				
.040		-.0874					
.045			-.1218				
.050	-.5800		-.3417	-.4102	-.4565	-.6103	
.060			-.3813				
.085			-.1600				
.086		-.0603					
.094	-.5164			-.4079	-.6351	-.6843	-.7347
.150		.0000					
.163		-.2402					
.177		-.1726					
.193							
.229	-.0489			-.3643	-.6624	-.7943	-.8336
.230		-.2937					
.274							
.339		-.2979					
.352	-.0106			-.2803	-.6246		-.8990
.400			-.2844				
.402				-.2400	-.3960		
.497	.0000		.0900				
.530							
.563							
.600							
.650							
.700	-.1330			-.2628	-.2915	-.4279	-.7831
.723							
.750			-.2891				
.780			-.2954	-.2565	-.3067		
.775							
.808							
.834	-.2878			-.2099	-.1769	-.0653	
.850			-.2648				
.857							
.865	-.2586						
.900	-.2497			-.1853			-.4529
.905			-.2154	-.0912			
.950			-.1304				-.0378
.955			-.1731				
.985	-.1821						



(RB1039) ( 15 FEB 74 )

ARC11-716 1A14 01+T12+S12N23+AT11 UPPER WING

PARAMETRIC DATA

MACH = 1.150 ELEVON = .000  
RUDDER = .000 SPOBRK = .000

REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
LREF = 38.7090 INCHES YMRP = .0000 INCHES  
BREF = 38.7090 INCHES ZMRP = .0000 INCHES  
SCALE = .0000 SCALE

ALPHA( 1 ) = -8.110 BETA( 1 ) = -8.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/ZW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0736	-.0706	.4237	.5745	.4828	.4119	.4380
.020			.5083	.6242	.5890	.6060	.6123
.025		.0856					
.040			.4371				
.045				.4178	.3993	.4149	.4137
.050	-.0017			.2450			
.080			.3120				
.085		.1793					
.095							
.094	-.0028			.0155	.0434	.0866	.0966
.150		.0000					
.163			.0861				
.177		.2300					
.193							
.229	.0432						
.250				-.1397	-.1364	-.1446	-.1634
.274			-.0383				
.339		-.0337					
.362	.1676						
.400				-.2872	-.3423		-.3901
.402			-.2320				
.497	.0000			.1187	-.0176		
.550			.0000				
.565							
.600							
.650							
.700	.2408						
.725				-.0677			
.750				-.1216			
.760			.0710	.0486	.1329	.0122	
.775			.2966	.2502	.1901		
.809							
.834	.1323			.3114	.3087	.3115	
.850			.3048				
.857							
.865	.3002						
.900	.2958			.3177	.2930		.2961
.905			.3028				

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(R81039)

ARC11-716 1A14 01+712+312M25+AT11 UPPER WING

ALPHA(1) = -0.110 BETA(1) = -0.020

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.930				.2802		.3014	
.933			.2642				
.935	.2436						

ALPHA(1) = -0.100 BETA(2) = -4.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.930	-.1498	-.1292	.3234	.4788	.3788	.3038	.3367
.920				.5834	.5502	.5797	.5866
.925			.4290				
.940		.0092					
.945			.3793				
.950	-.0701			.3961	.3954	.4233	.4285
.980				.2429			
.985			.2795				
.986		.1250					
.994	-.0565			.0452	.0893	.1258	.1379
.130		.0000					
.163			.1069				
.177		.1756					
.193							
.229	-.0179			-.1026	-.0961	-.0734	-.0996
.250			.0095				
.274		-.0117					
.339							
.362	.1099			.1049	.0701		-.1574
.400			-.0637				
.402							
.497	.0000			.1116	-.0015		
.530			.0003				
.565							
.600							
.650							
.700	.2468						
.725				.0271	-.1338		
.750							
.760			.0542				
.775				.0929	.0236		
.808			.1824				
.834	.0821						
.850				.2306	.1995	.1884	







(R01039)

ARC11-716 [A14 OL+T12+312M25+AT11 UPPER WING

ALPHA( 1 ) = -0.090 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.775				.0161	-.0348		
.808			.0529				
.834	.0143						
.850				.0444	-.0050	-.0011	
.857			.0944				
.865	.1069						.0925
.900	.1113		.1443	.1273	.1523		
.905						.1333	
.950			.1451	.1344			
.953							
.965	.1150						

ALPHA( 1 ) = -0.100 BETA( 4 ) = 4.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.800	-.2335	-.0407	.2237	.3947	.1944	.0867	.0910
.820				.5133	.4979	.5266	.4993
.825			.3189				
.840		.0341					
.847			.2375				
.850	-.1273		.3809	.4281	.4273	.4119	
.880			.2832				
.885			.2319				
.886		.1260					
.904	-.0986						
.150				.1712	.2091	.2196	.1836
.163		.0000					
.177			.1668				
.193		.1756					
.229	-.0215			.0703	.0760	.0598	-.0227
.250			.1318				
.274							
.339		.1067					
.362	.1276						
.400				.0813	.0188		-.1575
.402			.0759				
.497	.0000			.0572	-.0282		
.550			.0000				
.565							
.600							
.630							
							-.2371
							-.1512



1R01U39:

TABULATED PRESSURE DATA - 1A14A - VOL. 6

AB511-716 1A14 Q1+T12+S12N25+AT11 UPPER WING

DATE : 3 JAN 75

$$\text{ALPHA}(1) = -0.109 \quad \text{BETA}(4) = 4.085$$

SECTION : RUBBER WING

	2690	1640	.4270	.5347	.6730	.7809	.8870
--	------	------	-------	-------	-------	-------	-------

[illegible]
$$\text{ALPHA}(1) = -0.130 \quad \text{BETA}(5) = 0.160$$

SECTION / LINGER WING

2090	4540	4273	5340	.6730	.7800	.3070
------	------	------	------	-------	-------	-------

[illegible]

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81039)

ARC11-716 1A14 01+712+312N25+AT11 UPPER WING

ALPHA(1) = -8.130 BETA(1) = 8.160

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.550	.0228	-.0477				
.565	.0000					-.2482
.580						-.1670
.595						
.610						
.625	.1200					
.640						
.655						
.670						
.685						
.700						
.715						
.730						
.745						
.760						
.775						
.790						
.805						
.820						
.835						
.850						
.865						
.880						
.895						
.910						
.925						
.940						
.955						
.970						
.985						
.1000						

ALPHA(2) = -4.100 BETA(2) = -8.030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.550	-.0749	.0212	.5493	.6724	.6152	.5588	.5872
.565				.4971	.4810	.4906	.5026
.580							
.595							
.610							
.625							
.640							
.655							
.670							
.685							
.700							
.715							
.730							
.745							
.760							
.775							
.790							
.805							
.820							
.835							
.850							
.865							
.880							
.895							
.910							
.925							
.940							
.955							
.970							
.985							
.1000							





ARC11-716 1A14 01+112+312M25+AT11 UPPER WING

ALPHA( 2 ) = -4.110 BETA( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.229 -.0371  
 .250 -.2334 -.2367 -.2350 -.2727  
 .274 -.1340  
 .339 -.1139  
 .352 .0947  
 .400 -.3466 -.4112 -.4705  
 .492 -.2705  
 .497 .0000  
 .550 -.0035 -.3048  
 .565 .0000  
 .630 -.3285  
 .650 .1381  
 .700 -.1391  
 .725 -.0048  
 .750 .0231  
 .760 .0504 .0602  
 .775 .0830  
 .808 .1806  
 .834 .1719 .1934 .2303  
 .850 .2467  
 .857 .2260 .2361 .2487  
 .865 .1327  
 .900 .1533  
 .905 .2083  
 .930 .2083  
 .953 .1721  
 .965 .040

ALPHA( 2 ) = -4.110 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

Z/CW

.000 -.1875 -.0253 .4304 .5322 .4352 .5500 .5830  
 .020 .020 .4684 .4248 .4292 .4292  
 .025 .3631  
 .040 .0364  
 .045 .2947  
 .090 -.8119  
 .090 .2754 .2362 .2424 .2380  
 .091 .1271  
 .096 .1942  
 .094 .1061  
 -.1375



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING (R81039)

ALPHAO ( 2 ) = -4.110 BETA0 ( 3 ) = .040	
SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	
.2990 .3640 .4270 .5340 .6730 .7800 .8870	
X/CW	
.190	-.0548
.163	-.0625
.177	-.0353
.193	-.0405
.229	
.250	-.1705
.274	-.1970
.339	-.2329
.362	-.2610
.400	
.402	-.2122
.497	-.3435
.550	-.4508
.565	
.600	.0296
.650	-.0462
.700	
.725	.0300
.750	
.760	-.1650
.775	-.1539
.808	-.0367
.834	-.1159
.850	-.0082
.857	
.865	-.0154
.900	-.0698
.905	
.950	-.0070
.953	
.965	-.0227
	-.0432
	.0632
	.1801
	.0931
	.1567
	.0538
	.0314

ALPHAO ( 2 ) = -4.110 BETA0 ( 4 ) = 4.170	
SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	
.2990 .3640 .4270 .5340 .6730 .7800 .8870	
X/CW	
.000	-.2746
.020	-.0411
.025	.2746
.040	.4339
.045	.3180
.050	.2457
	.2959
	.4011
	.3507
	.3869
	.4230
	.2906
	.0495
	.2086
	.2265
	.1987
	.2165
	.2333

TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U39)

ARC11-716 IA14 01+T12+SIEM25+AT11 UPPER WING

ALPHA( 2 ) = -4.110 BETA( 4 ) = 4.170

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BL .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CM

.080	.0977					
.085	.1399					
.086	.1185					
.094	-.1262					
.150		-.0690	-.0620	.0031	.0017	
.163	.0000					
.177	.0092					
.193	.1063					
.229	-.0210					
.250		-.729	-.1911	-.1678	-.2116	
.274	-.0745					
.339	-.0597					
.362	.0936					
.403		-.0355	-.0633		-.2732	
.452	-.0508					
.497	.0000					
.550		.0377	-.0603			
.565	.0000					
.600						
.650						
.700	.1420					
.725		-.0616				
.750						
.760						
.775		-.0331				
.808		-.0480	-.1061			
.834	-.0564					
.842		-.0525	-.0936	-.0834		
.857		-.0489				
.865	-.0030					
.900	-.0083					
.905		-.0475				
.950		-.0128	-.0323		.0350	
.953		-.0001				
.965		-.0152				
	-.0250					



(R81039)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+112+512N23+AT11 UPPER WING

ALPHA( 2) = -4.130 BETA( 5) = 0.110

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6750	.7600	.8870
X/CW							
.050	-.4590	-.0711	.1264	.3793	.2467	.1443	.592
.025			.1441	.4051	.3496	.3571	.3452
.025		-.0417					
.040			.1553				
.045				.2774	.2332	.2318	.2148
.050	-.2379			.1770			
.080			.1588				
.085		.0516					
.086				.0378	.0247	.0210	-.0033
.094	-.1487						
.130		.0000					
.163			.1132				
.177		.0885					
.193							
.229	-.0268			-.0775	-.0994	-.1294	-.1915
.250			.0169				
.274		.0233					
.339							
.362	.0621			-.0670	-.1149		-.2737
.400			-.0529				
.402				-.0272	-.1044		
.497	.0050		.0000				-.3077
.550					-.2176		
.565				-.1063			
.600					-.1743	-.2267	
.650							
.700	.0756						
.725				-.0772	-.0893	-.1313	
.750				-.0977			
.760							
.775							
.808							
.834	-.0867			-.1121	-.1348	-.1407	
.850							
.857				-.1240			
.865	-.0762						-.0397
.900	-.1012			-.1231	-.1066		
.905				-.0990	-.0976	-.0580	
.910							
.933				-.0817			
.965	-.1023						

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(R81039)

ARC11-716 1A14 Q8+T18+S12N25+AT11 UPPER WING

ALPMAO( 3) = -.330 BETAO ( 1) = -4.020

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2890	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.0632	-.0349	.5028	.6783	.6492	.6269	.6670
.020				.3384	.3319	.3194	.2977
.025			.2853				
.040		.0106					
.045			.1794				
.050	-.1577			.1138	.0651	.0818	.0379
.080				-.0620			
.085			.0586				
.086		.0648					
.094	-.1594						
.150				-.2505	-.2737	-.2329	-.2132
.163		.0500					
.177			-.1411				
.193		.0467					
.229	-.0658						
.250				-.3611	-.4015	-.4122	-.4077
.274			-.2389				
.339		-.1844					
.362	.0727						
.400				-.4482	-.5286		-.5876
.402			-.3601				
.497	.0000						
.550				-.3716	-.5429		
.565			.0000				-.5976
.600							
.650					-.5711		
.700	-.0242			-.1205	-.1693		
.725						-.3919	-.4994
.750			-.0421				
.760				-.0066	-.0348		
.775			.0471				
.808							
.834	-.0145			.0400	.0696	-.2349	
.850			.0328				
.857							
.885	.0717						-.2219
.900	.0461			.0772	.1154		
.905			.0737				.0030
.910				.1115			
.953			.1207				
.965	.0877						



TABULATED PRESSURE DATA - TAIL - VOL. 3

ARC11-716 TAIL 01-112-512N25-AT11 UPPER WING

(881035)

ALPHA (3) = -.330 BETA (2) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2993	.3540	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1635	-.0608	.4465	.6256	.5555	.5015	.5623
.020			.2724	.3313	.3086	.2902	.2832
.040		-.0302					
.060			.1912				
.080	-.2362			.1314	.0658	.0702	.0404
.100			.0290	-.0220			
.120		.0533					
.140	-.1906						
.160		.0900		-.1959	-.2361	-.2214	-.2104
.180			-.0966				
.200		.0719					
.220	-.0948			-.2339	-.3611	-.3870	-.4067
.240		-.1002					
.260				-.4028	-.4845		-.5775
.280	.0834						
.300			-.3051				
.320	.0000			-.1651	-.4940		
.340			.0500				
.360							
.380							
.400							
.420							
.440							
.460							
.480							
.500							
.520							
.540							
.560							
.580							
.600							
.620							
.640							
.660							
.680							
.700	.0300			-.0643			
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.000							

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81039)

ARC11-716 1A14 Q1+T12+512M25+AT11 UPPER WING

ALPHA( 3 ) = -.340 BETA( 3 ) = 4.070

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP				
Y/BW	.2990	.3640	.4270	.5340	.6730 .7600 .8870
X/CW					
.000	-.2910	-.1233	.3158	.5244	.4418 .3951 .4410
.020				.2847	.2429 .2305 .2335
.025			.1846		
.040		.0278			
.045			.1230		
.050	-.2567		.0942	.0274	.0338 .0171
.080			-.0461		
.085			.0212		
.086		.0993			
.094	-.1639			-.2059	-.2349 -.2284 -.2143
.150		.0000			
.163		-.0840			
.177		.0582			
.193					
.229	-.0251			-.2801	-.3241 -.3800 -.3972
.250			-.1720		
.274		-.1356			
.339	.0760			-.3793	-.4398 -.5497
.362			-.3113		
.400				-.0873	-.2154
.402	.0000		.0000		
.497					-.5735
.550				-.2021	
.565				-.1803	
.600				-.0876	
.650	.0197				-.0397 -.1686
.700			-.0908		
.725			-.0719	-.0724	-.0823
.750					
.760					
.775					
.808					
.834	-.0937			-.0876	-.0308 .0856
.850			-.0901		
.857					
.885	-.0488				
.900	-.0661			-.0873	.0443 .1150
.905			-.0672		
.950			-.0356		.1259
.953			-.0584		
.965	-.0754				



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 OR+T12+S12N25+AT11 UPPER WING (R81U39)

ALPHA( 3 ) = -.340 BETA( 4 ) = 0.140

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.5377	-.1582	.1065	.3041	.3891	.3058	.3322
.020			.1998	.2471	.2197	.2036	
.025			.0806				
.040		-.0298					
.045			.0606				
.050	-.3497		.0899	.0822	.0496	.0072	
.080			-.0044				
.085			.0329				
.086		-.0480					
.094	-.2026						
.150				-.1157	-.1766	-.1882	-.1989
.163		.0000					
.177			-.0347				
.193		.0866					
.229	.0001						
.250				-.2063	-.2862	-.3360	-.3814
.274			-.1142				
.339		-.1050					
.362	.0590			-.2133	-.3593		-.5298
.400							
.402			-.1822				
.497	.0000						
.550				-.1091	-.2071		
.565			.0000				
.600							
.650							
.700	-.0064				-.2796		-.4632
.725				-.1579			
.750					-.1994		
.760			-.1105				
.775				-.1269	-.1616		
.808			-.1338				
.834	-.1346						
.850				-.1512	-.1578	-.1200	
.857			-.1608				
.865	-.1140						
.900	-.1434			-.1617			.0471
.905			-.1510		-.1119		
.910				-.1459		.0024	
.953			-.1437				
.965	-.1670						

(RB1039)

ARC11-716 1A14 01+712+512N25+AT11 UPPER WING

ALPMAO( 4) = 4.010 BETAO ( 1) = -8.060

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/W	2990	3640	4270	5340	6730	7800	8870
X/CW							
.000	.0704	-.0618	.9520	.7676	.7684	.7215	.6949
.020				.1535	.1477	.0632	-.0075
.025			.1800				
.040		-.0046					
.045			.0441				
.050	-.0646			-.0800	-.1323	-.1612	-.2510
.080				-.2524			
.085			-.1446				
.086		.0570					
.094	-.0631						
.150				-.4828	-.4731	-.4561	-.4456
.153		.0000					
.177			-.2989				
.193		-.0565					
.229	-.0879						
.250				-.5621	-.6127	-.6081	-.5795
.274		-.4071					
.339		-.3023					
.362	.0588			-.6372	-.7417		-.7293
.400		-.5320					
.402							
.497	.0000			-.6562	-.7192		
.530		.0000					
.565							
.590							
.650							-.7015
.700	-.4083				-.7222		
.725			-.2827	-.2428			
.750					-.5745	-.6294	
.760		-.1862					
.775			-.1970	-.5484			
.808		-.0546					
.834	-.C099						
.850			-.0593	-.5231	-.4827		
.857		-.0012					
.865	.0635						
.900	.0446		-.0250				-.4854
.925		.0375		-.3384			
.950			-.0248				-.4004
.955		.0313					
.965	-.0358						



(881059)

ARC11-716 1A14 01+712+S12N25+AT11 UPPER WING

ALPHA (4) = 4.010 BETA (2) = -4.010

SECTION: 1 UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0266	-.1496	.4561	.7216	.7084	.6641	.6484
.020				.1366	.1326	.0655	-.0111
.025			.1335				
.040		-.0640					
.045			.0107				
.050	-.1506			-.0762	-.1172	-.1426	-.2398
.080				-.2277			
.085			-.1443				
.086		-.0020					
.094	-.1557						
.150		.0090		-.4373	-.4360	-.4231	-.4242
.163							
.177			-.2531				
.193		-.0507					
.229	-.1157						
.250				-.4672	-.5740	-.5743	-.5591
.274		-.2819					
.339		-.2540					
.362	.0386						
.400				-.5238	-.6799		-.7012
.402		-.4317					
.497	.0000			-.5338	-.6739		
.550		.0000					
.565							
.600							
.633						-.6929	-.6712
.730	-.2812			-.1748	-.2146		
.725							
.750							
.760		-.1113					
.775			-.0961	-.2471			
.806		-.0072					
.834	-.0122						
.850			-.0562	-.0537	-.4544		
.857		-.0168					
.865	.0406						
.900	.0164		-.0290				-.4272
.935		-.0051	.0134				
.950			-.0057			-.3926	
.953		-.0073					
.965	-.0484						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81039)

ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING

ALPHA(4) = 4.000 BETA(3) = .050

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7650	.8670
X/CW							
.005	-.1812	-.2330	.4026	.6818	.6437	.5980	.5908
.020				.1527	.1420	.0670	-.0049
.025			.1392				
.040		-.0721					
.045			.0348				
.050	-.2346			.0985	-.1068	-.1358	-.2265
.060				-.2133			
.085			-.0835				
.086		-.0079					
.094	-.2448			-.3580	-.4192	-.4027	-.4143
.150		.0000					
.163			-.1737				
.177		.0110					
.193							
.229	-.1199			-.4136	-.5372	-.5545	-.5524
.250			-.2717				
.274		-.1331					
.339							
.362	.0756			-.4694	-.6200		-.6936
.400			-.4103				
.497	.0000			-.4321	-.6110		
.550			.0000				
.565							-.6651
.600							
.650					-.6780		
.700	-.1692			-.1661			
.725					-.1914		
.750						-.5537	-.5766
.760			-.1271	-.1030	-.1325		
.775			-.0562				
.808							
.834	-.0653			-.0858	-.0617	-.4004	
.850			-.0872				
.857							
.865	-.0390						-.3945
.900	-.0668			-.0740	.0408		
.905			-.0915	-.0545			-.1195
.950							
.955			-.1003				
.965	-.1201						



DATE 10 JAN 75  
 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+512N25+AT11 UPPER WING  
 (881039)

ALPHA (4) = 4.000 BETA (4) = 4.100		SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
X/CW	Y/BW				
.000	-.3067	-.3106	.2175	.5434	.5261
.020			.1108	.0973	.0312
.025			.0477		-.0259
.040		-.0309			
.045		-.0132			
.050	-.2859		-.0775	-.1312	-.1657
.080			-.2062		-.2352
.085		-.1091			
.086		.0184			
.094	-.2201		-.3547	-.4183	-.4061
.150		.0000			-.4160
.163			-.1929		
.177		-.0235			
.193					
.229	-.0584		-.4134	-.4989	-.5523
.250			-.2671		-.5511
.274		-.2060			
.339			-.4769	-.5903	-.6915
.362	.0367				
.400			-.3008	-.5766	
.402			.0000		-.6553
.497	.0000				
.550			-.6333		
.600			-.2006		
.650			-.1360		
.700	-.1063				
.725			-.4314	-.5321	
.750		-.1374			
.760			-.0856	-.1529	
.775		-.1001			
.808					
.834	-.0983		-.1014	-.0638	-.0678
.850					
.857		-.1317			
.865	-.0814				
.900	-.1061		-.1072	.0174	-.3218
.935		-.1307			
.950		-.0871			.0473
.953		-.1144			
.965	-.1345				



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+T12+S12N25+AT11 UPPER WING (RBIU39)

A_PMAO( 4 ) = 3.990 BETAO ( 5 ) = 8.140		SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW				
.2990	.3640	.4270	.5340	.6730	.6870
.000	-.6125	-.1210	.0501	.2332	.4505
.020				.4196	.4345
.025				-.0185	.0907
.040				.0301	-.0316
.045				.0142	
.050				-.0633	
.060				-.0072	
.065				-.1048	-.1144
.080				-.1579	-.2309
.085				-.1791	
.086				-.0430	
.094				-.0376	
.150				-.3502	
.163				-.2610	-.3606
.177				-.3850	-.3975
.193				.0000	
.229				-.1276	
.250				-.0090	
.274				-.1951	
.339				-.1584	
.362				.0381	
.403				-.3515	-.5354
.402				-.6875	
.497				-.3013	
.550				-.1702	-.3249
.565				.0000	
.600					-.6343
.650				-.3935	
.700				-.0222	
.725				-.1960	
.750				-.2194	
.760				-.2242	-.3920
.775				-.1574	
.808				-.1715	-.1952
.834				-.1851	
.850				-.1697	-.1708
.857				-.0789	
.865				-.2001	
.900				-.1842	-.1377
.905				-.1840	
.950				-.1872	-.0849
.955				-.1545	.0117
.965				-.1699	
.965				-.1913	



## TABULATED PRESSURE DATA - IALIA - VOL. 6

ARC11-716 IALIA 01+112+S12N25+AT11 UPPER WING (RB1059)

DATE 10 JAN 75

ALPHA( 5 ) = 7.920 BETA( 1 ) = -8.050

## DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	.0844	-.0434	.5002	.7990	.7434	.6628	.6032
.020				-.0220	-.0197	-.1412	-.2393
.025			.0058				
.040		-.0143					
.045			-.1234				
.050	-.0382			-.2388	-.2974	-.3451	-.4501
.060				-.4005			
.085			-.2987				
.086		.0066					
.094	-.0425			-.6227	-.6006	-.5978	-.6003
.110		.0000					
.163			-.4700				
.177		-.1593					
.193							
.229	-.0425			-.7227	-.7233	-.7253	-.6971
.250			-.5203				
.274		-.3923					
.339	.0223			-.7742	-.7611		-.7436
.352			-.5979				
.400		.0000		-.7592	-.7669		
.402			.0000				-.7243
.497	.0000			-.5539	-.2682		
.550						-.5883	-.6246
.565							
.600							
.630	-.4847						
.700				-.2587	-.4838	-.6271	
.725			-.1761				
.750							
.760							
.775							
.808							
.834	-.0141			-.2763	-.5939	-.5316	
.850			-.1206				
.897							
.865	.0349			-.1546			-.6320
.910	.0101				-.5248		
.905			-.0581				-.4809
.930				-.0930			
.933			-.0135				
.965	-.0807						

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+712+812M25+AT11 UPPER WING  
 ALPHA( 5 ) = 7.930 BETA( 2 ) = -3.990

(R91039)

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.0409	-.2152	.3662	.7170 .7010 .6134 .5530
.020				-.0552 -.0290 -.1541 -.2726
.025			-.0571	
.040		-.1570		
.045			-.1786	
.050	-.1357		-.2460	-.2917 -.3510 -.4714
.055			-.3933	
.085			-.5228	
.086		-.1043		
.094	-.1689			-.5813 -.5829 -.5946 -.6137
.150		.0000		
.165			-.3265	
.177		-.1708		
.193				
.229	-.1712			-.6379 -.7036 -.7166 -.7053
.250			-.3622	
.274		-.2120		
.339	-.0057			-.6422 -.7991 -.7999
.362			-.4914	
.400				-.6286 -.7668
.402				
.497	.0000			-.7742
.550			.0000	-.2473
.565				-.6056 -.6676
.600				
.650				
.700	-.4252		-.2484	
.725				
.750		-.1843		
.760			-.2204	-.6151
.775		-.1011		
.808				
.834	-.0141		-.1075	-.5469 -.5217
.850				
.897		-.0936		
.865	.0035			-.1469
.900	-.0417		-.0705	-.3072
.905				-.1015
.950			-.0648	-.4715
.953				
.965	-.1070			



DATE 10 JAN 75  
 TABULATED PRESSURE DATA - IA14A - VOL. 6  
 ARC11-716 IA14 01-T12-S12N25+AT11 UPPER WING  
 (R81U39)

ALPHA( 5) = 7.930 BETAD ( 3) = .040  
 SECTION 1 UPPER WING  
 DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	-.1080	-.4316	.3246	.6495	.6476	.5700	.5055
.000	.020	.025	.040	.045	.050	.055	.060
.065	.070	.075	.080	.085	.090	.095	.100
.105	.110	.115	.120	.125	.130	.135	.140
.145	.150	.155	.160	.165	.170	.175	.180
.185	.190	.195	.200	.205	.210	.215	.220
.225	.230	.235	.240	.245	.250	.255	.260
.265	.270	.275	.280	.285	.290	.295	.300
.305	.310	.315	.320	.325	.330	.335	.340
.345	.350	.355	.360	.365	.370	.375	.380
.385	.390	.395	.400	.405	.410	.415	.420
.425	.430	.435	.440	.445	.450	.455	.460
.465	.470	.475	.480	.485	.490	.495	.500
.505	.510	.515	.520	.525	.530	.535	.540
.545	.550	.555	.560	.565	.570	.575	.580
.585	.590	.595	.600	.605	.610	.615	.620
.625	.630	.635	.640	.645	.650	.655	.660
.665	.670	.675	.680	.685	.690	.695	.700
.705	.710	.715	.720	.725	.730	.735	.740
.745	.750	.755	.760	.765	.770	.775	.780
.785	.790	.795	.800	.805	.810	.815	.820
.825	.830	.835	.840	.845	.850	.855	.860
.865	.870	.875	.880	.885	.890	.895	.900
.905	.910	.915	.920	.925	.930	.935	.940
.945	.950	.955	.960	.965	.970	.975	.980
.985	.990	.995	1.000				

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ARC11-716 1A14 01+712+512M25+AT11 UPPER WING

ALPMAD(5) = 7.930 BETA0 (4) = 4.120

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3655	-.3686	.0812	.4844	.5426	.4946	.4565
.020				-.1085	-.0605	-.1680	-.2722
.040			-.0852				
.060		-.1539					
.080	-.3099		-.1286				
.100				-.2562	-.3006	-.3483	-.4654
.120				-.3750			
.140		-.0620	-.1994				
.160	-.2977			-.5263	-.5656	-.5759	-.5990
.180		.0000					
.200		-.0833	-.2887				
.220	-.0605			-.6220	-.6719	-.6891	-.6934
.240		-.2855	-.3409				
.260	.0088			-.6592	-.7382		-.7916
.280			-.4574				
.300	.0000			-.4959	-.7202		
.320		.0000					-.7629
.340					-.7635		
.360	-.1669			-.2584		-.6280	-.6221
.380			-.1680				
.400			-.1555	-.2055	-.2148		
.420							
.440		-.1394	-.1766	-.1938	-.1943	-.5664	
.460							
.480	-.1266						-.9031
.500	-.1517			-.2060	-.0797		
.520			-.1791		-.1835	-.3328	
.540			-.1791				
.560	-.1855						



(RB1U39)

TABULATED PRESSURE DATA - IA14A - VOL. 6

ARC11-716 1A14 01+T12+S12N23+AT11 UFFER WING

$$\text{BETAO} (S) = 7.919 \quad \text{BETAO} (S) = 6.203$$

DEPENDENT VARIABLE: NUMBER LONG

[illegible]

(R81U40) ( 15 FEB 74 )

ARC11-716 1A14 01+T12+S12N23+AT11 UPPER WING

PARAMETRIC DATA

MACH = 1.400 ELEVON = .000  
RUDDER = .000 SPDRK = .000

REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5000 INCHES  
LREF = 36.7090 INCHES YMRP = .0000 INCHES  
BREF = 36.7090 INCHES ZMRP = .0000 INCHES  
SCALE = .0300 SCALE

ALPHA( 1 ) = -7.980 BETA( 1 ) = -8.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.000	.1103	.0196
.020	.0537	.0534
.025	.0537	.0534
.040	.0537	.0534
.045	.0537	.0534
.050	.0537	.0534
.060	.0537	.0534
.085	.0537	.0534
.096	.0537	.0534
.094	.0537	.0534
.150	.0537	.0534
.163	.0537	.0534
.177	.0537	.0534
.193	.0537	.0534
.229	.0537	.0534
.250	.0537	.0534
.274	.0537	.0534
.339	.0537	.0534
.362	.0537	.0534
.400	.0537	.0534
.402	.0537	.0534
.497	.0537	.0534
.530	.0537	.0534
.565	.0537	.0534
.600	.0537	.0534
.650	.0537	.0534
.700	.0537	.0534
.725	.0537	.0534
.750	.0537	.0534
.760	.0537	.0534
.775	.0537	.0534
.838	.0537	.0534
.834	.0537	.0534
.850	.0537	.0534
.857	.0537	.0534
.863	.0537	.0534
.900	.0537	.0534
.905	.0537	.0534



DATE 10 JAN 74 TABULATED PRESSURE DATA - IAI4A - VOL. 6

(RBIU4D)

ARC11-716 IAI4 01+112+12N25+AT11 UPPER WING

ALPHA(1) = -7.920 BETA(1) = -8.040

SECTION: (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW  
.990  
.953  
.965  
.1623  
.1778  
-.0387

ALPHA(2) = -7.910 BETA(2) = -1.000

SECTION: (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW  
.000  
.020  
.025  
.040  
.045  
.050  
.080  
.085  
.094  
.150  
.163  
.177  
.193  
.229  
.230  
.274  
.339  
.362  
.470  
.492  
.497  
.510  
.565  
.603  
.623  
.700  
.725  
.750  
.760  
.775  
.808  
.834  
.850  
.0225  
.2242  
.3549  
.3122  
.4086  
.2695  
.2348  
.1234  
.0022  
.0000  
.0981  
.1829  
-.0410  
-.0130  
.1225  
.0000  
-.2809  
-.2599  
-.2349  
-.2125  
-.1800  
-.0000  
-.1145  
-.0327  
-.1145  
-.0428  
.1093  
.1634  
.1145  
-.0300  
-.3042  
-.2837  
-.3023  
-.1079  
.1399  
.0035  
-.1231  
.4096  
.3841  
.4193  
.5600  
.5872  
.5938  
.4397  
.4342  
.1003  
.1518  
.1843  
-.0320  
-.0300  
-.0263  
-.2125  
-.2215  
-.2599  
-.3100  
-.2837  
-.3023  
-.1079  
.1399  
.0035  
-.1231

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(R81U40)

ARC11-716 1A14 01+112+312N25+AT11 UPPER WING

ALPHAD( 1 ) = -7.910 BETA( 2 ) = -4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
	.857			.1492				
	.855	.1770						
	.900	.1600		.1274				-.2119
	.905		.1479		.1580			
	.930			.1213				-.0470
	.933			.1309				
	.965	.1025						

ALPHAD( 1 ) = -7.900 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
	.100							
	.020	-.0828	-.1139	.1467	.3728	.3009	.2430	.2777
	.025			.2833	.5204	.4690	.4966	.5021
	.040		-.0213					
	.045		.2468		.3640	.3740	.3791	.3788
	.050	-.1028		.2412				
	.090		.1942					
	.085		.0736					
	.094	-.0505			.0651	.0850	.1341	.1660
	.130		.0000					
	.163		.1322					
	.177							
	.193	.0070			-.0969	-.0543	-.0353	-.0367
	.229							
	.250							
	.274							
	.339		-.0258					
	.362	.0858						
	.400							
	.402							
	.497	.0000						
	.550							
	.565							
	.600							
	.650							
	.700	-.0558						
	.725							
	.750							
	.760							



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
ARC11-716 1A14 01+T12+S12M25+AT11 UPPER WING

(R81U40)

ALPHA( 1 ) = -7.900 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW				.1283	.0760		
.775			.1220				
.808							
.834	.0920			.1193	.1216	-.0490	
.850			.0992				
.857							
.865	.1296			.0923			-.1084
.900	.1171		.0942	.1430			
.905				.0767		.0154	
.930			.0851				
.953							
.965	.0554						

ALPHA( 1 ) = -7.910 BETA( 4 ) = 4.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1543	-.1071	.1160	.3035	.1987	.1278	.1910
.020			.4520	.3881	.4088	.4307	
.025		.2325					
.040		-.0532	.2143				
.045				.3222	.3154	.3203	.3285
.050	-.1398		.2197				
.080			.1630				
.085		.0284					
.086							
.094	-.0734			.0621	.0781	.1179	.1443
.150		.0000					
.163			.0786				
.177		.1181					
.193							
.229	-.0206			-.1235	-.0523	-.0388	-.0392
.250			.0056				
.274		-.0242					
.339	.0765			-.2560	-.1770		-.2058
.362							
.400			-.1512				
.402							
.497	.0000			-.2805	-.2269		
.550			.0900				
.565							
.600							
.650							-.2840
							-.2813

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ARC11-716 1A14 01+712+312N25+AT11 UPPER WING

ALPHA(1) = -7.910 BETA(4) = 4.100

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .700 -.0333 .0350 -.1206 -.2738

.725 .0424 .0734 .1322

.750 .0911 .0493 .1290 .0525

.775 .0324 .0137 .1165 -.0063

.800 .0381 .0430 .1165 .1577

.825 .0631 .0428

.850 .0381 .0428

.875 .0381 .0428

.900 .0381 .0428

.925 .0381 .0428

.950 .0381 .0428

.975 .0381 .0428

.990 .0381 .0428

.995 .0381 .0428

.999 .0381 .0428

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.999 .0381 .0428

.999 .0381 .0428



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6  
ARC11-715 IA14 01+712+512NGS:AT11 UPPER WING (R81U40)

ALPHAD ( 1 ) = -8.000 BETAD ( 5 ) = 0.190

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.550				-2131	-2072		
.565			.0000				-2640
.600							
.650							
.700							
.725							
.750							
.760							
.775							
.800							
.834							
.850							
.857							
.865							
.900							
.905							
.950							
.953							
.965							

ALPHAD ( 2 ) = -4.010 BETAD ( 1 ) = -0.050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000							
.020							
.025							
.040							
.045							
.050							
.080							
.095							
.086							
.094							
.150							
.165							
.177							
.195							
.229							
.250							
.274							
.339							

(R01U40)

ARC11-716 1A14 01+T12+S12N23+AT11 UPPER WING

ALPHA( 2 ) = -4.010 BETA( 1 ) = -0.050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8170

X/CW

.362	.0869					
.400						
.402						
.437	.0000					
.550						
.555						
.600						
.650						
.700						
.725						
.750						
.750						
.775						
.819						
.834	.0640					
.850						
.857						
.865	.1673					
.900	.1658					
.905						
.930						
.933						
.965	.1035					

ALPHA( 2 ) = -4.010 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8070

X/CW

.000	.0197	-.0464	.3357	.5726	.5296	.4870	.5292
.020				.5175	.5111	.5109	.5300
.025			.3116				
.040		.0152					
.045			.2590				
.050	-.0497			.3115	.3324	.3355	.3360
.060				.1571			
.065			.1374				
.066		.0904					
.094	-.0477						
.150							
.163		.0000					
.177							
.193							



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N25+AT11 UPPER WING (RB1040)

ALPHAO ( 2 ) = -1.010 BETAO ( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.229 -.0096  
 .250  
 .274  
 .339  
 .362 .0349  
 .400  
 .402  
 .497 .0000  
 .550  
 .565  
 .600  
 .650  
 .700  
 .725  
 .750  
 .775  
 .808  
 .834  
 .850  
 .857  
 .855  
 .900  
 .905  
 .933  
 .933  
 .955  
 .955

-.2052 -.1604 -.1369 -.1220  
 -.1219  
 -.1061  
 -.3084 -.3022 -.3083  
 -.2594  
 -.3619 -.3435  
 .0000  
 -.3864  
 -.1378  
 -.1524  
 -.0353 -.0390 -.2639  
 .0937  
 .1004  
 .0267 -.1616 -.2985  
 .0552 -.1507  
 .0707 -.1624  
 .0876

-.3555

-.3712 -.3664

-.2969

ALPHAO ( 2 ) = -3.930 BETAO ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000  
 .020  
 .025  
 .040  
 .045  
 .050  
 .080  
 .085  
 .086  
 .094

-.0969 -.0730 .2036 .4510 .4354 .3837 .4380  
 .4447 .4364 .4416 .4584  
 .2258  
 .1709  
 .2684 .2778 .2805 .2859  
 .1341  
 .0980  
 .0406

-.1263

ORIGINAL PAGE 13  
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DATE 15 JAN 75 TABULATED PRESSURE DATA - IAI14A - VOL. 6  
 ARC11-716 IAI14 01+112+512N25+AT11 UPPER WING (RB1040)

ALPHA(2) = -3.940 BETA(4) = 4.100

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.5730	.7800
.8870		
.080	.0861	.1031
.095		
.086	.0012	
.094	-.1373	
.150		-.0362
.163	.0000	-.0349
.177		-.0045
.193	.0525	.0224
.229	-.0611	
.250		-.0151
.274	-.0793	
.339	-.0897	
.362	.0205	
.400		-.1853
.402	-.2093	-.1507
.497		-.1470
.550	.0000	-.1523
.565		
.600	-.3194	-.2889
.650		-.2840
.700	-.1012	-.2644
.725		-.3346
.753		-.1327
.760	-.0140	-.0494
.775	.0579	-.3012
.808		-.3560
.834	.0167	
.850		.0453
.857	.0259	-.0659
.865		.0440
.900	.0368	.0319
.905		-.1293
.930	.0158	
.953	-.0057	-.0795
.965	.0067	-.0990
	-.0211	

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01\*112+312N25+AT11 UPPER WING  
 (R81U40)

ALPHAO( 2) = -3.930 BETA0 ( 5) = 0.130

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3077	-.1217	.0915	.3405	.2702	.1964	.2306
.020				.3065	.2942	.3056	.3092
.040			.1187				
.060		-.1025					
.080			.1212				
.100	-.2679			.1813	.1840	.1971	.1864
.120				.0849			
.140			.0902				
.160		.0056					
.180	-.1899			-.0397	-.0345	-.0055	.0126
.200		.0000					
.220			-.0082				
.240		.0492					
.260	-.0852			-.1731	-.1390	-.1318	-.1442
.280			-.0695				
.300		-.0891					
.320	.0316			-.2731	-.2485		-.2960
.340			-.2090				
.360	.0000			-.2958	-.2857		
.380			.0000				-.3510
.400				-.3143			
.420				-.1436			
.440	-.0841			-.0580			
.460					-.2816	-.3107	
.480			-.0225	.0353	-.0263		
.500		.0321					
.520	.0065			.0137	.0541	-.0745	
.540			.0000				
.560	.0356			-.0167			-.0845
.580	.0030			-.0155	.0663		
.600				-.0364		.0687	
.620			-.0283				
.640	-.0536						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1U40)

ARC11-715 IA14 01+T12+S12N23+AT11 UPPER WING

ALPHA(3) = -.379 BETA(1) = -.090

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/3W	X/CM			
.2990	.3640	.4270	.5340	.6730 .7800 .8870
.1118	.0085	.4571	.7524	.7300 .6806 .7304
.020			.4606	.4927 .4823 .4965
.025		.2879		
.040	.0445	.1810		
.045			.2251	.2611 .2576 .2608
.050	.0489		.0577	
.060		.0357		
.085				
.086	.0958			
.094	.0186			
.130			-.1606	-.1148 -.0443 -.0010
.163	.0090			
.177		-.0845		
.193	.0174			
.229	-.0061			
.250			-.2958	-.2643 -.2274 -.1969
.274		-.2059		
.339	-.1625			
.352	.0316			
.400			-.4009	-.3949 -.3706
.402		-.3093		
.497	.0090			
.530			-.4270	-.4388
.555		.0090		
.600				-.3862
.630				-.4285
.700	-.2334		-.1676	
.725			-.3567	
.750				-.4611 -.3923
.780		-.1075	-.1944	-.4229
.775		-.0324		
.808				
.834	-.0156		-.1304	-.2911 -.4142
.850		.0189		
.857				
.865	.1027		-.1066	
.900	.1194			-.3173
.909		.0622	-.2139	
.957			-.0677	-.2834
.953		.0795		
.965	.0628			



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI4A - VOL. 6  
ARC11-715 IAI4 01+T12+S12N25+AT11 UPPER WING  
(RBIU40)

ALPHA(3) = -.380 BETA(3) = .020

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP									
Y/B	.2990	.3640	.4270	.5340	.6730	.7800	.8870			
X/CW										
.070	-.0941	-.0974	.3086	.5738	.5420	.5057	.5438			
.020			.2052	.3778	.3860	.3738	.3721			
.025		-.0332								
.040			.1326							
.075				.1805	.1870	.1856	.1765			
.090	-.1224			.0361						
.085			.0294							
.086		.0286								
.094	-.1378									
.150		.0090		-.1480	.1357	-.0862	-.0533			
.163			-.0808							
.177		.0038								
.193										
.229	-.0321			-.2885	-.2595	-.2502	-.2319			
.250			-.1927							
.274		-.1652								
.339										
.362	-.0008			-.3644	-.3843		-.3914			
.400			-.2919							
.402										
.497	.0000			-.3942	-.4123					
.550			.0000							
.555										
.600										
.650										
.700	-.2262			-.1857	-.1499		-.4495			
.725										
.750			-.0851							
.750				-.1006	-.2845					
.775			.0307							
.808										
.834	-.0026			-.0413	-.2108	-.3019				
.850			.0331							
.857										
.865	.0348									
.900	.0450			-.0054						
.905			.0308	-.1855						
.950			.0146							
.953			.0170							
.965	-.0176									

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(R01U40)

ARC11-716 1A14 01-112-S12N25+AT11 UPPER WING

ALPHA( 3 ) = -.390 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/Z	.2990	.3640	.4270	.5340 .6730 .7600 .8670
X/CW				
.000	-.1996	-.1475	.2483	.5088 .4522 .3955 .4434
.020			.1411	.2844 .2950 .2759 .2854
.025		-.0504		
.040			.0822	.1230 .1182 .1075 .1099
.045	-.2069			-.0002
.050			.0066	
.060		-.0019		
.065				
.086		-.2043		-.1552 -.1703 -.1259 -.0922
.094				
.150		.0000		
.163			-.0902	
.177		.0168		
.193				
.229	-.0773			-.3012 -.2654 -.2603 -.2542
.250		-.1535		
.274				
.339		-.1320		
.362	.0102			
.400			-.3704 -.3553	
.402		-.2613		
.497	.0000			
.550		-.4049 -.3674		
.565		.0000		
.601				-.4138
.650			-.4176	
.700	-.1860		-.1549	
.725			-.1972	
.750				-.3646 -.4141
.760		-.0809		
.775			-.0738 -.2214	
.808		.0136		
.834	-.0187			
.850			-.0197 -.1522 -.2341	
.857		-.0012		
.865	.0157			
.900	.0035		-.0300	-.2667
.905		-.0107	-.0112	
.950			-.0411	-.1737
.953		-.0240		
.965	-.0321			



BETA (S) = 0.149

DEPENDENT VARIABLE CR

1782	2982	3643	4270	5340	6730	7800	8870
7/CM							
.000	-.3878	-.1856	.0708	.3551	.3740	.3202	.3548
.020			.0457	.2038	.2530	.2440	.2559
.040		-.1263					
.060			.0336	.0869	.0966	.0941	.0795
.080	-.3077			.0124			
.100			.0100				
.120		-.5462					
.140	-.2863			-.1278	-.1449	-.1202	-.0925
.160		.0000					
.180		.0336					
.200	-.2762		-.0523	-.2331	-.2289	-.2383	-.2151
.220			-.1023				
.240		-.1233					
.260	.0251			-.3160	-.3300		-.3611
.280			-.2423				
.300	.0043			-.3377	-.3497		
.320			.0000				
.340						-.3560	-.4000
.360	-.1742			-.1456			
.380						-.3302	-.372
.400			-.0790	-.0493	-.1270		
.420			-.0053				
.440	-.0300			-.0334	-.0356	-.1601	
.460			-.0307				
.480	-.0001			-.0340	.0347		-.181
.500	-.0234		-.0333				
.520			-.0411			-.1048	
.540							
.560	-.0669						

(R81U40)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01-112-S12N25+AT11 UPPER WING

DATE 10 JAN 75

ALPHA(4) = 4.080 BETAD (1) = -0.060

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/B	.2990	.640	.4270	.5340	.6720	.7800	.8870
Z/CM							
.000	.1104	.0657	.0006	.8240	.8466	.7880	.8263
.020				.3370	.3640	.3240	.3157
.025			.2071				
.040		.0373					
.045				.0885			
.050	.0563			.1115	.1263	.1032	.0777
.080				-.0637			
.085			-.0717				
.086		.0735					
.094	.0179			-.2819	-.2334	-.1862	-.1505
.130		.0000					
.163			-.1985				
.177		-.0500					
.193							
.229	-.0022			-.4330	-.3716	-.3355	-.3015
.250			-.2914				
.274		-.2323					
.339							
.352	.0133			-.4983	-.4861		-.4401
.400			-.3623				
.402							
.497	.0000			-.5310	-.5222		
.550			.0000				
.565							-.4468
.600							
.643							
.700	-.2847			-.4599	-.1930		-.4924
.725							
.730							
.760			-.1975				
.775				-.3511	-.5050		
.808			-.1517				
.834							
.850	-.1176			-.2339	-.4112	-.4693	
.857			-.1153				
.865	.0218						-.4063
.900	.0640			-.1928			
.905			-.0546	-.2678			
.925							-.4043
.950			-.1776				
.955			-.0031				
.965	.0168						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAL44 - VOL. 6  
 ARCL1-T16 IAL4 01+T12+512R25+AT11 UPPER WING (RBIU40)

ALPHA (A) = 4.080 BETA (B) = -4.020

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

WING	2990	3640	4270	5340	6730	7800	8870
W/CW							
.000	.0150	-.0186	.4046	.7252	.7659	.7122	.7225
.020			.1728	.2805	.1460	.2824	.2549
.040		-.0170					
.060			.0616				
.080	-.0312			.0726	.0952	.0749	.0311
.100			-.0733	-.0760			
.120		.0069					
.140	-.0620			-.2827	-.2454	-.2000	-.1792
.160		.0000					
.180		-.0608	-.1898				
.200	-.0620			-.4003	.3764	-.3465	-.3278
.220		-.2210	-.2803				
.240				-.4857	-.4837		-.4551
.260	-.3105		-.3673				
.280				-.5105	-.5161		
.300			.0000				
.320						-.4954	-.4539
.340	-.3036			-.4259	-.1905		
.360			-.1758			-.3038	-.4657
.380			-.1467	-.2795	-.4827		
.400	-.0822			-.2318	-.3566	-.4690	
.420			-.0933				
.440	.0403			-.2271			-.4165
.460	.0636		-.0343	-.2923			
.480				-.1978	-.2991		
.500			.0016				
.520	-.0050						



TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 75

(R81040)

ARC11-71.6 1A14 01+712+512N23+AT11 UPPER WING

ALPHAS(4) = 4.010 BETA(3) = .030

SECTION: (1) UPPER WING DEPENDENT VARIABLE CP

Y/Z	.2990	.3640	.4270	.5340	.6730	.7800	.8870
Y/CW							
.000	-.0936	-.1701	.3283	.6610	.6536	.5950	.6163
.020			.1145	.2325	.2723	.2151	.1886
.040		-.1122					
.060			.0294	.0409	.0491	.0249	-.0109
.080	-.1274			-.1024			
.100			-.0903				
.120		-.0171					
.140	-.1230			-.2880	-.2622	-.2224	-.2018
.160		.0000					
.180			-.1857				
.200		-.0616					
.220	-.0897			-.3748	-.3810	-.3614	-.3422
.240		-.2257					
.260				-.4549	-.4793		-.4680
.280			-.3629				
.300	-.0000			-.4834	-.5051		-.4683
.320						-.5017	
.340				-.2451			
.360					-.5224	-.4656	
.380			-.1595				
.400				-.2265	-.3906		
.420			-.0928				
.440							
.460				-.2162	-.3247	-.3807	
.480			-.0478				
.500							
.520				-.1615			-.4273
.540			-.0156	-.2944			
.560				-.0940			-.2850
.580			-.0092				
.600							
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.990							



(R81040)

DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI4A - VOL. 6  
ARC11-716 IAI4 01+712+S12R23+AT11 UPPER WING

ALPHA = 4.040 BETA = 4.090

SECTION 11 UPPER WING DEPENDENT VARIABLE C<sub>p</sub>

X/CW	Y/CW	CP
.000	.2247	-.3544
.020		.5353
.040		.1117
.060		.0315
.080		-.1141
.100		-.0370
.120		-.0347
.140		-.1106
.160		-.0390
.180		-.2062
.200		.0000
.220		-.1842
.240		-.0468
.260		-.2478
.280		-.2003
.300		-.0092
.320		-.3302
.340		-.4472
.360		-.4571
.380		.0000
.400		-.4979
.420		-.1790
.440		-.2519
.460		-.1455
.480		-.2948
.500		-.0601
.520		-.0453
.540		-.1470
.560		-.2405
.580		-.1084
.600		-.2089
.620		-.0438
.640		-.1084
.660		-.2671
.680		-.0374
.700		-.0867

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TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1040)

ARC11-716 IA14 OI+T12+312N25+AT11 UPPER WING

ALPHA(4) = 4.060 BETA(5) = 8.160

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.4271	-.2792	-.0162	.2338	.4064	.4170	.4966
.020				-.0230	.1079	.0860	.1030
.025			-.0401				
.040		-.1779					
.045			-.0617				
.050	-.3742			-.1055	-.0574	-.0730	-.0770
.060				-.1760			
.065			-.0869				
.066		-.1779					
.074	-.3078			-.2535	-.2846	-.2698	-.2297
.100		.0000					
.105			-.1465				
.177		-.0207					
.193			-.1933				
.229	-.0756			-.2932	-.3668	-.3007	-.3549
.253							
.274		-.1590					
.339				-.3529	-.4216		-.4677
.362	.0122		-.2862				
.400				-.3600	-.4303		
.402			.0000			-.4639	-.4589
.497	.0000			-.1880			
.530							
.565							
.600							
.650							
.700	-.2610			-.1752		-.4111	-.4584
.725							
.750			-.1362				
.760				-.1022	-.2200		
.775			-.0690				
.808							
.834	-.0550			-.0666	-.1579	-.2615	
.850			-.0827				
.857							
.865	-.0324			-.0696			-.2990
.900	-.0335		-.0720	-.0649			
.905			-.0679			-.2100	
.950							
.953			-.0779				
.965	-.1103						





## DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI14 - VOL. 6

(R81040)

ARC11-716 IAI14 01+712+512N25+AT11 UPPER WING

ALPHA( 5 ) = 7.980 BETAD ( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/B	.2990	.3640	.4270	.5340	.6730	.7800	.8870
Y/CW							
.000	.0121	-.0457	.4353	.6157	.7713	.7113	.6933
.020			.0907	.1621	.1872	.0933	.0280
.040		-.0685					
.065			-.0298				
.090	-.0344			-.0436	-.0538	-.0959	-.1727
.115				-.1932			
.140			-.1723				
.165		-.0241					
.190	-.0543			-.3936	-.3519	-.3341	-.3293
.215		.0000					
.240			-.3249				
.265		-.1605					
.290	-.0765			-.5403	-.4675	-.4492	-.4307
.315		-.3037					
.340			-.3780				
.365	-.0581						
.390				-.6060	-.5576		-.5281
.415		-.4421					
.440				-.6215	-.5538		
.465			.0000				
.490							
.515							
.540							
.565							
.590							
.615							
.640							
.665							
.690							
.715							
.740							
.765							
.790							
.815							
.840							
.865							
.890							
.915							
.940							
.965							



$$\beta_1 = 7.915 \quad \text{BETA2 (3)} = .030$$

DEPENDENT VARIABLE: C6

Year	1990	1991	1992	1993	1994
1990	2,990	3,540	4,210	5,323	6,730
1991					
1992					
1993					
1994					
1995					
1996					
1997					
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2082					
2083					
2084					

3-14

.000	-.0727	-.2161	.2920	.6774	.6757	.5185	.6148
.720				.0598	.1252	.9474	-.0055

025  
-3.79

- .749

394 -1971

0.00	- .1350	- .1093	-.0994	-.1341	-.1949
------	---------	---------	--------	--------	--------

66-11100-10000

1995 - 1996

1,169 -1,277

150	-.3905	-.3698	-.3536	-.3429
-----	--------	--------	--------	--------

.163 .0500

— 1914

193 -1511

.229 -.1300

25) -4953 -4753 -4653 -4416

-.3402

	- .2814
.939	- .0600

	- .0069		
.582		- .5370	- .5329

972  
-4184

49' .0000

-.550      -.5463    -.5636

.555  
.0000

-.5257

0.620  
-0.3317

1995 1996 1997 1998 1999 2000 2001 2002 2003 2004 2005 2006 2007 2008 2009 2010 2011 2012 2013 2014 2015 2016 2017 2018 2019 2020 2021 2022 2023 2024 2025 2026 2027 2028 2029 2030 2031 2032 2033 2034 2035 2036 2037 2038 2039 2040 2041 2042 2043 2044 2045 2046 2047 2048 2049 2050 2051 2052 2053 2054 2055 2056 2057 2058 2059 2060 2061 2062 2063 2064 2065 2066 2067 2068 2069 2070 2071 2072 2073 2074 2075 2076 2077 2078 2079 2080 2081 2082 2083 2084 2085 2086 2087 2088 2089 2090 2091 2092 2093 2094 2095 2096 2097 2098 2099 2100 2101 2102 2103 2104 2105 2106 2107 2108 2109 2110 2111 2112 2113 2114 2115 2116 2117 2118 2119 2120 2121 2122 2123 2124 2125 2126 2127 2128 2129 2130 2131 2132 2133 2134 2135 2136 2137 2138 2139 2140 2141 2142 2143 2144 2145 2146 2147 2148 2149 2150 2151 2152 2153 2154 2155 2156 2157 2158 2159 2160 2161 2162 2163 2164 2165 2166 2167 2168 2169 2170 2171 2172 2173 2174 2175 2176 2177 2178 2179 2180 2181 2182 2183 2184 2185 2186 2187 2188 2189 2190 2191 2192 2193 2194 2195 2196 2197 2198 2199 2200 2201 2202 2203 2204 2205 2206 2207 2208 2209 2210 2211 2212 2213 2214 2215 2216 2217 2218 2219 2220 2221 2222 2223 2224 2225 2226 2227 2228 2229 2230 2231 2232 2233 2234 2235 2236 2237 2238 2239 2240 2241 2242 2243 2244 2245 2246 2247 2248 2249 2250 2251 2252 2253 2254 2255 2256 2257 2258 2259 2260 2261 2262 2263 2264 2265 2266 2267 2268 2269 2270 2271 2272 2273 2274 2275 2276 2277 2278 2279 2280 2281 2282 2283 2284 2285 2286 2287 2288 2289 2290 2291 2292 2293 2294 2295 2296 2297 2298 2299 2300 2301 2302 2303 2304 2305 2306 2307 2308 2309 2310 2311 2312 2313 2314 2315 2316 2317 2318 2319 2320 2321 2322 2323 2324 2325 2326 2327 2328 2329 2330 2331 2332 2333 2334 2335 2336 2337 2338 2339 2340 2341 2342 2343 2344 2345 2346 2347 2348 2349 2350 2351 2352 2353 2354 2355 2356 2357 2358 2359 2360 2361 2362 2363 2364 2365 2366 2367 2368 2369 2370 2371 2372 2373 2374 2375 2376 2377 2378 2379 2380 2381 2382 2383 2384 2385 2386 2387 2388 2389 2390 2391 2392 2393 2394 2395 2396 2397 2398 2399 2400 2401 2402 2403 2404 2405 2406 2407 2408 2409 2410 2411 2412 2413 2414 2415 2416 2417 2418 2419 2420 2421 2422 2423 2424 2425 2426 2427 2428 2429 2430 2431 2432 2433 2434 2435 2436 2437 2438 2439 2440 2441 2442 2443 2444 2445 2446 2447 2448 2449 2450 2451 2452 2453 2454 2455 2456 2457 2458 2459 2460 2461 2462 2463 2464 2465 2466 2467 2468 2469 2470 2471 2472 2473 2474 2475 2476 2477 2478 2479 2480 2481 2482 2483 2484 2485 2486 2487 2488 2489 2490 2491 2492 2493 2494 2495 2496 2497 2498 2499 2500 2501 2502 2503 2504 2505 2506 2507 2508 2509 2510 2511 2512 2513 2514 2515 2516 2517 2518 2519 2520 2521 2522 2523 2524 2525 2526 2527 2528 2529 2530 2531 2532 2533 2534 2535 2536 2537 2538 2539 2540 2541 2542 2543 2544 2545 2546 2547 2548 2549 2550 2551 2552 2553 2554 2555 2556 2557 2558 2559 2560 2561 2562 2563 2564 2565 2566 2567 2568 2569 2570 2571 2572 2573 2574 2575 2576 2577 2578 2579 2580 2581 2582 2583 2584 2585 2586 2587 2588 2589 2590 2591 2592 2593 2594 2595 2596 2597 2598 2599 2600 2601 2602 2603 2604 2605 2606 2607 2608 2609 2610 2611 2612 2613 2614 2615 2616 2617 2618 2619 2620 2621 2622 2623 2624 2625 2626 2627 2628 2629 2630 2631 2632 2633 2634 2635 2636 2637 2638 2639 2640 2641 2642 2643 2644 2645 2646 2647 2648 2649 2650 2651 2652 2653 2654 2655 2656 2657 2658 2659 2660 2661 2662 2663 2664 2665 2666 2667 2668 2669 2670 2671 2672 2673 2674 2675 2676 2677 2678 2679 2680 2681 2682 2683 2684 2685 2686 2687 2688 2689 2690 2691 2692 2693 2694 2695 2696 2697 2698 2699 2700 2701 2702 2703 2704 2705 2706 2707 2708 2709 2710 2711 2712 2713 2714 2715 2716 2717 2718 2719 2720 2721 2722 2723 2724 2725 2726 2727 2728 2729 2730 2731 2732 2733 2734 2735 2736 2737 2738 2739 2740 2741 2742 2743 2744 2745 2746 2747 2748 2749 2750 2751 2752 2753 2754 2755 2756 2757 2758 2759 2760 2761 2762 2763 2764 2765 2766 2767 2768 2769 2770 2771 2772 2773 2774 2775 2776 2777 2778 2779 2780 2781 2782 2783 2784 2785 2786 2787 2788 2789 2790 2791 2792 2793 2794 2795 2796 2797 2798 2799 2800 2801 2802 2803 2804 2805 2806 2807 2808 2809 2810 2811 2812 2813

.78
- .927 - .5250
.5600

6559

779 -2787 -4574

618 - 2069

.634    -.0617

.050                    -.2316                    -.4663                    -.4690

-.057  
-.1436

	1968	1967	1966	1965	1964	1963	1962	1961	1960	1959	1958	1957	1956	1955	1954	1953	1952	1951	1950	1949	1948	1947	1946	1945	1944	1943	1942	1941	1940	1939	1938	1937	1936	1935	1934	1933	1932	1931	1930	1929	1928	1927	1926	1925	1924	1923	1922	1921	1920	1919	1918	1917	1916	1915	1914	1913	1912	1911	1910	1909	1908	1907	1906	1905	1904	1903	1902	1901	1900	1899	1898	1897	1896	1895	1894	1893	1892	1891	1890	1889	1888	1887	1886	1885	1884	1883	1882	1881	1880	1879	1878	1877	1876	1875	1874	1873	1872	1871	1870	1869	1868	1867	1866	1865	1864	1863	1862	1861	1860	1859	1858	1857	1856	1855	1854	1853	1852	1851	1850	1849	1848	1847	1846	1845	1844	1843	1842	1841	1840	1839	1838	1837	1836	1835	1834	1833	1832	1831	1830	1829	1828	1827	1826	1825	1824	1823	1822	1821	1820	1819	1818	1817	1816	1815	1814	1813	1812	1811	1810	1809	1808	1807	1806	1805	1804	1803	1802	1801	1800	1799	1798	1797	1796	1795	1794	1793	1792	1791	1790	1789	1788	1787	1786	1785	1784	1783	1782	1781	1780	1779	1778	1777	1776	1775	1774	1773	1772	1771	1770	1769	1768	1767	1766	1765	1764	1763	1762	1761	1760	1759	1758	1757	1756	1755	1754	1753	1752	1751	1750	1749	1748	1747	1746	1745	1744	1743	1742	1741	1740	1739	1738	1737	1736	1735	1734	1733	1732	1731	1730	1729	1728	1727	1726	1725	1724	1723	1722	1721	1720	1719	1718	1717	1716	1715	1714	1713	1712	1711	1710	1709	1708	1707	1706	1705	1704	1703	1702	1701	1700	1699	1698	1697	1696	1695	1694	1693	1692	1691	1690	1689	1688	1687	1686	1685	1684	1683	1682	1681	1680	1679	1678	1677	1676	1675	1674	1673	1672	1671	1670	1669	1668	1667	1666	1665	1664	1663	1662	1661	1660	1659	1658	1657	1656	1655	1654	1653	1652	1651	1650	1649	1648	1647	1646	1645	1644	1643	1642	1641	1640	1639	1638	1637	1636	1635	1634	1633	1632	1631	1630	1629	1628	1627	1626	1625	1624	1623	1622	1621	1620	1619	1618	1617	1616	1615	1614	1613	1612	1611	1610	1609	1608	1607	1606	1605	1604	1603	1602	1601	1600	1599	1598	1597	1596	1595	1594	1593	1592	1591	1590	1589	1588	1587	1586	1585	1584	1583	1582	1581	1580	1579	1578	1577	1576	1575	1574	1573	1572	1571	1570	1569	1568	1567	1566	1565	1564	1563	1562	1561	1560	1559	1558	1557	1556	1555	1554	1553	1552	1551	1550	1549	1548	1547	1546	1545	1544	1543	1542	1541	1540	1539	1538	1537	1536	1535	1534	1533	1532	1531	1530	1529	1528	1527	1526	1525	1524	1523	1522	1521	1520	1519	1518	1517	1516	1
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.967	- .0896	- .4184
.9914	- .2105	- .8075

Year	1923	1924	1925	1926	1927	1928	1929	1930	1931	1932	1933	1934	1935	1936	1937	1938	1939	1940	1941	1942	1943	1944	1945	1946	1947	1948	1949	1950	1951	1952	1953	1954	1955	1956	1957	1958	1959	1960	1961	1962	1963	1964	1965	1966	1967	1968	1969	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100
1923	1924	1925	1926	1927	1928	1929	1930	1931	1932	1933	1934	1935	1936	1937	1938	1939	1940	1941	1942	1943	1944	1945	1946	1947	1948	1949	1950	1951	1952	1953	1954	1955	1956	1957	1958	1959	1960	1961	1962	1963	1964	1965	1966	1967	1968	1969	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	

1150

965 - 0621

(R81U40)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 O1+T12+SI2N88+AT11 UPPER WING

DATE 19 JAN 75

ALPHA(5) = 0.000 BETA(4) = 0.210

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW	CP	Y/BW	CP	Y/BW	CP	Y/BW	CP
.2990	.3640	.4270	.5340	.6730	.7800	.8870	

Y/CW	CP	Y/CW	CP	Y/CW	CP	Y/CW	CP
.000	-.4007	-.2793	-.1031	.1070	.4080	.4303	.5120
.020				-.2185	-.0362	-.0367	-.0490
.025			-.1214				
.040		-.2217					
.045			-.1320				
.050	-.3702		-.2597	-.2167	-.2077	-.2231	
.080			-.3030				
.095			-.1518				
.006		-.1938					
.094	-.3140			-.3424	-.4108	-.3901	-.3552
.150		.0000					
.163			-.2044				
.177							
.193		-.1468					
.229	-.1190			-.4710	-.4806	-.4705	-.4503
.250			-.2497				
.274							
.339		-.2155					
.362	-.0204			-.5067	-.5210		-.5397
.400			-.3274				
.402							
.497	.0000			-.5050	-.4952		
.550				.0000			
.565							-.5296
.600						-.5353	
.610							
.700	-.3612			-.3016			
.725					-.2099		
.750						-.4989	-.5045
.760			-.1780				
.775				-.2438	-.2749		
.809			-.1290				
.834							
.850	-.0992			-.2073	-.2337	-.4230	
.857			-.1320				
.865	-.0801						-.3914
.900	-.0947			-.2059			
.905			-.1257		-.1399		
.950			-.2062			-.3222	
.993			-.1283				
.965	-.1278						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81041) (15 FEB 74)

UPPER WING

ARC11-716 IA14 Q1+T12+S12N25

PARAMETRIC DATA

MACH = .600 ELEVOM = .000  
RUDDER = .000 SPEEDBK = .000

REFERENCE DATA

SREF = 2.4210 SQ.FT. AREA = 29.5300 INCHES  
REF = 30.7090 INCHES AREA = .0000 INCHES  
SREF = 36.7090 INCHES AREA = .0000 INCHES  
SCALE = .0000 SCALE

ALPHA (1) = -7.940 BETA (1) = .020

SECTION: (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0304	-.0909	.0277	-.2137	-.4520	-.6075	-.4137
.020				.4334	.3879	.4092	.3530
.025			.2941				
.040		.0772					
.045			.2690				
.050	.0062		.2867	.2999	.2995	.2447	
.060			.1563				
.085			.1851				
.086		.1792					
.094	.0186						
.150		.0090		-.0529	-.0278	-.0047	-.0403
.163							
.177			-.0030				
.193		.1201					
.229	.0983						
.250				-.1760	-.1903	-.2020	-.2367
.274			.1089				
.339	-.0967						
.352	.0901						
.403							
.402			-.1174	-.2174		-.2489	
.497	.0000						
.550				-.1351	-.1693		
.565			.0000				
.600							
.650							
.700	-.0340				-.1462		
.725				-.1350			
.750				-.0753		.0372	.0034
.760			-.0886				
.775			.0364	.0364			
.808			.0341				
.834	-.0411						
.850				.0423	.0653	.0791	
.857			.0213				
.865	.0324						
.900	.0175			.0371			.0541
.905			.0269		.0752		





TABULATED PRESSURE DATA - IA14A - VOL. 6

UPPER WING

$$\text{BETA}(\text{I}) = -7.950 \quad \text{BETA}(\text{I} + 2) = 4.380$$

DEPENDENT VARIABLE C=

1954	.2990	.3540	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

0.57	-0.0001		
0.65	0.0056		
0.90	-0.0032	0.0014	0.0525
0.75		0.135	0.674
0.93		0.0225	0.077
0.95	-0.0335	0.011	

$$\text{ALPHA}(1) = -7.970 \quad \text{BETA}(3) = 0.150$$

SECTION / WING	DEPENDENT VARIABLE CF
----------------	-----------------------

1984	.2990	.3540	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

[illegible]

DATE 12 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N25  
 (RBU41)

UPPER WING

ALPHA( 1 ) = -7.970 BETAD ( 3 ) = 8.150

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/Cu							
.775				-.0090	.0049		
.808			-.0126				
.834	-.0093			.0061	.0344	.0556	
.850							
.857			-.0118				
.865	-.0135			.0309			.0551
.900	-.0155				.0560		
.935			-.0074	.0247		.0621	
.950							
.953			-.0171				
.965	-.0506						

ALPHA( 2 ) = -4.090 BETAD ( 1 ) = -8.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/Cu							
.000	.0560	.0948	.3543	.3903	.3204	.2086	.2572
.020				.3127	.3052	.3338	.3416
.025		.1748	.2656				
.040			.1519				
.045				.0441	.0491	.0683	.0505
.050	.0349			-.1276			
.080			-.0268				
.085		.1854					
.086							
.094	.0419						
.150		.0003		-.3112	-.3023	-.2958	-.2671
.163							
.177			-.2171				
.193		-.0281					
.229	.0805			-.3575	-.3149	-.4498	-.4593
.250							
.274			-.2581				
.339		-.2410					
.362	-.0145			-.2455	-.3140		-.3542
.400			-.2313				
.402							
.497	.0703			-.1277	-.1694		
.550			.0003				
.565							-.2178
.600							-.1487
.650							



(RB1U41)

UPPER WING

DATE 10 JAN 75

ARC11-716 1A14 01+112+512N25

ALPHA( 2 ) = -4.030 BETA( 1 ) = -0.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	CP
.700	.0074
.725	
.750	
.750	
.775	
.808	
.834	
.850	
.857	
.865	
.890	
.915	
.930	
.933	
.965	

ALPHA( 2 ) = -4.030 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	CP
.000	.0101
.020	.0469
.025	
.040	
.045	
.050	
.060	
.065	
.066	
.094	
.150	
.163	
.177	
.193	
.229	
.250	
.274	
.339	
.362	
.400	
.402	
.497	

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DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1U41)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 2 ) = -4.050 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE C<sub>p</sub>

Y/BW	X/CW	DEPENDENT VARIABLE C <sub>p</sub>
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.330		-.1394 -.2316
.363		.0000
.600		-.2139
.650		-.1561
.700	-.0338	-.1348
.725		-.0660
.750		-.0714
.760		.0493 .0436
.775		.0519
.806		.0540 .0751 .0866
.834	-.0170	.0413
.857		.0594
.865	.0594	.0738 .0896 .0854
.900	.0362	.0457 .0685 .0916
.905		.0907
.950		-.0102
.953		
.965		

ALPHA( 2 ) = -4.050 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE C<sub>p</sub>

Y/BW	X/CW	DEPENDENT VARIABLE C <sub>p</sub>
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.000	-.0244	.1936 .1563 .0194 -.1152 -.0456
.020		.3201 .2873 .3065 .2943
.025		.2396
.040	.1066	.1679
.045		.1045 .1185 .1176 .0746
.050	-.0284	-.0422
.080		.0359
.085	.1459	
.094	-.0151	-.2559 -.2265 -.2063 -.2169
.150		.0000
.163		-.1586
.177	.0043	
.193	.0505	
.229		-.3254 -.3331 -.3746 -.4020
.250		-.2299
.274		-.2122
.339		



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBU41)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 2 ) = -4.050 BETA( 3 ) = .030

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.362	.0020					
.400		-.2654	-.3141			-.3279
.402		-.2603				
.497	.0000					
.550		-.1771	-.2100			
.565		.0000				
.600						
.650					-.1695	-.2143
.700	-.0728		-.1359			
.725			-.0923			
.750				.0224		-.0094
.763		-.0991	.0230	.0241		
.775		.0196				
.808						
.834	-.0519		.0339	.0549	.0701	
.850						
.857		.0149				
.865	.0211					.0764
.900	.0084		.0555	.0785		
.905		.0243	.0575		.0804	
.950						
.953		.0199				
.965	-.0231					

ALPHA( 2 ) = -3.950 BETA( 4 ) = 4.070

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0634	.0032	.1320	.0817	-.0634	-.2149	-.1314
.020				.2889	.2400	.2658	.2418
.025		.2000					
.040		.0742					
.045		.1506					
.050	-.0493		.0944	.0978	.0973	.0499	
.060			-.0364				
.085		.0303					
.086		.1160					
.094	-.0330						
.100							
.163		.0000		-.2226	-.2138	-.2028	-.2184
.177							
.193		.0075					

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RS1U41)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA(2) = -3.950 BETA(4) = 4.070

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.229	.0356					
.250						
.274						
.339						
.362						
.400						
.452						
.497						
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.955						
.965						

ALPHA(2) = -3.950 BETA(5) = 9.110

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000						
.020						
.025						
.040						
.045						
.050						
.080						
.085						
.086						
.094						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1041)

UPPER WING

ARC11-716 IA14 01+12+S12N25

ALPHA( 2 ) = -3.950 BETA( 5 ) = 8.110

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3540	.4270
.3540	.5340	.6730
.6730	.7800	.8870
.8870		
.0000		
.1277		
.0070		
.0233		
.2226		
.3073		
.3322		
.3417		
.3557		
.2036		
.0004		
.3243		
.3271		
.2884		
.2286		
.2480		
.0050		
.1915		
.1614		
.1357		
.0022		
.0229		
.1497		
.0315		
.0123		
.0293		
.0049		
.0264		
.0486		
.0305		
.0176		
.0608		
.0639		
.0174		
.0200		
.0689		
.0174		
.0594		

ALPHA( 3 ) = -.310 BETA( 1 ) = -9.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3540	.4270
.3540	.5340	.6730
.6730	.7800	.8870
.8870		
.0000		
.1277		
.0070		
.0233		
.2226		
.3073		
.3322		
.3417		
.3557		
.2036		
.0004		
.3243		
.3271		
.2884		
.2286		
.2480		
.0050		
.1915		
.1614		
.1357		
.0022		
.0229		
.1497		
.0315		
.0123		
.0293		
.0049		
.0264		
.0486		
.0305		
.0176		
.0608		
.0639		
.0174		
.0200		
.0689		
.0174		
.0594		



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(R01041)

UPPER WING

ARC11-716 IA14 01+712+S12N25

ALPHA( 3 ) = -.310 BETA( 1 ) = -0.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.080						
.085						
.086						
.094						
.150						
.163						
.177						
.193						
.229						
.250						
.274						
.339						
.362						
.400						
.402						
.497						
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.834						
.890						
.897						
.885						
.900						
.905						
.950						
.953						
.965						



TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81041)

UPPER WING

DATE 10 JAN 73

ARC11-716 IA14 01+712+812N25

ALPHA ( 3 ) = -.320 BETA ( 2 ) = -4.020

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0089	.0632	.3332	.4564	.4302	.3807	.4220
.020				.1165	.0721	.0659	.0525
.025			.1292				
.040		.1116					
.045			.0058				
.050	-.0389			-.1342	-.1891	-.2013	-.2492
.080				-.3245			
.085			-.1645				
.086		.1016					
.094	-.0259			-.4721	-.4988	-.4974	-.4703
.150		.0000					
.163			-.3479				
.177		-.1347					
.193							
.229	.0210			-.4820	-.5580	-.6033	-.6179
.250			-.3702				
.274		-.3342					
.339	-.0902			-.3301	-.3984		-.4605
.362			-.3197				
.400				-.1849	-.2390		
.402			.0090				-.2472
.497	.0005			-.1729			
.550				-.0686	-.1354		
.565							
.600							
.650							
.700	-.0377						
.725							
.750			-.0716			.0281	-.0151
.760				.0460	.0348		
.775			.0478				
.808							
.834	-.0179			.0375	.0753	.0844	
.850			.0448				
.857							
.865	.0378						.0873
.900	.0416			.0731	.0957		
.903			.0904				
.930				.0745		.1020	
.933			.0392				
.965	.0030						

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$$\text{ALPHA}(\text{S}) = -.320 \quad \text{BETA}(\text{S}) = .040$$

SECTION (3) UPPER WING

DEPENDENT VARIABLE CP

Y/SW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.0535	.0289	.2848	.3790	.3344	.2811	.3254	.3254
.020			.1209	.1400	.0906	.0849	.0657	.0657
.025								
.040	.0900							
.045								
.050	-.0635							
.060								
.065								
.086	.0909							
.094	-.0446							
.130								
.160								
.177								
.193	-.1094							
.229	.0110							
.250								
.274								
.339								
.362	-.0819							
.400								
.402								
.497	.0000							
.550								
.565								
.600								
.650								
.700	-.1036							
.725								
.750								
.760								
.775								
.800								
.834	-.0501							
.850								
.857								
.865	.0097							
.900	.0053							
.905								
.950								
.955								
.965	-.0226							

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TABULATED PRESSURE DATA - IA14A - VOL. 6

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(881041)

UPPER WING

ARC11-716 IA14 01+T12+S12N25

ALPHA(3) = -.330 BETA(4) = 4.060

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1110	-.0146	.2307	.3014	.2388	.1887	.2402
.020				.1323	.0751	.0710	.0451
.025			.1083				
.040		.0357					
.045			.0311				
.050	-.0855		-.0927	-.1256	-.1475	-.2034	
.080			-.2300				
.085			-.1024				
.086		.0698					
.094	-.0641			-.3853	-.4030	-.4174	-.4179
.150		.0000					
.163			-.2703				
.177		-.0914					
.193							
.229	.0371			-.4412	-.4892	-.5242	-.5469
.250			-.3322				
.274		-.2981					
.339	-.0728			-.3513	-.4009		-.3860
.362			-.3407				
.400				-.2404	-.2680		
.402	.0000						
.497							
.550							
.565							
.600							
.630							
.700	-.1381						
.725				-.1208			
.730							
.760			-.1215				
.775				-.0061	.0012		
.808			-.0135				
.834	-.0898						
.830				.0128	.0438	.0852	
.837			-.0094				
.861	-.0275						
.900	-.0275						
.901			-.0030	.0375	.0748		.0837
.930							
.931			-.0009	.0449		.0855	
.965	-.0491						

(R81041)

UPPER WING

ARC11-716 1A14 Q1+T12+S12M25

ALPHA( 3 ) = -.330 BETA( 1 ) = 8.140

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1923	-.0371	.0948	.2386	.1313	.0925	.1471
.020				.1315	.0715	.0815	.0438
.025			.0462				
.040		-.0184					
.045			.0020				
.050	-.1260			-.0703	-.0882	-.1100	-.1791
.080				-.2025			
.085			-.0946				
.086		.0098					
.094	-.1012						
.150				-.3509	-.3628	-.3615	-.3718
.163		.0000					
.177			-.2410				
.193		-.0703					
.229	-.0104						
.250				-.4152	-.4570	-.4787	-.4999
.274			-.3114				
.339	-.2856						
.362	-.0693			-.3513	-.3919		-.3735
.400			-.3379				
.402				-.2579	-.2745		
.497	.0000						
.550			.0000				
.555							
.600							
.650							
.700	-.1787				-.1986		
.725				-.1396			
.750							
.760			-.1453				
.775				-.0351	-.0177		
.808			-.0416				
.834	-.1135			-.0094	.0254	.0547	
.850			-.0381				
.857							
.865	-.0314						
.900	-.0491			.0157	.0647		.0815
.905			-.0161				
.950				.0208		.0756	
.955			-.0191				
.985	-.0582						



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(RB1U41)

UPPER WING

ARC11-716 1A14 Q1+T12+S12+25

ALPHA( 4) = 4.130 BETA( 1) = -0.050

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP		
V/BW			
X/CW			
.000	-.0027	.0109	.3339
.020			.5454
.025			.5081
.040			.4365
.045			.4053
.050			.3769
.060			.3131
.085			-.1584
.086			-.3121
.094			-.5975
.150			-.7283
.163			-.8151
.177			-.9052
.193			-.7605
.229			-.4938
.250			-.0129
.274			-.7852
.339			-.8977
.362			-.9414
.400			-.8801
.402			-.5976
.497			-.3243
.550			-.5198
.565			-.6816
.600			-.8238
.630			-.9080
.700			-.9524
.725			-.5278
.750			-.3647
.760			-.3829
.775			-.4837
.808			-.5278
.834			-.1742
.850			-.2521
.857			.0000
.865			.3127
.900			-.1809
.930			-.1372
.955			-.0238
.965			.0228
			-.0629
			-.0124
			.0827
			.0554
			.0768
			.0898
			.0906
			.0890
			.0761
			.1058
			.1023
			.1037
			.1166
			.0585
			.0764
			.1166
			.1058
			.1900
			.0595
			.0393

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TABULATED PRESSURE DATA - IA14A - VOL. 6

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(R81U41)

UPPER WING

ARC11-716 IA14 01+T12+S12N25

ALPHA( 4) = 4.130 BETA( 2) = -4.010

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7000
.8070		
.000	-.0500	.3076
.020	-.0563	.5099
.040	.0261	.4774
.060		.4057
.080		.3787
.100		-.2330
.120		-.3581
.140		-.4690
.160		-.5931
.180		-.1106
.200		-.2460
.220		-.5027
.240		-.6193
.260		-.7190
.280		-.8173
.300		-.6637
.320		-.4310
.340		-.0084
.360		-.0765
.380		.0000
.400		.0000
.420		-.5394
.440		-.2067
.460		-.0326
.480		-.274
.500		-.339
.520		-.4620
.540		-.1917
.560		.0000
.580		.0000
.600		-.1982
.620		-.2639
.640		-.0000
.660		-.2992
.680		-.1859
.700		-.1316
.720		-.0570
.740		.0146
.760		-.0477
.780		-.0539
.800		.0543
.820		.0382
.840		.0496
.860		-.0117
.880		.0667
.900		.0822
.920		.0819
.940		.0497
.960		.0629
.980		.0629
.000		.0524
.020		.1102
.040		.0709
.060		.0880
.080		.1069
.100		.0479
.120		.0210



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81U41)

UPPER WING

ARC11-716 1A14 01+112+S12N23

ALPHA(4) = 4.139 BETA(3) = .060

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1136	-.0473	.2680	.4580	.4281	.3639	.3428
.020				-.1716	-.2917	-.4076	-.5046
.025			-.0780				
.040		.0153					
.045			-.1942				
.050	-.1127			-.4202	-.5414	-.6211	-.7222
.060				-.5680			
.085			-.3619				
.085		-.0023					
.094	-.0912						
.150				-.6497	-.7138	-.7597	-.7323
.163		.0000					
.177			-.4854				
.193		-.2512					
.229	-.0325			-.6187	-.7144	-.7867	-.8123
.250							
.274			-.4854				
.339		-.4372					
.362	-.1721						
.400			-.3966	-.4021	-.4732		-.5025
.402							
.497	.0000			-.2293	-.2785		
.550			.0000				
.565							
.600							
.650						-.1921	-.2805
.700	-.1293			-.1266			
.725			-.0804			.0149	-.0350
.750							
.760			-.0341	.0241	.0201		
.775			.0244				
.800							
.834	-.0331			.0492	.0663	.0782	
.850			.0242				
.857							
.865	.0178			.0684			.0785
.900	.0223						
.905			.0371	.0980			
.950				.0738		.1033	
.953			.0336				
.965	.0000						



(R81U41)

UPPER WING

ALPHA( 4 ) = 4.120 BETA( 4 ) = 4.080

ARC11-716 1A14 ON+112+312N25

## DEPENDENT VARIABLE CF

## SECTION ( 1 ) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7830	.8870
X/CW							
.000	-.1987	-.1287	.2021	.3944	.3680	.3089	.3046
.020				-.1338	-.2339	-.3456	-.4424
.025				-.0680			
.040		-.0144					
.045			-.1895				
.050	-.1406			-.3715	-.4804	-.5587	-.6602
.080				-.5031			
.085			-.3079				
.086		-.0149					
.094	-.1070						
.130				-.5998	-.6712	-.6955	-.6800
.163		.0000					
.177			-.4354				
.193		-.2217					
.229	-.0401			-.5876	-.6709	-.7339	-.7535
.250							
.274		-.4846					
.339				-.4216	-.4801		-.4895
.362	-.1576		-.4103				
.400				-.2824	-.2920		
.402							
.497	.0000						
.590							
.585							
.600							
.650							
.700	-.1630			-.1088		-.1994	-.2853
.725							
.750						.0011	-.0266
.760							
.775				-.0030	.0763		
.806							
.834	-.0904			-.0160			
.850							
.857				-.0067			
					.0221	.0538	.0707
.865	-.0280						
.900	-.0126				.0484		.0829
.908				.0046		.0899	
.950					.0552		.1013
.953				.0090			
.965	-.0327						

(R01U41)

UPPER WING

ARC11-716 1A14 01+T12+S12N23

$$\text{ALPHA}(4) = 4.110 \quad \text{BETA}(5) = 8.150$$

SECTION (1) UPPER WING

1984	.2990	.5640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

MS/X

.000	-.3127	-.1559	.0440	.2958	.3112	.2656	.2735
.020				-.0728	-.1764	-.2459	-.3383
.025			-.0925				
.040		-.0683					
.045			-.1640				
.050	-.1895			-.2846	-.3833	-.4494	-.5682
.080				-.4124			
.085			-.2493				
.086		-.0642					
.094	-.1887						
.150				-.5160	-.5765	-.6242	-.6176
.163		.0000					
.177			-.3597				
.193		-.1930					
.229	-.0391						
.250				-.2354	-.6128	-.6358	-.6759
.274			-.4066				
.339		-.3734					
.362	-.1337			-.4068	-.4587		-.4566
.400			-.3870				
.402							
.497	.0000			-.2727	-.2969		
.550			.0000			-.2000	-.2474
.565					-.1564		
.600							
.650	-.1922			-.1900		.0024	-.0104
.700							
.725			-.1452		-.0074		
.750				-.0399			
.760			-.0318				
.775							
.808							
.834	-.1187			-.0047	.0421	.0682	
.850							
.857			-.0387				
.865	-.0608						.0931
.900	-.0470			.0202	.0735		
.905			-.0175				
.950				.0258		.0981	
.955			-.0117				
.965	-.0311						

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U41)

UPPER WING

ARC11-716 1A14 01-112-912N25

ALPHA(1) = 7.990 BETA(1) = -8.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.000	-.0715	.2378
.020	-.1033	.4497
.025		.2978
.040		.0971
.045		-.0223
.050		-.7170
.060		-1.0190
.065		-1.3460
.080		-1.6530
.085		
.094		-.5692
.1062		-.9889
.150		-1.2810
.163		-1.5010
.177		-1.6860
.193		-1.1480
.229		-.7666
.274		
.339		-.1357
.362		-.1062
.400		-.162
.402		-.190
.497		-.0000
.550		-.7800
.565		
.600		-.6391
.650		-.5733
.700		-.2873
.725		-.4045
.750		-.4298
.775		-.5515
.806		-.7202
.834		
.850		-.1767
.857		-.2895
.865		.0000
.900		-.4002
.905		-.2036
.930		-.1379
.935		-.0167
.955		-.0138
		-.1603
		-.0151
		.0835
		.0467
		.0643
		.0920
		.0840
		.0575
		.0806
		.1014
		.1108
		.0674
		.0979
		.0831
		.0600
		.0506



## TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1U41)

UPPER WING

DATE 10 JAN 75

ARC11-716 IA14 01+112+S12N25

ALPHA( 5 ) = 0.000 BETA( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/8W .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1280	-.1302	.2219	.4444	.3183	.1057	-.0267
.020				-.5983	-.0784	-1.1800	-1.4750
.025			-.3351				
.040		-.0738					
.045			-.4766				
.050	-.1350			-.8485	-1.1180	-1.3260	-1.5360
.080				-1.0060			
.085			-.6630				
.096		-.1111					
.100	-.1242						
.150		.0000		-.9575	-1.1370	-1.2210	-1.1740
.163			-.7196				
.177		-.4253					
.193							
.229	-.0721			-.7843	-.9161	-1.0870	-1.1470
.250			-.6176				
.274		-.5503					
.339	-.2720			-.4498	-.5519		-.6675
.362			-.4344				
.400				-.2109	-.2858		
.402	.0000		.0000				-.3781
.497						-.2070	
.550				-.0446	-.1243		-.1193
.565							
.600							
.650	-.1128						
.700							
.725							
.750							
.760			-.0424	.0550	.0368		
.775			.0467				
.808							
.834	-.0066		.0474	.0720	.0784	.0836	
.850							
.857							
.865	.0662			.0905	.1095		.0192
.900	.0762		.0571				
.905				.0937		.0929	
.950			.0566				
.953							
.965	.0338						

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U41)

UPPER WING

ALPHA( 5 ) = 0.000 BETA( 3 ) = .010

ARC11-716 1A14 01+112+512N25

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1959	-.1734	.1876	.4206	.3091	.1199	-.0128
.025				-.5938	-.7515	-1.0280	-1.3120
.025			-.2774				
.040		-.0735					
.045			-.4077				
.050	-.1739			-.7478	-.9865	-1.1820	-1.3930
.380				-.8750			
.085			-.5693				
.086		-.0958					
.094	-.1321						
.150				-.8685	-1.0360	-1.1150	-1.0610
.163		.0000					
.177		-.3778					
.193							
.229	-.0680			-.7448	-.9105	-1.0190	-1.0390
.250							
.274			-.5896				
.339		-.5263					
.362	-.2509			-.4578	-.5474		-.6280
.400			-.4488				
.402							
.497	.0000			-.2426	-.3006		-.3540
.530			.0000			-.2104	
.565							
.620							
.650							
.700	-.1308			-.0759	-.1193		
.725						-.0065	-.0915
.750			-.0747				
.760				.0290	.0186		
.775			.0230				
.808							
.834	-.0448			.0308	.0675	.0624	
.850			.0340				
.857							
.865	.0280						
.900	.0418			.0689	.0999	.0404	
.915			.0452				
.950				.0808		.0941	
.953			.0414				
.965	.0195						



(RB1041)

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

UPPER WING

ARC11-716 1A14 01+712+S12N25

ALPHA( 5 ) = 7.870 BETAD ( 4 ) = 0.160

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/W .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.4370	-.2634	-.1192	.2356	.2941	.1666	.0583
.020				-.3383	-.4965	-.6994	-.9459
.025			-.2616				
.040		-.1315					
.045			-.3040				
.050	-.2557			-.5109	-.6837	-.8577	-1.0550
.080				-.6115			
.085			-.3616				
.086		-.1324					
.094	-.2673			-.6340	-.7868	-.8609	-.8653
.150		.0000					
.163			-.4446				
.177		-.2714					
.193							
.229	-.1034			-.6172	-.7442	-.8297	-.8729
.250			-.4463				
.274		-.4308					
.339							
.362	-.1855			-.4303	-.5126		-.5144
.400			-.4043				
.42							
.497	.0000			-.2736	-.3070		-.2907
.550			.0000			-.2034	
.565							
.600							
.650							
.700	-.1837			-.1236	-.1598		
.725						-.0066	-.0446
.750							
.760			-.1437	-.0450	-.0066		
.775			-.0666				
.808							
.834	-.1157			-.0149	.0394	.0632	
.850			-.0515				
.857							
.885	-.0610						.0704
.900	-.0463		.0111	.0809			
.905			-.0296	.0197		.0953	
.950							
.955			-.0131				
.965	-.0486						

ARC11-716 1A14 01+T12+S12N25

**UPPER WING**

(R01U42) ( 16 FEB 74 )

## REFERENCE DATA

9REF =	2.4210 SQ.FT.	YARP =	29.9000 INCHES
1REF =	30.7090 INCHES	YARP =	.0000 INCHES
8REF =	30.7090 INCHES	ZARP =	.0000 INCHES
SCALE =	.0000 SCALE		

ALFMAO( 1 ) = -7.099      BETAO ( 1 ) = -8.999

## SECTION (1) UPPER WING

## DEPENDENT VARIABLE CP

Y/8W	.2993	.3640	.4270	.5340	.6730	.7800	.8670
X/CW							
.000	.0319	.0219	.2112	.2041	.0809	-.0302	.0477
.020				.9025	.4852	.5013	.4683
.025			.3760				
.040		.1977					
.045			.3182				
.050	.0826			.2861	.3110	.3166	.2721
.060				.1234			
.085			.1773				
.086		.2606					
.094	.0885						
.150		.0000		-.1064	-.0734	-.0326	-.0760
.163							
.177			-.0402				
.193		.1283					
.229	.1439						
.250				-.2104	-.2503	-.2829	-.3283
.274			-.1367				
.339		-.1122					
.362	.1098			-.1599	-.2468		-.2909
.400			-.1492				
.402	.0000			-.0873	-.1609		
.497			.0200				-.1996
.550						-.1379	
.563							
.600							
.650	.0584						
.700				-.0274	-.1480		
.725							
.750							
.760			-.0398			.0829	.0273
.775			.1050	.1098	.0868		
.806							
.834	.0283						
.850							
.857			.0801	.0883	.1008	.1112	
.900	.1148						
.900	.0880			.0935			.0786
.905			.0682		.1130		

DATE 10 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

PAGE 3041

(R81042)

UPPER WING

ALPHA(1) = -7.890 BETA(1) = -8.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.950			.0752		.1089
.953		.0360			
.965	.0045				

ALPHA(2) = -7.780 BETA(2) = -3.990

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0090	-.0356	.1217	.0803	-.0520	-.1977	-.1071
.020				.4806	.4430	.4584	.4138
.025		.1478	.3538				
.040			.3113				
.045				.2954	.3078	.3065	.2526
.050	.0481			.1437			
.080		.2008					
.085							
.086		.2291					
.094	.0550			-.0713	-.0566	-.0326	-.0677
.150		.0000					
.163			-.0133				
.177		.1370					
.193							
.229	.1231			-.1988	-.2302	-.2611	-.3043
.250			-.1188				
.274		-.1031					
.339							
.362	.1107			-.1786	-.2563		-.3061
.400			-.1675				
.402							
.497	.0000			-.1252	-.1855		
.550			.0000				
.565							
.600							
.650							
.700	.0032			-.0644	-.1488		-.2154
.725							
.750							
.760							
.775			-.0783				
.775				.0738	.0595		
.808			.0710				
.834	-.0149						
.850				.0590	.0797	.0931	



(RB1U42)

UPPER WING

ARC11-716 1A14 01-112-512N25

ALPHA( 1 ) = -7.760 BETA( 2 ) = -3.990

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.857 .0510

.865 .0731

.900 .0475

.905 .0695

.930 .0425

.935 .0918

.965 .0578

.965 .0170

ALPHA( 1 ) = -7.760 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.000 -.0279

.020 -.0311

.025 .0624

.040 .0386

.045 .4507

.090 .3167

.095 .1049

.095 .2876

.094 .2903

.150 .2889

.163 .2001

.177 .1536

.193 .2001

.229 .1993

.250 .0000

.274 .0110

.339 .1420

.362 .1048

.400 .274

.402 .1089

.497 .0927

.550 .1122

.565 .362

.600 .400

.650 .402

.700 .497

.725 .550

.750 .565

.760 .600

.887 .650

.900 .700

.905 .725

.930 .750

.935 .760



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81042)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA(1) = -7.780 BETA(3) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.775 .0355 .0352  
 .808 .0344  
 .834 -.0819  
 .850 .0333 .0601 .0775  
 .857 .0167  
 .865 .0301  
 .900 .0120 .0490 .0765 .0558  
 .905 .0195  
 .930 .0417 .0827  
 .953 .0304  
 .965 -.0413

ALPHA(1) = -7.790 BETA(4) = 4.090

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.0367 -.0331 .0248 .1359 -.3293 -.4662 -.3551  
 .020 .4014 .3231 .3321 .2777  
 .025 .2742  
 .040 .0795  
 .045 .2584  
 .050 -.0049 .2714 .2606 .2394 .1768  
 .080 .1526  
 .085 .1878  
 .086 .1710  
 .094 .0083  
 .150 .0000  
 .163 .0216  
 .177 .1378  
 .193 .0928  
 .229 .1917 .2165 .2384 .2957  
 .250 .1005  
 .274 .1005  
 .339 -.0883  
 .382 .1127  
 .400 .2235 .2724 .3263  
 .402 .2150  
 .497 .0000  
 .550 .1968 .2413  
 .565 .0000  
 .600 .0000  
 .650 -.2410  
 .650 -.2058

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U42)

UPPER WING

ARC11-716 IA14 01+T12+S12N25

ALPHA( 1 ) = -7.790 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.700	-.0972			-.1390		.0146	-.0139
.725							
.750							
.760				-.1325	.0046	.0039	
.775				.0036			
.806							
.834		-.1016			.0151	.0408	.0571
.850				-.0034			
.857							
.865	-.0061						
.900	-.0213			.0263		.0591	.0458
.925				.0035		.0169	.0772
.950							
.953				-.0100			
.965	-.0572						

ALPHA( 1 ) = -7.970 BETA( 5 ) = 8.160

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0638	-.0409	.0157	-.1989	-.4455	-.5849	-.4727
.020				.3591	.2641	.2656	.2023
.025			.2318				
.040		.0322					
.045			.2245				
.050	-.0307			.2508	.2243	.2061	.1357
.080				.1472			
.085			.1733				
.086		.1409					
.094	-.0164						
.150		.0000		-.0408	-.0433	-.0386	-.1120
.163							
.177			.0228				
.193		.1231					
.229	.0719						
.250				-.1854	-.2120	-.2332	-.2934
.254			-.1014				
.339		-.0923					
.362	.0936						
.400				-.2380	-.2885		-.3215
.402			-.2291				
.497	.0000						



DATE 13 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81042)

UPPER WING

ARC11-716 IA14 Q1+T12+S12N25

ALPHA( 1 ) = -7.970 BETA( 5 ) = 0.160

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.590						
.565						
.600						
.650						
.700						
.725						
.750						
.780						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.935						
.950						
.953						
.965						

ALPHA( 2 ) = -4.020 BETA( 1 ) = -0.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000						
.020						
.025						
.040						
.045						
.050						
.080						
.085						
.086						
.094						
.150						
.163						
.177						
.193						
.229						
.250						
.274						
.339						

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881042)

UPPER WING

ARC11-716 1A14 01+T12+S12M25

ALPHA( 2 ) = -4.020 BETA( 1 ) = -0.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.362	.0040	
.400		
.402		
.497	.0050	
.550		
.565		
.600		
.650		
.700	.0088	
.725		
.750		
.760		
.775		
.808		
.834	.0183	
.850		
.857		
.865	.1096	
.900	.0790	
.905		
.950		
.953		
.965	.0030	

ALPHA( 2 ) = -4.030 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.000	.0129	
.020		
.025		
.040	.1578	
.045		
.050	.0120	
.060		
.065		
.066	.1863	
.074	.0131	
.150		
.163		
.177		
.193		



(R81U42)

UPPER WING

ALFMAO( 2) =	-4.030	BETAO ( 2) =	-4.030
--------------	--------	--------------	--------

SECTION (1) UPPER WING	DEPENDENT VARIABLE CF
1	1
2	2
3	3
4	4
5	5
6	6
7	7
8	8
9	9
10	10
11	11
12	12
13	13
14	14
15	15
16	16
17	17
18	18
19	19
20	20
21	21
22	22
23	23
24	24
25	25
26	26
27	27
28	28
29	29
30	30
31	31
32	32
33	33
34	34
35	35
36	36
37	37
38	38
39	39
40	40
41	41
42	42
43	43
44	44
45	45
46	46
47	47
48	48
49	49
50	50
51	51
52	52
53	53
54	54
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57	57
58	58
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60	60
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62	62
63	63
64	64
65	65
66	66
67	67
68	68
69	69
70	70
71	71
72	72
73	73
74	74
75	75
76	76
77	77
78	78
79	79
80	80
81	81
82	82
83	83
84	84
85	85
86	86
87	87
88	88
89	89
90	90
91	91
92	92
93	93
94	94
95	95
96	96
97	97
98	98
99	99
100	100

1984	.2990	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

**MJ/X**

.229	.0782
.250	-.3788    -.4334    -.4730    -.5161

1.235	-	.2701
.274	-	.274

**.339** **- .2443**

.362	.0125		
.400		-.2864	-.3659
			-.3957

**.452** **-.2765**

.497	.00000	
.590		-.1711
		-.2382

.333	18721	16395
.565		
.0000		- 2443

.699		- .2443
.699		- .1859

.700	-.0300	-.1547
.725		-.0766

.750	.0459	.0006
.760	-.0012	

.705	.0012	
.775	.0631	.0523
.608	.0570	

.608	.0579
.634	-.0259
	.0027

.859	.0658	.0796	.0903
.857		.0475	

.063	.0635	.0697	.0853
.909	.0425		

1992	.905	.0496	.0871
1993	.950	.0613	.0921

.957		.0673	.0324
.953		.0264	
.952			.0187

**.965    -.0123**

PHAC( 2 ) = -4.040    BETAD ( 3 ) = .040

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
1	1
2	2
3	3
4	4
5	5
6	6
7	7
8	8
9	9
10	10
11	11
12	12
13	13
14	14
15	15
16	16
17	17
18	18
19	19
20	20
21	21
22	22
23	23
24	24
25	25
26	26
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28	28
29	29
30	30
31	31
32	32
33	33
34	34
35	35
36	36
37	37
38	38
39	39
40	40
41	41
42	42
43	43
44	44
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62	62
63	63
64	64
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66	66
67	67
68	68
69	69
70	70
71	71
72	72
73	73
74	74
75	75
76	76
77	77
78	78
79	79
80	80
81	81
82	82
83	83
84	84
85	85
86	86
87	87
88	88
89	89
90	90
91	91
92	92
93	93
94	94
95	95
96	96
97	97
98	98
99	99
100	100

54	.2990	.3640	.4270	.5340	.6730	.7600	.8670
55							

NO/x

.003	-.0234	.0484	.2501	.2269	.1089	-.0004	.0626
.020				.3249	.2816	.3017	.2689

.025	.2392
.040	.240

0.040	1.1200		
0.045	.1722		
0.050		1.084	0.908
0.055		1.044	0.812

.030	-.0150	.1051	.1264	.0998	.0412
.060		-.0476			

.005	.0415
.006	.1627

**.994    -.0062**

1. The first step in the process of identifying a problem is to recognize that a problem exists. This is often done by comparing current performance with a desired state or goal. If there is a significant difference, a problem is identified.

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(R81U42)

UPPER WING

ARC11-716 1A14 01+712+312425

ALPHA( 2 ) = -4.040 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.190						
.163	.0000					
.177						
.193	.0225					
.229	.0659					
.250						
.274						
.339						
.362	.0191					
.400						
.452						
.497	.0300					
.530						
.565						
.600						
.650						
.700	.0842					
.725						
.750						
.780						
.775						
.808						
.834	.0723					
.850						
.857						
.865	.0184					
.900	.0285					
.905						
.930						
.933						
.965	.0317					

ALPHA( 2 ) = -4.040 BETA( 4 ) = 4.070

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0648	.0190	.2002	.1405	.0053	.1179	.0426
.050				.2976	.2374	.2590	.2226
.025			.2004				
.040		.0916					
.845			.1515	.1065	.0940	.0845	.0176
.050	.0440						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1U42)

UPPER WING

ARC11-716 1A14 01+112+S12N25

ALPHA( 2) = -4.040 BETA( 4) = 4.070

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8670

X/CW

.080 .085 .086 .094 .150 .163 .177 .193 .229 .250 .274 .339 .362 .400 .402 .497 .550 .565 .600 .650 .700 .725 .750 .760 .775 .808 .834 .850 .857 .865 .900 .905 .950 .953 .965

-.0273

.0483

.1332

-.0250

.0500

.0286

.0506

-.1330

.0286

.0506

-.1330

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-.1330

.0286

.0506



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R51U42)

UPPER WING

ARC11-716 1A14 01+712+512N25

ALPHA( 2 ) = -4.040 BETA( 5 ) = 0.130

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

V/BW	.2990	.3640	.4270	.5340	.6750	.7600	.8670
X/CW							
.000	-.1206	-.0250	.1049	.0884	-.1039	-.2240	-.1400
.020				.2654	.1990	.2105	.1713
.025			.1879				
.040		.0255					
.045			.1449				
.050	-.0772			.0986	.0858	.0702	.0072
.060				-.0234			
.065			.0477				
.066		.1054					
.094	-.0499						
.150				-.1953	-.2101	-.2157	-.2654
.163		.0000					
.177			-.1150				
.193		.0285					
.229	.0320						
.250				-.3198	-.3717	-.3976	-.4423
.274		-.2121					
.339		-.2020					
.362	.0161						
.400			-.3166	-.3329	-.3902		-.3983
.402							
.497	.0200			-.2650	-.3086		
.550		.0000					
.565							
.600							
.630							
.690							
.700	-.1977						
.725			-.1877	-.1809		-.2333	-.2552
.750							
.760		-.1860				-.0058	-.0213
.775				-.0382	-.0188		
.800			-.0481				
.834	-.1432						
.850				-.0174	.0218	.0514	
.857			-.0393				
.865	-.0557						
.900	-.0564			.0070			.0495
.905			-.0282		.0573		
.930				.0079			.0702
.935			-.0319				
.965	-.0604						



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(R81U42)

UPPER WING

ARC11-716 1A14 01+112+S12M23

ALPHA(3) = -.320 BETA(1) = -.060

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7000	.8870
X/CW							
.000	.0398	.1145	.4223	.5616	.5364	.4968	.5202
.020				.0895	.0487	.0210	-.0044
.025			.1167				
.040		.1452					
.045			-.0154				
.050	-.0097			-.2026	-.2527	-.2758	-.3412
.060				-.4110			
.065			-.2124				
.086		.1221					
.094	-.0061						
.150		.0000		-.5806	-.6298	-.6386	-.6007
.163							
.177			-.4305				
.193		-.1488					
.229	.0473			-.5959	-.7049	-.7959	-.8269
.250							
.274		-.4489					
.339		-.3954					
.362	-.0899			-.3461	-.4388		-.4592
.400			-.3453				
.402							
.497	.0000			-.1555	-.2316		
.550			.0000				
.565							-.2768
.600					-.1678		
.650				-.0329	-.1526		
.700	-.0128					.0642	-.0066
.725							
.750			-.0291	.1003	.0785		
.760							
.775		.0982					
.808							
.834	.0217			.0917	.1051	.1123	
.850			.0822				
.857							
.865	.1158						.1006
.900	.0955			.0591	.1151		
.905		.0746		.0855		.1146	
.950							
.955		.0512					
.965	.0263						

(R91042)

UPPER WING

ALPHA( 3 ) = -.343 BETA( 2 ) = -4.010

ARC11-716 IA14 Q1-T12-S12N23

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8670
X/CW							
.000	-.0111	.0741	.3690	.4846	.4463	.3986	.4231
.020				.1379	.0806	.0621	.0262
.025			.1273				
.040		.1237					
.045			.0222				
.050	-.0370			-.1452	-.1955	-.2152	-.2830
.080				-.3339			
.085			-.1637				
.086		.1169					
.094	-.0260			-.5016	-.5522	-.5654	-.5517
.193		.0000					
.163			-.3707				
.177		-.1180					
.193							
.229	.0332			-.5491	-.6455	-.7330	-.7624
.250			-.4189				
.274		-.3724					
.339							
.362	-.0741			-.3619	-.4607		-.4654
.400			-.3551				
.497	.0000			-.1993	-.2645		
.550			.0000				
.565							
.600							
.630							
.700	-.0462			-.1540			
.725				-.0744			
.750			-.0713			.0480	-.0129
.760				.0815	.0487		
.775			.0561				
.808							
.834	-.0262			.0629	.0603	.0930	
.850			.0502				
.857							
.865	.0600						
.900	.0526			.0763	.0922		.0942
.905			.0493				
.910				.0879		.1010	
.933			.0370				
.965	.0003						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U42)

UPPER WING

ARC11-716 1A14 Q1+T12+S12N25

ALPHA ( 3 ) = -.340 BETA ( 3 ) = .050

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0628	.0342	.3180	.4023	.3326	.3036	.3345
.020				.1524	.0916	.0733	.0317
.025			.1286				
.040		.0986					
.045			.0344				
.050	-.0635			-.1080	-.1355	-.1850	-.2562
.080				-.2735			
.085			-.1285				
.086		.1037					
.094	-.0474			-.4452	-.4779	-.5034	-.5104
.150		.0000					
.163			-.3226				
.177		-.0934					
.193							
.229	.0217			-.5106	-.5929	-.6659	-.7074
.274		-.3809					
.339		-.3445					
.362	-.0639			-.3837	-.4659		-.4694
.400			-.3753				
.402				-.2446	-.2933		
.497	.0000		.0000				
.550							
.600							
.650							
.700	-.1219			-.1316		-.2087	
.725			-.1119				
.750						.0293	-.0139
.760			-.1150				
.775				.0251	.0244		
.808			.0162				
.834	-.0727			.0323	.0596	.0790	
.850			.0134				
.857							
.865	.0035						
.900	.0004			.0496			.0084
.905			.0194		.0784		
.950				.0524		.0859	
.953			.0129				
.965	-.0279						

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(R81U42)

UPPER WING

ARC11-716 1A14 01+712+512N25

ALPHA( 3 ) = -.350 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000	-.1259	-.0191	.2544	.3205	.2519	.2039	.2427
.020				.1419	.0764	.0662	.0213
.025			.1091				
.040		.0631					
.045			.0250				
.050	-.0670			-.0847	-.1175	-.1617	-.2475
.060				-.2325			
.085			-.1006				
.086		.0602					
.094	-.0614						
.150		.0000		-.4011	-.4431	-.4607	-.4920
.163			-.2814				
.177		-.0798					
.193							
.229	.0124						
.250				-.4854	-.5654	-.6234	-.6668
.274			-.3516				
.339		-.3193					
.362	-.0575						
.400				-.4021	-.4825		-.4637
.402			-.3692				
.497	.0000			-.2691	-.3187		
.550			.0000				
.565							
.600							
.650							
.700	-.1604						
.725				-.1719			
.750				-.1905			
.760			-.1555			.0048	-.0204
.775				-.0142	-.0036		
.800			-.0261				
.834	-.1233						
.850				.0283	.0368	.0388	
.857			-.0213				
.865	-.0486						
.905	-.0405			.0266			.0837
.905			-.0106		.0657		
.950				.0307		.0758	
.955			-.0140				
.965	-.0619						





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(RB1U42)

UPPER WING

ARC11-716 1A14 01+712+S12N23

ALPMAO( 4) = 4.220 BETAO ( 1) = -8.060

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2090	.3640	.4870 .5340 .6730 .7800 .8870
.0000	.0281	.3838 .5786 -.2355 .0368 .2896
.0200		-.2520 -.2784 -.1140 -.0385
.0250		-.1204
.0400	.0543	
.0450		-.2661
.0500	-.0693	-.5685 -.3381 -.2033 -.0411
.0600		-.8011
.0650		-.4861
.0800	.0166	
.0900	-.0325	-.9246 -.2429 -.2027 .0108
.1100		.0000
.1600		-.7027
.1700	-.3175	
.1900		-.2640 -.1513 -.1756 .3394
.2200	.0015	
.2300		-.6548
.2700	-.5717	
.3300		-.3198 -.1545 .3540
.3600	-.1962	
.4000		-.4089
.4200	.0000	
.4900		.9757 -.2478
.5500		.0000
.5600		-.0977
.6000		-.1105
.6500		-.3362
.7000	-.0480	
.7200		.7872
.7500		-.0030
.7600		-.0406 -.1201
.7700		.9495 -.3815
.8000	.0963	
.8300	.0418	.6452 -.3882 .0970
.8500		.0942
.8570		
.8650	.1370	
.9000	.1206	.5153 -.3279 -.2843
.9050		.0918
.9300		.3966 .1912
.9350		.0685
.9650	.0478	



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TABULATED PRESSURE DATA - IA14A - VOL. 6

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(R81U42)

UPPER WING

ARC11-716 IA14 OL+T12+S12N25

ALPHA( 4 ) = 4.210 BETAO ( 2 ) = -4.030

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/W .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0624	.0076	.3606	.9399	.4972	.4191	.3792
.020				-.1810	-.3192	-.4525	-.5984
.025			-.0816				
.040		.0921					
.045			-.2084				
.050	-.0921			-.4830	-.6285	-.7502	-.9248
.060				-.7010			
.065			-.4039				
.086		.0216					
.094	-.0736						
.150		.0900		-.6410	-.9700	-1.0560	-.9875
.163							
.177			-.6208				
.193		-.2753					
.229	-.0110						
.250				-.8177	-1.0060	-1.1490	-1.1430
.274			-.6102				
.339		-.5406					
.362	-.1754						
.400				-.4363	-.5150		-.5063
.402			-.4396				
.497	.0000						
.550				-.2015	-.2730		
.565			.0000				
.600							-.3200
.650					-.1421		
.700	-.0971			-.0499			
.725						.0411	-.0407
.750							
.760			-.0457				
.775				.0709	.0629		
.808			.0662				
.834	-.0085						
.850				.0733	.0933	.1033	
.857			.0624				
.865	.0785						.0794
.900	.0735			.0665			
.905			.0617		.1158		
.950				.0665		.1129	
.955			.0499				
.965	.0246						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01042)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 4 ) = 4.210 BETAD ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1213	-.0449	.3190	.4824	.4454	.3706	.3330
.020				-.1330	-.2640	-.3880	-.5334
.025		.0374					
.040							
.045							
.050	-.1129		-.1697	-.4061	-.5321	-.6331	-.8316
.060				-.5930			
.065							
.066							
.086		.0212					
.094	-.0839						
.130				-.7253	-.8355	-.8943	-.9081
.163		.0000					
.177							
.193		-.2309					
.229	-.0162						
.250				-.7187	-.8829	-1.0230	-1.0160
.274			-.5544				
.339		-.4896					
.362	-.1538			-.4617	-.5351		-.5259
.400							
.402			-.4639				
.497	.0000						
.550				-.2542	-.3016		
.565							
.600							
.650							
.700	-.1489						
.725				-.0899	-.1462		
.750							
.760							
.775							
.808				-.0861	.0367	.0382	
.834	-.0563						
.850							
.857							
.865							
.900	.0240			.0311	.0756	.0910	
.920	.0272						
.905				.0659			.0874
.950				.0342	.0975		
.953				.0673			.1076
.965				.0312			
.965	.0001						



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U42)

UPPER WING

ARC11-716 1A14 ON+T12+S12N25

ALPHA( 4) = 4.200 BETA( 4) = 4.090

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/BW .2390 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2121	-.1276	.2331	.4138	.3796	.3177	.2940
.020			-.1011	-.2304	-.3375	-.4787	
.025			-.0369				
.040		-.00374					
.045			-.1593				
.050	-.1319		-.3551	-.4791	-.5820	-.7615	
.080			-.5155				
.085			-.2956				
.086		.0005					
.094	-.1071		-.6337	-.7537	-.8045	-.8322	
.150		.0000					
.163			-.4611				
.177		-.2025					
.193							
.229	-.0265		-.6610	-.7835	-.9244	-.9321	
.250							
.274		-.5011					
.339		-.4415					
.362	-.1360						
.400			-.4810	-.5585		-.5224	
.402			-.4680				
.497	.0000		-.2933	-.3288			
.550			.0000				
.565							-.2940
.600							
.650							
.700	-.1972		-.1279	-.1601			
.725							
.750							
.760			-.1136	.0017	.0172		
.775			-.0180				
.808							
.834	-.1124						
.850							
.857			-.0147	.0209	.0609	.0809	
.865	-.0281						
.900	-.0169		.0421	.0890		.0906	
.905			-.0005	.0472		.1038	
.950			.0033				
.953							
.965	-.0331						

(R81U42)

UPPER WING

ARC11-716 1A14 01-112-512M25

ALPHA( 4) = 4.190 BETA( 5) = 8.183

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000	-.3258	-.1371	.0294	.2787	.3268	.2704	.2659
.020			-.0721	-.1622	-.2498	-.3837	
.025		-.0696					
.040			-.1432				
.045				-.2852	-.3667	-.4692	-.6414
.050	-.1959			-.4019			
.080			-.2290				
.085							
.086		-.0531					
.094	-.2218						
.150				-.5262	-.6267	-.6935	-.7354
.163		.0000					
.177			-.3596				
.193		-.1707					
.229	-.0179			-.5738	-.6994	-.8077	-.8363
.250							
.274			-.4156				
.339		-.3876					
.362	-.1157			-.4581	-.5504		-.5069
.400			-.4309				
.402				-.3120	-.3417		
.497	.0000						
.550			.0000				
.565							
.600							
.650							
.700	-.2337			-.1900			
.725			-.1726			.0178	-.0100
.750							
.760			-.0490	-.0032			
.775			-.0638				
.808							
.834	-.1431			-.0154	.0363	.0733	
.850							
.857			-.0502				
.885	-.0784						
.900	-.0592			.0077	.0709		.0989
.905			-.0253				
.930				.0187		.0938	
.935			-.0183				
.965	-.0587						





(R01U42)

## UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALFMAO( 3) = 0.000 BETAO ( 2) = -3.990

DEPENDENT VARIABLE CP

SECTION 11 UPPER WING

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.1262	-.0959	.2971	.9214	.4061	.2332	.1303	
.020				-.4876	-.6807	-.9815	-1.1760	
.025				-.2357				
.040	-.0422			-.4181				
.045								
.050	-.1442				-.7780	-1.0200	-1.1970	-1.4640
.060				-1.0200				
.085				-.6335				
.086				-.0727				
.094	-.1162							
.150					-.1190	-1.4460	-1.5050	-1.4890
.163			.0000					
.177			-.8236					
.193			-.4034					
.229	-.0457							
.250					-.9826	-1.4390	-1.4360	-1.3590
.274				-.7754				
.339			-.7213					
.362	-.2367			-.4900				
.400					-.4612	-.4825		-.6946
.497	.0000							
.550					-.1965	-.2796		
.565				.0005				
.600								-.3706
.650							-.2090	
.700	-.1176				-.1370			
.723					-.0350			
.750				-.0362			-.0316	-.1761
.760								
.775				.0591		.0694	.0414	
.806								
.834	-.0103							
.850								
.857					.0619	.0665	.0374	
.865				.0626				
.900	.0609							-.0332
.920	.0667				.0905	.1112		
.953				.0661				
.950				.0919			.0722	
.953				.6560				
.965	.0439							

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81U42)

UPPER WING

ARC11-716 1A14 01-712-512N25

ALPHA( 5 ) = 7.970 BETA( 3 ) = .050

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
Y/BW	.2990 .3640 .4270 .5340 .6730 .7800 .8870
X/CW	.000 .020 .025 .040 .045 .050 .060 .065 .086 .094 .150 .163 .177 .193 .229 .274 .339 .362 .400 .402 .497 .550 .565 .600 .650 .700 .725 .750 .760 .775 .808 .834 .850 .857 .865 .900 .905 .930 .953 .965
	-.1985 -.1629 .2431 .4686 .3846 .2229 .1049 -.3894 -.6032 -.8845 -1.1220 -.2156 -.0301 -.3483 -.6794 -.9171 -1.1150 -1.3920 -.8815 -.5386 -.0650 -.1271 .0000 -.7264 -.3322 -.0510 -.7155 -.6302 -.2259 -.5024 -.5319 -.5903 -.2393 -.2894 .0000 -.1351 -.0666 -.0682 .0361 .0361 .0406 -.0543 .0501 .0500 .0508 .0726 .0763 .0443 .0061 -.1037 .0328 .0761 .0699 .1064 .0422 .0976

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(R81U42)

## UPPER WING

ALPHA( 3 ) = 7.969    BETA( 4 ) = 4.130

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP
1	0.0000
2	0.0000
3	0.0000
4	0.0000
5	0.0000
6	0.0000
7	0.0000
8	0.0000
9	0.0000
10	0.0000
11	0.0000
12	0.0000
13	0.0000
14	0.0000
15	0.0000
16	0.0000
17	0.0000
18	0.0000
19	0.0000
20	0.0000
21	0.0000
22	0.0000
23	0.0000
24	0.0000
25	0.0000
26	0.0000
27	0.0000
28	0.0000
29	0.0000
30	0.0000
31	0.0000
32	0.0000
33	0.0000
34	0.0000
35	0.0000
36	0.0000
37	0.0000
38	0.0000
39	0.0000
40	0.0000
41	0.0000
42	0.0000
43	0.0000
44	0.0000
45	0.0000
46	0.0000
47	0.0000
48	0.0000
49	0.0000
50	0.0000
51	0.0000
52	0.0000
53	0.0000
54	0.0000
55	0.0000
56	0.0000
57	0.0000
58	0.0000
59	0.0000
60	0.0000
61	0.0000
62	0.0000
63	0.0000
64	0.0000
65	0.0000
66	0.0000
67	0.0000
68	0.0000
69	0.0000
70	0.0000
71	0.0000
72	0.0000
73	0.0000
74	0.0000
75	0.0000
76	0.0000
77	0.0000
78	0.0000
79	0.0000
80	0.0000
81	0.0000
82	0.0000
83	0.0000
84	0.0000
85	0.0000
86	0.0000
87	0.0000
88	0.0000
89	0.0000
90	0.0000
91	0.0000
92	0.0000
93	0.0000
94	0.0000
95	0.0000
96	0.0000
97	0.0000
98	0.0000
99	0.0000
100	0.0000

Y/YW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3118	-.2859	.1382	.3985	.3526	.2125	.0943
.020				-.3377	-.5326	-.7897	-1.0630
.025							
.040				-.0864			
.045							
.050	-.1878			-.3195			
.060					-.5960	-.8144	-1.0150
.085					-.7752		
.086				-.4810			
.094				-.0848			
.150	-.1459				-.8469	-1.0900	-1.2090
.163				.0000			
.177							
.193				-.6519			
.229	-.0762			-.3089			
.250					-.8431	-1.0710	-1.2590
.274				-.6280			
.339				-.5445			
.362	-.2124				-.5150	-.5728	-.5197
.400							
.497	.0000			-.5234			
.550					-.2784	-.5100	
.565				.0000			
.600							
.650							
.700	-.2115				-.1426	-.2071	-.3312
.725				-.1024			
.750				-.1026			
.760					.0040	.0241	
.775				-.0122			
.808							
.834	-.0921						
.850					.0328	.0654	.0801
.857				-.0042			
.865	-.0020						
.900	.0103				.0512	.0994	.0648
.905				-.0021			
.950					.0695		.1040
.955				.0063			
.965	-.0008						





## UPPER WING

### PARAMETRIC DATA

```

WACH = .000 ELEVON = .000
RUDDER = .000 SPCBRK = .000

```

## REFERENCE DATA

SREF = 2.4210 INCHES YARP = 29.5000 INCHES  
 LREF = 30.7000 INCHES YARP = .0000 INCHES  
 BREF = 30.7000 INCHES ZARP = .0000 INCHES  
 SCALE = .0000 SCALE

ALPHA( 1 ) = -7.095      BETA( 1 ) = -0.025

SECTION (1) UPPER WING

Y/8W	.2993	.3640	.4270	.5340	.6730	.7900	.8870
------	-------	-------	-------	-------	-------	-------	-------

MS/X

.000	.0427	.0971	.2720	.3062	.2004	.0875	.1475
.020				.5169	.4957	.5069	.4735
.025			.4035				
.040		.2213					
.045			.3369				
.050	.0870			.2933	.3150	.3180	.2665
.060				.1244			
.065			.1979				
.066		.2640					
.094	.0916						
.150		.0000		-.1001	-.0790	-.0532	-.0943
.163							
.177			-.0354				
.193		.1498					
.229	.1625						
.250				-.2315	-.2694	-.3100	-.3626
.274			-.1341				
.339		-.1107					
.362	.1332						
.400				-.1721	-.2696		-.3564
.402			-.1569				
.497	.0000			-.0905	-.1751		-.2291
.550			.0000			-.1503	
.565							
.600							
.650							
.700	.0676			-.0347	-.1569		
.725							
.750							
.760			-.0469			.1062	.0422
.775							
.806			.1241				
.834	.0317						
.850							
.857			.0934				
.865	.1377						
.900	.1029			.0905	.1136	.1246	
.905							
.909			.0766	.1003	.1027		.0957



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81043)

UPPER WING

ARC11-716 1A14 01+712+512N25

ALPHA(1) = -7.890 BETA(1) = -0.020

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.950 .0639 .1138

.953

.0293

.965

.0034

ALPHA(1) = -7.780 BETA(2) = -4.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 .0064 .1936 .1983 .0709 -.0496 .0120

.020

.3781

.025

.1752

.040

.3288

.045

.2139

.050

.3001

.060

.3113

.085

.1501

.086

.2536

.094

.0814

.150

.0000

.163

.0000

.177

.0089

.193

.1590

.220

.1399

.250

.1139

.274

.0976

.339

.1358

.362

.0000

.400

.1932

.402

.2022

.497

.1777

.550

.0000

.565

.2192

.600

.0000

.650

.2494

.700

.1830

.725

.1606

.750

.0814

.760

.0765

.775

.0233

.808

.0988

.834

.0727

.850

.0764

.0372

.0846

.050

.0948

.1013

.1013

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(RB1U43)

UPPER WING

ARC11-718 IA14 01+T12+S12M25

ALPHA ( 1 ) = -7.780 BETA ( 2 ) = -4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.857 .0489

.865 .0732

.900 .0488

.905 .0368

.950 .0440

.955 .0021

.965 -.0284

.0729

.0820

.0926

ALPHA ( 1 ) = -7.780 BETA ( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.000 -.0133

.020 -.0128

.025 .3409

.040 .1316

.045 .3155

.050 .0265

.080 .3013

.085 .2925

.086 .2812

.094 .2144

.150 .1667

.163 .2204

.177 .0000

.193 .0247

.229 .1639

.250 .1218

.274 .0961

.339 -.0810

.362 .1349

.400 .0000

.402 .1863

.497 .0000

.550 .0000

.565 .0000

.600 .0000

.650 .0000

.700 .0000

.725 .0000

.750 .0000

.780 .0000

.1276

.0903

.4693

.3939

.3013

.2925

.2812

.2144

.1667

.2204

.0000

.0247

.1639

.1218

.0961

.1349

.0000

.1863

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U43)

UPPER WING

ARC11-716 1A14 01-T12-S12N2:

ALPHA( 1 ) = -7.780 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.775			.0311	.0302		
	.608		.0219				
	.834	-.1028					
	.850			.0202	.0495	.0630	
	.857		-.0016				
	.865	.0147					.0340
	.900	.0025	.0016	.0331	.0405		
	.905			.0143		.0376	
	.950		-.0229				
	.953						
	.965	-.0380					

ALPHA( 1 ) = -7.850 BETA( 4 ) = 4.110

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	-.0229	-.0028	.0816	-.0066	-.1967	-.3446
	.020				.4294	.3330	.3372
	.025		.2913				.2724
	.040	.1029					
	.045		.2756	.2911	.2637	.2408	.1680
	.050	.0041		.1793			
	.060		.2118				
	.065		.1851				
	.066						
	.094	.0179			-.0247	-.0326	-.0365
	.150		.0000				-.1240
	.163			.0487			
	.177		.1562				
	.193						
	.229	.0998			-.1734	-.2211	-.2655
	.250						-.3722
	.274		-.0784				
	.339		-.0643				
	.362	.1287					
	.400				-.2395	-.3276	-.4537
	.452		-.2021				
	.497	.0000					
	.550			-.2710	-.3304		
	.565		.0000				
	.600						-.2952
	.650						-.2580

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(881043)

UPPER WING

ARC11-716 1A14 01-T12-S12N25

ALPHA(1) = -7.890 BETA(4) = 4.110

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.700	-.1354			-.1902			
.725				-.1893		.0077	-.0361
.750							
.760			-.2086				
.775				-.0090	-.0047		
.808			-.0295				
.834		-.1590					
.850				-.0145	.0188	.0412	
.857			-.0379				
.865		-.0445					
.900		-.0463		.0060			.0239
.905			-.0373		.0173		
.950				-.0127		.0416	
.953			-.0500				
.955		-.0868					

ALPHA(1) = -7.890 BETA(5) = 8.170

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.900	-.0618	-.0264	.0827	-.0655	-.3040	-.4603	-.4055
.920				.3896	.2785	.2723	.1951
.925			.2619				
.940		.0850					
.945			.2470				
.950	-.0195			.2758	.2337	.2058	.1189
.980				.1719			
.985			.1943				
.986		.1634					
.994	-.0024						
.150		.0000		-.0112	-.0325	-.0401	-.1486
.163							
.177		.0405					
.193		.1351					
.229	.0803						
.250							
.274			-.0768				
.339		-.0783					
.368	.0951						
.400				-.1641	-.2132	-.2632	-.3896
.482							
.497			-.2174				
.0000				-.2575	-.3510		-.5024



## TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U43)

UPPER WING

DATE 10 JAN 75

ARC11-716 1A14 01-T12+S12N25

ALPHA( 1 ) = -7.890 BETA( 5 ) = 0.170

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.550	-3.437	-4.036				
.565	.0000					
.600						
.650						
.700	-2.213	-2.086				
.725						
.750						
.760	-2.2566					
.775	-0.0614					
.808						
.834	-0.0289	0.0109	0.0423			
.850						
.857	-0.0632					
.865						
.900	-0.0046	0.0273	0.0539			
.905	-0.0451					
.950	-0.0160					
.953	-0.0578					
.965	-0.0669					

ALPHA( 2 ) = -3.680 BETA( 1 ) = -0.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	0.0875	0.1425	0.3972	0.4924	0.4327	0.3581	0.3906
.020				0.3494	0.3151	0.3275	0.3129
.025			0.2944				
.040		0.2197					
.045			0.1916				
.050	0.0904			0.0747	0.0626	0.0643	0.0136
.060				-0.1214			
.085			0.0119				
.086		0.2301					
.094	0.0337						
.150		0.0000		-0.3464	-0.3564	-0.3459	-0.3506
.163			-0.2399				
.177		0.0182					
.193							
.229	0.1146						
.250				-0.4714	-0.5409	-0.5945	-0.6449
.274			-0.3293				
.339		0.2826					

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(RB1043)

UPPER WING

ALPHA(2) = -3.880 BETA(1) = -0.060

ARC11-716 1A14 01+112+312M25

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.362	.0356	
.400		
.402		
.497	.0000	
.550		
.565		
.800		
.550		
.730	.0207	
.725		
.750		
.760		
.775		
.808		
.834	.0219	
.850		
.857		
.865	.1303	
.900	.0969	
.905		
.950		
.953		
.965	.0079	

ALPHA(2) = -3.880 BETA(2) = -4.020

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.000	.0216	.0927
.020		
.025		
.040	.1835	
.045		
.050	.0184	
.080		
.085		
.086	.2099	
.094		
.150	.0265	
.165		
.177		
.195		



## TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBLU43)

UPPER WING

DATE 13 JAN 75

ARC11-716 1A14 01+T12+312N25

ALPHA( 2 ) = -3.853 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.229	.0973					
.250						
.274						
.339						
.362	.0421					
.400						
.432						
.497	.0000					
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.775						
.830						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.953						
.965						

ALPHA( 2 ) = -3.640 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0136	.0625	.2703	.2930	.1789	.0888	.1481
.020				.3437	.2846	.3034	.2576
.025							
.040							
.045							
.050							
.080							
.085							
.086							
.094							



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R01043)

UPPER WING

ARC11-716 IA14 OL+T12+S12N25

ALPHA( 2 ) = -3.640 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.150 .163 .177 .193 .229 .250 .274 .339 .362 .400 .402 .497 .550 .565 .600 .650 .700 .725 .750 .760 .775 .808 .834 .850 .857 .865 .900 .905 .950 .953 .965

.0900 .1489 .0484 .0820 .0000

-.2405 -.2527 -.2467 -.3186

-.3736 -.4237 -.4884 -.5932

-.3323 -.4493 -.6382

-.3137

-.2772 -.3417

.0000

-.2331

-.1769

-.1451

.0340 .0355

.0286 .0565 .0759

.0051

.0393 .0515 .0676

.0117 .0240

-.0073

-.0471

-.0000

-.0000

-.0000

-.0000

-.0000

ALPHA( 2 ) = -3.660 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 .0513 .0404 .2171 .1965 .0622 -.0422 .0304

.080 .025 .040 .1106 .1736 .1290 .1069 .0889 .0110



DATE 19 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U43)

UPPER WING

ARC11-716 IA14 03+112+S12M25

ALPHA( 2) = -3.880 BETAD ( 4) = 4.090

SECTION ( 1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

-.0126

.080

.0681

.085

.1472

.086

-.0115

.094

.150

.183

.177

.193

.229

.250

.274

.339

.362

.400

.402

.497

.530

.565

.600

.650

.700

.725

.750

.760

.775

.808

.834

.850

.857

.865

.900

.905

.950

.953

.965

-.0815

-.0333

.0105

.0455

.0640

.0718

.0296

-.0095

.0110

.0081

.0292

.0336

.0646

.0303

.0109

.0267

.0172

.0000

-.3514

-.4698

-.2523

-.1891

.0296

-.0095

.0110

.0081

.0292

.0336

.0646

.0303

.0109

.0267

.0172

.0000

-.3514

-.4698

-.2523

-.1891

.0296

-.0095

(RB1V43)

**UPPER WING**

TABLE 1. TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+312N25

DATE 10 JAN 75

$$\text{ALPHA}(2) = -3.999 \quad \text{BETA}(3) = 0.150$$

SECTION 1 - JUMPER WING

Y/CM	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CM							
.000	-.1171	-.0106	.0937	.1507	-.0297	-.1541	-.0801
.020				.2918	.2135	.2234	.1653
.025			.2035				
.045		.0171	.1672				
.050	-.0714			.1273	.1066	.0769	-.0041
.060			.0786	.0580			
.065							
.066		.1176					
.094	-.0469						
.150		.0900		-.1670	-.1975	-.1997	-.2983
.163			-.0901				
.177		.0441					
.193							
.229	.0421						
.250			-.1921	-.2990	-.3600	-.4240	-.5571
.274		-.1757					
.359							
.362	.0303		-.3039	-.3434	-.4612		-.6689
.400							
.497	.0000		.0000	-.4148	-.3417		
.550						-.2470	-.2370
.565							
.600							
.650							
.700	-.2607			-.2307	-.2058		
.725							
.750			-.2498			.0270	-.0021
.760				-.0433	.0000		
.775			-.0802				
.806							
.834	-.2319			-.0271	.0271	.0629	
.850			-.0670				
.857							
.865	-.1323						.0790
.900	-.0969			-.0087	.0373		
.905			-.0483	-.0107		.0676	
.950							
.999			-.0532				
.985	-.0940						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U43)

UPPER WING

ARC11-716 1A14 Q1-T12-S12N25

ALPHA(3) = -.310 BETA(1) = -0.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW	.000	.0421	.1337	.4348	.5990	.5660	.5276
	.020			.1353	.0763	.0489	.0052
	.025		.1422				
	.040		.1733				
	.045		.0165				
	.050	-.0034		-.1670	-.2149	-.2487	-.3410
	.080			-.3830			
	.085		-.1901				
	.086		.1519				
	.094	.0079					
	.150		.0000	-.6100	.6755	-.6468	-.6275
	.163						
	.177		-.4354				
	.193		-.1163				
	.229	.0712					
	.250			-.6032	-.7836	-.6812	-.6784
	.274		-.4845				
	.339		-.4703				
	.362	-.0341					
	.400			-.5808	-.5122		-1.0700
	.402		-.3948				
	.497	.0000		-.1311	-.1988		
	.550		.0000				
	.565						
	.600						
	.650						
	.705	-.0117					
	.725			-.0204	-.1601	-.1040	-.1818
	.750						
	.760		-.0141				
	.775		.1202	.1274	.1085		
	.808						
	.834	.0334		.1111	.1221	.1300	
	.850		.1047				
	.857						
	.865	.1404					
	.900	.1186		.1028	.0997		.1097
	.905		.0832				
	.930		.0753			.1059	
	.933		.0480				
	.965	.0340					

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01043)

UPPER WING

ARC11-716 1A14 C1112+S12M25

ALPHA(3) = -.330 BETA(2) = -5.040

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
//CW							
.000	.0033	.1009	.3937	.5405	.5014	.4601	.4770
.020				.1474	.0880	.0646	.0164
.025			.1481				
.040		.1497					
.045			.0339				
.050	-.0241			-.1360	-.1873	-.2236	-.3190
.080				-.3485			
.085			-.1488				
.086		.1424					
.094	-.0192			-.5574	-.6099	-.5994	-.6031
.150		.0000					
.163			-.4000				
.177		-.0984					
.193							
.229	.0361			-.6585	-.7632	-.8271	-.8462
.250			-.4774				
.274		-.4336					
.339							
.362	-.0475			-.4082	-.5415		-1.0230
.400			-.4225				
.402				-.2086	-.2424		
.497	.0000		.0000				-.1983
.550							
.565							
.600							
.650							
.700	-.0633			-.0581	-.1560		
.725							
.790			-.0568			.0861	.0108
.780				.0987	.0872		
.775			.0854				
.808							
.834	-.0112			.0856	.0990	.1132	
.850			.0675				
.857							
.865	.0918			.0866			.0989
.900	.0765				.0812		
.905			.0815				
.950			.0634			.0913	
.953			.0394				
.965	.0113						



DATE 10 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81043)

UPPER WING

ARC11-716 IA14 01+112+S12N25

ALPHA(3) = -.340 BETA(3) = .040

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.000	-.0558	.3212	.4398	.3764
.020			.1729	.0989
.025		.1453		.0272
.040		.1235		
.045		.0915		
.050	-.0458		-.0828	-.1311
.080			-.2598	
.085		-.1044		
.086		.1272		
.094	-.0318		-.4484	-.5082
.150		.0000		-.5280
.163				-.5629
.177		-.3223		
.193		-.0692		
.229	.0443		-.5700	-.6842
.250				-.7450
.274		-.4022		-.8106
.339	-.3585			
.362	-.0340		-.4371	-.5782
.400				-.9670
.402		-.4288		
.497	.0000		-.3241	-.3563
.550		.0000		-.2335
.565				
.600				
.650				
.700	-.1583		-.1660	-.1962
.725			-.1265	
.750				.0524
.760		-.1330		-.0009
.775		.0336	.0468	
.808		.0247		
.834	-.0972		.0376	.0659
.850		.0183		.0855
.857				
.865	-.0017		.0477	.0881
.900	-.0021			
.905		.0169	.0337	.0507
.950			.0370	
.953		-.0002		
.965	-.0348			

DATE 10 JAN 79 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881043)

UPPER WING

ARC11-716 1A14 01+112+512N25

ALPHA( 3 ) = -.350 V. 10 ( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1239	-.0109	.2593	.3548	.2787	.2236	.2651
.020				.1610	.0857	.0734	.0152
.025			.1175				
.040		.0736					
.045			.0432				
.050	-.0768			-.0624	-.1034	-.1472	-.2559
.080				-.2195			
.085			-.0857				
.086		.0988					
.094	-.0515			-.3882	-.4356	-.4674	-.5454
.150		.0000					
.163			-.2661				
.177		-.0521					
.193							
.229	.0328			-.3027	-.6012	-.6843	-.7816
.250			-.3461				
.274		-.3056					
.339							
.362	-.0214			-.4445	-.5795		-.8540
.400			-.4127				
.402							
.497	.0000			-.4188	-.5400		
.550			.0000				
.565							
.600							
.650							
.700	-.2246						
.725				-.1695			
.750					-.2120		
.760			-.1826				
.775				-.0032	.0185		
.808			-.0370				
.834	-.1580			.0010	.0433	.0710	
.850			-.0292				
.857							
.885	-.0637						
.900	-.0531			.0198	.0515		.0816
.905			-.0125				
.950				.0184		.0653	
.955			-.0161				
.965	-.0673						



DATE 15 JAN 75 TABULATED PRESSURE DATA - IAI14 - VOL. 6

(R81043)

UPPER WING

ARC11-716 IAI14 01-712-312N25

ALPHA(3) = -.340 BETA(3) = 0.150

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2143	-.0459	.0968	.2643	.1873	.1012	.1584
.020				.1624	.0874	.0914	.0249
.025			.0490				
.040		-.0072					
.045			.0206				
.050	-.1346			-.0276	-.0670	-.1010	-.2132
.060				-.1592			
.085				-.0425			
.086		-.0052					
.094	-.1249			-.3158	-.3667	-.3851	-.4934
.150		.0000					
.163				-.1951			
.177		-.0163					
.193							
.229	.0241			-.4246	-.5399	-.5893	-.7181
.250				-.2828			
.274		-.2607					
.339	-.0235			-.4238	-.5507		-.7894
.362				-.3743			
.400							
.402				-.0000			
.497	.0000			-.4851	-.5959		
.850							
.565							
.800							
.650						-.2217	-.2204
.700	-.2823			-.2293	-.2037		
.725						.0415	.0187
.750				-.2451			
.760				-.0628	.0093		
.775				-.0885			
.808							
.834	-.2305			-.0279	.0350	.0722	
.850				-.0870			
.857							
.900	-.1405			-.0077			.0873
.905	-.1037			-.0410			
.908				-.0089			.0697
.910				-.0396			
.933							
.965	-.0917						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U43)

UPPER WING

ARC11-716 1A14 01+112+312N25

ALPHA(4) = 4.130 BETA(1) = -8.090

SECTION (1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.0103	.0521	.4095	.6212 .5855 .5126 .4767
.020				-.1370 -.2249 -.3445 -.4606
.025				-.0445
.040		.0871		
.045			-.1950	
.050	-.0702			-.4373 -.5401 -.6259 -.7770
.080				-.6712
.085				-.4148
.086		.0538		
.094	-.0482			
.150				-.9174 -1.0020 -.9924 -.9779
.163		.0000		
.177			-.6404	
.193		-.2528		
.229	.0291			
.250				-.9898 -1.1800 -1.0640 -1.0960
.274		-.7393		
.339		-.6476		
.362	-.1395			
.400			-.6103 -.7808	-.8294
.432		-.6064		
.497	.0500		-.1284 -.4043	
.530		.0000		
.565				-.3867
.600				
.650				-.4397
.700	-.0388			
.725			-.0361	
.750				-.2489 -.8085
.780		-.0043		
.775		.0982		
.808				
.834	.0411		.0832 .0715	-.1171
.850				
.857		.0903		
.865	.1469			
.900	.1271		.0939	-.0720
.905		.0881	.1257	
.930			.0854	-.0057
.933		.0535		
.965	.0483			



DATE 10 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

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ARC11-716 IA14 01+T12+S12M25

UPPER WING

(R81043)

ALPHA( 4) = 4.050 BETA( 2) = -4.020

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0654	.0190	.3734	.5720	.5335	.4802	.4244
.020				-.0871	-.1936	-.3133	-.4432
.025			-.0160				
.040		.0824					
.045			-.1382				
.050	-.0912			-.3695	-.4920	-.5934	-.7419
.080				-.5938			
.085							
.086			-.3367				
.086		.0577					
.094	-.0594						
.150				-.8259	-.9328	-.9226	-.9529
.163		.0000					
.177			-.5607				
.193		-.2124					
.229	.0191						
.250				-.8728	-1.0600	-1.1440	-1.0770
.274			-.6682				
.339		-.5947					
.362	-.1239						
.400				-.5898	-.9613		-.7302
.402		-.7074					
.497	.0000			-.1967	-.2807		
.565			.0500				
.600							
.650					-.3648		-.3737
.700	-.1168			-.0801	-.1494		
.725						-.1041	-.1597
.750							
.780		-.0440		.0564	.0407		
.775		.0653					
.808							
.834	-.0184			.0876	.0905	.0403	
.850			.0671				
.857							
.885	.0889						
.900	.0828		.0773				-.0196
.905		.0619	.1028				
.950			.0718			.0091	
.955		.0383					
.985	.0258						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881043)

UPPER WING

ARC11-716 1A14 01+712+912M25

ALPHA( 4 ) = 4.950 BETA( 3 ) = .030

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW	X/CW	CP
.000	-.1176	-.0310
.020		.3103
.025		.5096
.040		.4706
.045		.3976
.050		.3649
.060		-.0533
.065		-.1767
.066		-.2912
.067		-.4270
.070		.0017
.075		.0646
.080		-.1169
.085		-.3230
.090		-.4446
.095		-.5643
.100		-.7126
.105		-.5236
.110		-.2690
.115		.0516
.120		-.1069
.125		-.7294
.130		-.8530
.135		-.9673
.140		-.9072
.145		.0000
.150		-.1807
.155		-.4965
.160		-.1003
.165		-.7629
.170		-.9094
.175		-1.0460
.180		-.5903
.185		-.1073
.190		-.5432
.195		-.362
.200		-.400
.205		-.402
.210		-.6269
.215		-.6374
.220		-.9424
.225		-.9876
.230		.497
.235		.0000
.240		-.3192
.245		-.2975
.250		.565
.255		.600
.260		.600
.265		-.1775
.270		-.3545
.275		-.1949
.280		-.1614
.285		-.0997
.290		.0148
.295		-.0541
.300		.0388
.305		.0403
.310		.0274
.315		.0574
.320		.0746
.325		.0996
.330		.0321
.335		.0134
.340		.865
.345		.900
.350		.0281
.355		.0284
.360		.0608
.365		.0904
.370		.0360
.375		.0491
.380		.1022
.385		.0190
.390		.965
.395		-.0078



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1043)

UPPER WING

ARC11-716 1A14 01-712-912N25

ALPHA(4) = 4.140 BETA(4) = 4.140

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2223	-.1259	.2286	.4352	.3994	.3329	.3137
.020				-.0458	-.1650	-.2705	-.4094
.025			-.0185				
.040		.0169					
.045			-.1135				
.050	-.1229			-.2934	-.4148	-.5208	-.6769
.080				-.4622			
.085			-.2592				
.086		.0227					
.094	-.0878						
.150		.0000		-.6307	-.7318	-.7988	-.8519
.163							
.177			-.4322				
.193		-.1611					
.229	-.0066						
.250				-.6870	-.8402	-.9367	-1.0250
.274			-.4838				
.339		-.4646					
.362	-.0875			-.5805	-.7748		-1.1200
.400			-.3567				
.402				-.4435	-.5120		
.497	.0000		.0000				
.550							
.565							
.600							
.650							
.700	-.2745			-.1678			-.3185
.725				-.1373			
.750			-.1455		.0548	.0121	
.760				.0051	.0429		
.775			-.0907				
.808							
.834	-.1622			.0231	.0603	.0947	
.850			-.0204				
.857							
.865	-.0517						.0956
.900	-.0288			.0341	.0716		
.905			-.0087				
.950				.0341			.0882
.955			-.0109				
.965	-.0451						

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81043)

UPPER WING

ARC11-716 IA14 01+T12+S12M25

ALPHA(4) = 4.150 BETA(5) = 0.190

SECTION (1) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3411	-.1159	.0298	.2564	.3501	.2746	.2773
.020				-.0593	-.1103	-.1981	-.3171
.025			-.0653				
.040		-.0537					
.045			-.1199				
.050	-.2030			-.2277	-.3256	-.4210	-.5787
.060				-.3579			
.085			-.1941				
.086		-.0394					
.094	-.2497						
.150				-.4937	-.6121	-.6678	-.7767
.163		.0000					
.177			-.3206				
.193		-.1504					
.229	.0046						
.250				-.5629	-.7230	-.8326	-.9409
.274			-.4025				
.339		-.3645					
.362	-.0459						
.400				-.5220	-.6690		-1.0800
.402			-.4517				
.497	.0000			-.4684	-.6273		
.550			.0700				
.565							-.1975
.600						-.1876	
.650				-.1853			
.700	-.2959			-.1800			
.725					.0560	.0391	
.750							
.760			-.2164				
.775			-.0915	-.0616	.0835		
.808							
.834	-.2007			-.0234	.0502	.0950	
.850			-.0700				
.857							
.885	-.1171						.1101
.900	-.0422			-.0036	.0604		
.905			-.0409				
.950				-.0031	.0903		
.953			-.0331				
.963	-.0796						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U43)

UPPER WING

ARC11-716 1A14 01+112+512N25

ALFMAO( 5) = 0.040 BETA0 ( 1) = -0.070

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.4340	.6750	.7800	.8870
X/CW							
.000	-.0567	-.0143	.4045	.6269	.5270	.3840	.3191
.020				-.3471	-.4837	-.6784	-.8353
.025							
.040		.0178					
.045							
.050	-.1162						
.060				-.6448	-.7952	-.9196	-1.0580
.085				-.8643			
.086							
.094	-.0810						
.150							
.163		.0000					
.177							
.193		-.3454					
.229	.0070						
.250							
.274							
.339							
.362	-.1793						
.400							
.402							
.497	.0000						
.530							
.565							
.600							
.650							
.700	-.0485						
.725							
.750							
.775							
.806							
.834	-.0414						
.850							
.857							
.865	.0623						
.900	.0672						
.935							
.950							
.953							
.965	.0237						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81043)

UPPER WING

ARC11-716 .A14 01+T12+S12N25

ALPHAC( 5) = 0.010 BETA( 2) = -4.020

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.000	-.1281	-.0569	.3615	.5880	.9003	.3539	.2780
.020				-.2844	-.4431	-.6559	-.5389
.025				-.1455			
.040		.0142					
.045			-.2940				
.050	-.1417			-.5005	-.7345	-.8913	-1.0980
.060				-.7910			
.085			-.5233				
.086		-.0132					
.094	-.1000			-1.0960	-1.1710	-1.0510	-.9812
.150		.0000					
.163			-.7111				
.177		-.3075					
.193							
.229	-.0544			-1.1010	-1.1670	-.8083	-.6716
.290			-.8162				
.274		-.7055					
.339							
.362	-.1686			-.6276	-.7093		-.6124
.400			-.8972				
.402	.0000			-.2716	-.6251		
.497			.0000				
.550						-.5320	
.565							
.600							
.650				-.1755		-.5830	
.700	-.1119			-.1890			
.725							
.750						-.4597	-.4193
.760			-.1127	-.0751	-.3129		
.775			-.0019				
.808							
.834	-.0640			-.0173	-.1749	-.3322	
.850			.0096				
.857							
.865	.0392			.0246			-.3152
.900	.0479						
.905		.0601		-.0518			
.930			.0335			-.2242	
.935		.0211					
.965							
		.0134					



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U43)

UPPER WING

DATE 13 JAN 75

ARC11-716 1A14 Q1+T12+S12N25

ALPHA (3) = 8.910 BETA (3) = .030

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

X/CW	Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.2041	-.1317	.2809	.5155	.4550	.3168	.2276	
.020				-.2387	-.4126	-.6376	-.8270	
.025				-.1364				
.040			-.0094					
.045				-.2649				
.050	-.1533			-.5198	-.7151	-.8674	-1.0880	
.080				-.7263				
.085				-.4422				
.086			-.0164					
.090	-.1125							
.150				-.9347	-1.1150	-1.1310	-1.1210	
.163			.0050					
.177			-.2762					
.193			-.6391					
.229	-.0110			-.9806	-1.2480	-.9454	-.7934	
.250				-.7307				
.274			-.6470					
.339								
.362	-.1557			-.7539	-.7290		-.8608	
.400				-.8352				
.402				-.2675	-.5592			
.497	.0000			.0000				
.550								
.565								
.600								
.650								
.700	-.1940			-.1636				
.725				-.1731				
.750								
.760				-.1105				
.775				-.0553	-.1650			
.808				-.0128				
.834	-.1080							
.850				.0012	-.0513	-.2689		
.857				-.0043				
.865	.0056							
.900	.0190			.0185	.0542		-.2532	
.905				.0119				
.950				.0392			-.1259	
.955				.0113				
.965	-.0071							



DATE 15 JAN 75 TABULATED PRESSURE DATA - IAI14A - VOL. 6

(R81043)

UPPER WING

ARC11-716 IAI14 OI+T12-S12N23

ALPHA( 9 ) = 9.000 BETA( 4 ) = 4.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.3385	.2764	.1748
.4220	.3939	.2817
.2319	.3830	-.5965
.1576		
.0528		
.2623		
.4812	.6732	-.8204
.6727		
.4012		
.0435		
.1403		
.0000		
.5732		
.2437		
.0455		
.274		
.359		
.362		
.402		
.0000		
.3450		
.0000		
.7647		
.6967		
.3450		
.0000		
.1729		
.1356		
.1334		
.0302		
.0360		
.0072		
.0136		
.0261		
.0180		
.0081		
.0000		
.0303		
.0499		
.0392		
.0277		
.0096		
.0319		



(RB1043)

UPPER WING

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14-01-112-S12N25

DATE 10 JAN 75

ALPHA(5) = 8.080 BETA(5) = 8.250

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.4992	-.1269	.2279	.3596	.2444	.1608
.020		-.2156	-.3077	-.3548	-.3360	-.7667
.025						
.040		-.1126				
.045			-.2427			
.050	-.2671		-.4481	-.6189	-.7525	-.9902
.080			-.5867			
.085		-.3080				
.086						
.094	-.2573	-.1141				
.150		.0000	-.6857	-.9181	-.9739	-1.1010
.163						
.177		-.4190				
.193		-.2434				
.229	-.1102		-.7268	-.9129	-1.0980	-1.2090
.230			-.4952			
.274		-.4792				
.339	-.1332		-.3698	-.7977		-1.0590
.362			-.5045			
.403				-.4398	-.4930	
.402	.0000	.0000				-.4888
.497					-.1837	
.530				-.1952		
.565			-.1684		.0119	-.1428
.600						
.650	-.3070					
.700						
.725		-.2001				
.750			-.0690	.0062		
.760						
.775		-.0997				
.808						
.834	-.1804		-.0239	.0467	.0655	
.850						
.857		-.0747				
.865	-.1118		-.0114			.0377
.900	-.0753		-.0522	.0646		
.935			-.0045			.0926
.950						
.953		-.0444				
.965	-.0743					

DATE 10 JAN 74

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 3092

(RB1044) (15 FEB 74)

UPPER WING

ARC11-715 1A14 01+712+312N25

## REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
 LREF = 38.7390 INCHES YMRP = .0000 INCHES  
 SREF = 38.7390 INCHES ZMRP = .0000 INCHES  
 SCALE = .0300 SCALE

## PARAMETRIC DATA

MACH = .930 ELEVON = .000  
 RUDDER = .000 SPDBRK = .000

ALPHA(1) = -7.970 BETA(1) = -8.090

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7890	.8870
X/CW	.0000	.0561	.0841	.3184	.3623	.2607	.1307
	.020			.5237	.4956	.5142	.4886
	.025		.2428				
	.040			.3959			
	.045				.3093	.3218	.3242
	.050				.1378		.2770
	.060						
	.088			.2218			
	.086		.3208				
	.094						
	.130						
	.163						
	.177						
	.193						
	.229						
	.250						
	.274						
	.339						
	.362						
	.400						
	.402						
	.497						
	.550						
	.563						
	.600						
	.650						
	.700						
	.723						
	.750						
	.760						
	.775						
	.808						
	.834						
	.890						
	.857						
	.865						
	.900						
	.905						



DATE 13 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881044)

UPPER WING

ALPHAO ( 1 ) = -7.970 BETAO ( 1 ) = -8.050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7300	.8870
X/CW	.990		.0254		.0923		
	.993		.0044				
	.985	-.0075					

ALPHAO ( 1 ) = -7.960 BETAO ( 2 ) = -4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	.0259	.0353	.2346	.2594	.1271	.0046
	.020			.5122	.4571	.4734	.4342
	.025		.3980				
	.040	.1959					
	.045		.3478				
	.050			.3170	.3194	.3167	.2595
	.080	.0687		.1688			
	.085		.2268				
	.086	.2738					
	.094	.0793		-.0909	-.0414	-.0203	-.0772
	.150	.0000					
	.163		.0159				
	.177		.1836				
	.193	.1556					
	.229			-.2016	-.2327	-.2684	-.3638
	.290		-.1048				
	.274						
	.339	-.0824					
	.362	.1493		-.1861	-.2902		-.4871
	.409		-.1653				
	.402			-.1687	-.2811		
	.497	.0000		.0000			-.2298
	.550						
	.565						
	.600						
	.630						
	.700	.0031		-.1792			
	.725		-.1094				
	.750					.0830	.0251
	.760		-.1167				
	.775			.0905	.0798		
	.808		.0779				
	.834	-.0841					
	.850			.0556	.0770	.0958	

(RBLU44)

DATE 10 JAN 75      TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01+112+512N25      UPPER WING

ALPHAO ( 1 ) = -7.960    BETAO ( 2 ) = -4.000

SECTION ( 1 ) UPPER WING      DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.857	.0430	
.865	.0781	
.900	.0435	.0501
.905	.0232	.0350
.930	-.0022	.0616
.953	-.0282	
.965	-.0471	

ALPHAO ( 1 ) = -7.960    BETAO ( 3 ) = .050

SECTION ( 1 ) UPPER WING      DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.000	.0224	.1760
.020	.0235	.1477
.025		.4731
.040	.3584	.4097
.045	.1581	.4205
.050	.3237	.3159
.080		.3104
.085	.2563	.2957
.086	.2416	.2324
.094	.0581	.1868
.150		
.163	.0000	-.0174
.177	.1807	-.0136
.193		-.0056
.229	.1386	-.0792
.250		
.274	-.0726	-.1837
.339	-.0580	-.2004
.362	.1567	-.2391
.400		-.3462
.402	-.1681	
.497	.0000	-.1954
.550		-.2915
.565		-.4901
.600		
.650		-.2240
.700		-.3525
.725		.0000
.750		-.3311
.760		-.3160
		-.1954
		-.1753
		.0885
		.0222
		-.1908



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1044)

UPPER WING

ALPHAX(1) = -7.960 BETAX(3) = .050

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.775 .0340 .9407

.808 .0129

.834 -.1311

.850 .0114

.857 -.0110 .0435 .0697

.865 .0002

.900 -.0148 .0084 .0487

.905 .0161

.950 -.0299 .0265

.953 -.0548

.965 -.0784

ALPHAX(1) = -7.970 BETAX(4) = 4.100

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.0139 .0391 .1410 .0736 -.1238 -.2632 -.2203

.020 .4575 .3587 .3599 .2997

.025 .3274

.040 .1306

.045 .3043

.050 .0350 .3190 .2893 .2611 .1930

.060 .2013

.065 .2581

.086 .2069

.094 .0538 .0101 -.0036 -.0063 -.0695

.150 .0000

.163 .0418

.177 .1738

.193 .1243

.229 .1459 -.1848 -.2254 -.3393

.250 .0511

.274 -.0379

.339 .1392

.362 .1983 -.2983 -.4925

.400 .1694

.402 .0000

.497 -.2743 -.3851

.550 .0000

.565 .0000

.600 -.5432

.650 -.5151



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U44)

UPPER WING

ARC11-716 1A14 01+T12+S12H25

ALPHA(1) = -0.000 BETA(1) = 0.170

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.590		-.3022 -.3975
.565		.0000
.600		-.9220
.650		-.5900
.700	-.1763	-.2611
.725		-.3952
.790		-.3088 -.1840
.760		-.3697
.775		-.2023 -.2402
.808		-.1775
.834	-.3067	-.1059 -.0707 -.0484
.850		-.1512
.865	-.2068	-.0694
.900	-.1556	-.1114 -.0099
.905		-.0562
.930		-.0034
.933		-.1015
.965	-.1421	

ALPHA(2) = -4.070 BETA(1) = -0.060

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.000	.0673	.4234 .5200 .4500 .3885 .4222
.020		.3721 .3395 .3539 .3353
.025		.3168
.040	.2355	
.045		.2164
.050	.0339	.1036 .0688 .0376
.060		-.0934
.065		.0361
.086	.2472	
.094	.0611	-.3939 -.3304 -.3110 -.3209
.150		.0000
.163		-.2178
.177		.0442
.193		
.229	.1303	-.4634 -.5386 -.5694 -.6053
.250		-.3220
.274		
.339	-.2763	



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81044)

UPPER WING

ARC11-716 IA14 01-712-512N25

ALPHA( 2 ) = -4.070 BETA( 1 ) = -0.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6750 .7800 .8870
.362	.0623	
.400		
.402		
.497	.0000	
.550		
.565		
.600		
.650		
.700	.0223	
.725		
.750		
.760		
.775		
.808		
.834		
.850	.0089	
.857		
.865	.1373	
.900	.1032	
.905		
.950		
.953		
.965	-.0007	

ALPHA( 2 ) = -3.970 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6750 .7800 .8870
.000	.0364	
.020	.1135	
.025		
.040	.3143	
.045	.2062	
.050		
.060	.0343	
.065		
.086	.2324	
.094	.0451	
.150		
.165		
.177		
.195		



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U44)

UPPER WING

ARC11-716 1A14 01+112+512N25

ALPHA( 2 ) = -3.970 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.229	.1176							
.250								
.274								
.339								
.362	.0700							
.400								
.402								
.497	.0000							
.550								
.565								
.600								
.650								
.700								
.725								
.750								
.760								
.775								
.834								
.850								
.857								
.865								
.900								
.905								
.950								
.955								
.965								

ALPHA( 2 ) = -3.080 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	.0106							
.020								
.025								
.040								
.045								
.050								
.065								
.086								
.094								

DATE 13 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U44)

UPPER WING

ARC11-716 IA14 01+T12+S12N25

ALPHA(D 2) = -3.800 BETA(D 3) = .030

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .5730 .7800 .8870

X/CW .150 .163 .177 .193 .229 .250 .274 .339 .362 .400 .432 .497 .550 .565 .600 .650 .700 .725 .750 .760 .775 .808 .834 .850 .867 .885 .900 .935 .950 .953 .965

-.2054 -.2132 -.2055 -.2864

-.1213

.0781

.1041

-.3493 -.4060 -.4522 -.5496

-.2208

-.1929

.0796

-.3059 -.4239 -.6560

-.2919

.0000

-.3093 -.4526

.0000

-.6090

-.1153

-.2228

.725

-.2803

.750

.0866 .0399

.760

-.2596

.775

.0309 .0649

.808

-.0029

.834

.0177 .0662 .0966

.850

.0143

.867

.0796

.885

-.0486

.900

.0170 .0163

.935

-.0190

.950

-.0232

.953

.0429

.965

-.0532

ALPHA(D 2) = -3.920 BETA(D 4) = 4.080

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .5730 .7800 .8870

X/CW .000 .0480 .0615 .2514 .2419 .1076 .0051 .0758

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050

.020 .025 .040 .045 .050



(R81U44)

TABULATED PRESSURE DATA - IA104 - VOL. 6

UPPER WING

ARC11-716 IA14 O1+T12+S12N25

ALPHAD ( 2 ) = -3.920 BETA D ( 4 ) = 4.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/RW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.080	.0222					
.085	.1025					
.086	.1746					
.094	.0258					
.150						
.163						
.177						
.193						
.229						
.250						
.274						
.339						
.362						
.400						
.402						
.497						
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.953						
.965						

-.1630 -.1845 -.1871 -.2674

-.3022 -.3522 -.4109 -.5281

-.2978 -.4141 -.6403

-.3397 -.4695 -.6414

-.4177 -.2520 -.8016

-.1739 -.1539

-.1887 -.1300

-.0619 -.0203 .0065

-.0940

-.0330 -.0262 .0246

-.0755 -.0417 .0034

-.0707

-.1023

(R81U44)

UPPER WING

ALPHA( 2 ) = -3.930 BETA( 5 ) = 0.150

ARC11-716 IA14 Q1-T12-S12N25

DEPENDENT VARIABLE CF

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CM

.000	-.1773	.0103	.1186	.1928	.0128	-.1089	-.0495
.020				.3265	.2476	.2574	.2043
.025			.2293				
.040		.0314					
.045			.1944				
.050	-.0627			.1629	.1446	.1200	.0421
.060				.0418			
.085			.1046				
.086		.1411					
.094	-.0214						
.150		.0000		-.1254	-.1495	-.1469	-.2453
.163							
.177			-.0516				
.193		.0741					
.29	.0732						
.250				-.2558	-.3033	-.3557	-.4936
.274			-.1402				
.339		-.1196					
.362	.0827						
.400				-.2880	-.3889		-.6184
.402			-.2502				
.497	.0000						
.550				-.3577	-.4723		
.565			.0000				
.600							-.6637
.650						-.6175	
.730	-.1998				-.2783		
.725				-.4552			
.750						-.3743	-.2071
.760			-.4047				
.775				-.3630	-.3908		
.808			-.2788				
.834	-.3495						
.890				-.1261	-.0995	-.1020	
.897			-.1955				
.885	-.3135						
.900	-.2185			-.0709			-.0373
.975			-.1247		-.0128		
.950				-.0493		-.0103	
.953			-.0979				
.965	-.1505						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U44)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 3 ) = .000 BETA( 1 ) = -6.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/W .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.000	.0364	.1516	.4664	.6501	.5982	.5543	.5635
.020			.1753	.1122	.0731	.0287	
.040		.1940					
.060			.0556				
.080	.0028			-.1173	-.1745	-.2189	-.3113
.100				-.3356			
.120			-.1447				
.140		.1749					
.160	.0171			-.5981	-.6388	-.6168	-.5969
.180		.0000					
.200		-.3940					
.220		-.0788					
.240	.0913			-.6729	-.8034	-.8473	-.8323
.260		-.5146					
.280		-.4550					
.300	-.0192			-.7201	-.9479		-1.0
.320		-.6406					
.340	.0000			-.1931	-.2371		
.360		.0000					
.380						-.2330	-.4244
.400	-.0262			-.0252	-.1914		
.420							
.440		-.0219					
.460		.1265		.1299	.1117		
.480							
.500	.0212		.1036	.1018	.1167	.1522	
.520							
.540	.1424			.0886			.0692
.560	.1178			.0779	.0668		
.580				.0430		.0978	
.600			.0149				
.620							
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.000							



DATE 12 JAN 75

$$\text{BETAD} ( 3 ) = -.329$$

DEPENDENT VARIABLE CF

SECTION (1) UPPER WING

	2697	1643	.4270	.5340	.6730	.7800	.8870
--	------	------	-------	-------	-------	-------	-------

218

.000	-.0358	.0706	.1307	.4657	.4017	.3575	.3887
120				.1950	.1275	.1069	.0614

.323	.1433	.9797
.040		
.045		

152

0.551	-0.0325	0.498
0.080		-0.2367
		0.5703

...

.086	.1901
.094	-.0120
	.0010
	.1714
	-.4679
	-.3240

.163	.0000
.177	-.2940

**.0376**

.229	.0663		
.259		-.5305	-.6303
			-.6937
			-.7562

1274 - 3591  
1339 - 3591  
1423 - 3591

.493
- .4732
.492
.486

.550	- .550
.565	.0000
	- .6103

.650	- .6089
.700	- .2199
- .1502	

.750	.1024	.0769
.760	-.2109	

- .0769

.078 - 2096

974 - 2096

0.037  
0.037  
- .0021  
0.037

	0.000	-0.0542	-0.0094	.0267	.0339
0.000					
0.005					
0.010					
0.015					
0.020					
0.025					
0.030					
0.035					
0.040					
0.045					
0.050					
0.055					
0.060					
0.065					
0.070					
0.075					
0.080					
0.085					
0.090					
0.095					
0.100					
0.105					
0.110					
0.115					
0.120					
0.125					
0.130					
0.135					
0.140					
0.145					
0.150					
0.155					
0.160					
0.165					
0.170					
0.175					
0.180					
0.185					
0.190					
0.195					
0.200					
0.205					
0.210					
0.215					
0.220					
0.225					
0.230					
0.235					
0.240					
0.245					
0.250					
0.255					
0.260					
0.265					
0.270					
0.275					
0.280					
0.285					
0.290					
0.295					
0.300					
0.305					
0.310					
0.315					
0.320					
0.325					
0.330					
0.335					
0.340					
0.345					
0.350					
0.355					
0.360					
0.365					
0.370					
0.375					
0.380					
0.385					
0.390					
0.395					
0.400					
0.405					
0.410					
0.415					
0.420					
0.425					
0.430					
0.435					
0.440					
0.445					
0.450					
0.455					
0.460					
0.465					
0.470					
0.475					
0.480					
0.485					
0.490					
0.495					
0.500					
0.505					
0.510					
0.515					
0.520					
0.525					
0.530					
0.535					
0.540					
0.545					
0.550					
0.555					

.957	- .0936
.965	- .097





DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R01U44)

UPPER WING

ARC11-716 IA14 01-T12-S12N25

ALPHA( S ) = -.330 BETAC ( S ) = 0.150

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	-.2847	-.0318	.1029	.2836	.2185	.1397	.1881
.025			.0619	.1864	.1162	.1258	.0670
.040		.0045					
.045			.0365				
.050	-.1446		.0041	-.0324	-.0593	-.1608	
.060			-.1214				
.065			-.0174				
.066		.0017					
.094	-.1004						
.150		.0000		-.2772	-.3245	-.3506	-.4368
.163							
.177			-.1555				
.193		.0096					
.229	.0523						
.250				-.3034	-.4688	-.5397	-.6586
.274			-.2406				
.339		-.2229					
.362	.0083						
.400			-.3728	-.4837			-.7264
.402			-.3200				
.497	.0000			-.3950	-.5296		
.550			.0000				
.565							
.600							
.650							
.700	-.2292			-.2864			
.725				-.4553			
.750					-.3217	-.2141	
.760			-.4227				
.775			-.4146	-.3028	-.3553		
.808							
.834	-.3813						
.850			-.2277	-.1478	-.1127	-.1112	
.857							
.865	-.4097						
.900	-.3165			-.0071	-.0220	-.0312	
.905			-.1313				
.950				-.0513		-.0172	
.955			-.0883				
.965	-.1720						

TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 75

(R81U44)

UPPER WING

ARC11-716 1A14 Q1+T12+S12N25

ALPHA( 4 ) = 4.200 BETA( 1 ) = -0.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	-.0312	.0640	.4361	.6330	.6167	.5477
.020				-.0646	-.1425	-.2510	-.3606
.025			.0014				
.040		.1082					
.045			-.1372				
.050	-.0699			-.3550	-.4531	-.5238	-.6595
.060				-.5727			
.065			-.3611				
.066		.0820					
.094	-.0430			-.6350	-.6948	-.6814	-.6750
.110		.0000					
.163			-.5688				
.177		-.2034					
.193	.0495			-.9116	-1.0630	-.9910	-.9114
.229			-.6817				
.250		-.5857					
.274				-.9672	-.9126		-.6575
.339	-.0918			-.8459			
.400				-.2347	-.5817		
.402	.0000						
.497							
.550							
.565							
.600							
.650							
.700	-.1069			-.1092		-.4127	-.4514
.725					-.2333		
.750				-.0476	-.0093	-.2472	
.760				.0768			
.775							
.808	.0061						
.834				.0207	-.1285	-.3260	
.850							
.857				.0602			
.865	.1270						
.900	.0937			.0313			-.3921
.915				.0527	-.0169		
.950				.0068		-.2100	
.953				.0138			
.965	.0111						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81044)

UPPER WING

ARC11-716 1A14 01+712+S12N25

ALPHA( 4 ) = 4.190 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0674	.0298	.3938	.5979	.5617	.4906	.4576
.020				-.0314	-.1189	-.2335	-.3562
.025			.0249				
.040		.1000					
.045			-.1023				
.050	-.0723			-.3079	-.4104	-.5008	-.6404
.080				-.5215			
.085			-.2982				
.086		.0329					
.094	-.0484						
.150				-.7416	-.8293	-.8265	-.8607
.163		.0000					
.177			-.5114				
.193		-.1717					
.229	.0451						
.250				-.8233	-.9781	-1.0360	-.9735
.274			-.6175				
.339	-.5437						
.362	-.0794						
.400				-.9007	-1.0700		-.7172
.402			-.7832				
.497	.0000			-.3324	-.4677		
.550			.0000				
.585						-.5965	-.5486
.600							
.650					-.2048		
.700	-.1872			.1235			
.725						-.4392	-.4576
.750			-.0731				
.760				.0232	-.1187		
.775			.0372				
.806							
.834	-.0386			.0406	-.0488	-.2736	
.850			.0409				
.857							
.865	.0739						-.2328
.900	.0361			.5370			
.905		.0256		.0404			
.930			.0169			-.1464	
.950		-.0507					
.953							
.965	-.0549						

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81J44)

UPPER WING

ARC11-716 1A14 01+112+512N25

ALPHA( 4 ) = 4.060 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7800
.8870		
.000	-.1270	.3433
.020	.0205	.5370
.025	.0349	.4930
.040	.0867	.4234
.045	.0699	.3973
.050	-.0875	-.0014
.060		-.1093
.065		-.2222
.085	-.2276	-.3320
.095	.0837	
.094	-.0500	-.2613
.150		-.360E
.163	.0000	-.4739
.177	-.4421	-.6190
.193	-.1444	-.4622
.229	.0433	
.250		-.6691
.274	-.5459	-.7838
.339	-.4909	-.7773
.362		-.8227
.400	-.0844	
.402		-.7421
.497	.0000	-.8849
.550		-.9790
.565		-.9956
.600		
.650		-.8016
.700	-.2202	-.9863
.725		-.9052
.750		
.760		-.4054
.775		-.5212
.808		.0050
.84	-.2629	-.6134
.850		-.5189
.857		-.1957
.865		-.2082
.900		-.3348
.905		-.4576
.933		
.955		-.2760
		.0115
		-.0424
		-.0560
		.0215
		.0145
		-.1307
		-.0005
		.0266
		.0202
		-.3135
		.0016
		-.0243
		-.0322
		-.0447



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 3111

(R81044)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA(4) = 4.100 BETA(4) = 4.110

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .5640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2258	-.1144	.2425	.4556	.4194	.3562	.3423
.020				-.0032	-.1135	-.2122	-.3354
.025		.0377					
.040							
.045							
.050	-.1144			-.2471	-.3646	-.4531	-.5940
.060				-.4116			
.065							
.066							
.094	-.0716		.0445	-.2122			
.150							
.163			.0000	-.6057	-.6938	-.7380	-.7856
.177							
.193			-.1226				
.229	.0159						
.250				-.6607	-.7763	-.8890	-.9826
.274				-.4922			
.339			-.4287				
.362	-.0603						
.400				-.6232	-.8433		-1.1210
.402			-.5614				
.497	.0000			-.4418	-.6044		
.550							
.565			.0000				
.600							
.650							
.700	-.2613						
.725				-.4061	-.2523		
.750							
.760			-.3152				
.775				-.2361	-.0656		
.800			-.1923				
.834	-.3393						
.850				-.0617	.0027	.0080	
.857			-.0906				
.865	-.3969						
.900	-.1360			-.0324			-.1824
.905				-.0587	-.0075		
.950				-.0205		.0018	
.955			-.0483				
.955	-.0732						

DATE 10 JAN 75

UPPER WING

ARC11-716 1A14 01+Y12+S12N25

$$\text{ALPHA}(4) = 4.109 \quad \text{BETA}(5) = 0.190$$

SECTION (1) UPPER WING	DEPENDENT VARIABLE C/F
1	1.00
2	1.00
3	1.00
4	1.00
5	1.00
6	1.00
7	1.00
8	1.00
9	1.00
10	1.00
11	1.00
12	1.00
13	1.00
14	1.00
15	1.00
16	1.00
17	1.00
18	1.00
19	1.00
20	1.00
21	1.00
22	1.00
23	1.00
24	1.00
25	1.00
26	1.00
27	1.00
28	1.00
29	1.00
30	1.00
31	1.00
32	1.00
33	1.00
34	1.00
35	1.00
36	1.00
37	1.00
38	1.00
39	1.00
40	1.00
41	1.00
42	1.00
43	1.00
44	1.00
45	1.00
46	1.00
47	1.00
48	1.00
49	1.00
50	1.00
51	1.00
52	1.00
53	1.00
54	1.00
55	1.00
56	1.00
57	1.00
58	1.00
59	1.00
60	1.00
61	1.00
62	1.00
63	1.00
64	1.00
65	1.00
66	1.00
67	1.00
68	1.00
69	1.00
70	1.00
71	1.00
72	1.00
73	1.00
74	1.00
75	1.00
76	1.00
77	1.00
78	1.00
79	1.00
80	1.00
81	1.00
82	1.00
83	1.00
84	1.00
85	1.00
86	1.00
87	1.00
88	1.00
89	1.00
90	1.00
91	1.00
92	1.00
93	1.00
94	1.00
95	1.00
96	1.00
97	1.00
98	1.00
99	1.00
100	1.00

Y/8W	.2990	.3640	.4270	.5340	.6730	.7890	.8870
------	-------	-------	-------	-------	-------	-------	-------

**MJ/X**

.020	-.3657	-.0960	.0378	.2462	.3731	.3027	.3078
.020				-.0421	-.0623	-.1374	-.2664
.025			-.0567				
.040		-.0348					
.045			-.0899				
.050	-.2054			-.2010	-.2769	-.3598	-.4936
.080			-.3118				
.085			-.1606				
.086		-.0222					
.094	-.2480			-.4460	-.5646	-.6217	-.6928
.150		.0000					
.163			-.2770				
.177		-.1168					
.193							
.229	.0286		-.3616				
.250		-.3310		-.5206	-.6726	-.7644	-.8620
.274							
.339							
.362	-.0342						
.400			-.4140	-.4602	-.6352		-1.0220
.402							
.497	.0000			-.4473	-.6010		
.550			.0000				
.565							
.600							
.650							
.700	-.2695			-.3799	-.2827	-.6938	-.6722
.725							
.750			-.3323			-.1683	-.1722
.760				-.2808	-.1779		
.775			-.2485				
.808							
.834	-.3818						
.850			-.1642	-.1110	-.0466	-.0235	
.857							
.865	-.3726						
.900	-.2171			-.0619	.0154		-.0289
.935			-.1002				
.950			-.0472			.0160	
.953			-.0642				
.967	-.1309						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 ARC11-716 1A14 01-712-512N25 UPPER WING (R81U44)

ALPHAO( 5) = 0.040 BETAO( 1) = -0.080

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.4340 .6730 .7800 .8870
X/CW				
.000	-.0806	-.0237	.4091	.6392 .5582 .4569 .3709
.020				-.2931 -.3916 -.5501 -.7008
.025				-.1484
.040		.0309		
.045			-.3214	
.050	-.1299			-.5622 -.6851 -.7939 -.9370
.080				-.7687
.085			-.5140	
.086		-.0025		
.094	-.0916			
.150		.0000		-1.0510 -1.0800 -.9161 -.9744
.163				
.177			-.6939	
.193		-.3138		
.229	.0279			
.250			-1.0900 -1.1120 -.7489 -.7986	
.274			-.8077	
.339		-.6717		
.362	-.1345			
.400				
.402		-.9542		-.8163 -.6878 -.5749
.497	.0000			
.550			.0000	-.5328 -.6991
.565				
.600				
.650				-.5506
.700	-.1278			-.5944
.725			-.3804	-.2470
.750				-.4029 -.4819
.760			-.1729	
.775			.0047	-.3171 -.4866
.808				
.834	-.0846			
.850			-.1827	-.4188 -.4450
.857			-.0163	
.865	.0285			
.900	.0245		-.1040	
.905			.0038	-.3244
.950			-.0496	-.3908
.955			-.0161	
.965	-.0215			



(RBIU44)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+112+312N25

UPPER WING

ALPHA( 5) = 7.980 BETA( 2) = -4.020

DEPENDENT VARIABLE CP

SECTION 1 (1) UPPER WING

Y/SW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1498	-.0572	.3543	.4820	.5182	.4070	.3276
.020				-.335	-.3581	-.5389	-.7006
.025				-.1243			
.040		.0034					
.045			.2612				
.050	-.1561		-.5150	-.6459	-.7704	-.9327	
.080			-.7114				
.085			-.4731				
.086		-.0043					
.094	-.1194			-.9451	-1.0450	-.9914	-1.0120
.150		.0000					
.163			-.6273				
.177		-.2744					
.193							
.229	.0116			-1.0040	-1.1960	-.8396	-.9314
.250			-.7475				
.274		-.6320					
.339	-.1304						
.352			-.9010	-1.0320	-.7675		-.6333
.400							
.402				-.3311	-.7029		
.497	.0000						
.550			.0000				
.565							-.5610
.600						-.5843	
.650							
.700	-.1808			-.3343	-.2314		
.725							
.750			-.1828	-.2492	-.4702		
.760							
.775			-.0388				
.808							
.814	-.1150						
.850				-.1396	-.3678	-.4487	
.857			-.0405				
.865	-.0001						
.900	.0024		-.0790				-.9020
.905			-.0244				
.910				-.2267			
.953			-.0452				-.4008
.965	-.0281		-.0405				



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 3115

(R810441)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHAD( 5) = 7.970 BETAD ( 3) = .020

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8070
X/CW							
.000	-.2165	-.1320	.2946	.5284	.4494	.3388	.2713
.020				-.1879	-.3324	-.5368	-.7036
.025			-.0991				
.040		.0068					
.045			-.2190				
.050	-.1644			-.4532	-.6141	-.7558	-.9326
.080				-.8463			
.085			-.4012				
.086		.0066					
.094	-.1119						
.150		.0000		-.8606	-.9938	-1.0350	-1.0350
.163							
.177			-.3768				
.193		-.2341					
.229	.0092			-.8987	-1.1330	-.9765	-.9569
.250							
.274			-.6816				
.339		-.5868					
.362	-.1162			-.9624	-.9729		-.6452
.400							
.402		-.8267					
.497	.0000			-.3352	-.6743		
.550			.0000				
.565							
.600							
.650							
.700	-.2454				-.5789		-.6043
.725				-.2725	-.2286		
.750							
.760		-.1891		-.1538	-.4109		
.775							
.808		-.0576					
.834	-.1608			-.0998	-.2915	-.4092	
.850							
.857		-.0534					
.865	-.0332						
.900	-.0137			-.0680			-.5220
.905			-.0342		-.1498		
.950				-.0556			-.3404
.955			-.0296				
.965	-.0396						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01044)

UPPER WING

ARC11-716 1A14 01-712-512N25

ALPHAO ( 5 ) = 0.020 BETA0 ( 4 ) = 4.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.003	-.3836	-.2690	.1614	.4384	.4143	.2804	.1995
.010				-.1854	-.3027	-.5124	-.7052
.025			-.1438				
.040		-.0416					
.045			-.2277				
.050	-.1970			-.4254	-.5776	-.7157	-.9336
.060				-.5991			
.085			-.3641				
.086		-.0288					
.094	-.1489						
.157				-.7723	-.9320	-.9865	-1.0490
.163		.0000					
.177			-.5146				
.193		-.2090					
.229	-.0319						
.250				-.8054	-1.0310	-1.0580	-1.0300
.274		-.5210	-.6046				
.339							
.362	-.1163			-.8371	-1.0700		-.6816
.400							
.402			-.7503				
.497	.0000						
.550				-.4332	-.5425		
.565			.0000				
.600							
.650							
.700	-.3159			-.2779	-.2290	-.3303	-.6079
.725							
.750							
.760			-.2293				
.775				-.1291	-.2426		
.800			-.0744				
.834	-.2426						
.850				-.0777	-.1673	-.3186	
.857			-.0605				
.865	-.0840						
.900	-.0470			-.0496			-.5064
.905			-.0444		-.0782		
.940				-.0397		-.2546	
.955			-.0419				
.965	-.0575						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1044)

UPPER WING

ARC11-716 IA14 OI+T12+S12N25

ALPHA(5) = 3.080 BETA(5) = 8.230

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7000 .8673

X/C

.053	-.5469	-.1717	-.0817	.2641	.4017	.2927	.1746
.020				-.2736	-.2712	-.4340	-.6604
.025			-.1888				
.040		-.1098					
.045			-.2144				
.050	-.3017			-.3952	-.5233	-.6612	-.8856
.063				-.5283			
.085			-.2679				
.086		-.0946					
.094	-.3060			-.6378	-.8458	-.9035	-1.0090
.150		.0000					
.163							
.177			-.3854				
.193		-.2141					
.229	-.1037			-.6797	-.8846	-1.0480	-1.1120
.250							
.274			-.4709				
.339		-.4366					
.362	-.1074			-.5854	-.8433		-.9673
.430			-.4994				
.402							
.497	.0000			-.4911	-.6509		
.550			.0000				-.6267
.565							
.600							
.650							
.700	-.3070			-.2367		-.3719	
.725				-.2660			
.750							
.760			-.2685				
.775			-.1545	-.1318	-.0592		
.808							
.834	-.2519			-.0619	-.0384	-.1317	
.850			-.1319				
.857							
.855	-.1782						
.900	-.1230			-.0423	.0099		-.3523
.915			-.0850				
.950				-.0400			-.0449
.953			-.0748				
.965	-.0915						

(RB1U43) (20 FEB 74)

UPPER WING

ARC11-716 IA14 Q8+T18+312N25

## REFERENCE DATA

SREF = 2.4210 30 FT. XMRP = 29.5000 INCHES  
 LREF = 36.7090 INCHES YMRP = .0000 INCHES  
 SREF = 36.7090 INCHES ZMRP = .0000 INCHES  
 SCALE = .0300 SCALE

ALPHA(1) = -7.870 BETA(1) = -8.040

## PARAMETRIC DATA

MACH = .950 ELEVON = .000  
 RUDDER = .000 SPOBRK = .000

## DEPENDENT VARIABLE CP

## SECTION (1) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870	
X/CW	.000	.0723	.1209	.3829	.4180	.5135	.2134	.2461
.020				.5480	.5136	.5350	.5106	
.025			.2714	.4495				
.040				.3827				
.045				.3302	.3404	.3485	.3006	
.050		.1063		.1609				
.060			.2466					
.065			.3289					
.086		.1181						
.094			.0020					
.150			.0043					
.163			.1978					
.177								
.193		.1997						
.229								
.250								
.274								
.339								
.362		.1873						
.400								
.402								
.497		.0000						
.550								
.565								
.600								
.650								
.700		.0905						
.725								
.750								
.760								
.775								
.806								
.834		.0339						
.850								
.857								
.865		.1748						
.900		.1404						
.905								



DATE 10 JAN 74 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBLU49)

UPPER WING

ARC11-716 1A14 01+T12+312N23

ALPHA( 1 ) = -7.873 BETA( 1 ) = -8.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW  
.930 .0612 .0987  
.933 .0431  
.935 .0344

ALPHA( 1 ) = -7.773 BETA( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW  
.000 .0651 .0865 .2904 .3197 .1975 .0792 .1110  
.020 .029 .025 .4317 .5384 .4814 .4980 .4580  
.040 .2381 .3922 .3461 .3434 .3423 .2903  
.045 .1057 .2697 .1959  
.060 .3110  
.085 .1200  
.086 .0000 .0468  
.094 .2152  
.130 .1960  
.163 .177  
.177 .0715  
.193 .0475  
.229 .1980  
.250 .1451 .2414 .4454  
.274 .1321  
.339 .0000  
.362 .1445 .2781  
.400 .0000  
.402 .0000  
.497 .0000  
.530 .0000  
.565 .0000  
.600 .0000  
.650 .0000  
.700 .0000  
.725 .0000  
.750 .0000  
.775 .0000  
.808 .0000  
.834 .0000  
.850 .0000

TABULATED PRESSURE DATA - IA14A - VOL. 6

DATE 19 JAN 75

(RB1U45)

UPPER WING

ARC11-716 IA14 01+712+512N25

ALPHA( 1 ) = -7.770 BETA( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.057 .0809

.065

.0902

.0781

.903

.935

.930

.953

.965

-.0070

.1106

.0914

.0633

.0679

.0388

.0137

.0825

ALPHA( 1 ) = -7.760 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.090

.020

.025

.040

.045

.050

.060

.085

.086

.094

.150

.163

.177

.193

.229

.250

.274

.339

.362

.400

.402

.497

.550

.563

.650

.700

.725

.790

.780

.0464

.0653

.2360

.2184

.5131

.4015

.2109

.3671

.3508

.2196

.2785

.2888

.1228

.0050

.0865

.2287

.1933

.0110

.2093

.400

.0050

.497

.550

.563

.650

.700

.725

.790

.780

.2184

.0640

-.0731

-.0412

.4394

.3920

.3342

.3205

.2648

.0238

.0247

.0368

-.0304

-.1241

-.1533

-.1850

-.2897

-.0283

-.1389

-.2319

-.4321

-.1208

-.1767

-.2984

.0000

-.4848

-.4285

-.2397

-.2699

-.1402

-.0984

-.3008

-.4848

-.4285

-.2397

-.2699

-.1402

-.0984

-.3008

**UPPER WING**

ARC11-716 1A14 C1+Y12+S12N25

SECTION ( 1 ) UPPER WING

ALPHA( 1 ) = -7.900 BETA( 4 ) = 4.110

SECTION ( ) UPPER WING	DEPENDENT VARIABLE CP
1	0.000
2	0.000
3	0.000
4	0.000
5	0.000
6	0.000
7	0.000
8	0.000
9	0.000
10	0.000
11	0.000
12	0.000
13	0.000
14	0.000
15	0.000
16	0.000
17	0.000
18	0.000
19	0.000
20	0.000
21	0.000
22	0.000
23	0.000
24	0.000
25	0.000
26	0.000
27	0.000
28	0.000
29	0.000
30	0.000
31	0.000
32	0.000
33	0.000
34	0.000
35	0.000
36	0.000
37	0.000
38	0.000
39	0.000
40	0.000
41	0.000
42	0.000
43	0.000
44	0.000
45	0.000
46	0.000
47	0.000
48	0.000
49	0.000
50	0.000
51	0.000
52	0.000
53	0.000
54	0.000
55	0.000
56	0.000
57	0.000
58	0.000
59	0.000
60	0.000
61	0.000
62	0.000
63	0.000
64	0.000
65	0.000
66	0.000
67	0.000
68	0.000
69	0.000
70	0.000
71	0.000
72	0.000
73	0.000
74	0.000
75	0.000
76	0.000
77	0.000
78	0.000
79	0.000
80	0.000
81	0.000
82	0.000
83	0.000
84	0.000
85	0.000
86	0.000
87	0.000
88	0.000
89	0.000
90	0.000
91	0.000
92	0.000
93	0.000
94	0.000
95	0.000
96	0.000
97	0.000
98	0.000
99	0.000
100	0.000

Y/W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
K/CW							
.000	-.0939	.0429	.2013	.1373	-.0412	-.1839	-.1661
.020				.4915	.3957	.3868	.3259
.025			.3537				
.040		.1704					
.045			.3290				
.050	-.0100			.3491	.3232	.2940	.2274
.060				.2340			
.065			.2571				
.066		.2407					
.094	.0422						
.120		.0000		.0506	.0377	.0391	-.0407
.163			.1020				
.177							
.193		.2102					
.229	.1606						
.250				-.0957	-.1319	-.1668	-.2850
.274			.0039				
.339		.0165					
.362	.1818						
.400			-.1067	-.1424	-.2318		-.4385
.497	.0000			-.2119	-.3179		
.550			.0000				
.565							
.600							
.650						-.4555	-.5266



(R81U45)

UPPER WING

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+T12+S12N25

ALPHA( 1 ) = -7.900 BETA( 4 ) = 4.110

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.700	-.0619			-.3074	-.2490		
.725						-.4026	-.3406
.750							
.760							
.775							
.808							
.834							
.850							
.857							
.865							
.900							
.925							
.930							
.953							
.965							

X/CW

.700

.725

.750

.760

.775

.808

.834

.850

.857

.865

.900

.925

.930

.953

.965

ALPHA( 1 ) = -7.930 BETA( 5 ) = 8.180

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2331	.0069	.1569	.0990	-.1151	-.2854	-.2750
.020				.4337	.3710	.3513	.2753
.025							
.040							
.045							
.050							
.060							
.085							
.094							
.130							
.163							
.177							
.193							
.229							
.250							
.274							
.339							
.362							
.400							
.402							
.497							

X/CW

.000

.020

.025

.040

.045

.050

.060

.085

.094

.130

.163

.177

.193

.229

.250

.274

.339

.362

.400

.402

.497



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U43)

UPPER WING

ARC11-716 1A14 OI+712+312N25

ALPHAO ( 1 ) = -7.930 BETAO ( 5 ) = 8.180

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .8990 .3840 .4270 .3340 .6730 .7800 .8870

X/CW

.590 -.2205 -.3113  
 .565 .0000  
 .600  
 .650 -.4587  
 .700 -.1114  
 .725  
 .750 -.3302  
 .760  
 .775 -.3389  
 .806 -.2744 -.3500  
 .834 -.2170  
 .850  
 .857 -.1928 -.2418 -.2831  
 .865 -.2017  
 .900 -.1440  
 .905 -.1769 -.0969  
 .950 -.1306  
 .913 -.0752  
 .965 -.1699  
 .2001

ALPHAO ( 2 ) = -4.020 BETAO ( 1 ) = -8.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3840 .4270 .3340 .6730 .7800 .8870

X/CW

.000 .0620 .1892 .4649 .5615 .4950 .4371 .4676  
 .020 .3986 .3680 .3794 .3644  
 .025 .3471  
 .040 .2645  
 .045 .2479  
 .050 .0617  
 .060 .1416 .1216 .1298 .0837  
 .065 -.0387  
 .085 .0745  
 .086 .2761  
 .094 .0758  
 .150 .0000  
 .163 -.1726  
 .177 .0830  
 .193 .1593  
 .229  
 .230  
 .274  
 .339  
 -.4203 -.4823 -.5084 -.5373  
 -.2801  
 -.2634

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1U45)

UPPER WING

ARC11-716 1A14 ON+T12+312N25

ALPHA( 2 ) = -4.020 BETA( 1 ) = -8.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.362	.1021					
	.400			-.3166	-.6229		-.7740
	.402		-.3041				
	.497	.0000					
	.550			-.1629	-.3314		
	.565		.0000				
	.600						-.7538
	.650						
	.700	.0443			-.3222		
	.725			-.0496	-.1977		
	.750					.2165	.1546
	.760		-.0470	.1674	.1720		
	.775		.1909				
	.808						
	.834	.0054		.1310	.1927	.1722	
	.850						
	.857		.1249				
	.865	.1559					.1286
	.900	.1316		.1147	.0726		
	.905		.0989			.0911	
	.920			.0523			
	.953		.0379				
	.965	.0344					

ALPHA( 2 ) = -3.960 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.000	.0444	.1943	.4025	.4664	.3807	.3059
	.080				.4062	.3566	.3711
	.025		.3449				.3495
	.040	.2407					
	.045		.2619		.1672	.1563	.1501
	.050	.0639			-.0146		.0949
	.060			.1123			
	.065						
	.066	.2660					
	.094	.0802					
	.159				-.2367	-.2307	-.2171
	.163		.0000				-.2406
	.177		-.1256				
	.193	.1039					



TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 75

(R81U45)

UPPER WING

ARC11-716 1A14 01+112+512N25

ALPHA( 2 ) = -3.900 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.229	.1962							
.290								
.274								
.339								
.362	.1146							
.400								
.402								
.497	.0000							
.550								
.565								
.600								
.650								
.700								
.725								
.750								
.760								
.775								
.808								
.834								
.850								
.857								
.865	.0337							
.900	.0455							
.935								
.950								
.953								
.965								

ALPHA( 2 ) = -3.870 BETA( 3 ) = .010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.000	.0259							
.020								
.025								
.040								
.045								
.050								
.060								
.065								
.066								
.094								



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+512N25 (R81045)

UPPER WING

ALPHA( 2 ) = -3.940 BETA( 4 ) = 4.590

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0609					
.005	.1306					
.006	.2110					
.094	.0322					
.150						
.163	.0000					
.177	-.0320					
.193	.1227					
.229	.1336					
.250						
.274						
.339	-.1025					
.362	.1200					
.400						
.402	-.2163					
.497	.0000					
.550						
.565						
.600						
.650						
.700	-.0561					
.725						
.750						
.760						
.775						
.808						
.834	-.3094					
.850						
.897						
.865	-.2663					
.905	-.2403					
.905						
.950						
.953						
.965	-.1994					

-.1146

-.1327

-.1252

-.1984

-.2470

-.2997

-.3349

-.4425

-.2361

-.3265

-.5321

-.2574

-.3022

-.5160

-.2660

-.3325

-.4460

-.3841

-.3592

-.4026

-.3007

-.2660

-.2746

-.3255

-.2382

-.1660

-.1041

-.1177

-.1243

-.1546

-.1397

TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 75

(R81045)

UPPER WING

ARC11-716 1A14 01+712+812M25

ALPHA(2) = -3.950 BETA(3) = 0.170

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

V/W	2990	3640	4270	5340	6730	7800	8870
X/CW							
.000	-.2803	.0179	.1301	.2591	.0850	-.0461	.0069
.025				.3625	.2948	.3064	.2589
.050			.2454				
.075		.0310	.2253				
.100	-.1577			.2021	.1927	.1742	.1106
.125				.0843			
.150			.1479				
.175		.1446					
.200	-.0666			-.0691	-.0637	-.0856	-.1650
.225		.0000	.0229				
.250		.1169					
.275	.1000		-.0790	-.1695	-.2366	-.2813	-.3993
.300		-.0606					
.325	.1011			-.2085	-.3094		-.5242
.350			-.1737				
.375	.0000			-.2721	-.3783		
.400			.0000				-.3669
.425						-.5221	
.450	-.1363			-.3725			
.475					-.2729		
.500			-.3616			-.4744	-.4483
.525			-.3079		-.3591	-.4247	
.550				-.2900	-.3635	-.3402	
.575	-.3131		-.2774				
.600				-.2306			-.2065
.625	-.2839			-.2283	-.1802		
.650	-.2639			-.1637		-.1549	
.675			-.1883				
.700							
.725							
.750							
.775							
.800							
.825							
.850							
.875							
.900							
.925							
.950							
.975							
.995							



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81045)

UPPER WING

ARC11-716 1A14 OR-T112-S12N25

ALPHA( 3 ) = -.300 BETA( 1 ) = -.6100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

X/CW	Y/BW	.2990	.3840	.4270	.5340	.6730	.7800	.8870
.000	-.0198	.1823	.5021	.6633	.6259	.5869	.5991	
.020				.5333	.1828	.1487	.1062	
.025		.2263						
.040		.2310						
.045			.1041					
.050	-.0058			-.0432	-.1062	-.1367	-.2079	
.060				-.2614				
.065			-.0808					
.066		.2176						
.094	.0383							
.150		.0000		-.5115	-.5643	-.5187	-.4973	
.163								
.177		-.3809						
.193		-.0806						
.229	.1317							
.250				-.5989	-.7084	-.7389	-.7280	
.274		-.4397						
.339		-.3791						
.362	.0401			-.7133	-.8584		-.9479	
.400			-.6145					
.402	.0000		.0000	-.2643	-.5120			
.497								
.550								
.565								
.600								
.650								
.700	-.0279			-.0572	-.2149	-.5149	-.6497	
.725								
.750								
.760			-.0459	.1434	.1189			
.775			.1361					
.908								
.934	-.0006							
.950				.1244	.1284	.1278		
.957			.1183					
.965	.1472							
.970	.1284			.1065	.0504		-.1697	
.975			.0948					
.979				.0442		.1232		
.983			.0380					
.985	.0383							



(881043)

UPPER WING

ARC11-716 1A14 OL+T12-S12N25

ALPHA( 3 ) = -.320 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.000	-.0091	.1558
.020		.4545
.025		.5889
.040	.2404	.5403
.045	.2184	.4956
.050		.4697
.060		.1839
.085		.1145
.086		
.094	.2169	
.150		
.165	.0000	
.177		
.193	.0073	
.229	.1272	
.255		
.274		
.339	-.3364	
.362	.0516	
.400		
.402	-.5300	
.497	.0000	
.550		
.565		
.600		
.650		
.700	-.0478	
.725		
.750		
.760		
.775		
.800		
.834	-.2482	
.850		
.857		
.865	-.1059	
.900	-.0133	
.905		
.950		
.955		
.965		

(R81U48)

UPPER WING

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 O1+T12+S12N25

ALPHA (3) = -.330 BETA (3) = .040

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0516	.1185	.3937	.5974	.4379	.3933	.4295
.020				.2612	.1823	.1726	.1274
.025			.2294				
.040		.1890					
.045			.1386				
.050	.0124			.0166	-.0355	-.0719	-.1561
.080			-.0042	-.1653			
.085							
.086		.2024					
.094	.0420			-.3599	-.4036	-.4001	-.4316
.150		.0000					
.163			-.2172				
.177		.0266					
.193				-.4680	-.5485	-.5899	-.6487
.229	.1222		-.3267				
.250							
.274		-.2976					
.339				-.4640	-.6168		-.8573
.362	.0616						
.400			-.4277				
.432				-.2582	-.4119		
.497	.0000		.0000				-.8187
.550					-.5130		
.565				-.3363	-.2804		
.630							
.650	-.0799						
.700							
.725							
.750							
.760			-.3424				
.775			-.3041				
.808				-.3236	-.3591		
.834							
.850	-.2678			-.2936	-.2057	-.1290	
.857			-.0082				
.865	-.2603						-.0613
.900	-.2695			-.1534	-.0946		
.935			-.1946				-.0944
.950				-.0960			
.955			-.1169				
.965	-.1626						

ORIGINAL PAGE IS  
OF POOR QUALITY

(R81049)

UPPER WING

ALPHA( 3 ) = -.330 BETA( 4 ) = 4.090

ARC11-716 1A14 01+712+512N25

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3840	.4270	.5340	.5730	.7600	.8670
X/CW							
.000	-.1625	.0493	.3175	.4235	.3428	.2926	.3356
.020				.2390	.1620	.1588	.1133
.025			.1852				
.040		.1319					
.050			.1174	.0201	-.0180	-.0448	-.1424
.060	-.0400			-.1377			
.085			.0066				
.086		.1559					
.094	.0087			-.2928	-.3301	-.3450	-.4097
.150							
.163		.0000					
.177			-.1832				
.193		.0316					
.229	.1035			-.4049	-.4968	-.5502	-.6301
.250							
.274		-.2481					
.339		-.2368					
.362	.0599			-.3673	-.4869		-.8039
.400			-.3464				
.402							
.497	.0000			-.2881	-.4266		-.5724
.565			.0000				
.600							
.650					-.5437		
.700	-.1275			-.3700			
.725							
.730							
.760			-.3862	-.3704	-.4144		
.775							
.808			-.3526				
.834	-.3193			-.3620	-.3411	-.2275	
.850							
.857			-.3597				
.865	-.3095						-.1416
.900	-.3224			-.3102			
.905			-.3077	-.1052			
.950				-.1924		-.1525	
.953			-.2319				
.965	-.2925						



DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81045)

UPPER WING

ARC11-716 1A14 01-712-512N25

ALPHA(3) = -.330 BETA(5) = 0.160

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8970

X/CW

.000	-.3626	-.0186	.1162	.2914	.2651	.1793	.2310
.020				.2263	.1652	.1784	.1295
.025			.0818				
.040		.0184					
.045			.0563				
.050	-.2010			.0556	.0257	-.0051	-.0924
.060				-.0689			
.065			.0173				
.066		.0054					
.094	-.1379			-.2203	-.2637	-.2638	-.3512
.190		.0000					
.163			-.0976				
.177		.0544					
.193							
.229	.0876			-.3153	-.4048	-.4594	-.5671
.250			-.1839				
.274		-.1659					
.339							
.362	.0475			-.2977	-.4037		-.6447
.400			-.2614				
.402				-.3084	-.4333		
.497	.0000		.0000				-.6167
.590					-.5677		
.565				-.2910			
.600				-.3973			
.650	-.1632				-.4909	-.4249	
.703							
.725							
.750							
.760			-.3954				
.775			-.3841				
.806							
.834	-.3441			-.3693	-.4160	-.3153	
.850							
.857			-.3757				
.865	-.3491						-.8126
.903	-.3507			-.3205	-.1944		
.925			-.3103				
.950				-.2073	-.1960		
.955			-.2190				
.965	-.2805						

ORIGINAL PAGE IS  
OF POOR QUALITY

(R81U45)

UPPER WING

ALPHA( 4) = 4.170 BETA( 1) = -0.100

ARC11-716 1A14 OI+T12+312N25

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1000	.0913	.4718	.6780	.6485	.5866	.5499
.020				-.0002	-.0677	-.1754	-.2730
.025		.1501	.0547				
.040							
.045			-.0853				
.050	-.1109			-.2792	-.3659	-.4314	-.5616
.080				-.4822			
.085			-.2970				
.086		.1321					
.094	-.0459						
.150				-.7258	-.7906	-.7780	-.7768
.163		.0000					
.177			-.4745				
.193		-.1322					
.229	.1027			-.8278	-.9516	-.9849	-.8941
.274		-.5921					
.339		-.4894					
.362	-.0209			-.8869	-1.0670		-.6871
.400			-.7617				
.402				-.3964	-.6947		
.497	.0000		.0000				-.6127
.550							
.565				-.1322	-.2645		
.600						-.4945	-.5386
.650	-.2513						
.700			-.0324	-.0268	-.4051		
.725			.1294				
.750							
.760							
.775							
.808							
.834	-.0147			.0281	-.2997	-.4277	
.850			.0990				
.857							
.865	.1377			.0376			-.4912
.900	.1155						
.929		.0780		-.1402			
.950			.0100				-.3700
.953			.0225				
.965	.0264						



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 3135

(R81049)

UPPER WING

ALPHA( 4 ) = 4.230 BETA( 2 ) = -4.030

ARC11-716 1A14 Q1+T12+S12N25

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2146	.0456	.4346	.6348	.9932	.9294	.5028
.020				.0376	-.0423	-.1494	-.2584
.025			.0763				
.040		.1342					
.045			-.0402				
.050	-.0920			-.2516	-.3185	-.3983	-.3299
.080				-.4333			
.085			-.2442				
.086		.1237					
.094	-.0258						
.150		.0000		-.6929	-.7215	-.7167	-.7467
.183							
.177		-.4320					
.193		-.1116					
.229	.0889			-.7255	-.8792	-.9142	-.9084
.250							
.274		-.5356					
.339		-.4615					
.362	-.0195			-.8138	-.9919		-1.0440
.400							
.402		-.7076					
.497	.0000			-.4414	-.9687		
.550		.0000					
.565							
.600							
.650							
.700	-.1897				-.7145		
.725				-.3045	-.2792		
.750							
.780		-.3258				-.5400	-.5999
.775				-.1405	-.2105		
.806		-.1892					
.834	-.2581						
.850				-.0457	-.1112	-.4692	
.857		-.0434					
.865	-.2279						
.900	-.0732			-.0148			-.5791
.905		-.0150			-.0871		
.950				-.0418		-.3668	
.953		-.0427					
.965	-.0336						

(R81U45)

UPPER WING

ARC11-716 IA14 CR112+312M25

ALPHA(1) = 4.150 BETA(3) = .010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/W	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.1824	-.0044	-.365	.5663	.5280	.4605	.4386
.020			.0720	.0495	-.0407	-.1493	-.2587
.025		.1145					
.040			-.0359				
.045							
.050	-.0809		-.2006	-.3066	-.3891	-.5208	
.060			-.3950				
.085			-.1874				
.086		.1168					
.094	-.0247						
.150		.0000		-.5921	-.6982	-.6850	-.7262
.163			-.3763				
.177		-.0698					
.193							
.229	.0819			-.6627	-.8008	-.8833	-.8926
.230							
.274		-.4164					
.339							
.362	-.0084			-.7479	-.9092		-1.0720
.400			-.6430				
.452	.0000						
.497			.0000	-.3603	-.6267		
.550							
.565							
.600							
.653							
.700	-.1708						
.725				-.2033			
.750			-.3610				
.760							
.775			-.3529	-.3930	-.2747		
.808			-.3392				
.834	-.3001						
.850				-.2565	-.1366	-.0877	
.857			-.3385				
.865	-.3043						
.900	-.3156			-.1599			-.5577
.905			-.2299		-.1216		
.950				-.1555		-.0993	
.955			-.1532				
.965	-.1838						



ADMIN ATEC PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 73

ARC11-716 1A14 01+T12+S12N25

ALPHA( 4) = 4.140 BETA( 4) = 4.110

SECTION 1 RUBBER WING

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/84	.2990	.3640	.4270	.5340	.6730	.7800	.8670
------	-------	-------	-------	-------	-------	-------	-------

153/18

1000	-3116	-0050	.2462	.4028	.4496	.3940	.3863
.020				.0447	-.0320	-.1425	-.2590
							.0361

0613

0.00	-0.0410	-0.1836	-0.2952	-0.3732	-0.4998
0.05	-0.1159	-0.3543			
0.10					

-1505

.063			
.006	.0750		
.094	-.0705	- .512	- .6145
			- .6516
			- .6993

...

.163	.0000
.177	-.3289
.193	-.3669

[illegible][illegible][illegible]

.790	-.1799		-.2760
.725		-.3972	
.750		-.3498	-.4761

**-.377**

775	- .4367	- .4266
.026	- .3901	
274		2.1195

.050	- .4030	- .1594
.057		
.065	- .3339	

3

.929	-.3623	-.1772	-.1566
.953			
.933	-.2643		

100



(R01045)

UPPER WING

ALPHA( 3 ) = 0.150 BETA( 1 ) = -0.090

ARC11-716 1A14 01-712+512N25

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.000	-.1200	-.0505
.020	.4203	.6457
.025	-.0989	-.2287
.040	.0908	-.3044
.045	-.2662	-.4592
.050	-.4817	-.5811
.060	-.6663	-.6454
.065	-.4640	-.7859
.066	.0356	-.8243
.094	-.1702	-.9322
.150	.0000	-.9511
.163	-.6217	-.9545
.177	-.2655	-.9454
.193	.0710	-.9322
.229	-.7164	-.9511
.250	-.5832	-.9545
.274	-.0676	-.9454
.339	-.8463	-.7859
.362	-.0000	-.6457
.402	-.3682	-.4592
.497	.0000	-.3044
.550	-.4453	-.2287
.565	-.2652	-.0989
.600	-.4411	.0505
.650	-.0794	.6457
.700	-.1235	.6457
.725	-.0545	.6457
.750	-.0043	.6457
.760	-.0056	.6457
.775	-.0316	.6457
.78	-.0316	.6457
.80	-.0316	.6457
.837	-.0316	.6457
.865	-.0316	.6457
.900	-.0316	.6457
.905	-.0316	.6457
.950	-.0316	.6457
.955	-.0316	.6457
.965	-.0316	.6457



DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1045)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 5 ) = 0.150 BETA( 2 ) = -4.030

## SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .9340 .6730 .7800 .8870
-.2337	-.0934	.3945 .5990 .5382 .4296 .3711
.020		-.1743 -.2819 -.4498 -.5942
.023		-.0657
.040	.0416	
.045		-.2129
.050		-.4310 -.5327 -.6645 -.8368
.060		-.6133
.065		-.3759
.086	.0382	
-.0967		-.8522 -.9204 -.9502 -.9803
.150	.0000	
.163		-.5528
.177	-.2177	
.193		
.229	.0582	
.250		-.8943 -1.0690 -.9331 -.9443
.274	-.6519	
.339	-.3338	
.362		
.400		-.9294 -.9765 -.8434
.452	-.8019	
.497	.0000	
.550		-.4048 -.7744
.565	.0000	
.600		
.630		-.6582
.700	-.4022	
.725		-.2872
.750		-.4358
.780	-.2920	
.775		-.5747 -.6337
.808	-.1301	
.834		
.850		-.2301 -.5092 -.5092
.857	-.1285	
.865		
.900	-.1375	
.905	-.0890	
.905		-.1494 -.4065 -.6338
.930	-.0880	
.953		-.1209 -.4770
.965	-.0858	

(RBLU45)

UP, CR WING

ARC11-716 1A14 01+712+S12N25

ALPHA(1) = 0.145 BETA(1) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2590	.3640	.4270	.5340	.6730	.7890	.8870
X/CW							
.020	-.3029	-.1207	.3345	.5878	.5046	.3778	.3101
.040				-.1002	-.2436	-.4350	-.5966
.060		.0442	-.0399				
.080			-.1552				
.100	-.1554		-.3807	-.5081	-.6382	-.8232	
.120			-.5465				
.140			-.3289				
.160		.0516					
.180	-.0439			-.7577	-.8651	-.9087	-.9655
.200		.0000					
.220		-.1609	-.4747				
.240	.0508						
.260		-.5801		-.8000	-1.0010	-1.0200	-1.0270
.280		-.4820					
.300	-.0329			-.8592	-1.0710		-.9076
.320			-.7388				
.340	.0000			-.4950	-.7639		
.360			.0000				
.380						-.7099	-.6931
.400	-.3371			-.2794			
.420			-.3354			-.5631	-.6742
.440			-.3380	-.2746	-.4970		
.460			-.2458				
.480			-.2558				
.500	-.2073			-.2496	-.3850	-.4983	
.520							
.540							
.560	-.2756			-.1959			-.6799
.580	-.2293			-.1711	-.2767		
.600				-.1607		-.4645	
.620				-.1250			
.640							
.660							
.680							
.700							
.720							
.740							
.760							
.780							
.800							
.820							
.840							
.860							
.880							
.900							
.920							
.940							
.960							
.980							
.995							



DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1045)

UPPER WING

ARC11-716 1A14 01+712+S12N25

ALPHA( 5 ) = 8.140 BETA( 4 ) = 4.150

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.000	-.4329	-.1902	.1664	.5036	.4768	.3537	.2646
.020				-.1030	-.2123	-.4016	-.5936
.025			-.0829				
.040		-.0016					
.045			-.1528				
.050	-.1926			-.3329	-.4771	-.6015	-.8134
.060				-.5068			
.065			-.2663				
.066		.0096					
.094	-.1903						
.190		.0000		-.6754	-.8245	-.8716	-.9437
.163			-.4150				
.177		-.1390					
.229	-.0069						
.250				-.7114	-.8344	-1.0140	-1.0900
.274			-.5097				
.339		-.4280					
.362	-.0592						
.400				-.7828	-.9780		-.9962
.402			-.6346				
.497	.0000						
.500				-.4543	-.6563		
.565			.0000				
.600							-.7408
.650					-.2858		
.700	-.2601			-.3501			
.725							
.750					-.5361	-.7222	
.760			-.3751				
.775				-.3355	-.3115		
.808			-.3830				
.834	-.3595						
.850				-.2860	-.2609	-.4868	
.857			-.3371				
.865	-.3670						-.6854
.900	-.3922			-.2360			
.905			-.2502	-.2389			
.930				-.2006		-.4502	
.933			-.1873				
.965	-.2206						

(R01045)

UPPER WING

ARC11-716 1A14 01-712-912M25

ALPHA2( S) = 0.120 BETA2( S) = 0.270

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CU							
.000	-.5366	-.1069	-.0968	.1185	.4494	.3569	.2595
.020				-.2680	-.1817	-.3340	-.5334
.025			-.1215				
.040		-.0640					
.045			-.1609				
.090	-.3603			-.3668	-.4285	-.5440	-.7310
.080				-.4435			
.085			-.2044				
.086		-.0374					
.094	-.4181			-.5160	-.7299	-.7949	-.8897
.150		.0000					
.163			-.3129				
.177		-.1811					
.193							
.229	-.0309			-.5769	-.7862	-.9401	-1.0010
.250			-.3050				
.274		-.3886					
.339	-.0613			-.4699	-.7281		-1.0400
.362			-.3996				
.400				-.4149	-.5827		
.497	.0000			.0000			
.590						-.7699	
.600				-.2962			
.650	-.2462			-.4307			
.700						-.2527	-.7146
.725			-.4147				
.750				-.4310	-.2880		
.775			-.4453				
.800							
.834	-.3981			-.2793	-.2166	-.2698	
.850			-.3620				
.857							
.865	-.4272						
.900	-.4163			-.2031	-.1789		-.9017
.905			-.2741				
.950				-.1572		-.2411	
.953			-.1867				
.965	-.2253						



(RBU46) ( 15 FEB 74 )

UPPER WING

ARC11-716 1A14 01+12+312N25

PARAMETRIC DATA

MACH = .975 ELEVON = .000  
RUDDER = .000 SPOBRK = .000

REFERENCE DATA

REF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
LREF = 38.7990 INCHES YMRP = .0000 INCHES  
BREF = 38.7990 INCHES ZMRP = .0000 INCHES  
SCALE = .0300 SCALE

ALPMAO ( 1 ) = -7.970 BETAO ( 1 ) = -0.050

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0631	.1391	.4030	.4349	.3518	.2530	.2817
.020				.5701	.5342	.5539	.5318
.025			.4735				
.040		.2990					
.045			.4040				
.050	.0652			.3528	.3644	.3684	.3296
.060				.1870			
.065			.3523	.2626			
.066	.0937						
.094			.0000				
.150			.2270				
.163				-.0390	-.0125	.0194	-.0154
.177							
.193							
.229	.2215						
.250				-.1931	-.2151	-.2387	-.3036
.274							
.339			-.0322				
.32	.2139						
.400				-.1041	-.2179		-.4534
.412					-.1010		
.497	.0000						
.550				-.0596	-.2076		
.565					.0000		
.600							-.3532
.690							
.700	.1323						
.725				-.0105	-.1789		
.750						.2293	.1733
.760				-.0142			
.775					.2162	.2042	
.808			.2007				
.834	.0622						
.850							
.857			.1697		.1727	.1890	.2042
.865	.2071						
.900	.1733				.1598		.1743
.905				.1445	.1206		

ORIGINAL PAGE IS  
OF POOR QUALITY

DATE 10 JAN 75 TABULATED PRESSURE DATA - 7A14A - VOL. 6

(R01U46)

UPPER WING

ALPHAO ( 1 ) = -7.970 BETAO ( 1 ) = -0.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.950

.935

.965 .0734

.1012

.1376

ALPHAO ( 1 ) = -7.960 BETAO ( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.900

.020

.025

.040

.045

.050

.080

.085

.086

.094

.150

.163

.177

.193

.229

.250

.274

.339

.362

.400

.402

.497

.950

.965

.600

.650

.700

.725

.750

.760

.775

.808

.834

.850

.1233

.3323

.3567

.2311

.1127

.1426

.5648

.5081

.5202

.4829

.4616

.2693

.4129

.3751

.3727

.3707

.3166

.2260

.2937

.3387

.0942

.0000

.0868

.2517

.2208

.0290

-.1295

-.1479

-.1850

-.2712

-.0294

-.1058

-.2048

-.4044

-.0890

-.1092

-.2397

.0000

-.4102

-.3555

-.2260

-.1507

.1159

.0773

-.1506

.1703

.1501

.1425

.1320

.1338

.1654







(R81U46)

UPPER WING

ALPHA( 1 ) = -7.890 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.775		
.858		
.834		
.890		
.857		
.865		
.900		
.905		
.930		
.933		
.965		

ALPHA( 1 ) = -7.950 BETA( 4 ) = 4.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.775		
.858		
.834		
.890		
.857		
.865		
.900		
.905		
.930		
.933		
.965		



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81U46)

UPPER WING

ARC11-716 1A14 01+112+S12N25

ALPHA( 1 ) = -7.990 BETA( 4 ) = 4.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.700	-.0128			-.2803		
.725			-.2641			
.750					-.3541	-.3445
.760		-.2945				
.775			-.2465	-.3016		
.808		-.2349				
.834	-.2404					
.850			-.1323	-.2241	-.1029	
.857		-.1158				
.865	-.2041					-.0710
.900	-.1158		-.0495			
.905		-.0726		-.0081		
.950			-.0345		-.0186	
.953		-.0680				
.965	-.1030					

ALPHA( 1 ) = -7.980 BETA( 5 ) = 8.190

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.3192	.0281	.1946	.1264	-.1001	-.2524	-.2906
.020				.4941	.3925	.3727	.2962
.025			.3540				
.040		.1178					
.045			.3363				
.050	-.1230			.3786	.3491	.3153	.2406
.060				.2876			
.085			.2786				
.086		.2346					
.094	-.0444						
.150				.1198	.1087	.1020	.0123
.163		.0000					
.177			.1507				
.193		.2204					
.229	.1447						
.250				-.0147	-.0512	-.0887	-.2000
.274			.0614				
.339		.0308					
.362	.1737						
.400				-.0088	-.1678		-.3887
.402							
.497	.0000		-.0355				

TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 75

(R81046)

UPPER WING

ARC11-716 1A14 01+712+312N25

ALPHA( 1 ) = -7.980 BETA( 5 ) = 0.190

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.550				-.1777	-.2654		
.565			.0000				
.600						-.4894	
.650						-.4103	
.700	-.0631			-.2914	-.2558		
.725						-.3794	-.4186
.750							
.760							
.775							
.808							
.834	-.2591						
.850							
.857							
.865							
.900	-.1825						
.905	-.1646						
.925							
.950							
.953							
.965	-.1733						

ALPHA( 2 ) = -3.920 BETA( 1 ) = -0.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1221	.2129	.4980	.5945	.3245	.4690	.4936
.020				.4228	.3666	.4013	.3859
.025							
.040							
.045							
.050	.0141						
.060							
.065							
.086							
.094	.0619						
.150							
.163							
.177							
.193							
.229	.1793						
.250							
.274							
.339							



(RBIU46)

UPPER WING

ARC11-716 1A14 01+T12+312N25

ALPHA( 2 ) = -3.920 BETA( 1 ) = -8.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .362 .1322

.400 .3769 -.8099 -.7269

.402 .3763

.497 .0000

.550 -.1276 -.2901

.565 .0000

.800 -.3876 -.7254

.650 .0758

.700 .0758

.725 .0758

.750 .0758

.760 .0758

.775 .0758

.800 .0758

.834 .0758

.850 .0758

.857 .0758

.865 .0758

.900 .0758

.905 .0758

.950 .0758

.953 .0758

.965 .0758

ALPHA( 2 ) = -3.920 BETA( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .000

.000 .2323 .1889 .4414 .5036 .4179 .3464 .3851

.020 .020

.025 .025

.040 .040

.045 .045

.050 .050

.060 .060

.065 .065

.086 .086

.094 .094

.150 .150

.163 .163

.177 .177

.193 .193

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(R81U46)

UPPER WING

ARC11-716 IA14 CR-T12+812N25

ALPHA( 2 ) = -3.920 BETA( 2 ) = -4.010

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.289 .1881 -.3212 -.3862 -.4124 -.4727

.274

.339

.1599

-.1847

-.1029

.362

.403

.452

.497

.539

.585

.630

.653

.693

.700

.723

.750

.775

.808

.834

.850

.837

.865

.900

.925

.950

.953

.965

ALPHA( 2 ) = -3.850 BETA( 3 ) = .020

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.3084 .1041 .3871 .4173 .2965 .2035 .2603

.020 .025 .023 .040 .045 .030 .0838 .085

.086 .094 .2443 .2936 .2230 .1984 .1941 .1304

.0728 .1676 .2803 .0714



(881046)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

UPPER WING

ARC11-716 1A14 01+712+912N25

ALPHA( 2 ) = -3.850 BETA( 3 ) = .020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.150 -.1249 -.1371 -.1317 -.1867

.163

.0000

.177

-.0318

.193

.1615

.229

.1826

.259

-.2654

.274

-.3230

.339

-.3564

.382

-.4367

.409

-.1318

.432

-.1985

.497

-.2127

.530

-.1873

.565

-.3275

.600

.0000

.639

-.4500

.700

-.2736

.725

-.2739

.750

-.3371

.760

-.2689

.775

-.2911

.808

-.2578

.834

-.3094

.857

-.2303

.865

-.1782

.900

-.0259

.905

-.0476

.959

-.0193

.933

-.0210

.965

-.0411

-.0336

-.0462

ALPHA( 2 ) = -3.970 BETA( 4 ) = 4.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.2983 .1182 .3158 .3299 .1991 .0950 .1473

.020

.020

.025

.4231

.040

.3429

.045

.3492

.050

.2862

.2416

.2130

.1944

.1303

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UPPER WING

(R01U46)

ALPHA(2) = -3.970 BETA(4) = 4.100

ARC11-716 IA14 OR+T12+S12M25

DEPENDENT VARIABLE CP

SECTION 11 UPPER WING

Y/BW	X/CW	CP
.2990	.3640	.4270
.3340	.5340	.6730
.7000	.7000	.8070
.080	.1090	
.085	.1878	
.086	.2423	
.094	.0355	
.150		-.0616
.163	.0000	-.0843
.177	.0150	-.0809
.193	.1747	-.1500
.229	.1642	
.250		-.1976
.274	-.0756	-.2511
.339	-.0590	-.2835
.362	.1603	-.3041
.400		-.1061
.402	-.1715	-.2669
.497	.0000	-.4914
.550		-.1985
.565	.0000	-.3209
.600		
.650		-.4510
.700	-.0399	-.2723
.725		-.2964
.750		-.3906
.760	-.3154	-.3306
.775	-.2812	-.3464
.808		
.834	-.2643	-.2924
.850		-.3062
.857	-.2760	-.2176
.865	-.2469	
.900	-.2299	-.2251
.905		-.1259
.950	-.1966	-.1232
.953	-.1661	-.1163
.965	-.2076	



(R0146)

**UPPER WING**

ARC11-716 1A14 01+T12+S12N25

$$\text{ALPHAC}(2) = -3.999 \quad \text{BETAO}(5) = 0.160$$

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CF
1	1
2	2
3	3
4	4
5	5
6	6
7	7
8	8
9	9
10	10
11	11
12	12
13	13
14	14
15	15
16	16
17	17
18	18
19	19
20	20
21	21
22	22
23	23
24	24
25	25
26	26
27	27
28	28
29	29
30	30
31	31
32	32
33	33
34	34
35	35
36	36
37	37
38	38
39	39
40	40
41	41
42	42
43	43
44	44
45	45
46	46
47	47
48	48
49	49
50	50
51	51
52	52
53	53
54	54
55	55
56	56
57	57
58	58
59	59
60	60
61	61
62	62
63	63
64	64
65	65
66	66
67	67
68	68
69	69
70	70
71	71
72	72
73	73
74	74
75	75
76	76
77	77
78	78
79	79
80	80
81	81
82	82
83	83
84	84
85	85
86	86
87	87
88	88
89	89
90	90
91	91
92	92
93	93
94	94
95	95
96	96
97	97
98	98
99	99
100	100

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

MS/X

.990	-.4262	.0255	.1203	.2918	.1145	-.0112	.0304
.020				.4049	.3290	.3344	.2095
.023			.2752				
.040		.0251					
.045			.2671				
.990	-.1609			.2450	.2320	.2073	.1431
.080				.1293			
.985			.1935				
.086		.1616					
.094	-.0844						
.150		.0000		-.0264	-.0433	-.0422	-.1231
.163							
.177			.0405				
.229	.1106	.1949					
.290				-.1457	-.1919	-.2309	-.3497
.274			-.0417				
.339		-.0290					
.362	.1361						
.400				-.1712	-.2621		-.4723
.402		-.1421					
.497	.0000						
.590				-.2251	-.3283		
.565			.0000				
.600							-.5367
.690						-.4704	
.700	-.0919			-.3251	-.2704		
.725						-.4272	-.4095
.790		-.3342					
.780				-.3161	-.3786		
.775			-.2797				
.808							
.834	-.2090						
.890				-.2742	-.3362	-.3227	
.857		-.2503					
.865	-.2543			-.2806			-.1007
.900	-.2319						
.905		-.2154		-.1736			
.930				-.1753		-.1591	
.933		-.1985					
.963	-.2319						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U46)

UPPER WING

ARC11-716 1A14 01+T12+S12M25

ALPHA( 3 ) = -.300 BETA( 1 ) = -8.100

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CM							
.000	-.0925	.1877	.3238	.6814	.6565	.6156	.6280
.020				.2523	.2158	.1822	.1488
.025			.2435				
.040		.2416					
.045			.1277				
.050	-.1855			-.0213	-.0690	-.0967	-.1726
.080				-.2325			
.085			-.0609				
.086		.2326					
.094	.0027						
.150				-.4759	-.5179	-.4702	-.4485
.163		.0000					
.177			-.2869				
.193		.0035					
.229	.1489						
.250				-.5523	-.6592	-.6900	-.6732
.274			-.4088				
.359		-.3475					
.362	.0682			-.6676	-.8051		-.8905
.400							
.402				-.2496	-.6762		
.497	.0000		.0000				
.550							
.565							
.600							
.650							
.700	-.0153			-.1084	-.2269	-.5740	-.9102
.725							
.750							
.760							
.775							
.808				-.0785	.1395	.1026	
.834			.1371				
.850	-.0436						
.857			.1273				
.865	.1424						
.900	.1394						
.905							
.950				.1286	.0744		-.1667
.955				.1092			
.958				.0674			.1505
.965	.0568		.0510				





(R01046)

UPPER WING

ARC11-716 1A14 CR+T12+S12N25

ALPHAO ( S ) = -.338 BETAO ( S ) = .040

DEPENDENT VARIABLE CP

SECTION ( ::UPPER WING

Y/BW .2993 .3640 .4270 .5340 .6730 .7000 .8870

X/CW

.000	-.3065	.1353	.4220	.3319	.4684	.4224	.4573
.020				.2874	.2252	.2386	.1676
.025		.2142	.2597				
.040							
.045			.1705	.0512	.0017	-.0233	-.1074
.050	-.0061			-.1189			
.060			.0309				
.065		.2248					
.066							
.094	.0403			-.3110	-.3529	-.3561	-.3802
.150		.0000					
.163			-.1690				
.177		.0630					
.193							
.229	.1494			-.4117	-.4976	-.5389	-.5958
.250			-.2783				
.274		-.2523					
.339				-.4136	-.5661		-.7979
.362	.0984		-.3784				
.400				-.2124	-.3616		
.402		.0000					
.497							
.550							
.565							
.600							
.650							
.700	-.0432			-.2922	-.2784		
.725							
.750							
.760			-.3107			-.3236	-.2648
.775				-.2782	-.3155		
.808			-.2649				
.834	-.8845						
.850				-.2565	-.1795	-.0697	
.857			-.2689				
.865	-.2218						
.900	-.2261			-.1951			-.0211
.903			-.1865	-.0733			
.930				-.0753		-.0538	
.953			-.1151				
.985	-.1767						



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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

UPPER WING

ARC11-716 1A14 01+112+312N25

ALPHAO (3) = -.330 BETA0 (4) = 4.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.3474	.0700	.3337	.4431	.3644	.3167	.3509
.020				.2734	.1990	.1933	.1567
.025			.2132				
.040		.1403					
.045			.1469				
.050				.0559	.0225	-.0091	-.0917
.080	-.0643			-.0920			
.085			.0458				
.086		.1607					
.094	-.0114						
.150		.0000		-.2413	-.2632	-.3023	-.3587
.163			-.1366				
.177		.0738					
.193							
.229	.1274						
.250				-.3602	-.4460	-.4992	-.5785
.274			-.2095				
.339		-.1869					
.362	.0971			-.3159	-.4140		-.7474
.403			-.2941				
.402	.0000			-.2424	-.3784		
.550			.0000				
.565							-.5234
.600							
.650							
.700	-.0841				-.2871		
.725				-.3249			
.750							-.3949
.760							-.3286
.775				-.3443	-.3247	-.3705	
.806			-.3132				
.834	-.2860						
.850				-.3245	-.3093	-.1959	
.857			-.3207				
.865	-.2732						
.900	-.2835			-.2929			-.1030
.905			-.2892		-.1715		
.950				-.2547			-.1223
.953			-.2430				
.965	-.2817						

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UPPER WING

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 OF 112+312M25

ALPHA(3) = -.330 BETA(3) = 0.160

DEPENDENT VARIABLE CP

SECTION (3) UPPER WING

X/CW .2999 .3640 .4250 .5340 .6750 .7800 .8870

Y/CW

.000	-.5183	-.0767	.1232	.2896	.2862	.2066	.2465
.025				.2511	.2568	.2066	.1870
.025		.0944					
.040		.0278					
.045			.0771	.0950	.0648	.0347	-.0493
.090	-.2430			-.0228			
.080			.0447				
.085		.0020					
.086	.1577						
.094				-.1724	-.2188	-.2434	-.3108
.150		.0000					
.163			-.0574				
.177		.0888					
.193							
.229	.1014			-.2796	-.3619	-.4135	-.5236
.250			-.1476				
.274		-.1345					
.339							
.362	.0790			-.2599	-.3604		-.6040
.400			-.2282				
.402				-.2659	-.3833		
.497	.0000		.0000				-.5721
.590							
.565							
.600							
.650							
.700	-.1239			-.3563			
.750							
.775			-.3572	-.3640	-.4116		
.800			-.3443				
.834	-.3191			-.3561	-.3773	-.2878	
.850			-.3477				
.857							
.865	-.3170			-.3109			-.1810
.900	-.3141			-.2889	-.2194		
.905				-.2409		-.1752	
.950							
.953			-.2347				
.965	-.3001						

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R01U46)

UPPER WING

ARC11-716 IA14 01+T12+S12N23

ALPHAO(4) = 4.100 BETA0(1) = -0.100

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BA .2990 .3640 .4270 .5340 .6730 .7890 .8870

X/CU

.000	-.0830	.0872	.4988	.7008	.6746	.6157	.5803
.020				.0308	-.0236	-.1265	-.2218
.025			.0854				
.040		.1546					
.045			-.0608				
.050	-.2081			-.2428	-.3141	-.3749	-.4995
.080				-.4353			
.085			-.2587				
.086		.1481					
.094	-.1885						
.150		.0000		-.6751	-.7241	-.7111	-.7069
.163			-.4298				
.177		-.1060					
.193							
.229	.1211			-.7569	-.6792	-.8907	-.8627
.250			-.5463				
.274		-.4473					
.339							
.362	.0123			-.8209	-1.0060		-.9038
.400			-.6984				
.402							
.497	.0000			-.4641	-.7832		
.550			.0000				
.565							
.600							
.6571						-.6246	-.7063
.700	-.2672			-.1729	-.2888		
.725						-.5141	-.5928
.750			-.1021				
.760				-.0413	-.4471		
.775			.1117				
.808							
.834	-.1019			.0300	-.2836	-.4523	
.850			.1065				
.857							
.865	.1033						
.900	.1162			.0418			-.5424
.905			.0967		-.0697		
.930			.0088				-.4210
.953			.0459				
.965	.0502						

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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UPPER WING

ARC11-716 1A14 01+12+312N25

ALPHA( 4 ) = 4.090 BETA( 2 ) = -4.040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1960	.0576	.4370	.6614	.6172	.5343	.5226
.020				.0627	-.0106	-.1179	-.2276
.025			.1031				
.040		.1532					
.045			-.0191				
.050	-.2991			-.2007	-.2872	-.3604	-.4871
.080				-.3965			
.085			-.2137				
.086		.1511					
.094	-.0301						
.150		.0000		-.6093	-.6784	-.6721	-.7030
.163							
.177			-.3903				
.193		-.0775					
.229	.1175						
.250				-.6862	-.8378	-.8670	-.8584
.274		-.4123					
.339							
.362	.0202						
.403				-.7652	-.9497		-1.0280
.402			-.6553				
.497	.0000			-.3902	-.9348		
.530			.0000				
.565							
.600							
.650							
.700	-.1591			-.3225			
.725					-.2833		
.750							
.760			-.3098				
.775			-.1991		-.1922	-.1787	
.808							
.834	-.2174			-.0794	-.0801	-.4370	
.850							
.857			-.1394				
.865	-.2075						
.900	-.1901			-.0221			-.6216
.905			-.0296		-.0539		
.930				-.0346		-.3093	
.933			-.0329				
.965	-.0299						



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 TABULATED PRESSURE DATA - 1A14A - VOL. 6  
 (RB1U46)

UPPER WING

ARC11-716 1A14 Q1-T12-S12N25

ALPHA( 4 ) = 4.093 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.3483	.0028	.3878	.9904	.5543	.4890	.4634
.020				.0741	-.0030	-.1031	-.2203
.025			.0954				
.040			.1325				
.045							
.050	-.0856		-.0039	-.1663	-.2645	-.3410	-.4740
.080				-.3559			
.085			-.1568				
.086			.1386				
.094	-.0186			-.5474	-.6475	-.6305	-.6742
.150			.0000				
.163							
.177			-.0527				
.193							
.229	.1052			-.6085	-.7570	-.8324	-.8316
.250							
.274			-.4323				
.339			-.3693				
.362	.0260			-.6937	-.8566		-1.0060
.400							
.402			-.9901				
.497	.0000			-.3174	-.5998		
.550			.0000				
.565							
.600							
.690							
.700	-.1301			-.3197	-.2848		-.9711
.725							
.760			-.3378				
.775			-.2921				
.808				-.3075	-.2912		
.834	-.2595						
.850				-.2685	-.1405	-.0595	
.857			-.3173				
.865	-.2603						
.900	-.2677			-.1895	-.0959		-.4240
.905			-.2620				
.950				-.1364			-.0464
.953			-.1902				
.965	-.2796						

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(R81U48)

UPPER WING

ARC11-716 1A14 01+T12+S12M25

ALPMAD( 4) = 4.070 BETA0 ( 4) = 4.120

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/DW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.5398	-.0595	.2599	.4924	.4846	.4305	.4188
.020				.0904	.0017	-.0863	-.1944
.025							
.040		.0773					
.045			.0015				
.050	-.1639			-.1262	-.2344	-.3127	-.4336
.060					-.2856		
.085			-.0983				
.086		.0417					
.094	-.0903						
.150				-.4626	-.5538	-.5859	-.6281
.163		.0000					
.177			-.2667				
.193		-.0346					
.229	.0864						
.250				-.5281	-.6343	-.7330	-.7903
.274			-.3682				
.339		-.3091					
.362	.0260						
.400				-.5326	-.7458		-.9439
.402			-.4614				
.497	.0000			-.2908	-.4372		
.550							
.565			.0000				
.600							
.650							
.700	-.1393			-.3477	-.2990	-.5032	-.9090
.725							
.750							
.760			-.3378				
.775				-.3475	-.3790		
.808			-.3411				
.834	-.3062						
.850				-.3465	-.2898	-.1849	
.857			-.3691				
.885	-.3109						
.900	-.3208			-.3126			-.1251
.935			-.3394		-.1735		
.950				-.2386		-.1099	
.953			-.2886				
.965	-.3470						



(R81U46)

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UPPER WING

ARC11-716 1A14 01+712+312N25

ALPHA( 4 ) = 4.060 BETA( 5 ) = 0.200

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.9502	-.0392	.0665	.2282	.4284	.3849	.3736
.020				.0079	.0432	-.0380	-.1441
.025			.0066				
.040		.0078					
.045			-.0208				
.050	-.3285			-.1136	-.1844	-.2475	-.3681
.080				-.2094			
.095			-.0719				
.096		.0217					
.094	-.3676						
.150		.0000		-.3267	-.4555	-.5034	-.5560
.163							
.177		-.0146					
.193			-.1815				
.229	.0952						
.250				-.4031	-.5486	-.6225	-.7191
.274			-.2647				
.339		-.2389					
.362	.0256			-.3431	-.4999		-.8552
.400							
.402		-.2978					
.497	.0000						
.550				-.3135	-.4512		
.565		.0000					
.600							
.650							
.700	-.1524						
.725				-.3782		-.5613	
.750							
.760		-.3819					
.775				-.3985	-.4295		
.808		-.3961					
.834	-.3385						
.850				-.3853	-.3740	-.2619	
.857			-.4003				
.865	-.3490						
.900	-.3676						
.905							
.950			-.3478		-.2041		-.1897
.953			-.2454				
.965	-.3545		-.2808				

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(881U46)

UPPER WING

ARC11-716 1A14 OL+T12+S12N25

ALPHA( 5) = 0.040 BETA( 1) = -0.080

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	X/CW			
.2993	.3640	.4270	.5340	.6730 .7800 .8970
.000	-.0833	-.0696	.4323	.6811 .6100 .5117 .4547
.020			-.2092	-.2690 -.4141 -.5272
.025			-.0836	
.040		.0483		
.045			-.2437	
.050	-.2154		-.4465	-.5434 -.6367 -.7722
.060			-.6330	
.065			-.4408	
.066		.0434		
.094	-.2034			
.150		.0000	-.0852	-.9021 -.9142 -.9262
.163			-.5759	
.177		-.2370		
.193			-.6734	
.229	.0846		-.9530	-1.0470 -.8847 -.9054
.250				
.274		-.5353		
.339	-.0377		-.9740	-.9023
.352			-.7912	
.400			-.4646	-.7867
.497	.0000		.0000	
.550				-.6873
.565			-.3134	
.600			-.4829	
.630			-.2670	-.5946 -.6674
.700	-.6146		-.4853	-.6245
.725			-.1139	
.750			-.3349	-.5767 -.5268
.760			-.0812	
.775				
.808				
.834	-.1274		-.2204	-.6553
.850			-.0306	-.4875
.857			-.1372	-.4991
.863	-.0136		-.0202	
.900	-.0125			
.905				
.950				
.953				
.965	-.0236			



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ALPHA( 5 ) = 7.920 BETA( 2 ) = -4.020

ARC11-716 1A14 01+712+312M25

UPPER WING

(RB1U46)

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/B	X/CW	CP
.000	-.2156	-.0796
.020	.4224	.6478
.025	-.1312	-.2341
.040	-.0260	
.045	.0679	
.050	-.1744	
.060	-.3842	-.5031
.065	-.5624	
.066	-.3361	
.094	.0637	
.150	-.2504	
.163	.0500	
.177	-.3040	
.193	-.1752	
.229	.0871	
.250		
.274	-.6050	
.339	-.4647	
.362	-.0250	
.400		
.402	-.8786	-1.0290
.497	-.7457	
.550	.0000	
.565	-.4805	-.6257
.600	.0000	
.650		
.700	-.4821	
.725		
.750		
.760	-.2796	
.775	-.1262	
.808		
.834	-.1885	
.850		
.857		
.865	-.1470	
.900	-.1006	
.905		
.950		
.993		
.965		

DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI4A - VOL. 6

(R81U46)

UPPER WING

ARC11-716 IAI4 OI+T12+312M25

ALPHA( 5) = 7.910 BETA( 3) = .080

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3686	-.1155	.3540	.6564	.5479	.4362	.3565
.020				-.0699	-.1989	-.3791	-.5384
.025			-.0107				
.040		.0650					
.045			-.1249				
.050	-.3822			-.3221	-.4642	-.5862	-.7670
.080					-.4985		
.085			-.2882				
.086		.0760					
.094	-.5638			-.7021	-.8171	-.8567	-.9079
.150							
.163		.0500					
.177			-.4328				
.193		-.1225					
.229	.0718			-.7541	-.9499	-.9961	-1.0130
.250			-.5320				
.274		-.4342					
.339							
.362	-.0195			-.8118	-1.0310		-1.0400
.400			-.6862				
.432				-.5094	-.8992		
.497	.0000					-.7710	
.550			.0500				-.8059
.565							
.600							
.650							
.700	-.2989			-.3350		-.5928	-.7538
.725					-.3015		
.750							
.760			-.3358		-.2680	-.4181	
.775				-.2662			
.808							
.834	-.2700				-.2339	-.2870	-.5280
.850			-.2756				
.857							
.865	-.2808						-.7495
.900	-.2932			-.2095			
.905			-.2326		-.1775		
.950				-.1926		-.5441	
.953			-.1796				
.965	-.1820						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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/R81U46)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 5) = 8.040 BETA( 4) = 4.150

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.8135	-.1564	.1455	.5224	.5000	.3933	.3158
.020				-.0739	-.1728	-.3505	-.5362
.025			-.0715				
.040		.0006					
.045			-.1271				
.050	-.2237			-.2979	-.4297	-.5538	-.7563
.080				-.4626			
.085			-.2194				
.086		.0200					
.094	-.1528						
.150				-.6228	-.7726	-.8242	-.8930
.163		.0000					
.177			-.3649				
.193		-.1303					
.229	.0377			-.6576	-.8840	-.9639	-.9997
.250			-.4513				
.274							
.339		-.3607					
.362	-.0257			-.7194	-.9234		-1.1130
.400			-.5835				
.402				-.3994	-.6914		
.497	.0000		.0000				
.550							
.565							
.600							
.650							-.9054
.700	-.2141			-.3609	-.2934	-.7693	
.725							
.750			-.3708			-.4619	-.8413
.760							
.775			-.3863	-.3913	-.2987		
.808							
.834	-.3328			-.3457	-.2065	-.2394	
.850			-.4003				
.857							
.865	-.3366			-.2515			-.7751
.900	-.3610			-.3515	-.1885		
.925				-.2108		-.2035	
.930							
.953			-.2590				
.965	-.3259						

(R81U46)

UPPER WING

ALPHA(5) = 0.030 BETA(5) = 0.220

ARC11-716 1A14 OI+T12+312M25

## SECTION (1) UPPER WING

## DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.5804	-.0860	-.0639	.0549	.4383	.3775	.2982
.020				-.2836	-.1971	-.3113	-.4882
.025				-.1097			
.040				-.0483			
.045				-.1363			
.050	-.3962			-.3480	-.4253	-.3151	-.7115
.080				-.4149			
.085				-.1833			
.086				-.0385			
.094				-.4614			
.150				-.4672	-.6920	-.7646	-.8432
.163				.0000			
.177				-.2850			
.193				-.1581			
.229	.0094						
.250				-.4928	-.7250	-.8919	-.9322
.274				-.3338			
.339				-.3472			
.362	-.0279						
.400				-.4005	-.6222		-1.0380
.402				-.3538			
.497	.0000			-.3701	-.5312		
.550				.0700			
.565							
.600							
.650							
.700	-.2070			-.4107	-.3085	-.6097	-.9858
.725							
.750				-.4097		-.4356	-.5747
.760							
.775				-.4443	-.4566		
.808				-.4597			
.834	-.3761			-.3650	-.3132	-.2272	
.850				-.4090			
.857							
.865	-.3979						
.900	-.4240			-.2765			-.3442
.935				-.3158	-.1824		
.950				-.2175			-.1836
.955				-.2316			
.965	-.2827						



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TABULATED PRESSURE DATA - IALIA - VOL. 6

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ARC11-716 IALIA OL+12+S12N25

UPPER WING

(RB1U47) ( 15 FEB 74 )

## REFERENCE DATA

## PARAMETRIC DATA

SREF = 2.4210 SQ.FT. XMRP = 29.5800 INCHES  
 LREF = 30.7090 INCHES YMRP = .0000 INCHES  
 BRP = 30.7090 INCHES ZMRP = .0000 INCHES  
 SCALE = .0300 SCALE

MACH = 1.050 ELEVON = .000  
 RUDDER = .000 SPDBRK = .000

ALPHA( 1 ) = -7.960 BETA( 1 ) = -8.070

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1328	-.1080	.4113	.5083	.4213	.3308	.3673
.020				.5616	.5340	.5642	.5584
.025			.4523				
.040		.2204					
.045			.3830				
.050	-.0569			.3393	.3529	.3784	.3524
.060				.1594			
.065			.2445				
.086		.3041					
.094	-.0276			-.0642	-.0138	.0266	.0136
.150			.0000				
.163							
.177			.1871				
.193							
.229	.0114			-.2070	-.2250	-.2203	-.2375
.250							
.274				-.1115			
.339			-.0969				
.362	.1506			-.0368	-.1024		-.4699
.400							
.402				-.1278			
.497	.0000						
.530				.0059	-.1385		
.565				.0000			
.600							
.650							
.700	.1868						-.3442
.725							
.750				-.1702			
.760				-.0404		.1132	.0540
.775							
.808				-.0237			
.834	.0879			.2640	.2096		
.850				.2540			
.857							
.865				.2336			
.900	.2512			.2387	.2436	.2350	
.900	.2339						
.905				.2355			.2243
				.2172	.2019		



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(R01U47)

UPPER WING

ARC11-716 1A14 01+712+912N25

ALPHAO (1) = -7.969 BETAO (1) = -6.070

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.959 .1852 .2026  
.953 .1680  
.963 .1524

ALPHAO (1) = -7.969 BETAO (2) = -4.010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.900 -.2385 -.1433 .3491 .4212 .3066 .2043 .2285  
.920 .5717 .5236 .5474 .5247  
.925 .4519  
.940 .1202  
.945 .3996  
.950 -.1195 .3803 .3849 .3971 .3627  
.980 .2299  
.985 .2907  
.986 .2790  
.994 -.0941 .0315 .0572 .0942 .0580  
.990 .0000  
.963 .0837  
.933 .2254  
.929 -.0381  
.920 .1140 .1280 .1375 .1969  
.974 -.0160  
.939 -.0041  
.962 .1732  
.900 .0000  
.902 .0576  
.997 .0261 .1170 .3267  
.950 .0203 .1470  
.965 .0000  
.900 .2709  
.9613  
.925 .1001 .2184  
.950 .1810 .1756  
.960 .1346  
.975 .0132 .0658  
.908 .1207  
.934 .1408 .1304 .1112  
.950



ALPHAD( 1 ) = -7.969 BETA0 ( 2 ) = -4.010

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CF
1	1
2	2
3	3
4	4
5	5
6	6
7	7
8	8
9	9
10	10
11	11
12	12
13	13
14	14
15	15
16	16
17	17
18	18
19	19
20	20
21	21
22	22
23	23
24	24
25	25
26	26
27	27
28	28
29	29
30	30
31	31
32	32
33	33
34	34
35	35
36	36
37	37
38	38
39	39
40	40
41	41
42	42
43	43
44	44
45	45
46	46
47	47
48	48
49	49
50	50
51	51
52	52
53	53
54	54
55	55
56	56
57	57
58	58
59	59
60	60
61	61
62	62
63	63
64	64
65	65
66	66
67	67
68	68
69	69
70	70
71	71
72	72
73	73
74	74
75	75
76	76
77	77
78	78
79	79
80	80
81	81
82	82
83	83
84	84
85	85
86	86
87	87
88	88
89	89
90	90
91	91
92	92
93	93
94	94
95	95
96	96
97	97
98	98
99	99
100	100

Y/BV	.2990	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

M2/x

.057	.1368		
.1239			
.063	.1615		
.970		.1461	.1092
.935	.1480		
.950	.1329		.1419
.953	.1196		
.965	.0927		

$$\text{ALPHA}(1) = -7.963 \quad \text{BETA}(3) = .030$$

SECTION 6 UPPER WING

DEPENDENT VARIABLE CP

Y184	.2990	.3640	.4270	.5340	.6730	.7900	.8870
------	-------	-------	-------	-------	-------	-------	-------

AD/x

1000	-.3118	-.0595	.2975	.3270	.1093	.0716	.0933
200				.5605	.4967	.5142	.4790
300		.1403					
400							
500	-.1713	.3918		.4000	.3992	.3989	.3585
600				.2742			
700		.3099					
800		.2950					
900	-.1375			.0942	.1109	.1379	.0866
1000		.0000					
1100			.1415				
1200		.2455					
1300	-.0109						
1400				-.0379	-.0475	-.0622	-.1531
1500		.0473					
1600		.0503					
1700	.1922						
1800				-.0136	-.0970		-.2077
1900		-.0230					
2000	.0000						
2100			.0900	-.0336	-.1425		-.3544
2200							
2300							
2400							
2500							
2600							
2700	.1143						
2800				-.1267	-.2200		-.2201
2900							-.2525
3000		-.1640					

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(R81U47)

UPPER WING

ARC11-716 1A14 01+112+512N25

ALPHAO ( 1 ) = -7.960 BETAO ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.775 -.1076 -.1657

.898

.834

.850

.857

.865

.900

.905

.910

.933

.955

.9239

.951

ALPHAO ( 1 ) = -7.960 BETAO ( 4 ) = 4.120

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.2977 -.0402 .2328 .2306 .0598 -.0626 -.0491

.020

.025

.040

.045

.050

.060

.085

.086

.094

.150

.153

.177

.193

.229

.250

.274

.336

.332

.400

.402

.497

.550

.565

.890

.850

.850

.850

.850

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.850



TABULATED PRESSURE DATA - 1A14A - VOL. 6

(N81047)

UPPER WING

ARC11 16 1A14 Q1+T12+S12N23

DATE 10 JAN 73

ALPHA(1) = -7.960 BETA(4) = 4.120

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8670

X/CW						
.700	.0773					
.725						
.750						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.953						
.965						

ALPHA(1) = -7.990 BETA(5) = 8.210

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8670

X/CW						
.000						
.020						
.025						
.040						
.045						
.050						
.080						
.085						
.086						
.094						
.150						
.163						
.177						
.193						
.229						
.250						
.274						
.339						
.362						
.400						
.402						
.497						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1U47)

UPPER WING

ARC11-716 1A14 01+712+S12N25

ALPHAO ( 1 ) = -7.990 BETAO ( 5 ) = 0.210

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.590				-.0565	-.1340		
.565			.0000				-.3430
.600					-.2644		
.650					-.2343		
.700	.0452			-.1627			
.725					-.2379	-.2910	
.750			-.2003	-.1650	-.2047		
.760			-.1732				
.775							
.808							
.834	-.1335			-.1704	-.1866	-.1904	
.850			-.1901				
.857							
.865	-.1029			-.1295			-.0865
.900	-.1333		-.0330	-.0829			
.905				-.0152	-.0805		
.950			-.0295				
.953							
.965	-.0761						

ALPHAO ( 2 ) = -3.820 BETAO ( 1 ) = -0.120

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.090	-.0911	.0241	.4941	.6192	.9580	.5070	.5583
.020				.4135	.3939	.4172	.4227
.025			.3506				
.040		.2105					
.045			.2453				
.050	-.1629			.1589	.1517	.1750	.1461
.060				-.0260			
.065			.0730				
.066		.2468					
.094	-.0882			-.2854	-.2569	-.2037	-.1811
.150		.0000					
.163			-.1322				
.177							
.193		.0672					
.229	-.0175						
.250				-.3829	-.4249	-.3509	-.4343
.274			-.2794				
.339		-.2406					



DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBU47)

UPPER WING

ARC11-716 1A14 OI+T12+S12N25

ALPHAO ( 2 ) = -3.820 BETA0 ( 1 ) = -8.120

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.362	.0729					
.400						
.402						
.497	.0000					
.550						
.565						
.600						
.650						
.700	.1113					
.725						
.750						
.760						
.775						
.808						
.834	.0492					
.850						
.857						
.865	.2249					
.900	.2256					
.905						
.950						
.953						
.965	.1520					

ALPHAO ( 2 ) = -3.830 BETA0 ( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1837	-.0113	.4426	.5411	.4601	.3905	.4360
.020				.4303	.3856	.4039	.4210
.025			.3497				
.040		.1614					
.045			.2662				
.050	-.2164			.1989	.1813	.1935	.1795
.060				.0247			
.065			.1290				
.066		.2330					
.094	-.1573						
.130							
.163		.0000					
.177							
.193		.1083					

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U47)

UPPER WING

ARC11-716 1A14 OL+T12+S12M25

ALPHAO( 2 ) = -3.830 BETAO ( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.229	-.0347					
.250						
.274						
.339						
.362						
.400						
.402						
.497						
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.806						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.955						
.965						

ALPHAO( 2 ) = -3.910 BETAO ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2492	.0003	.3857	.4540	.3550	.2694	.3027
.020				.4317	.3805	.4022	.3968
.025				.3408			
.040				.1548			
.045				.2725			
.050	-.2516			.2286	.2227	.2310	.1918
.060				.0753			
.085				.1614			
.086							
.094	-.1726	.2205					



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 3177

(RB1047)

UPPER WING

ARC11-716 1A14 01-712+512M25

ALPHAO( 2 ) = -3.910 BETAO ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.150 .163 .177 .193 .229 .250 .274 .339 .362 .400 .402 .497 .550 .565 .600 .650 .700 .725 .750 .760 .775 .808 .834 .850 .857 .865 .900 .905 .930 .935 .965

.0000

.1365

-.0150

-.1059

-.1085

.1191

.0000

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ALPHAO( 2 ) = -3.910 BETAO ( 4 ) = 4.100

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 .020 .025 .040 .045 .050

-.3729 -.0125 .3000 .3678 .2509 .1966 .1967

.020 .025 .040 .045 .050 .055 .060

.020 .025 .040 .045 .050 .055 .060

.020 .025 .040 .045 .050 .055 .060

.020 .025 .040 .045 .050 .055 .060

.020 .025 .040 .045 .050 .055 .060

.020 .025 .040 .045 .050 .055 .060

.020 .025 .040 .045 .050 .055 .060

.020 .025 .040 .045 .050 .055 .060



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U47)

UPPER WING

ARC11-716 1A14 01-712-912M25

ALPHA(2) = -3.910 BETA(4) = 4.190

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.8730	.7000
.8870		
.080	.1022	
.085	.1613	
.086	.1633	
.094	-.1613	
.150		-.0322
.163	.0000	-.0555
.177	.0131	-.0461
.193	.1333	-.0728
.229	.0144	
.250		-.1620
.274	-.0500	-.2022
.339	-.0350	-.2213
.362		-.3034
.403	.1091	
.402		-.1246
.497		-.1766
.590	.0000	-.3720
.565		-.1227
.600		-.0956
.650		-.2036
.700	.0373	
.725		-.3312
.750		-.2396
.760		-.1841
.775	-.2197	
.808	-.1764	-.2792
.834		-.2404
.850		-.1039
.857	-.1873	-.2373
.865		-.1920
.900	-.1484	-.2153
.905	-.1361	-.1486
.950		-.1012
.953	-.1340	-.1071
.965	-.0960	-.0374
	-.1123	
	-.1416	





(原書1047)

**UPPER WING**

ARC11-716 1A14 01+Y12+S12N25

ALPHA( 3 ) = .020 BETA( 1 ) = -0.130

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

V78W	.2990	.3840	.4270	.5340	.6730	.7000	.8670
------	-------	-------	-------	-------	-------	-------	-------

**AD/X**

.000	-.0478	.0246	.3220	.7058	.6947	.6606	.6766
.020				.2543	.2460	.2234	.1887
.025			.2226				
.040		.1636					
.045			.0969				
.050	-.1558			-.0122	-.0224	-.0455	-.1078
.060				-.2116			
.085			-.0990				
.086		.1782					
.094	-.1626						
.150				-.4394	-.4405	-.3909	-.3722
.163		.0000					
.177			-.2650				
.183		-.0382					
.229	-.0682			-.5289	-.6007	-.6064	-.5721
.274			-.3725				
.339		-.3046					
.362	.0202						
.400				-.5972	-.7412		-.7771
.402			-.5307				
.497	.0000			-.3108	-.7518		
.590			.0000				-.6112
.585						-.6704	
.600					-.2242		
.650				-.1761			
.700	-.0635					-.4100	-.5645
.725							
.750			-.1666				
.760			.1064	.0472	-.1449		
.775							
.808			.1647				
.834	-.0711			.1809	-.0251	-.3332	
.850							
.857							
.900	.1016			.1845	.0615		-.3586
.905	.1656						
.909			.1717				
.950				.1484		-.2034	
.953			.1318				
.965	.1288						

ALPHAO( 3 ) = .000 BETA0 ( 2 ) = -4.060

SECTION ( UPPER WING	DEPENDENT VARIABLE CP
1	
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Y/BW	.2995	.3645	.4275	.5340	.6735	.7655	.8670
------	-------	-------	-------	-------	-------	-------	-------

**x/c**

.0000	-.1317	-.0302	.4812	.6469	.6076	.5751	.6044
.0050				.2914	.2538	.2379	.1968
.0250		.1500	.2480				
.0450			.1456				
.0550	-.2508		.0413	.0030	-.0172	-.0827	
.0600			-.1432				
.0850			-.0165				
.0886		.1900					
.0940	-.2144						
.1050				-.3650	-.3998	-.3991	-.3469
.1630		.0500					
.1770			-.2027				
.1950		.0268					
.2290	-.0732						
.2550			-.3111	-.4531	-.5277	-.5490	-.5541
.2740		-.2601					
.3390							
.3620	.0580			-.5534	-.6612		-.7446
.4050			-.4769				
.4020				-.1927	-.4689		
.4970	.0000		.0000				-.7692
.5500					-.2390	-.5419	
.5650							
.6000							
.6900				-.1452			-.4806
.7050	.0217						
.7250			-.1859				
.7550				-.0982	-.0924		
.7750			-.1000				
.8006							
.8340	-.1106			-.0114	.0310	-.0096	
.8550			-.0521				
.8650	-.0304			.0883			-.2907
.9000	-.0387						
.9050			.0710		.0797		
.9300				.0899		.0560	
.9530			.0767				
.9650	.6504						



DATE 19 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1047)

UPPER WING

ARC11-716 IA14 01+T12+S12M25

ALPHA( 3 ) = .080 BETA( 4 ) = 4.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.4117	-.0706	.3065	.4763	.4047	.3749	.4376
.020			.1702	.2598	.1976	.2387	.1950
.025		.0650					
.040			.1081				
.045				.0562	.0216	.0096	-.0449
.050	-.3155			-.0844			
.080			.0177				
.085		.1145					
.086							
.094	-.1808			-.2630	-.2826	-.2699	-.2850
.150		.0000					
.163			-.1308				
.177		.0396					
.193							
.229	.0246			-.3409	-.3960	-.4134	-.4784
.250			-.2239				
.274		-.1811					
.339	.0592			-.2590	-.3846		-.6544
.362			-.2832				
.400				-.1331	-.2448		
.402		.0000					
.497	.0000						
.550			.0000				
.565							
.650							
.650							
.700	.0002			-.2091	-.2489	-.3602	-.3980
.725							
.750			-.2448				
.760				-.2084	-.2533		
.775			-.1912				
.800							
.834	-.1882			-.2176	-.2087	-.0890	
.850			-.2133				
.857							
.865	-.1593			-.2086			.0073
.900	-.1840						
.905			-.1897	-.0807			
.950				-.1468		-.0210	
.953			-.1885				
.965	-.2005						

(K01047)

**UPPER WING**

ARC11-716 1A14 01+Y12+S12M23

ALPHAC( 3) = .070 BETA( 3) = 0.100

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP					
Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8670
X/CW							
.000	-.6544	-.0980	.0065	.2946	.3428	.2918	.3425
.020				.2090	.2146	.2359	.2162
.025			.0627				
.040		-.0358					
.045			.0393				
.050	-.3619			.0780	.0757	.0697	.0064
.080				-.0285			
.085			.0041				
.086		-.0747					
.094	-.2031						
.150				-.1516	-.1775	-.1756	-.2170
.163		.0000					
.177			-.0570				
.193		.0791					
.229	.0167						
.250				-.2220	-.2928	-.3334	-.4215
.274			-.1109				
.339		-.1196					
.362	.0451						
.400				-.1757	-.2692		-.5113
.402		-.1655					
.497	.0000						
.550				-.1914	-.2358		
.565			.0000				
.607							
.650						-.3755	-.4298
.700	-.0181				-.2547		
.725				-.2413			
.750			-.2482			-.3044	-.2533
.780				-.2395	-.2778		
.775			-.2214				
.806							
.834	-.2081						
.850				-.2543	-.2518	-.1637	
.857			-.2502				
.865	-.2007						
.900	-.2074			-.2397	-.1596		-.0547
.925			-.2153				
.950				-.1682		-.0656	
.953			-.1843				
.965	-.2389						

DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI/A - VOL. 6

(R81047)

UPPER WING

ARC11-716 IAI14 OI-T19-212125

ALPHAO( 4) = 4.120 BETAO ( 1) = -8.110

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0314	-.0839	.4927	.7200	.7105	.6631	.6333
.020			.0764	.0312	.0339	-.0542	-.1376
.025		.0616					
.040			-.0676				
.045							
.050	-.1387			-.1990	-.2390	-.2868	-.3937
.080				-.3775			
.085			-.2637				
.086		.0874					
.094	-.1737						
.150				-.6174	-.6213	-.5995	-.5925
.163		.0000					
.177		-.1330					
.193			-.3689				
.229	-.1069						
.250				-.6738	-.7693	-.7648	-.7355
.274			-.1853				
.339		-.3734					
.362	.0097			-.7194	-.8971		-.6983
.400			-.6143				
.402				-.7262	-.9000		
.497	.0000		.0000				
.550							
.565							
.600							
.650							
.700	-.4180			-.2885	-.2835	-.7363	-.8215
.725							
.750							
.760			-.2137				
.775			-.0253	-.1866	-.5397		
.808							
.834	-.0072						
.850				-.0595	-.3787	-.4181	
.857			.0312				
.865	-.0426						
.900	.0016			.0249			-.6834
.905			.1046		-.1800		
.950				.0565			-.4040
.955			.1088				
.965	.0980						



(RB1U47)

UPPER WING

ARC11-716 IA14 OI+712+312+25

ALPHA( 4 ) = 4.100 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/B	X/C	CP
.2990	.3640	.4270
.3340	.6730	.7800
.6870		
.000	-.1411	-.1858
.020	.4846	.6880
.025	.6352	.8323
.040	.1159	.1031
.045	.0302	-.1294
.050	.1819	
.055	-.0106	
.060	-.1478	-.2052
.065	-.2184	-.3720
.066	-.3186	
.067	-.1828	
.068	.1333	
.069	-.2522	
.070	-.3298	-.5725
.071	-.5618	-.5800
.072	.0000	
.073	-.3172	
.074	-.0361	
.075	-.0347	
.076	-.3985	-.7217
.077	-.7537	-.7248
.078	-.4185	
.079	-.3335	
.080	.362	.0595
.081	-.6574	-.8288
.082	-.8778	
.083	-.5672	
.084	.0000	
.085	-.4435	-.8315
.086	-.7716	
.087	-.2610	
.088	-.3111	
.089	-.5185	-.8327
.090	-.2382	
.091	-.1826	-.2693
.092	-.1024	
.093	-.0788	-.1927
.094	-.4733	
.095	-.1168	
.096	-.0240	
.097	-.5335	
.098	-.0813	-.074
.099	-.0055	-.4092
.100	-.3433	
.101	-.1133	



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U47)

UPPER WING

ARC11-718 1A14 C1+712+512N25

ALPHA( 4 ) = 4.100 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2720	-.1041	.3969	.1144	.5917	-.0009	.5230
.020				.1136	.0696	-.0181	-.1228
.025				.1113			
.040		.0962					
.045			.0106				
.050	-.3270			-.1147	-.1781	-.2371	-.3535
.060				-.2709			
.085			-.1364				
.086		.1292					
.094	-.3133			-.4595	-.2004	-.5139	-.5506
.100		.0500					
.163			-.2657				
.177		-.0332					
.193							
.229	.0656			-.3176	-.6354	-.6949	-.6957
.250							
.274			-.3621				
.339		-.2933					
.362	.0559						
.400			-.6066				-.8339
.402		-.5206					
.497	.0000			-.3217	-.7351		
.550			.0000				
.565							
.600							
.630							-.8275
.700	-.1171			-.2200	-.2921	-.7368	
.725							
.750						-.8457	-.6209
.760			-.2563				
.775				-.1651	-.2588		
.800			-.1620				
.834	-.1512						
.850				-.1508	-.0850	-.1124	
.857			-.1960				
.865	-.1455						
.900	-.1502			-.1185			-.3895
.905			-.1692		-.0143		
.950				-.0732		-.0323	
.953			-.1310				
.965	-.2003						



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI4A - VOL. 6

(R81U47)

UPPER WING

ARC11-716 IAI4 01+712+312N25

ALPHA( 4) = 4.080 BETA( 5) = 8.230

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/8W .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.7655	-.1117	.0313	.2156	.4432	.4183	.4436
.020				.0011	.0776	.0166	-.0680
.025			.0013				
.040		-.0586					
.045			-.0276				
.050	-.4696			-.1077	-.1176	-.1855	-.2829
.060				-.1925			
.065			-.0709				
.066		-.0530					
.094	-.3557			-.2976	-.3932	-.4182	-.4636
.150		.0000					
.163			-.1713				
.177		.0089					
.193							
.229	.0317			-.3767	-.4804	-.5460	-.6128
.250			-.2598				
.274		-.2160					
.339							
.362	.0303			-.2270	-.3513		-.7394
.400			-.2433				
.402				-.1867	-.3085		
.497	.0000		.0000				
.550							
.565							
.600							
.650							
.700	-.0570						
.725				-.2643	-.2686		
.750							
.760			-.2780				
.775				-.2579	-.2903		
.806			-.2554				
.834	-.2251						
.850							
.857			-.2764				
.865	-.2308			-.2647	-.2391	-.1312	
.900	-.2406						
.905				-.2296			-.0711
.950			-.2376		-.1137		
.953				-.1818			-.0605
.965	-.2503		-.2029				

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U47)

UPPER WING

ALPHA( 3 ) = 8.050 BETA( 1 ) = -8.090

ARC11-716 1A14 01+712+312M25

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW	.0000	.0091	-.1624	.4381	.7371	.6804	.5763
.020	.020				-.1935	-.1727	-.3001
.025	.025						-.4074
.040	.040						
.045	.045						
.050	.050						
.060	.060						
.085	.085						
.086	.086						
.094	.094						
.150	.150						
.163	.163						
.177	.177						
.193	.193						
.229	.229						
.250	.250						
.274	.274						
.339	.339						
.362	.362						
.403	.403						
.452	.452						
.530	.530						
.565	.565						
.600	.600						
.650	.650						
.700	.700						
.725	.725						
.790	.790						
.775	.775						
.800	.800						
.834	.834						
.850	.850						
.857	.857						
.865	.865						
.900	.900						
.905	.905						
.950	.950						
.953	.953						
.965	.965						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U47)

UPPER WING

ARC11-716 1A14 OL+T12+S12N25

ALPHA( 5) = 0.030 BETA( 2) = -4.040

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1200	-.3277	.4290	.7025	.6556	.5486	.4807
.020				-.0822	-.1307	-.2752	-.4103
.025			-.0099				
.040		-.1120					
.045			-.1450				
.050	-.2373			-.3059	-.3936	-.4641	-.6290
.060				-.4695			
.085			-.3124				
.086		.0625					
.094	-.2442						
.150		.0000		-.7046	-.7273	-.7475	-.7765
.163							
.177			-.4159				
.193		-.1279					
.229	-.1681			-.7449	-.6630	-.6817	-.8753
.250							
.274			-.5069				
.339		-.3828					
.362	.0483			-.7449	-.9572		-.9104
.400			-.6309				
.402				-.7610	-.8306		
.497	.0000		.0000				
.550							
.565							
.600							
.650							-.7445
.700	-.5526			-.2991			
.725				-.3240			
.750							-.5972
.760			-.2531				
.775				-.2744	-.6121		
.808			-.0621				
.834	-.1211			-.2045	-.5250	-.4645	
.890			-.1156				
.857							
.865	-.1273						-.6889
.900	-.0036			-.1362			
.905			-.0993		-.3840		
.950				-.0960			-.4504
.953			-.0811				
.965	-.1388						

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(R01U47)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

UPPER WING

ARC11-716 1A14 CR+712+512N25

ALPHA(1) = 0.030 BETA(1) = 0.020

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/W .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2758	-.2816	.3531	.6272	.6096	.5081	.4390
.020				-.0340	-.1105	-.2636	-.4105
.025			-.0030				
.040		.0377					
.045			-.1187				
.050	-.3314			-.2588	-.3644	-.4642	-.6221
.080				-.4207			
.085			-.2455				
.086		.0711					
.094	-.3200						
.150		.0005		-.6203	-.6964	-.7235	-.7671
.163			-.3598				
.177		-.0964					
.193							
.229	.0657			-.6552	-.8224	-.8548	-.8638
.250		-.4518					
.274			-.3473				
.339							
.362	.0286			-.6905	-.9089		-.9770
.400			-.5798				
.402	.0000			-.6056	-.6812		
.497							
.550							
.565			.0000				
.600							-.6286
.650					-.7878		
.700	-.3735			-.3334	-.2838		
.725							
.750						-.5840	-.7637
.760			-.2879				
.775				-.2556	-.3765		
.808			-.1893				
.834	-.1516			-.1826	-.2369	-.5177	
.850			-.2178				
.857							
.865	-.1872						-.7606
.900	-.1972			-.1331			
.905			-.1955		-.1207		
.950				-.1077		-.5464	
.953			-.1686				
.965	-.2325						







(R81U47)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

UPPER WING

ARC11-716 1A14 Q1+T12+812/25

DATE 10 JAN 75

ALPHAO ( 5 ) = 0.020 BETAO ( 5 ) = 0.260

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7870 .8870

X/CW

.000	-.7681	-.1369	-.0548	.0426	.4183	.4261	.3656
.020				-.2506	-.1627	-.2384	-.3944
.025			-.0734				
.040		-.0862					
.045			-.0979				
.050	-.7152			-.2985	-.3582	-.4273	-.5944
.080				-.3513			
.085			-.1413				
.086		-.0644					
.094	-.6171			-.3982	-.5984	-.6618	-.7228
.150		.0000					
.163			-.2358				
.177		-.1440					
.193							
.229	.0065			-.4496	-.6459	-.7776	-.8250
.250			-.3180				
.274		-.2830					
.339							
.362	-.0016			-.3227	-.6452		-.9288
.400			-.3082				
.497	.0000			-.2518	-.3917		
.550			.0000				
.565							-.8672
.600					-.4426		
.650							
.700	-.1035			-.3014	-.2940		
.725						-.3137	-.8037
.750			-.3252				
.760			-.3119	-.3215			
.775			-.3816				
.808							
.834	-.2754			-.2795	-.2170	-.1095	
.850			-.3129				
.857							
.865	-.2926			-.2223			-.2644
.900	-.7993			-.2609	-.1023		
.905				-.1945			
.950							
.953			-.2047				
.965	-.2364						



(RB1U48) ( 15 FEB 74 )

UPPER WING

PARAMETRIC DATA

MACH = 1.100 ELEVON = .000  
RUDDER = .000 SPOBRK = .000

REFERENCE DATA

SREF = 4210 SQ.FT. XMRP = 29.5800 INCHES  
LREF = 38.7590 INCHES YMRP = .0000 INCHES  
BREF = 38.7590 INCHES ZMRP = .0000 INCHES  
SCALE = .0300 SCALE

ALPHA( 1 ) = -7.930 BETA( 1 ) = -8.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/Cu							
.000	-.1111	-.0495	.4303	.5374	.4905	.3740	.4090
.020				.5924	.5533	.5713	.5707
.025			.4873				
.047		.0732					
.045			.4189				
.050	-.0082			.3737	.3733	.3678	.3632
.080				.1979			
.085			.2773				
.086		.2812					
.094	-.0113			-.0475	-.0109	.0375	.0336
.150		.0000					
.163			.0446				
.177		.2189					
.193							
.229	.0319			-.1907	-.2085	-.2066	-.2377
.250			-.0861				
.274		-.0816					
.339	.1747			-.3369	-.3990		-.4716
.400			-.2886				
.402		.0000		.0760	-.0486		
.497			.0000				
.550							
.565							
.600							
.650							
.700	.2097						
.725				-.0135	-.1596		
.750						.0239	-.0352
.760			-.0351				
.775				.2134	.1346		
.808		.2820					
.834	.0865			.2344	.2458	.2280	
.850							
.857		.2568					
.865	.2651						
.900	.2567			.2815			.2433
.905		.2499			.2302		

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U48)

UPPER WING

ARC11-716 1A14 01-712-512M25

ALPHA(1) = -7.930 BETA(1) = -8.589

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .0990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .930 .2110 .2236 .2263

.953  
.965 .1934

ALPHA(2) = -7.950 BETA(2) = -4.020

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW  
.000 -.1861 -.1299 .3514 .4440 .3418 .2532 .2877  
.020 .5739 .5165 .5477 .5584  
.025 .4520  
.040 .0125  
.045 .4012  
.050 -.0701 .3758 .3715 .3913 .3680  
.060 .2162  
.085 .2889  
.086 .1910  
.094 -.0516 .0014 .0415 .0819 .0683  
.130 .0000  
.163 .0799  
.177 .2156  
.193 -.0211  
.229 -.1471 -.1497 -.1412 -.1805  
.250  
.274 -.0461  
.339 -.0487  
.362 .1368  
.400  
.408 -.0670  
.497 .0000  
.510  
.565 .0403 -.0841  
.620 .0000  
.630  
.709 .1932  
.725  
.750 -.0428  
.760 -.1976  
.773  
.780 -.0606  
.795 -.0074 -.0757  
.808 .0608  
.834  
.856 -.0174  
.886 .1246 .0563 .0204

-.3056

-.2131

-.1519 -.2001

-.0074 -.0757

.0608

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

.1246 .0563 .0204

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U48)

UPPER WING

ARC11-716 1A14 OR-T12-S12N25

ALPHA( 1 ) = -7.950 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.857 .1361

.865 .1110

.900 .1370

.905 .1629

.950 .1669

.953 .1480

.965 .1184

ALPHA( 1 ) = -7.925 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -2.476

.020 -.1048

.025 .2814

.040 .3990

.045 .3660

.050 .3660

.060 .2820

.085 .1965

.094 .0890

.150 .0000

.163 .1154

.177 .2153

.193 .2153

.229 -.0196

.250 .0504

.274 .0251

.339 .1610

.362 .0276

.400 -.0307

.402 -.0109

.497 .0000

.550 .0228

.565 .0837

.602 .0000

.650 .0000

.700 .0000

.725 .0000

.750 .0000

.760 .0000

.0000

.0000

.0000

.0000

.0000

.0000

.0000

DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81U48)

UPPER WING

ALPHA( 1 ) = -7.920 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEFENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.775				-.0540	-.1069		
.806			-.0292				
.834	-.0380						
.850				-.0394	-.0618	-.0819	
.857			-.0085				
.965	.0224			.0566			.0069
.900	.0417		.0641	.0636			
.905				.0934		.0439	
.950			.0737				
.933							
.965	.0488						

ALPHA( 1 ) = -7.930 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEFENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2815	-.0634	.2244	.2626	.1129	.0040	.0190
.020				.5265	.4679	.4860	.4510
.025			.3465				
.040		.0246					
.045			.3318				
.050	-.1626			.3276	.4052	.4026	.3634
.080				.2943			
.085			.2918				
.086		.1336					
.094	-.1299						
.150		.0000		.1442	.1679	.1916	.1392
.163							
.177			.1690				
.193		.2022					
.229	-.0164			.0351	.0347	.0207	-.0706
.290			.1037				
.274							
.339		.0910					
.362	.1362						
.400				.0345	-.0216		-.2072
.402			.0257				
.497	.0000						
.550				.0843	-.0830		
.565							
.609			.0000				-.2976
.698							-.2123



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81040)

UPPER WING

ARC11-716 1A14 01-112-512N25

ALPHA( 1 ) = -7.950 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.700 .1236

.725

.750

.750

.775

.800

.834

.850

.857

.865

.900

.905

.950

.953

.965

-.2156

-.0905

-.1795

-.2427

-.1354

-.0851

-.1343

-.0954

-.0929

-.1235

-.1317

-.0908

-.0598

-.0500

-.0289

-.0297

.0045

ALPHA( 1 ) = -7.950 BETA( 5 ) = 0.230

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000

.020

.025

.040

.045

.050

.060

.065

.066

.094

.100

.163

.177

.193

.229

.250

.274

.339

.362

.400

.402

.497

.1034

.1125

.1156

.4779

.4155

.4202

.3025

.3852

.3030

.3369

.1737

.1921

.2073

.1509

.0677

.0691

.0520

-.0425

.0411

-.0124

-.1866

.0459

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U48)

UPPER WING

ARC11-716 IA14 01+712+812N25

ALPHA( 1 ) = -7.930 BETA( 5 ) = 0.230

SECTION ( 1 ) UPPER WING DEFENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.550				-.0169	-.0873		
.565			.0000			-.2999	
.613					-.2117		
.650	.0828						
.700							
.725				-.1109			
.750						-.1851	-.2618
.760			-.1492				
.775				-.1139	-.1504		
.808			-.1299				
.834	-.1012						
.850				-.1240	-.1494	-.1553	
.857					-.1672		
.865							
.900	-.1115			-.1341			-.0973
.920	-.1261				-.1101		
.935						-.0792	
.950				-.0618			
.953					-.0296		
.965	-.0690						

ALPHA( 2 ) = -3.940 BETA( 1 ) = -0.110

SECTION ( 1 ) UPPER WING DEFENDENT VARIABLE CF

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0901	.0040	.5210	.6372	.5687	.5125	.5470
.020				.4756	.4474	.4609	.4681
.025			.4034				
.040		.1188					
.045			.3075				
.050	-.1034			.2297	.5173	.2302	.2088
.060				.0464			
.085			.1358				
.096		.2694					
.094	-.0827						
.150		.0000		-.2158	-.1931	-.1412	-.1142
.165							
.177							
.193		.1303					
.229	-.0077						
.240				-.3347	-.3672	-.3647	-.3663
.274							
.339					-.2236		
							-.1731



TABULATED PRESSURE DATA - 1414A - WZL 6

(R81U48)

UPPER WING

ARC11: 716 1A:4 31+112+512H25

ALPHA( 2 ) = -3.940 BETA( 1 ) = -6.110

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7830 .8870

X/CW

.362 .1285  
 .400 .4619 -.5360 -.5084  
 .422 -.3980  
 .497 .0000  
 .550 -.1655 -.5773  
 .565 .0000  
 .600  
 .635 -.6547  
 .700 .0779 -.1946  
 .725 .0779 -.1946  
 .750 .0779 -.1946  
 .760 .0779 -.1946  
 .775 .0779 -.1946  
 .800 .0779 -.1946  
 .834 .0779 -.1946  
 .852 .0779 -.1946  
 .857 .0779 -.1946  
 .865 .0779 -.1946  
 .900 .2299 .2066  
 .905 .2210 .2025  
 .930 .2025 .1930  
 .953 .1930  
 .965 .1726

ALPHA( 2 ) = -3.960 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7830 .8870

X/CW

.0000 -.1262 -.0440 .4472 .5552 .4734 .4034 .4409  
 .020 .025 .025 .3747  
 .040 .0642  
 .045 .2914  
 .050 -.1328  
 .060 .2338 .2143 .2170 .1900  
 .065 .0629  
 .066 .1549  
 .094 .1933  
 .1011  
 .150 -.1660 -.1615 -.1221 -.1140  
 .165 .0000  
 .177 -.0520  
 .195 .1265



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1U48)

UPPER WING

ARC11-716 1A14 01+112+312N25

ALPHA( 2 ) = -3.860 BETA( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.229	-.0213					
.250						
.274						
.339						
.352						
.400						
.402						
.497						
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.955						
.965						

ALPHA( 2 ) = -3.860 BETA( 3 ) = .05C

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE C<sub>p</sub>

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.1271	-.0760	.3813	.4634	.3827	.2832	.3359
.020				.4373	.3708	.3844	.4117
.025							
.040							
.045							
.050							
.060							
.085							
.086							
.094							



DATE 10 JAN 79 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U48)

UPPER WING

ALPHA ( 2 ) = -3.890 BETA ( 3 ) = .050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3540 .4270 .5340 .6730 .7800 .8870

X/CW .150 .163 .177 .193 .209 .224 .239

.1200 -.1284 -.0682 -.0662

.0000

-.0153

.1257

-.0476

-.1303

-.1167

.1061

-.2656

.0000

-.2416

-.1967

-.0912

-.1252

-.0828

-.0619

-.0626

-.0576

-.0184

.0139

-.0014

.0633

.0733

.0907

.0905

.0950

.0933

.0965

ALPHA ( 2 ) = -3.890 BETA ( 4 ) = 4.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3540 .4270 .5340 .6730 .7800 .8870

X/CW .000 .020 .025 .040 .045 .050

-.3150 -.0392 .3044 .3888 .2770 .1947 .2298

.020 .025 .025 .025 .025 .025

.025 .025 .025 .025 .025 .025

.025 .025 .025 .025 .025 .025

.025 .025 .025 .025 .025 .025

.025 .025 .025 .025 .025 .025

.025 .025 .025 .025 .025 .025

.025 .025 .025 .025 .025 .025

.025 .025 .025 .025 .025 .025

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DATE 12 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U48)

UPPER WING

ARC11-716 1A14 01+712+512N25

ALPHAD (2) = -3.890 BETAD (4) = 4.130

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8070

X/CW						
.080		.1076				
.085	.1631					
.086	.1582					
.094	-.1522					
.150		-.0494	-.0319	-.0195	-.0334	
.163	.0000					
.177	.0229					
.193	.1337					
.229	-.0168					
.250		-.1536	-.1793	-.1816	-.2522	
.274		-.0819				
.339	-.0559					
.362	.1121					
.400		-.1038	-.1552		-.3707	
.402	-.0941					
.497	.0000					
.550		-.0456	-.1401			
.565		.0000				
.603						
.650						
.700	.0777					
.723		-.1219				
.750						
.760		-.1641				
.773		-.1212	-.1776			
.808		-.1218				
.834	-.1136					
.850		-.1423				
.857		-.1409	-.1688	-.1447		
.865	-.0936					
.900	-.0897					
.903		-.1445				
.903		-.1061	-.1078		-.0218	
.950		-.0903				
.953		-.0878			-.0234	
.965	-.1112					



(R81048)

UPPER WING

TABULATED PRESSURE DATA - IA14A - VOL. 6

ARC11-716 IA14 OL+T12+S12N25

ALPHA(2) = -3.890 BETA(5) = 8.190

SECTION (1) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3647	.4270	.5340 .6730 .7800 .8870
-.4709	-.0617	.1033	.3355 .1727 .0700 .1072
.020		.1866	.3752 .3298 .3648 .3585
.040	-.0316	.2058	.2341 .2464 .2523 .2245
.045			.1261
.050	-.2253		.1793
.080			.0929
.085			-.0076 .0269 .0288 .0067
.086	-.1316		
.094		.0000	.0545
.130		.1013	
.163			-.0687 -.1118 -.1232 -.1085
.177			-.0263
.193	.0001		-.0704 -.1171
.229			-.0650
.250			-.0605 -.1372
.274			.0000
.339			-.2704
.362	.0772		-.2075
.400			
.402			
.497	.0000		
.530			
.565			
.600			
.630			
.700	.0443		
.725			
.750			
.780			
.775			
.808			
.834	-.1357		
.850			
.857			
.865	-.1364		
.900	-.1456		
.935			
.950			
.953			
.965	-.1550		

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## TABULATED PRESSURE DATA - IA14A - VOL. 6

(R01U46)

UPPER WING

ARC11-716 IA14 01+712+312N25

ALPHA( 3 ) = -.310 BETA( 1 ) = -0.140

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	.0110	-.0249	.5399	.7123 .6872 .6639 .6975
.020			.3203	.3164 .3123 .2888
.025		.2801		
.040		.0627		
.045		.1552		
.050	-.1327		.0615	.0559 .0481 -.0018
.080			-.1327	
.085		-.0320		
.086		.1510		
.094	-.1229			
.150			-.3569	-.3579 -.2990 -.2720
.163		.0000		
.177		-.2232		
.193		.0092		
.220	-.0494		-.4854	-.5165 -.5100 -.4775
.250			-.3612	
.274		-.2914		
.339	.0591			
.362			-.5618	-.6655 -.6749
.400		-.5090		
.402	.0050		-.2031	-.6877
.497		.0000		
.550				
.565				
.630				
.650				
.700	-.0375		-.1536	-.2169 -.6943 -.7156
.725				
.750				
.760				
.775		-.1369	-.0025	-.1945
.800		.0359		
.834	-.0187		.0759	-.0576 -.3247
.850				
.857		.0750		
.865	.0682			
.900	.0653		.1574	.0410 -.3336
.905		.1557		
.950			.1584	-.2120
.955		.1546		
.965		.1566		



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881048)

UPPER WING

ARC11-716 1A14 01+112+S12N25

ALPHA( 3 ) = -.330 BETA( 2 ) = -4.040

DEPENDENT VARIABLE CF

SECTION ( 1 ) UPPER WING

Y/B	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.0626	-.0643	.4818	.6397	.5809	.5431	.6019
.020				.3309	.2964	.3039	.2829
.025			.2739				
.040		.0297					
.045			.1732				
.050	-.1615			.0901	.0558	.0581	.0144
.060				-.0938			
.085		.0056					
.086		.0920					
.094	-.1456						
.150				-.3063	-.3391	-.2815	-.2642
.163		.0000					
.177		-.1670					
.193		.0297					
.229	-.0658			-.4142	-.4644	-.4675	-.4731
.250							
.274		-.2775					
.339		-.2304					
.362	.0569			-.5090	-.6017		-.6631
.400							
.432		-.4343					
.497	.0000			-.2213	-.6237		
.530		.0000					
.555							
.600							
.650							
.700	-.0146						
.725				-.1176	-.2226		
.750							
.760		-.1339					
.775				-.0334	-.1583		
.806		-.0210					
.834	-.0589						
.850				-.0121	-.0209	-.1641	
.857		-.0365					
.855	.0159						
.900	-.0119			.0265	.0821		-.3208
.905		.0110					
.950		.0831					-.0647
.953		.0612					
.965	.0120						

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DATE 10 JAN 75

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## TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U48)

UPPER WING

ARC11-716 1A14 Q1+T12+S12N25

ALPHA(3) = -.340 BETA(3) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.2055	-.0670	.4118	.5451	.4741	.4221	.5218
.020				.3144	.2617	.2724	.2765
.025			.2464				
.040		.0345					
.045			.1618				
.050	-.2593			.0913	.0501	.0589	.0253
.060				-.0681			
.085			.0201				
.086		.1441					
.094	-.1906			-.2699	-.2842	-.2479	-.2411
.150		.0000					
.163			-.1326				
.177		.0443					
.193							
.229	-.0681			-.3582	-.4003	-.4208	-.4494
.250			-.2314				
.274		-.2028					
.339							
.362	.0654			-.4635	-.5324		-.6319
.400			-.3935				
.492							
.497	.0000			-.1030	-.2999		
.550							
.565			.0000				
.600						-.3400	-.6483
.650				-.1084			
.700	.0447						
.725							
.750				-.1901			-.3350
.763			-.1471				
.775				-.0920	-.1187		
.808			-.0771				
.834	-.0280			-.0970	-.0390	.0339	
.850			-.0948				
.857							
.865	-.0358			-.0693			-.0057
.903	-.0467						
.905			-.0647		.0488		.0891
.950				.0213			
.953			-.0323				
.965	-.0757						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(R81U48)

UPPER WING

ARC11-716 1A14 01+712+512N25

ALPHA( 3 ) = -.340 BETA( 4 ) = 4.110

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7000 .8870

X/CW

.000	-.3468	-.1096	.3019	.4942	.4049	.3345	.3904
.020				.3160	.2492	.2529	.2471
.025			.1790				
.040		.0349					
.045			.1161				
.050	-.3010			.1211	.0801	.0639	.0264
.080				-.0113			
.085			.0212				
.085		.1003					
.094	-.1912						
.130		.0000		-.1875	-.2171	-.2091	-.2085
.163							
.177		.0705					
.193		.0465					
.229	-.0201						
.250				-.2747	-.3351	-.3619	-.3967
.274		-.1530					
.339		-.1161					
.362	.0631						
.400				-.3392	-.4296		-.5679
.402		-.2696					
.497	.0600			-.0869	-.1744		
.550		.0000					
.555							
.600							
.650							
.700	.0171						
.725				-.1330	-.2244		
.750							
.760							
.775		-.1949				-.1891	-.0891
.808				-.1337	-.1727		
.834	-.1380						
.850		-.1326					
.857							
.865		-.1603		-.1510	-.1498	-.0322	
.865							
.900	-.1190						
.905				-.1605			.0723
.930		-.1375		-.0421			
.933		-.1221					.0357
.945		-.1255					
.945	-.1479						



(R81U48)

UPPER WING

ARC11-716 1A14 01+T12+312N25

ALPHA( 3 ) = -.340 BETA( 5 ) = 0.180

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3847	-.0797	.1013	.3127	.3177	.2542	.3394
.020			.0699	.2478	.2264	.2596	.2576
.040		-.0315					
.045			.0374	.1017	.0926	.0925	.0661
.050	-.3737			-.0037			
.080			.0307				
.085		-.0375					
.086							
.094	-.1960			-.1382	-.1723	-.1269	-.1501
.150		.0000					
.163			-.0277				
.177		.0761					
.193							
.229	.0206			-.2352	-.2905	-.2814	-.3551
.250			-.1112				
.274		-.1283					
.339	.0335			-.1540	-.1798		-.4419
.362							
.400			-.1565				
.497	.0000			-.0850	-.1845		-.3689
.550			.0000				
.565							
.600							
.650				-.3124			
.700	.0189			-.2404			
.725			-.1830				
.750				-.2595			-.2207
.760			-.1965				
.775			-.1774	-.2200			
.800			-.1657				
.834	-.1559						
.850			-.2044				
.857				-.1963	-.2084	-.1590	
.865	-.1448						
.900	-.1584						
.935			-.1769	-.1998	-.1419		-.0338
.950				-.1705			
.953			-.1706				-.0433
.965	-.1894						



(RB1U48)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

**UPPER WING**

ARC11-716 1A14 01+Y12+S12N25

$$\text{ALPHA}(4) = 4.170 \quad \text{BETA}(1) = -8.250$$

DEPENDENT VARIABLE: LING.

Y/8W	.2999	.3640	.4270	.5340	.6730	.7803	.8870
X/2W							
.000	.0192	-.0675	.5097	.7355	.7325	.6897	.6693
.020				.1009	.0902	.0111	-.0675
.025			.1164				
.040		-.0168					
.045			-.0283				
.050	-.0974			-.1448	-.1001	-.2177	-.3159
.060				-.3260			
.065			-.2211				
.066		.0432					
.094	-.1369						
.150				-.5666	-.5522	-.5243	-.5181
.163		.6100					
.177			-.3050				
.193		-.1135					
.229	-.1142						
.250				-.6360	-.7010	-.6857	-.6562
.274			-.4911				
.339		-.3681					
.362	.7143						
.400			-.8146		-.0261		-.8099
.402							
.497	.0000						
.550				-.7299	-.8378		
.565			.0000				
.600							
.650							
.700	-.0344			-.2887		-.6717	-.7696
.725							
.750			-.1615			-.5187	-.5929
.760							
.775			.0297	-.1976	-.5193		
.808							
.834	-.0173			-.0452	-.5061	-.4244	
.850							
.857			.0336				
.865	.0335			.0384			-.4455
.900	.0335						
.935			.0609	-.3293			
.950				.0764		-.3831	
.953			.0732				
.965	.0365						

(R81048)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

DATE 10 JAN 75

UPPER WING

ARC11-716 1A14 D1+T12+S12N25

ALPHA( 4 ) = 4.160 BETA( 2 ) = -4.050

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

X/CW	Y/W	2990	3640	4270	5340	6730	7000	8870
.000								
.020								
.040								
.060								
.080								
.100								
.120								
.140								
.160								
.180								
.200								
.220								
.240								
.260								
.280								
.300								
.320								
.340								
.360								
.380								
.400								
.420								
.440								
.460								
.480								
.500								
.520								
.540								
.560								
.580								
.600								
.620								
.640								
.660								
.680								
.700								
.720								
.740								
.760								
.780								
.800								
.820								
.840								
.860								
.880								
.900								
.920								
.940								
.960								
.980								
1.000								

.000								
.020								
.040								
.060								
.080								
.100								
.120								
.140								
.160								
.180								
.200								
.220								
.240								
.260								
.280								
.300								
.320								
.340								
.360								
.380								
.400								
.420								
.440								
.460								
.480								
.500								
.520								
.540								
.560								
.580								
.600								
.620								
.640								
.660								
.680								
.700								
.720								
.740								
.760								
.780								
.800								
.820								
.840								
.860								
.880								
.900								
.920								
.940								
.960								
.980								
1.000								

.000								
.020								
.040								
.060								
.080								
.100								
.120								
.140								
.160								
.180								
.200								
.220								
.240								
.260								
.280								
.300								
.320								
.340								
.360								
.380								
.400								
.420								
.440								
.460								
.480								
.500								
.520								
.540								
.560								
.580								
.600								
.620								
.640								
.660								
.680								
.700								
.720								
.740								
.760								
.780								
.800								
.820								
.840								
.860								
.880								
.900								
.920								
.940								
.960								
.980								
1.000								

.000								
.020								
.040								
.060								
.080								
.100								
.120								
.140								
.160								
.180								
.200								
.220								
.240								
.260								
.280								
.300								
.320								
.340								
.360								
.380								
.400								
.420								
.440								
.460								
.480								
.500								
.520								
.540								
.560								
.580								
.600								
.620								
.640								
.660								
.680								
.700								
.720								
.740								
.760								
.780								
.800								
.820								
.840								
.860								
.880								
.900								
.920								
.940								
.960								
.980								
1.000								

.000								
.020								
.040								
.060								

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81040)

UPPER WING

ARC11-716 1A14 01-712-512N25

ALPHA(4) = 4.17° BETA(3) = .030

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8870

X/CW

.000	-.2040	-.2154	.3895	.6069	.9880	.5490	.5525
.020				1228	.0964	.0279	-.0567
.025			.1137				
.040		-.0317					
.045			.0019				
.050	-.2650			-.1065	-.1443	-.1868	-.2855
.060				-.2662			
.065			-.1277				
.066		.0717					
.094	-.2754						
.150		.0200		-.4378	-.4880	-.4617	-.4795
.163							
.177		-.0434	-.2677				
.193							
.229	-.0924			-.5111	-.6094	-.6275	-.6259
.250							
.274			-.3624				
.339		-.2887					
.362	.0402			-.5732	-.7040		-.7742
.400							
.402			-.4975				
.497	.0000			-.4080	-.6691		
.550			.0000				
.565							
.600							
.650							-.7522
.700	-.2085				-.6857		
.725				-.2233	-.2404		
.750							
.760							
.775			-.2003				
.808			-.0968	-.3089			
.834			-.0801				
.850	-.1303						
.857							
.865			-.1137				
.900				-.0790	-.1606	-.3488	
.903	-.0712						
.903	-.0764						
.905			-.1030		.0022		-.4033
.950							
.953			-.0440				-.1700
.965			-.0864				
.965	-.1318						

## TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U48)

UPPER WING

ARC11-716 1A14 01+112+S12N25

ALPHA(4) = 4.173 BETA(4) = 4.130

SECTION / UPPER WING DEPENDENT VARIABLE CP

Y/RW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
Y/CW							
.000	-.3913	-.2600	.2316	.5362	.5235	.4884	.5015
.020			.1119	.0808	.0181	-.0551	
.025		.0443					
.040		-.0257					
.15			-.0162				
.050	-.3613		-.0902	-.1390	-.1816	-.2732	
.080				-.2325			
.085			-.1150				
.086		.0309					
.084	-.2953			-.3913	-.4504	-.4329	-.4566
.150		.0000					
.163			-.2140				
.177		-.0432					
.193	-.0348			-.4218	-.5277	-.5993	-.6044
.229		-.2633					
.274		-.2206					
.339	.0111			-.4749	-.5979		-.7495
.362			-.3931				
.550	.0000			-.2025	-.4530		
.565			.0000				
.600							-.7282
.650				-.2445			
.700	-.0361			-.1736			
.725					-.2405	-.4722	
.750		-.2335		-.1519	-.1787		
.760			-.1522				
.775							
.808							
.834	-.1454			-.1702	-.1209	-.0792	
.850			-.1881				
.857							
.865	-.1396			-.1743			-.2249
.900	-.1454				-.0286		
.925		-.1842					
.950			-.1369			.0174	
.953		-.1772					
.985	-.8006						



(R81U48)

UPPER WING

TABULATED PRESSURE DATA - IA14A - VOL. 6  
ARC11-716 IA14 OI+T12+S12N25

DATE 10 JAN 75

ALPHA( 4 ) = 4.160 BETA( 5 ) = 0.240

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7900 .8870

X/CW

.000	-.7134	-.1293	.0533	.2232	.4313	.4332	.4662
.020				.0064	.0764	.0479	-.0315
.025			-.0003				
.040		-.0314					
.045			-.0242				
.050	-.4677			-.1018	-.1121	-.1305	-.2317
.080							-.1862
.085			-.0637				
.086		-.0769					
.094	-.3986						
.150				-.2847	-.3649	-.3803	-.4071
.163		.0000					
.177		.0177	-.1673				
.193							
.229	.0502			-.3699	-.4468	-.4937	-.5561
.250							
.274		-.2454					
.339		-.2007					
.362	.0211			-.2956	-.4339		-.6867
.400							
.402		-.323					
.497	.0000			-.1226	-.2214		
.530							
.563		.0000					
.600							-.4828
.650							
.700	-.0049			-.2084		-.3237	
.725							
.750		-.2246				-.2301	-.1910
.760							
.775				-.1863	-.2184		
.800		-.1804					
.834	-.1662			-.2089	-.1837	-.0837	
.850							
.857		-.2235					
.865	-.1705						-.0346
.900	-.1812			-.1970			
.905		-.1977			-.0806		
.930				-.1637		-.0075	
.933		-.1832					
.965	-.2180						

C-10

DATE 19 JAN 75

PAGE 3216

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U48)

UPPER WING

ARC11-716 1A14 CL+T12+S12R25

ALPHA( 5 ) = 8.139 BETA( 1 ) = -4.040

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	-.0645	-.2503	.3800	.6934	.6710	.5738	.5123
.020				-.0793	-.0903	-.2159	-.3422
.025			-.0520				
.040		-.1830					
.045			-.1676				
.050	-.1657			-.2885	-.3438	-.4187	-.5483
.060				-.4484			
.085			-.3210				
.086		-.1211					
.094	-.1949			-.6651	-.6676	-.6726	-.6959
.150		.0000					
.163			-.4377				
.177							
.193		-.1722					
.229	-.1966			-.7509	-.7946	-.8025	-.7931
.250			-.4900				
.274		-.3712					
.339	-.0009			-.7206	-.8982		-.8988
.362							
.400			-.5921				
.402							
.497	.0000			-.7105	-.6840		-.8279
.550			.0000			-.8133	
.565				-.3301	-.2975		
.600						-.8325	-.7604
.650	-.3177						
.700				-.3202	-.6544		
.725			-.2787				
.750			-.1235				
.760				-.2595	-.3794	-.3560	
.775			-.1186				
.808	-.0866						
.834				-.1859			-.7492
.850	-.0457						
.857	-.0921			-.3			
.905			-.1083			-.3102	
.950			-.1110				
.953			-.1077				
.965	-.1713						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1048)

UPPER WING

ARC11-716 1A14 01+112+312N25

ALPHA( 3 ) = 8.130 BETA( 2 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2145	-.4376	.3218	.6241	.6135	.5277	.4692
.020			-.0370	-.0784	-.2102	-.3466	
.025			-.0139				
.040		-.1475					
.045			-.1222				
.050	-.2671		-.2414	-.3252	-.4059	-.5473	
.080			-.3990				
.085			-.2721				
.086		-.0138					
.094	-.2694			-.6061	-.6381	-.6536	-.6885
.130		.0000					
.163			-.3635				
.177		-.1105					
.193	-.1275			-.6387	-.7604	-.7777	-.7824
.250			-.4469				
.274		-.3432					
.339	.0209			-.6624	-.8531		-.8878
.362			-.5605				
.400			.0000	-.6420	-.8468		-.8422
.402	.497	.0500					
.550				-.2748		-.5914	-.7287
.565				-.2526			
.600			-.1207	-.2671	-.3703		
.650							
.700	-.4841			-.3328			
.725							
.750							
.760							
.775							
.806							
.834	-.0806			-.1824	-.2654	-.3291	
.850			-.1641				
.857							
.865	-.0944						
.900	-.1311			-.1263			-.7236
.905			-.1603	-.1824			
.930			-.0718				-.5402
.933			-.1697				
.985	-.2019						



DATE 10 JAN 75

TABULATED PRESSURE DATA - IAI14A - VOL. 6

(R81048)

UPPER WING

ARC11-716 IAI14 01+712+312N25

ALPHA( 5) = 0.120 BETA( 3) = 4.160

SECTION ( 1) UPPER WING

DEPENDENT VARIABLE CP

1/8W .2990 .3640 .4270 .5340 .6730 .7000 .8070

X/CW

.000	-.4199	-.3360	.0750	.4844	.5390	.4805	.4220
.020				-.1165	-.0906	-.2091	-.3477
.025			-.1060				
.040		-.1280					
.045			-.1521				
.050	-.3713		-.2600	-.3249	-.3970	-.5411	
.080			-.3994				
.085			-.2270				
.086		-.0646					
.094	-.3467			-.5420	-.6160	-.6311	-.6737
.150		.0000					
.163			-.2805				
.177		-.1151					
.193							
.229	-.0373			-.3289	-.7266	-.7520	-.7698
.250			-.3490				
.274		-.3030					
.339							
.362	-.0329			-.5728	-.7848		-.8776
.400			-.4550				
.402				-.3483	-.7134		
.497	.0000		.0000				-.8442
.550					-.2750		
.650							
.700	-.1288			-.2430			
.725							
.750			-.2871				-.6472
.760				-.1935	-.3137		
.775			-.2104				
.808							
.834	-.1739			-.1996	-.1670	-.2151	
.880			-.2287				
.937							
.965	-.1771			-.1955			-.6118
.990	-.1741				-.0376		
.995			-.2225				
.998			-.1550				-.0749
.999			-.2160				
.999	-.2339						



DATE 19 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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ARC11-716 1A14 OL+T12+S12N25

UPPER WING

(R81048)

ALPHA( 9 ) = 6.110 BETA( 4 ) = 8.275

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2995	.3640	.4270	.5340 .6730 .7600 .8870
X/CW				
.000	-.7402	-.1592	-.0588	.0260 .4101 .4430 .3937
.020				-.2514 -.1499 -.1990 -.3243
.025				-.0765
.040				-.1076
.045				-.1077
.050				-.2884 -.3411 -.3774 -.5191
.060				-.3306
.085				-.1448
.086				-.0818
.094				-.6180
.150				.0000
.163				-.2308
.177				-.1370
.193				
.229				.0224
.250				-.4133 -.6036 -.7051 -.7404
.274				-.3046
.339				-.2656
.362				-.0120
.430				-.2745 -.6204
.402				-.3238
.497				-.1910 -.3170
.550				.0000
.565				
.600				-.7805
.650				-.3609
.700				-.0472
.725				-.2463
.750				-.2778
.760				-.2531 -.5706
.775				-.2742
.806				-.2311 -.2524
.834				-.2456
.850				-.2238 -.1650 -.0958
.857				-.2483
.865				-.2199
.900				-.2222
.905				-.1979
.950				-.2179 -.0822
.953				-.1789
.965				-.1841
				-.0028
				-.2715
				-.2126

**UPPER WING**

ARC11-716 1A14 CI+Y12+S12N25

## PARAMETRIC DATA

MACH	=	1.150	ELEVON	=	.000
RUDDER	=	.000	SPOBRK	=	.000

**REFERENCE DATA**

XREF = 2.4215 50.FT. XMAP = 29.5000 INCHES  
 YREF = 30.7090 INCHES YMAP = .0000 INCHES  
 ZREF = 30.7090 INCHES ZMAP = .0000 INCHES  
 SCALE = .0003 SCALE

ALPHA2( 1 ) = -7.862      BETA0 ( 1 ) = -0.062

SECTION ( 1 ) UPPER WING

	Y/BV	.2990	.3640	.4270	.5340	.6730	.7800	.8670
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AD/x

000	-.0716	-.0103	.4079	.5634	.4900	.4150	.4533
020				.6264	.5875	.6101	.6103
025			.4970				
040		.0870					
045			.4343				
050	-.0066			.4100	.4143	.4295	.4090
060			.2472				
085			.3085				
086		.2002					
094	-.0060			.0276	.0449	.0895	.0897
1150		.0200					
163		.2223	.0855				
193							
229	.0465						
250				-.1348	-.1494	-.1439	-.1725
274			-.0475				
339	-.0397						
362	.1844						
405				-.2792	-.3320		-.3994
432			-.2312				
497	.0200			.1036	-.0499		
550			.0000				
565							
600							-.4779
650					-.1161		
700	.2373			.0390	-.1360		
725							
750							
760							
775			-.0011				
806			.2693	.1873	.1136		
834	.1294					.0460	-.1478
850							
857			.2815	.2863	.2436		
865	.2770		.2861				
900	.2821		.2978				.2363
905			.2865		.2722		



DATE 19 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(RB1U49)

UPPER WING

ARC11-716 IA14 01+712+S12M25

ALPHA( 1 ) = -7.860 BETA( 1 ) = -0.060

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .950 .2672 .2677

.953 .2546

.965 .2347

ALPHA( 1 ) = -7.860 BETA( 2 ) = -4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.0000 -.1558 -.1197 .2867 .4690 .3754 .3012 .3439

.020 .025 .040 .045 .050 .080 .085 .086

.094 .150 .163 .177 .193 .229 .240 .274

.339 .362 .400 .452 .497 .550 .565 .600

.650 .700 .725 .750 .760 .775 .808 .834

.850 .859 .8619 .8102 .8543 .8101

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DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81U49)

UPPER WING

ARC11-716 1A14 Q1+112+512N25

ALPHA( 1 ) = -7.865 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.0141	-.0375	
.0403		
.0133		
.0410	-.0106	-.0094
.0670		
.0845		
.1301		.0720
.0975		
.1276		.1524
.1634		.1153
.1331		
.1018		

ALPHA( 1 ) = -7.070 BETA( 4 ) = 4.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.2946	.1741	.0745
.5293	.5068	.5290
.4976		
.3186		
.3041		
.4074	.4436	.4450
.4060		
.2993		
.2644		
.0974		
.1830	.2129	.2335
.1881		
.0000		
.1801		
.1691		
.0836	.0833	.0670
-.0230		
.1435		
.1144		
.362	.1140	
.400		
.402	.0822	
.497		
.550		
.563	.0603	-.0287
.600	.0000	
.630		
-.2411		
-.1537		

DATE 10 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U49)

UPPER WING

ARC11-716 1A14 01+712+312M25

ALPHA( 1 ) = -7.870 BETA( 4 ) = 4.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.700	.1804	-.1692
.725		-.0316
.750		-.1217 -.1918
.760		-.0688
.775		-.0255 -.0755
.808		-.0259
.834	-.0120	-.0304 -.0641 -.0783
.850		-.0273
.857		
.865	.0002	
.900	.0343	-.0025 -.0002 -.0131
.905		.0555
.950		.0759 .0169
.953		.0590
.965	.0323	

ALPHA( 1 ) = -7.890 BETA( 5 ) = 8.200

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.000	-.3415	.1908 .2354 .0484 -.0613 -.0682
.020		.4912 .4209 .4306 .3951
.025		.3352
.040		
.045	-.0364	.3166
.050		.3633 .3839 .3844 .3496
.060	-.1834	.3012
.065		.2498
.066	.0691	
.094	-.1435	
.150		.1776 .1974 .2133 .1704
.163	.0000	
.177		.1821
.193	.2006	
.229	-.0567	.0606 .0750 .0689 -.0137
.250		
.274		.1483
.339	.0871	
.382	.1837	
.403		.0603 .0181 -.1489
.452		.0950
.497	.0000	



DATE 19 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(881U49)

UPPER WING

ARC11-716 IA14 01+712+512N25

ALPHA( 1 ) = -7.890 BETA( 5 ) = 0.200

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.590		.0274 -.0452
.565		.0000
.650		-.2500
.630		-.1678
.700	.1173	-.1853
.725		-.0613
.750		-.1323 -.2216
.760		-.1008
.775		-.0654 -.1033
.808		-.0862
.834	-.0540	-.0811 -.1086 -.1194
.850		-.1249
.865	-.0676	
.900	-.0930	-.1046 -.0802 -.0759
.905		-.0864
.950		-.0569 -.0526
.955		-.0231
.965	-.0554	

ALPHA( 2 ) = -3.930 BETA( 1 ) = -6.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.000	-.0675 .0186	.5334 .6628 .6040 .5500 .5867
.020		.4974 .4762 .4955 .5048
.025		.4243
.040	.0649	
.045		.3289
.050	-.0726	.2570 .2553 .2721 .2535
.080		.0776
.085		.1922
.086	.1473	
.094	-.0888	
.190		-.1659 -.1369 -.0849 -.0594
.163	.0000	
.177		-.0354
.193	.1304	
.229	-.0245	
.250		-.2892 -.3113 -.3015 -.3028
.274		-.1843
.339	-.1522	



DATE 18 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881049)

UPPER WING

ARC11-716 1A14 01+112+512N25

ALPHA( 2 ) = -3.930 BETA( 1 ) = -8.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.362 .1210  
 .400 .400  
 .402 .402  
 .497 .0000  
 .510 .510  
 .565 .565  
 .600 .600  
 .630 .630  
 .700 .0345  
 .725 .725  
 .750 .750  
 .760 .760  
 .775 .775  
 .806 .806  
 .834 .0386  
 .890 .890  
 .897 .897  
 .865 .1463  
 .900 .1638  
 .905 .905  
 .950 .950  
 .953 .953  
 .965 .1936

-.4050 -.4716 -.5197  
 -.3555  
 -.2693 -.5142  
 .0000  
 -.5927  
 -.1690  
 -.0642  
 -.2528 -.3396  
 -.0372  
 .0725 -.1132  
 .1130  
 .1508 .0311 -.1615  
 .1640  
 .2330  
 .2319 .2007  
 .2346 -.0458  
 .2235

ALPHA( 2 ) = -3.880 BETA( 2 ) = -4.120

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.0000 -.1450 -.0386 .4487 .5847 .5175 .4548 .4917  
 .020 .020  
 .025 .025  
 .040 .040  
 .043 .0143  
 .2966  
 .2541 .2555 .2658 .2488  
 .0950  
 .1506  
 .0988  
 .094 -.1285  
 .150  
 .163  
 .177  
 .193

-.1315 -.1067 -.0693 -.0482  
 -.0081  
 .1137



DATE 19 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RB1U49)

UPPER WING

ARC11-716 1A14 01-T12+S12N25

ALPHA( 2 ) = -3.880 BETA( 2 ) = -4.120

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.229	-.0651					
.250						
.274						
.339						
.362						
.400						
.402						
.497						
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.955						
.965						

ALPHA( 2 ) = -3.890 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2191	-.0713	.4072	.5202	.4201	.3440	.3743
.020							
.025							
.040							
.045							
.050							
.080							
.085							
.086							
.094							

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U49)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHAO ( 2 ) = -3.890 BETAO ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .150 .163 .177 .193 .229 .250 .274 .339 .362 .400 .497 .550 .565 .600 .650 .700 .725 .750 .760 .775 .806 .834 .850 .857 .865 .900 .905 .950 .953 .965

-.0495 -.0627 -.0334 -.0367

.0000 .0528 .1488

-.1738 -.2146 -.2269 -.2596

-.0606

-.2345 -.3294 -.4518

-.1889

.0405 -.0502

.0000

-.4001

-.1768

-.0362

-.1533

-.1224 -.0867

-.0726

-.0278 -.0715

-.0104

-.0327 -.0520 .0112

-.0188

.1365

-.0199

.0597

.1213

.0574

.0411

.0231

ALPHAO ( 2 ) = -3.890 BETAO ( 4 ) = 4.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.008 -.2614 -.0614 .2915 .4032 .3083 .2322 .2939

.052

.025

.040

.045

.059

.2767

.4131

.3549

.3793

.4116

.2416

.2277

.2341

.2509



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 3259

(R81U49)

UPPER WING

ARC11-716 1A14 01+718+512N25

ALPHA(1,2) = -3.690 BETA(4) = 4.080

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8670

X/CW

.080			.1081			
.085		.1387				
.086		.0979				
.094	-.1328					
.150			-.0608	-.0546	.0126	.0597
.163		.0000				
.177			.0167			
.193		.1134				
.229	-.0424					
.250			-.1648	-.1908	-.1496	-.2088
.274			-.0687			
.339		-.0615				
.362	.0796					
.400			-.0488	-.0246		-.2642
.402			-.1161			
.497	.0000					
.550			.0000	.0518	-.0451	
.565						
.600						
.650						
.700	.1262			-.1712	-.1945	-.2807
.725			-.0524			
.750					-.1496	-.1923
.760		-.0943				
.775			-.0540	-.1071		
.809		-.0337				
.834	-.0501					
.850			-.0666	-.0988	-.0943	
.857			-.0469			
.865	.0012					
.900	.0024		-.0565			.0180
.925			-.0112	-.0376		
.930						
.933		.0015	.0032		.0228	
.965	-.0209					

ORIGINAL PAGE IS  
OF POOR QUALITY

TABULATED PRESSURE DATA - 1A14A - VOL. 6

(081U49)

UPPER WING

ALPHA( 2 ) = -3.890 BETA( 3 ) = 0.100

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP				
Y/BW		.4270	.5340	.6730	.7800	.8870
X/CW						
.000	-.4330	-.0860	.1256	.4276	.2167	.1067
.020			.1515	.4349	.3411	.3545
.040		-.0618				
.060			.1730			
.080	-.2552			.2776	.2413	.2416
.100			.1797	.1699		
.120		.0469				
.140	-.1784			.0129	.0036	.0173
.160		.0000				-.0009
.180			.1102			
.200	-.0384			-.0934	-.1183	-.1235
.220		.0130				-.1890
.240						
.260	.0646			-.0693	-.1185	-.2757
.280			-.0563			
.300	.0000			-.0266	-.1069	
.320		.0000				-.3186
.340				-.2242		
.360	.0797			-.2002		
.380			-.1324			-.2494
.400				-.0984	-.1415	
.420		-.1078				
.440						
.460	-.0845		-.1459	-.1206	-.1402	-.1531
.480						
.500						-.0603
.520	-.0863			-.1400	-.1231	
.540	-.1066		-.1229	-.1221	-.0769	
.560						
.580			-.0982			
.600						
.620						
.640						
.660						
.680						
.700						
.720						
.740						
.760						
.780						
.800						
.820						
.840						
.860						
.880						
.900						
.920						
.940						
.960						
.980						
.1000						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81049)

UPPER WING

ARC11-716 1A14 OR+712+S12M25

ALPHA( 3 ) = -.310 BETA( 1 ) = -.8100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2993 .3640 .4270 .5340 .6730 .7000 .8070

X/CW

.000	.0148	-.0132	.5562	.7306	.7010	.6700	.7146
.025			.3110	.3613	.3606	.3586	.3436
.040		.0335					
.045			.1952				
.050	-.0902			.1096	.1074	.1021	.0651
.080				-.0804			
.085			.0163				
.086		.0822					
.094	-.0847			-.2924	-.2945	-.2334	-.2131
.150		.0000					
.163			-.1763				
.177		.0478					
.193	-.0367						
.229				-.4107	-.4465	-.4410	-.4140
.250		-.2247					
.274			-.4500				
.339	.0833			-.4974	-.5980		-.6091
.362							
.400			.0000	-.5493	-.6354		-.6502
.402							
.497	.0000			-.2062		-.6379	
.550							
.565							
.600							
.690	-.2250			-.2190		-.4426	-.5971
.700							
.725							
.750							
.760							
.775							
.808							
.834	.0110						
.850							
.857							
.865	.1055						
.900	.0880						
.905							
.950							
.955							
.965							

(R21149)

**UPPER WING**

ARC11-716 1A14 01+Y12+S12N25

ALPHA( 3 ) = -.339 BETA( 2 ) = -4.030

SECTION (1) UPPER WING

**DEPENDENT VARIABLE CP**

Y/6W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

**MJ/X**

.000	-.0996	-.0660	.4907	.6563	.6268	.6013	.6613
.020				.3500	.3450	.3920	.3425
.040		.0552	.2887				
.045			.1819				
.050	-.1822			.1197	.1082	.1112	.0796
.080				-.0594			
.085			.0191				
.066		.0494					
.094	-.1616						
.150				-.2569	-.2756	-.2107	-.1890
.163		.0000					
.177			-.1367				
.193		.0408					
.229	-.0854						
.250				-.3401	-.3949	-.3910	-.3907
.274			-.2393				
.339		-.1862					
.362	.0933						
.400				-.4227	-.5212		-.5735
.402			-.3552				
.497	.0000						
.550				-.3749	-.5391		
.565			.0000				
.600							-.6103
.650					-.5987		
.700	-.0442				-.1814		
.725				-.1051			
.750						-.3413	-.4886
.760			-.1001				
.775				.0159	-.1421		
.828			.0344				
.834	-.0136						
.890				.0481	-.0100	-.1878	
.857			.0199				
.865	.0609						-.2483
.900	.0427			.0789	.1139		
.905			.0435				
.950				.1212		-.0509	
.953			.0901				
.965	.0341						

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(N81049)

UPPER WING

ARC11-716 IA14 Q1+T12+S12M25

ALPHA(3) = -.340 BETA(3) = .030

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1604	-.1008	.4660	.6175	.5511	.4880	.5369
.020				.3809	.3226	.3107	.3184
.025			.2949				
.040		-.0130					
.045			.2047				
.050	-.2470			.1442	.1109	.0920	.0782
.080				-.0245			
.085			.0631				
.088		.0645					
.094	-.1973						
.150				-.2122	-.2263	-.2055	-.1870
.163		.0000					
.177			-.0598				
.193		.0879					
.229	-.1015						
.250				-.2712	-.3500	-.3783	-.3878
.274			-.1540				
.339		-.1191					
.362	.1068						
.400				-.3851	-.4744		-.5675
.402			-.3178				
.497	.0000						
.550				-.1136	-.4886		
.565			.0000				
.600							-.5868
.650					-.1697		
.700	.0226			-.0649			
.725						-.1753	-.3245
.750			-.1074				
.760				-.0429	-.0713		
.775			-.0448				
.808							
.834	-.0631						
.850				-.0577	-.0171	.0079	
.857			-.0746				
.865	-.0195						-.1738
.900	-.0328			-.0549		.0687	
.905			-.0500				
.950				-.0121		.1067	
.955			-.0331				
.965	-.0688						

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(R811.443)

**UPPER WING**

ARC11-716 1A14 01+712+912M23

$$\text{ALPHA}(\text{S}) = -.340 \quad \text{BETA}(\text{I}) = 4.100$$

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	2990	3640	4270	5340	6730	7600	8870
X/CW							
.000	-.2875	-.1347	.3393	.3106	.4281	.3647	.4139
.025			.2066	.3117	.2517	.2428	.2489
.040		.0349					
.045			.1366				
.050	-.2718			.1179	.0687	.0369	.0338
.060				-.0222			
.085			.0375				
.086		.0700					
.094	-.1795						
.150							
.163		.0000			-.1975	-.2196	-.2044
.177			-.0727				
.193		.0707					
.229	-.0827						
.250			-.1663		-.2821	-.3376	-.4003
.274		-.1317					
.339							
.362	.0904				-.3766	-.4589	-.5406
.400			-.3104				
.497	.0000				-.0795	-.2268	
.550			.0000				
.565							
.620							
.650						-.2426	-.5699
.700	.0105				-.0684		
.725			-.1429				
.750				-.0611	-.1016	-.3679	-.2119
.760			-.0791				
.775							
.806							
.834	-.0922		-.1104		-.0963	.0465	
.850							
.857							
.869	-.0995						
.920	-.0694		-.0937		-.1028	.0380	.1426
.935							
.950			-.0715		-.0540	.1002	
.953							
.965	-.1003						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U49)

UPPER WING

ARC11-716 1A14 01-712-512N23

ALPHA(3) = -.340 BETA(5) = 0.170

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.5387	-.1388	.1009	.4000	.3533	.2707	.3056
.020				.2852	.2523	.2447	.2269
.025			.0619				
.040		-.0722					
.045			.0394				
.050	-.3547			.1334	.1027	.0800	.0442
.055				.0204			
.065			.0468				
.086		-.0632					
.094	-.2284						
.150		.0000		-.1104	-.1634	-.1593	-.1761
.163			-.0097				
.177		.0650					
.193	-.0314						
.229				-.1987	-.2827	-.3224	-.3647
.250			-.0716				
.274		-.0649					
.339	.0551			-.2125	-.3553		-.3261
.362			-.1795				
.400				-.1133	-.2039		
.402			.0000				-.4329
.497	.0000					-.2791	
.550						-.2191	
.565				-.1425			
.600			-.1568			-.2148	-.1474
.650	-.0106		-.1302	-.1330	-.1599		
.700							
.725				-.1517	-.1856	-.1279	
.750			-.1758				
.760							
.775							
.806							
.834	-.1832						
.850							
.857							
.865	-.1135						.0182
.900	-.1340			-.1731	-.1358		
.905			-.1548				
.950				-.1686			-.0125
.955			-.1552				
.965	-.1849						

(R81U49)

UPPER WING

ARC11-716 1A14 01-T12-S12N25

ALPHA(4) = 4.130 BETA(1) = -0.000

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		
.0000	.0698	.5228
.0200		.7625
.0250		.7785
.0400	.1549	.7338
.0450		.7118
.0500		.1560
.0600		.1592
.0650		.0882
.0800		.0174
.0850		
.0940		
.1500		
.1630		
.1770		
.1930		
.2290		
.2500		
.2740		
.3390		
.3620		
.4030		
.4020		
.4970		
.5500		
.5650		
.6000		
.6300		
.7000		
.7250		
.7500		
.7750		
.8000		
.8340		
.8500		
.8650		
.9000		
.9250		
.9500		
.9530		
.9650		

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 5

(RB1U49)

UPPER WING

ARC11-716 IA14 01+112+S12N25

ALPHA(4) = 4.010 BETA(2) = -4.010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2993 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.0335	-.1720	.4589	.7179	.7076	.6718	.6584
.020				.1559	.1576	.0860	.0053
.025			.1268				
.040		-.0750					
.045			.0061				
.050	-.1550			-.0588	-.0918	-.1310	-.2275
.060				-.2242			
.085			-.1501				
.086		-.0204					
.094	-.1646						
.130		.0000		-.4336	-.4395	-.4142	-.4214
.163			-.2211				
.177		-.0685					
.193							
.229	-.1250			-.4706	-.5764	-.5739	-.5606
.250			-.2937				
.274		-.2090					
.339							
.362	.0236			-.5344	-.6807		-.7036
.400			-.4537				
.402				-.5490	-.6775		
.497	.0500		.0000				
.550							-.6854
.565							
.600							
.650						-.7114	
.700	-.3427			-.2183			
.725			-.2349			-.5119	-.6558
.750							
.760			-.1388	-.1759	-.3128		
.775							
.808			-.0078				
.834	-.0127						
.850				-.0879	-.1668	-.4341	
.837			-.0214				
.865	.0279						
.900	.0173			-.0236	-.0980		-.4233
.905			-.0093				
.950			.0253				-.3937
.955							
.965	-.0572		-.5153				

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U49)

UPPER WING

ARC11-716 1A14 CL+T12+S12M25

ALPHA( 4 ) = 4.000 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1716	-.2278	.4216	.6594	.6410	.5962	.5962
.020				.1720	.1587	.0864	.0051
.025			.1449				
.040		-.0909					
.045			.0506				
.050	-.2376			-.0596	-.0824	-.1258	-.2177
.080				-.1943			
.085			-.0835				
.086		-.0257					
.094	-.2484						
.150		.0000		-.3497	-.4146	-.3954	-.4163
.163			-.1682				
.177		.0334					
.193							
.229	-.1349			-.4231	-.5259	-.5551	-.5378
.250			-.2674				
.274		-.1969					
.339	.0907						
.362				-.4866	-.6246		-.6987
.402		-.4103					
.402				-.4697	-.6199		
.497	.0000		.0000				
.590						-.6805	
.565							
.600							
.650							
.700	-.1961			-.2046			
.723				-.2002			
.750							
.760			-.1884			-.4942	-.6122
.775				-.0924	-.2745		
.808			-.0551				
.834	-.0597						
.850			-.0696	-.0727	-.1693	-.4046	
.857							
.885	-.0502						
.900	-.0438			-.0711			-.3929
.905			-.0937		.0202		
.930				-.0909		-.2153	
.933			-.1005				
.985	-.1280						



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1U49)

UPPER WING

ARCL1-716 1A14 Q1+T12+512N25

ALPHA (4) = 4.030 BETA (4) = 4.110

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3251	-.3070	.2573	.5572	.5272	.4963	.5344
.020				.1468	.1119	.0567	-.0003
.025			.0924				
.040		-.0521					
.045			.0270				
.050	-.2987			-.0516	-.1049	-.1426	-.2179
.060				-.1892			
.085			-.0825				
.086		.0352					
.094	-.2706						
.150		.0000		-.3447	-.4098	-.3903	-.4049
.165							
.177			-.1785				
.195		.0027					
.229	-.0822						
.250				-.4063	-.4976	-.5409	-.5446
.274		-.2109					
.339			-.2666				
.362	.0522			-.4772	-.5833		-.6879
.400			-.3943				
.402				-.2854	-.5752		
.497	.0000		.0000				-.6769
.550							
.565							
.600							
.650							
.700	-.1136			-.6347			
.725				-.1365			
.750					-.3003	-.5000	
.760			-.1694				
.775				-.0863	-.1638		
.808			-.1054				
.834	-.1032						
.850			-.1346	-.1010	-.0743	-.1695	
.857							
.865	-.0952						-.2999
.900	-.1014			-.1158	.0036		
.905			-.1312				
.950				-.0965		-.0407	
.953			-.1255				
.965	-.1506						

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(RB1U49)

UPPER WING

ARC11-716 1A14 01+T12+S12M25

ALPHA(4) = 4.030 BETA(5) = 0.210

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CU

.000	-.0253	-.1692	.0780	.2830	.4406	.4149	.4422
.020				.0339	.1191	.0677	.0037
.025			.0269				
.040		-.0573					
.045			.0047				
.050	-.4556			-.0337	-.0727	-.1108	-.1971
.060					-.1444		
.065							
.086		-.1036					
.094	-.3704			-.2483	-.3329	-.3640	-.3756
.150			.0000				
.163							
.177			-.1338				
.193		.0557					
.229	.0173			-.3209	-.4209	-.4895	-.5284
.250			-.2055				
.274		-.1621					
.339							
.362	.0472			-.3783	-.5238		-.6519
.400							
.402							
.497	.0000			-.1804	-.3945		
.550			.0000				-.6436
.600							
.650				-.1921	-.2345	-.4283	
.700	-.0610						
.725							
.750							
.760							
.775				-.2071	-.1831	-.1865	
.800			-.1733				
.834	-.1447						
.850				-.1847	-.1658	-.0935	
.857			-.2082				
.865	-.1322						
.900	-.1805			-.1942			-.2051
.905				-.1927	-.0954		
.910				-.1799		.0087	
.915							
.930							
.935							
.965	-.8320						



(R81U49)

DATE 13 JAN 75      TABULATED PRESSURE DATA - 1A14A - VOL. 5      UPPER WING

ARC11-716 1A14 01+112+312N25

ALPHA( 3) = 8.040    BETA( 1) = -8.090

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/W    .2990    .3640    .4270    .5340    .6730    .7800    .8870

X/CW

.009	.0763	-.0492	.5094	.8076	.7579	.6707	.6202
.020				-.0062	-.0177	-.1383	-.2319
.025			.0213				
.040		-.0198					
.045			-.1217				
.050	-.0439			-.2348	-.2795	-.3403	-.4514
.080				-.3992			
.085			-.2864				
.086		.0094					
.094	-.0837			-.6335	-.6060	-.5999	-.6062
.150		.0000					
.163			-.4657				
.177		-.1955					
.193	-.3857			-.7298	-.7317	-.7289	-.7937
.229			-.5276				
.250		-.3999					
.274				-.7862	-.8341		-.8118
.339	.0240						
.367			-.6144				
.409				-.8057	-.8075		
.432	.0000		.0000				
.497							
.550							
.565							
.609							
.650							
.700	-.5287			-.4668			
.725							
.750			-.3259				
.760				-.4130	-.6466		
.775			-.2125				
.808							
.834	-.0910			-.2728	-.6005	-.5789	
.850			-.1621				
.857							
.865	-.0121			-.2029			
.900	.0186						
.905			-.0841				
.930				-.5147			
.933			-.1414				
.965	-.1047		-.0287				

-.5115

-.6562



(881049)

UPPER WING

ARC11-716 1A14 01+712+912M25

ALPHA( 5) = 7.970 BETA( 2) = -4.030

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7800
.8870		
.002	-.0399	-.2006
.020	.3776	.7337
.025	.7106	.8279
.040	-.0391	-.0217
.045	-.1396	-.2809
.050	-.0553	
.060	-.1639	
.065	-.1710	
.080	-.2386	-.2609
.085	-.3590	-.4630
.086	-.3918	
.094	-.3277	
.150	-.1084	
.163	-.1710	
.177	.0000	
.193	-.5885	-.5834
.229	-.5871	-.6073
.250	-.4322	
.274	-.1773	
.339	-.1731	
.362	-.8412	-.7047
.403	-.7114	-.7008
.402	-.3687	
.497	-.2630	
.550	-.0296	
.565	.453	
.600	-.5827	-.8034
.650	-.4971	
.705	-.6272	-.7081
.725	.0000	
.750	-.7852	
.760	-.7798	
.775	-.2661	
.806	-.6230	-.6739
.834	-.2365	
.850	-.2616	-.6158
.890	-.0994	
.897	-.2063	-.4657
.865	-.5278	
.900	-.0905	
.905	-.1315	
.950	-.0733	-.8703
.955	-.0920	
.965	-.4744	
.965	-.0759	
.965	-.1187	



(R81U49)

DATE 13 JAN 73 TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARCL1-716 1A14 01+112+312425 UPPER WING

ALPHA (S) = 0.060 BETA (S) = .040

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7830 .8870

X/CW

.000	-.1938	-.4403	.3487	.6594	.6542	.5718	.5143
.020				.0566	-.0203	-.1454	-.2736
.025			.0198				
.040		-.2664					
.045			-.0804				
.050	-.2431			-.1934	-.2657	-.3427	-.4754
.060					-.3476		
.065			-.2256				
.066		-.1180					
.094	-.2523			-.9509	-.5770	-.5873	-.8179
.150		.0000					
.163			-.3127				
.177		-.0467					
.193	-.1890			-.5781	-.6936	-.7106	-.7140
.229			-.3947				
.250		-.2646					
.274				-.6080	-.7903		-.8163
.339	.0064						
.362			-.5049				
.400				-.5969	-.7891		
.432	.0000		.0000				-.7937
.497						-.7909	
.550					-.2564		
.565							
.600							
.650	-.4375			-.2913			
.700							
.725							
.750							
.760				-.2222	-.2408	-.3762	
.775							
.806				-.0961			
.834	-.0370						
.850				-.1777	-.2623	-.3358	
.857			-.1307				
.885	-.0467						
.900	-.0954			-.1429			-.6504
.905				-.1335	-.1588		
.930					-.0995	-.5106	
.953				-.1525			
.985	-.1884						

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DATE 10 JAN 75

TABULATED PRESSURE DATA - IA14A - VOL. 6

PAGE 3244

(RBIU48)

UPPER WING

ALPHA( 3) = 8.060 BETAO ( 4) = 4.140

ARC11-716 IA14 01+712+312+25

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3635	-.4412	.0904	.3298	.3603	.5092	.4677
.025			-.0629	-.0479	-.1909	-.2734	
.050		-.0750					
.075		-.1705					
.100			-.1145				
.125	-.3174		-.2291	-.2717	-.3365	-.4673	
.150			-.3569				
.175		-.0444	-.1924				
.200	-.3052			-.5217	-.5591	-.5702	-.6031
.225		.0000					
.250		-.0746	-.2842				
.275	-.0804		-.3532	-.5281	-.6495	-.6868	-.6980
.300		-.2809					
.325	.0053			-.5023	-.7399		-.8018
.350			-.4598				
.375	.0000			-.4401	-.6992		
.400						-.7750	
.425	-.2112			-.2402			
.450			-.2237			-.6275	-.8281
.475		-.2229		-.1371	-.3780		
.500		-.1523					
.525	-.1382		-.1320	-.2537	-.5439		
.550		-.1779					
.575	-.1301			-.1515			-.5442
.600	-.1388		-.1797	-.0584			
.625			-.1399			-.3480	
.650			-.1841				
.675	-.1971						



(R81049)

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01+712+312423

UPPER WING

DATE 15 JAN 75

ALPHA (  $\alpha$  ) = 0.055 BETA (  $\beta$  ) = 0.270

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/DW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.6835	-.2121	-.0290	.1106	.4393	.4376	.4118
.020				-.1834	-.0934	-.1330	-.2680
.025			-.0334				
.040		-.1058					
.045			-.0799				
.050	-.3456			-.2411	-.2894	-.3335	-.4566
.060				-.2930			
.065			-.1170				
.086		-.1064					
.094	-.5085			-.3474	-.5356	-.5557	-.5899
.150		.0000					
.163							
.177		-.0650					
.193			-.1903				
.229	.0357			-.4022	-.5831	-.6657	-.6891
.250			-.2710				
.274		-.2243					
.339	.0121			-.4367	-.6306		-.7884
.362			-.0701				
.400				-.2430	-.4747		
.497	.0000						
.550							
.565							
.600							
.650					-.6206		-.7520
.700	-.0311			-.2673			
.725				-.2426		-.3493	-.6141
.750			-.2576				
.760				-.2256	-.2570		
.775			-.2305				
.808							
.834	-.1001			-.2370	-.1961	-.1956	
.850			-.2537				
.857							
.865	-.1996			-.2323			-.4839
.900	-.2176			-.2378	-.1038		
.905				-.2072		-.0711	
.950			-.2277				
.953							
.965	-.2659						

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DATE 13 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(881050)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 1 ) = -0.010 BETA( 1 ) = -0.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .950 .933 .965 .2091 -.0398

ALPHA( 1 ) = -0.010 BETA( 2 ) = -4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .000 -.0793 -.1598 .2072 .4554 .3745 .3133 .3694

.020 .025 .040 .045 .090 .085 .086 .094

.150 .163 .177 .193 .229 .250 .274 .339

.362 .400 .432 .497 .550 .565 .600 .650

.700 .725 .750 .760 .775 .808 .834 .850

.086 .094 .150 .163 .177 .193 .229 .250

.274 .339 .362 .400 .432 .497 .550 .565

.600 .650 .700 .725 .750 .760 .775 .808

.834 .850 .086 .094 .150 .163 .177 .193

.229 .250 .274 .339 .362 .400 .432 .497

.550 .565 .600 .650 .700 .725 .750 .760

.775 .808 .834 .850 .086 .094 .150 .163

.177 .193 .229 .250 .274 .339 .362 .400

.432 .497 .550 .565 .600 .650 .700 .725

.750 .760 .775 .808 .834 .850 .086 .094

.150 .163 .177 .193 .229 .250 .274 .339

.362 .400 .432 .497 .550 .565 .600 .650

.700 .725 .750 .760 .775 .808 .834 .850

.086 .094 .150 .163 .177 .193 .229 .250

.274 .339 .362 .400 .432 .497 .550 .565

.600 .650 .700 .725 .750 .760 .775 .808

.834 .850 .086 .094 .150 .163 .177 .193

.229 .250 .274 .339 .362 .400 .432 .497

.550 .565 .600 .650 .700 .725 .750 .760

.775 .808 .834 .850 .086 .094 .150 .163

.177 .193 .229 .250 .274 .339 .362 .400

.432 .497 .550 .565 .600 .650 .700 .725

.750 .760 .775 .808 .834 .850 .086 .094

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(081050)

UPPER WING

ARC11-716 IA14 01+712+S12N25

ALPHAO ( 1 ) = -0.010 BETAO ( 2 ) = -4.000

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.097			.1095				
.095	.1373						
.900	.1381		.1137				-.0350
.935		.1299	.1374				
.950			.1422			.1856	
.953			.1587				
.965	.1425						

ALPHAO ( 1 ) = -0.030 BETAO ( 3 ) = .030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1342	-.2005	.1172	.3422	.2552	.2135	.2552
.020				.5304	.4702	.5102	.5057
.025			.3008				
.040		-.0623					
.045			.2718				
.050	-.1203			.3775	.3805	.3882	.3728
.080			.1946	.2429			
.085							
.086		.0546					
.094	-.0665						
.150		.0200		.0513	.0858	.1221	.1226
.163							
.177			.0953				
.193		.1240					
.229	-.0401						
.250				-.0832	-.0833	-.0744	-.1019
.274			-.0569				
.339		-.0214					
.362	.0858						
.400			-.2107	-.2310			-.2943
.402			-.1721				
.497	.0000			.0598	-.2003		
.550			.0000				
.565							
.600							
.650							
.700	.1310						
.725				-.0821			
.750				-.1214			
.760				.0603		.0317	-.1042
			.0104				



$$\text{ALPHAO}(1) = -0.000 \quad \text{BETAO}(3) = .030$$

SECTION ( UPPER WING	DEPENDENT VARIABLE CP
1	0.000
2	0.000
3	0.000
4	0.000
5	0.000
6	0.000
7	0.000
8	0.000
9	0.000
10	0.000
11	0.000
12	0.000
13	0.000
14	0.000
15	0.000
16	0.000
17	0.000
18	0.000
19	0.000
20	0.000
21	0.000
22	0.000
23	0.000
24	0.000
25	0.000
26	0.000
27	0.000
28	0.000
29	0.000
30	0.000
31	0.000
32	0.000
33	0.000
34	0.000
35	0.000
36	0.000
37	0.000
38	0.000
39	0.000
40	0.000
41	0.000
42	0.000
43	0.000
44	0.000
45	0.000
46	0.000
47	0.000
48	0.000
49	0.000
50	0.000
51	0.000
52	0.000
53	0.000
54	0.000
55	0.000
56	0.000
57	0.000
58	0.000
59	0.000
60	0.000
61	0.000
62	0.000
63	0.000
64	0.000
65	0.000
66	0.000
67	0.000
68	0.000
69	0.000
70	0.000
71	0.000
72	0.000
73	0.000
74	0.000
75	0.000
76	0.000
77	0.000
78	0.000
79	0.000
80	0.000
81	0.000
82	0.000
83	0.000
84	0.000
85	0.000
86	0.000
87	0.000
88	0.000
89	0.000
90	0.000
91	0.000
92	0.000
93	0.000
94	0.000
95	0.000
96	0.000
97	0.000
98	0.000
99	0.000
100	0.000

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8670
------	-------	-------	-------	-------	-------	-------	-------

MJ/X

0000 0000 0000

**.9763**

	.0516	.0624	.0541	.0731
--	-------	-------	-------	-------

**.0584**

000	.0932	.1371
.865	.1101	

0180'

0265 .1450

**.1129**

ALPHA( 2 ) = -8.010    BETAC ( 4 ) = 4.100

DEPENDENT VARIABLE CP

7/8W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
------	-------	-------	-------	-------	-------	-------	-------

**X/CW**

Variable	Mean	Standard deviation	Skewness	Kurtosis
Age	35.27	10.12	-.001	3.06
Gender	1.00	1.00	-.001	3.06
Marital status	1.00	1.00	-.001	3.06
Education	12.50	2.50	-.001	3.06
Income	1.00	1.00	-.001	3.06
Occupation	1.00	1.00	-.001	3.06
Religion	1.00	1.00	-.001	3.06
Political affiliation	1.00	1.00	-.001	3.06
Health status	1.00	1.00	-.001	3.06
Life satisfaction	1.00	1.00	-.001	3.06
Stress	1.00	1.00	-.001	3.06
Depression	1.00	1.00	-.001	3.06
Loneliness	1.00	1.00	-.001	3.06
Self-esteem	1.00	1.00	-.001	3.06
Resilience	1.00	1.00	-.001	3.06
Optimism	1.00	1.00	-.001	3.06
Gratitude	1.00	1.00	-.001	3.06
Forgiveness	1.00	1.00	-.001	3.06
Empathy	1.00	1.00	-.001	3.06
Prosocial behavior	1.00	1.00	-.001	3.06
Altruism	1.00	1.00	-.001	3.06
Compassion	1.00	1.00	-.001	3.06
Kindness	1.00	1.00	-.001	3.06
Generosity	1.00	1.00	-.001	3.06
Helpfulness	1.00	1.00	-.001	3.06
Cooperativeness	1.00	1.00	-.001	3.06
Teamwork	1.00	1.00	-.001	3.06
Leadership	1.00	1.00	-.001	3.06
Communication	1.00	1.00	-.001	3.06
Conflict resolution	1.00	1.00	-.001	3.06
Problem solving	1.00	1.00	-.001	3.06
Decision making	1.00	1.00	-.001	3.06
Goal setting	1.00	1.00	-.001	3.06
Time management	1.00	1.00	-.001	3.06
Organization	1.00	1.00	-.001	3.06
Productivity	1.00	1.00	-.001	3.06
Efficiency	1.00	1.00	-.001	3.06
Effectiveness	1.00	1.00	-.001	3.06
Quality of work	1.00	1.00	-.001	3.06
Job satisfaction	1.00	1.00	-.001	3.06
Work-life balance	1.00	1.00	-.001	3.06
Work engagement	1.00	1.00	-.001	3.06
Work commitment	1.00	1.00	-.001	3.06
Work motivation	1.00	1.00	-.001	3.06
Work performance	1.00	1.00	-.001	3.06
Work productivity	1.00	1.00	-.001	3.06
Work efficiency	1.00	1.00	-.001	3.06
Work effectiveness	1.00	1.00	-.001	3.06
Work quality	1.00	1.00	-.001	3.06
Work life	1.00	1.00	-.001	3.06
Work balance	1.00	1.00	-.001	3.06
Work engagement	1.00	1.00	-.001	3.06
Work commitment	1.00	1.00	-.001	3.06
Work motivation	1.00	1.00	-.001	3.06
Work performance	1.00	1.00	-.001	3.06
Work productivity	1.00	1.00	-.001	3.06
Work efficiency	1.00	1.00	-.001	3.06
Work effectiveness	1.00	1.00	-.001	3.06
Work quality	1.00	1.00	-.001	3.06
Work life	1.00	1.00	-.001	3.06
Work balance	1.00	1.00	-.001	3.06
Work engagement	1.00	1.00	-.001	3.06
Work commitment	1.00	1.00	-.001	3.06
Work motivation	1.00	1.00	-.001	3.06
Work performance	1.00	1.00	-.001	3.06
Work productivity	1.00	1.00	-.001	3.06
Work efficiency	1.00	1.00	-.001	3.06
Work effectiveness	1.00	1.00	-.001	3.06
Work quality	1.00	1.00	-.001	3.06
Work life	1.00	1.00	-.001	3.06
Work balance	1.00	1.00	-.001	3.06
Work engagement	1.00	1.00	-.001	3.06
Work commitment	1.00	1.00	-.001	3.06
Work motivation				

	.4735	.4079	.4333	.4603
--	-------	-------	-------	-------

**0262'**

2307  
-0730

	.6391	.3447	.3431	.3353	.3612
--	-------	-------	-------	-------	-------

**.1676**

750344

1560  
- .0951

6000.  
6000.

1467

1162

**10551-2**

- 0038

-0217

0000.

-1562

**0000.**

**.1045 .0071**

0000



(R81U50)

UPPER WING

ALPHA(1) = -0.010 BETA(4) = 4.100

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.700	.1397			.0590		.0204	-.0423
.725							
.750							
.760							
.775							
.808							
.834	.0331			.0461	.0371	.0402	
.857							
.865	.0617						.0743
.900	.0619			.0281			
.925				.1081	.0516		
.930				.0686		.0917	
.953				.1252			
.965	.1037						

ALPHA(1) = -0.030 BETA(5) = 0.180

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8870
X/CW							
.000	-.2940	-.1184	.1142	.2351	.0960	.0037	.0239
.020				.4367	.3406	.3735	.3625
.025				.2525			
.040							
.045							
.050	-.1809			.3199	.3100	.3096	.2873
.080				.2146			
.085				.1926			
.086							
.094	-.1326	.0773		.0681	.0668	.1041	.1196
.150		.0000					
.163							
.177							
.193		.1232					
.229	-.0255						
.250							
.274							
.339		-.0114					
.362	.1099						
.400							
.402							
.497							



DATE 19 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBLU90)

UPPER WING

ARC11-716 1A11 01-T12-S12N25

ALPHA(1) = -0.930 BETA(1) = 0.160

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.550				.0910	.0482		
.565			.0000				
.600							
.690							
.700	.1247						
.725							
.750							
.760							
.775							
.808							
.834							
.850							
.857							
.865							
.900							
.935							
.950							
.953							
.965							

ALPHA(2) = -3.930 BETA(2) = -0.050

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.050	.0060	-.0577	.3761	.6490	.6157	.5641	.5963
.020				.5256	.5166	.5179	.5350
.025			.3541				
.040		.0680					
.045			.2756				
.050	-.0271			.2951	.3072	.3063	.2975
.060				.1170			
.065			.1454				
.066		.1044					
.094	-.0492						
.150		.0000					
.165							
.177							
.193		.0046					
.229	-.0187						
.250							
.274							
.339							

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1050)

UPPER WING

ARC11-716 1A14 OR+112+512M25

ALPHAO ( 2 ) = -3.930 BETAO ( 1 ) = -8.050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.362	.0798	
.403		-.3577 -.4102 -.4274
.402		-.3276
.497	.0050	
.550		-.4109 -.4554
.565		.0000
.600		-.4457
.650		-.4893
.700	-.2350	
.725		-.1161
.750		-.4929 -.4870
.760		
.775		-.0376
.806		-.0145 -.2679
.834	.0491	
.850		.1174
.857		.0604 -.1754 -.2630
.865	.1532	
.900	.1402	-.126
.905		.0858
.930		.1152
.933		-.0116
.965	.0888	
		.1071
		-.1857
		.1091

ALPHAO ( 2 ) = -3.930 BETAO ( 2 ) = -4.020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270 .5340 .6730 .7800 .8870
.362	.0798	
.403		-.3577 -.4102 -.4274
.402		-.3276
.497	.0050	
.550		-.4109 -.4554
.565		.0000
.600		-.4457
.650		-.4893
.700	-.2350	
.725		-.1161
.750		-.4929 -.4870
.760		
.775		-.0376
.806		-.0145 -.2679
.834	.0491	
.850		.1174
.857		.0604 -.1754 -.2630
.865	.1532	
.900	.1402	-.126
.905		.0858
.930		.1152
.933		-.0116
.965	.0888	
		.1071
		-.1857
		.1091



DATE 10 JAN 75 TABULATED PRESSURE DATA - IAI4A - VOL. 6

(RB1050)

UPPER WING

ARC11-716 IAI4 OI+T12+S12MES

ALPHA( 2 ) = -3.930 BETA( 2 ) = -4.320

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2930	.3640	.4270
.5340	.6730	.7800
.8870		

.229	-.0341	-.2267	-.2304	-.2271	-.2238
.290					
.274		-.1518			
.339	-.1265				
.362	.0340				
.400		-.3397	-.3083		-.4213
.432		-.3092			
.497	.0000				
.530		-.3578	-.4202		
.565		.0000			
.600					
.630					
.700	-.1227		-.1512		-.4519
.725		-.0695			
.750				-.4107	-.4611
.760		-.0494			
.775		.0537	-.1190		
.808		.0904			
.834	.0279				
.850		.0715	-.0387	-.1991	
.857		.0660			
.865	.1030				-.2369
.900	.0876	.0632	.0652		
.935		.0697			
.950		.0712	-.1604		
.953		.0625			
.955	.0538				

ALPHA( 2 ) = -3.940 BETA( 3 ) = .050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	X/CW	CP
.2990	.3640	.4270
.5340	.6730	.7800
.8870		

.000	-.1995	-.1137	.2822	.4843	.4252	.3623	.3984
.020				.4802	.4252	.4315	.4362
.025			.2886				
.040		-.0584					
.045			.2278				
.050	-.1702			.2704	.2694	.2569	.2434
.060				.1262			
.095							
.096			.1264				
.099	.0099						
.094	-.1475						

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01U50)

UPPER WING

ALPHA( 2 ) = -3.940 BETA( 3 ) = .050

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .8990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .130 .163 .177 .193 .229 .250 .274 .339 .362 .402 .497 .550 .563 .600 .650 .700 .725 .750 .775 .836 .834 .850 .857 .863 .900 .905 .933 .965

.0000 .0744 .0021 .1014 .0906 .0276 .0000

-.0619 -.0620 -.0253 -.0041

-.1843 -.1969 -.1989 -.2141

-.3028 -.3429 -.4015

-.2632

-.2956 -.3841

.0000

-.4447

-.0015

-.0332

.0531 -.0015

.0399 .0302 -.1033

.0109

.0192 .0554

.0232 .1021

.0205

.0005

ALPHA( 2 ) = -3.940 BETA( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .8990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW .000 .000 .020 .025 .040 .045 .050

-.2347 -.1072 .2444 .4163 .3820 .2560 .3425

.0000 .0000 .0000 .0000 .0000 .0000 .0000

.0000 .0000 .0000 .0000 .0000 .0000 .0000





(R01U50)

UPPER WING

ARC11-71F 1A14 01+T12+S12M25

ALPHA(2) = -3.950 BETA(3) = 0.110

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.3574	-.0497	.1014	.3820	.2434	.1677	.1932
.020				.3747	.3209	.3309	.3237
.025			.1839				
.040		-.0100					
.045			.1854				
.050	-.2851			.2331	.2156	.2562	.1878
.080				.1265			
.085			.1364				
.086		.0920					
.094	-.1932			-.0352	-.0302	-.0102	-.0137
.150		.0500					
.163			.0189				
.177		.1097					
.193							
.229	-.0480			-.1308	-.1654	-.1744	-.2078
.250			-.0688				
.274		-.0768					
.339							
.362	.0928			-.2425	-.2735		-.3718
.400			-.2177				
.497	.0500			-.0448	-.0976		
.550			.0500				
.565							
.600						-.3686	
.650					-.1048		
.700	.0119			-.0038	-.1662		
.725						-.0354	-.1032
.750			-.0590				
.760				.0510	-.0591		
.775			-.0150				
.808							
.854	-.0146			-.0220	-.0295	-.0888	
.890			-.0491				
.897							
.865	.0035						.0538
.900	-.0201			-.0475	-.0345		
.905			-.0424				
.950				-.0443		.0232	
.953			-.0297				
.965	-.0498						



TABULATED PRESSURE DATA - IA14A - Vol. 6

**(R01USQ)**

**UPPER WING**

ARC11-716 1A14 Q14T12+S12N25

ALPHAO( 3) = -.345 BETA0( 1) = -8.060

SECTION ( 1 ) UPPER WING

Y/8W	.2390	.3640	.4270	.5340	.6730	.7600	.8070
------	-------	-------	-------	-------	-------	-------	-------

**MJ/X**

.000	.0066	.0259	.4732	.7401	.7012	.6414	.6776
.020				.4177	.4151	.3980	.4142
.025		.0261					
.040							
.045			.1720				
.050	-.0334			.1720	.1716	.1630	.1560
.060				-.0147			
.085			.0236				
.086	.0676						
.094	-.0799						
.150		.0090		-.2370	-.2250	-.1569	-.1206
.163							
.177			-.1523				
.193		.0158					
.229	-.0589						
.250				-.3585	-.3828	-.3546	-.3261
.274			-.2615				
.339	-.1999						
.362	.0330			-.4377	-.5200		-.5110
.400			-.3983				
.402							
.497	.0090			-.4909	-.5572		
.550			.0090				-.5196
.565						-.5455	
.600							
.650							
.700	-.3229			-.1873	-.2132		
.725							
.750			-.1230			-.5655	-.4972
.760							
.775				-.1369	-.4079		
.808			.0016				
.834	-.0022						
.850				-.0694	-.2941	-.3572	
.857			.0447				
.865	.1101						
.900	.1027			-.0590	-.2621		-.4157
.905			.0656				
.950				-.0080		-.2673	
.953			.0621				
.965	.0345						



(0811/50)

UPPER WING

ALPHA( 3 ) = -.350 BETA( 2 ) = -4.020

ARC11-716 1A14 ON-T12-812M3

DEPENDENT VARIABLE CP

SECTION ( 3 ) UPPER WING

Y/BW	.2990	.3840	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.0661	-.0512	.3992	.6397	.6139	.5619	.5897
.020				.3952	.3889	.3689	.3713
.025			.2586				
.040		-.0096					
.045			.1530				
.050	-.1027			.1644	.1634	.1461	.1357
.080				-.0106			
.085			.0318				
.086		.0386					
.094	-.1385						
.150							
.163		.0003					
.177			-.1488				
.193		.0083					
.229	-.0467						
.250							
.274			-.2476				
.339		-.2048					
.362	.0140						
.400							
.402							
.497	.0000						
.550							
.565							
.600							
.650							
.700	-.2855						
.725							
.750							
.760							
.775							
.808							
.834	-.0041						
.850							
.857							
.885	.0784						
.900	.0615						
.905							
.930							
.933							
.965	-.0051						





TABULATED PRESSURE DATA - IAI14A - VOL. 6

(R01050)

UPPER WING

DATE 10 JAN 75

ARC11-756 IAI14 01+718+512M25

ALPHA(3) = -.380 BETA(4) = 4.080

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.4340	.6730	.7800	.8870
X/CW							
.000	-.2492	-.1315	.2742	.4974	.4289	.4122	.4482
.020				.3133	.2873	.2988	.2982
.025			.1807				
.040		-.0394					
.045			.1204				
.050	-.2711		.1221	.1027	.1141	.1019	
.080			-.0121				
.085			.0248				
.086		.0106					
.094	-.2190						
.130			-.1682	-.1844	-.1472	-.1323	
.163		.0000					
.177		.0431	-.0758				
.193							
.225	-.0940			-.2853	-.2949	-.3034	-.3197
.250			-.1611				
.274		-.1340					
.339	.0317			-.3514	-.3965		-.4854
.362			-.3084				
.400				-.3137	-.4260		
.402	.0000						-.4844
.497	.0000				-.1655		
.550				-.0355		-.2866	-.3338
.600							
.650	-.0929						
.700			-.0920	.0118	-.0472		
.725							
.750			.0031				
.760							
.775			-.0262				
.800	-.0469			-.0020	-.0076	-.1384	
.834							
.850							
.857							
.885	.0127						
.900	-.0088			-.0302	.0290		-.1720
.905			-.0235				
.930			-.0239			.0687	
.933			-.0227				
.965	-.0434						



DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RB1050)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 3) = -.360 BETA( 5) = 0.120

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8873

X/CW

.000	-.4797	-.1061	.1127	.3853	.3461	.2835	.3143
.020				.2596	.2624	.2351	.2204
.025			.0910				
.040		-.0794					
.045			.0725				
.050	-.3607			.1026	.0979	.0827	.0577
.060				-.0152			
.085			.0312				
.086		-.0379					
.094	-.2436						
.150		.0090		-.1417	-.1649	-.1529	-.1496
.163			-.0503				
.177		.0606					
.193							
.229	.0030			-.2359	-.2711	-.2943	-.3219
.250			-.1405				
.274		-.1420					
.339							
.362	.0549			-.3204	-.3813		-.4703
.400			-.2765				
.402							
.497	.0050			-.1591	-.3709		
.550			.0000				
.565							-.4682
.600							
.650							
.700	-.0605			-.1936		-.4219	
.725			-.0744				
.750						-.1801	-.3084
.760			-.1067				
.775				-.0343	-.0531		
.806			-.0466				
.834	-.0626						
.850				-.0561	-.0521	-.0315	
.857			-.0791				
.885	-.0304						
.900	-.0578			-.0771			-.0952
.905			-.0807		-.0592		
.950				-.0792		.0515	
.953			-.0852				
.965	-.1150						

ALPHA ( 1 ) = 4.019      BETA ( 1 ) = -8.080

(RBI 1450)

**UPPER WING**

ARC11-716 1A:4 01+T12+S12N25

SECTION ( 1 ) UPPER WING

Y/RW	X/CW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
.030		.1402	-.0069	.4623	.7983	.7804	.7369	.7649
.020				.1631	.2261	.2724	.2176	.1707
.025								
.040			-.0174					
.045				.0469				
.050		-.0157			-.0141	.0033	-.0071	-.0730
.080					-.1668			
.085				-.1219				
.086			.0243					
.094		-.0235						
.150					-.4076	-.3617	-.3148	-.2964
.163			.0000					
.177				-.2641				
.193			-.0655					
.229		-.0630						
.250				-.3704	-.4917	-.5043	-.4763	-.4448
.274			-.2910					
.339		-.0121			-.5633	-.6223		-.5909
.400				-.4743				
.402		.0000			-.5721	-.6369		-.5882
.497				.0000			-.6106	
.550					-.3619	-.2367		
.565							-.5577	-.5690
.620								
.650		-.3950						
.700								
.725				-.1992				
.750					-.2332	-.5133		
.760				-.1472				
.775								
.800								
.834		-.0520			-.1604	-.4526	-.4744	
.850				-.0697				
.857								
.865		.0604			-.0867			-.5197
.900		.0634		-.0209	-.3996			
.909					-.0413		-.3819	
.950				.0246				
.953								
.965		-.0099						

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 ARC11-716 1A14 01+T12+S12N25 UPPER WING  
 (881050)

ALPHA( 4 ) = 4.019 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING	DEPENDENT VARIABLE CP									
Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870			
X/CW										
.000	.0055	-.1096	.4110	.7010	.6916	.6377	.6099			
.020			.2192	.2289	.1773	.1380				
.025			.1345							
.040		-.0894								
.045			.0193							
.050	-.1030		-.0097	-.0200	-.0371	-.0955				
.080			-.1739							
.085			-.1244							
.086		-.0272								
.094	-.1098									
.150			-.3829	-.3620	-.3242	-.3113				
.163		.0000								
.177		-.2560								
.193		-.0912								
.229	-.1207									
.250			-.4755	-.5016	-.4788	-.4582				
.274		-.3706								
.339		-.2916								
.362	-.0391		-.5492	-.6112		-.5932				
.400			-.4772							
.452										
.497	.0000		-.5641	-.6390						
.550			.0000							
.565										
.600										
.650										
.700	-.3751		-.2188		-.6289					
.725			-.2386							
.750										
.760		-.1861								
.775			-.1720	-.4616						
.808		-.3851								
.834	-.0280									
.850		-.0347	-.1480	-.4224	-.4378					
.857										
.865	.0444									
.900	.0358		-.1264							
.905		-.0067	-.2986							
.918		-.0967								
.933		-.0158								
.965	-.0482									

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DATE 10 JAN 75

TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RB1U30)

UPPER WING

ARC11-716 1A14 01+712+912M25

ALPHA( 4 ) = 4.058 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING

DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8670

X/CW

.000	-.1341	-.2154	.3471	.6404	.6089	.5667	.6135
.020				.1992	.1915	.1367	.1133
.025			.1005				
.040		-.1116					
.045			-.2047				
.050	-.1840			-.0450	-.0672	-.1046	
.080				-.1903			
.085			-.1345				
.086		-.0516					
.094	-.1867						
.150				-.3794	-.3713	-.3335	-.3147
.163		.0000					
.177		-.2386					
.193		-.0824					
.229	-.1276						
.230				-.4516	-.4969	-.4845	-.4611
.274		-.3158					
.339		-.2467					
.362	-.0415						
.400				-.4792	-.5090		-.5939
.402		-.4151					
.497	.0000						
.550				-.4925	-.5954		
.565		.0000					
.600							-.5869
.650					-.6111		
.700	-.3321			-.2064			
.725				-.1986			
.750					-.5497	-.5890	
.760		-.1432		-.1634	-.3049		
.775		-.0501					
.808							
.834	-.0258			-.1171	-.2250	-.4182	
.850		-.0550					
.897							
.865	-.0051			-.0823			-.4524
.900	-.0182						
.905		-.0533		-.1634			
.950				-.0546		-.3639	
.953		-.0647					
.965	-.0905						



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TABULATED PRESSURE DATA - 1A14A - VOL. 6

PAGE 3265

(R81050)

UPPER WING

ARC11-716 1A14 01+712+512N25

ALPHA(4) = 4.050 BETA(4) = 4.097

SECTION (1) UPPER WING	DEPENDENT VARIABLE CF			
Y/BW	.2990	.3640	.4270	.5340 .6730 .7800 .8870
X/CW				
.000	-.3048	-.2922	.1999	.5439 .5165 .5309 .5640
.020				.1198 .1421 .1207 .0788
.025		.0228		
.040		-.1107		
.045			.1453	
.050	-.2650		-.0466	-.0753 -.0703 -.1142
.080			-.1771	
.085			-.1232	
.086		-.0564		
.094	-.2626		-.3254	-.3576 -.3161 -.3137
.150		.0000		
.163			-.1982	
.177		-.0485		
.193	-.1131			
.229			-.3788	-.4445 -.4569 -.4519
.250		-.2580		
.274				
.339	-.0096	-.2069		
.362			-.4394	-.5176 -.5848
.402		-.3812		
.497	.0000		-.4328	-.5150
.550		.0000		
.563				-.5675
.600				
.650				
.700	-.2903		-.1709	-.5731
.725			-.1974	
.750			-.5102	-.5382
.780		-.1272		
.775		-.0797	-.2185	
.808		-.0437		
.834	-.0551		-.0480	-.1735 -.2732
.850		-.0646		
.857				
.865	-.0421		-.0571	-.3678
.900	-.0402		-.0699	-.0382
.905			-.0560	-.2071
.950			-.0731	
.955				
.961	-.0881			

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(RBIUSD)

UPPER WING

ALPHA(4) = 4.060 BETA(5) = 8.170

ARC11-716 1A14 CR+T12+312N25

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW	.2993	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.000	-.5345	-.1833	.0435	.3008	.4456	.4130	.4343
.020	.020			.0551	.1419	.0836	.0393
.025			.0104				
.040		-.1178					
.045			-.0192				
.050	-.4296			-.0634	-.0573	-.0904	-.1553
.060				-.1622			
.065			.3516				
.066		-.1468					
.094	-.3233						
.150				-.2602	-.3275	-.3235	-.3201
.163		.0000					
.177			-.1487				
.193		.0310					
.229	-.0139						
.250				-.3255	-.4093	-.4539	-.4602
.274		-.2133					
.339		-.1833					
.362	.0378						
.400			-.3344	-.3869	-.4941		-.3851
.402							
.497	.0000			-.3045	-.4837		
.550		.0000					
.565							
.600							-.5645
.650					-.5659		
.700	-.1892			-.2199			
.725				-.1509			
.750					-.3651	-.4348	
.760		-.1568					
.775			-.0986	-.1753			
.808		-.0937					
.834	-.0746						
.850				-.1071	-.1129	-.2206	
.857		-.1263					
.865	-.0767						-.2589
.900	-.1037			-.1262	-.0546		
.905		-.1260					
.950				-.1291	-.0507		
.953		-.338					
.965	-.1378						

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(R01050)

UPPER WING

ARC11-716 1A14 01+712+512N25

ALPHA( 5) = 0.005 BETA( 1) = -0.030

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/BW	.2990	.3640	.4270	.5340
X/CW	.500	.1263	-.0231	.5369
	.020		.0327	.7960
	.025		.0892	.1132
	.040		.0827	.0512
	.045		-.0140	-.0821
	.050		-.0585	
	.060		-.1361	-.1573
	.080		-.3021	-.2073
	.085		-.2443	-.3020
	.086		-.0034	
	.094		-.0303	
	.150			-.5166
	.163		.0000	-.4842
	.177		-.4069	-.4685
	.193		-.1562	-.4677
	.229		-.0644	
	.230			-.6168
	.274		-.4726	-.6095
	.339		-.3532	-.5967
	.362		-.0090	
	.400			-.6821
	.402		-.5440	-.6761
	.497		.0000	-.6617
	.550			-.6866
	.565		.0000	-.6600
	.600			
	.650			-.6466
	.700		-.4557	-.5521
	.725			-.2656
	.750			
	.760		-.2413	-.5314
	.775		-.5263	-.5035
	.808		-.2110	
	.834			-.4282
	.850		-.1438	-.5390
	.857		-.1957	-.3114
	.865			
	.900		-.0029	-.3638
	.905		.0669	-.4901
	.950		-.1615	-.4669
	.955		-.2054	
	.993		-.1284	
	.985		-.0823	

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81050)

UPPER WING

ARC11-716 1A14 OL+T12+S12N25

ALPHA( 5) = 0.000 BETA( 2) = -3.980

DEPENDENT VARIABLE CP

SECTION ( 1) UPPER WING

Y/SW	X/CW	CP
.2990	.3640	.4270
.3340	.6730	.7800
.8870		
.0000	.0150	-.1503
.0200	.4042	.7468
.0250	.0378	.0842
.0400	.0965	-.0290
.0450	-.1240	-.1235
.0500	-.1201	
.0600	-.1163	-.1752
.0800	-.3187	-.2254
.0850	-.2776	-.3280
.0860	-.0327	
.0940	-.1125	-.5227
.1500	.0000	-.4894
.1630	-.3831	-.4770
.1770	-.1990	-.4827
.1930		
.2290	-.1295	-.6135
.2740	-.4562	-.6104
.3390	-.3481	-.5987
.3620		-.5833
.4000		
.4020	-.5259	-.6390
.4970	.0000	-.7042
.5500	-.6763	-.6803
.5650	-.0000	
.6000	-.6859	-.6570
.6500	-.2469	
.7000	-.3808	-.5718
.7250	-.2714	-.5948
.7500	-.2809	
.7600	-.5875	
.7750	-.2049	
.8000	-.1473	
.8340	-.0518	-.2504
.8500		-.3508
.8570		
.8650	.0099	
.9000	.0034	-.1995
.9050	-.0803	-.4774
.9200	-.1631	-.5021
.9330	-.0443	-.4636
.9650	-.0646	



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(R81U50)

UPPER WING

ARC11-716 1A14 OL+T12+S12M25

ALPHA( 3) = 7.910 BETA( 3) = .040

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7300	.8070
X/CW							
.006	-.1507	-.3115	.2708	.6296	.6379	.5840	.5591
.020				-.0180	.0382	-.0558	-.1490
.025			-.0517				
.040		-.2276					
.045			-.1605				
.050	-.1961		-.1971	-.2057	-.2494	-.3409	
.080			-.3321				
.085			-.2731				
.086		-.1502					
.094	-.1983			-.5165	-.4984	-.4852	-.4902
.150		.0000					
.163			-.3736				
.177		-.1707					
.193	-.1637			-.5814	-.6096	-.6026	-.5884
.229			-.4226				
.274		-.3083					
.339	-.0767			-.6037	-.6942		-.6867
.362							
.400			-.4588				
.402	.0000			-.5435	-.6766		
.497							
.550			.0000				
.565							
.600							-.6585
.650					-.6769		
.700	-.3997			-.2288			
.725			-.2419				
.750					-.6104	-.6185	
.760			-.2342				
.775				-.1998	-.5585		
.808			-.1393				
.834	-.0351			-.2017	-.5154	-.5391	
.850			-.1096				
.857							
.865	-.0157						
.900	-.0404			-.1839			-.4983
.905			-.0837		-.3611		
.950				-.1541		-.4718	
.955			-.0875				
.965	-.1222						

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DATE 10 JUL 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R81USD)

UPPER WING

ARC11-716 1A14 OR+T12+812N25

ALPHAO (S) = 8.000 BETAO (A) = 4.130

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/OW .2995 .3640 .4270 .5340 .6730 .7000 .8670

X/CW

.000	-.3425	-.4931	.0391	.5091	.573	.5410	.5200
.020				-.1039	.0053	-.0672	-.1515
.025				-.1569			
.040		-.2502					
.045			-.2115				
.050	-.2890			-.2301	-.2120	-.2497	-.3409
.080				-.3430			
.085			-.2683				
.086		-.1545					
.094	-.2157			-.4769	-.472C	-.4724	-.4820
.150		.0000					
.163			-.2703				
.177							
.193		-.1231					
.229	-.1641			-.4876	-.5753	-.5836	-.5801
.250			-.3241				
.274		-.2605					
.339	-.3361			-.4990	-.6556		-.6780
.362			-.4303				
.400				-.5036	-.6460		
.402							
.497	.0000						
.550		.0000					-.6524
.565						-.6597	
.600							
.650							
.700	-.3716			-.2556		-.6227	-.6237
.725							
.750			-.1780				
.760				-.1917	-.2000		
.775		-.0696					
.908							
.934	-.0536			-.1802	-.2006	-.5340	
.950			-.1025				
.957							
.965	-.0536			-.0987			-.4940
.990	-.0700		-.1096	-.1018			
.995				-.0909		-.4855	
.990							
.953			-.1260				
.965	-.1342						



(R81U90)

UPPER WING

TABULATED PRESSURE DATA - 1A14A - VOL. 6

ARC11-716 1A14 01-712-S12N25

ALPHA (S) = 7.980 BETA (S) = 0.220

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8070
X/CW							
.000	-.5456	-.2236	-.0614	.1779	.4730	.4454	.4475
.020				-.1065	-.0110	-.0870	-.1564
.025			-.0790				
.040		-.1035					
.045			-.1005				
.050	-.4667			-.2409	-.2233	-.2565	-.3346
.060				-.3047			
.065			-.1136				
.066		-.1476					
.094	-.4303			-.3698	-.4751	-.4726	-.4769
.150		.0000					
.163			-.2131				
.177		-.1364					
.193							
.229	-.0717			-.3945	-.5576	-.5610	-.5784
.250			-.2745				
.274		-.2330					
.339	.0044			-.4352	-.5815		-.6774
.362							
.400			-.3706				
.402				-.3424	-.5661		
.497	.0000		.0000				
.550					-.6498		-.6424
.565							
.600							
.650							
.700	-.2892			-.2471			
.725				-.2055			
.750			-.2164				
.760				-.1804	-.2803		
.775			-.1669				
.800							
.834	-.1281			-.1565	-.1994	-.2567	
.850			-.1910				
.857							
.865	-.1289			-.1669			-.4451
.900	-.1526						
.935			-.1768	-.0916			
.950			-.1873				-.1689
.953			-.1843				
.965	-.2058						

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ARC11-716 1A14 31+T12+S12N25 UPPER WING

(R01051) ( 15 FEB 74 )

## REFERENCE DATA

SREF = 2.4210 SQ.FT. XMRP = 29.9800 INCHES  
 LREF = 38.7090 INCHES YMRP = .0000 INCHES  
 SREF = 38.7090 INCHES ZMRP = .0000 INCHES  
 SCALE = .0300 SCALE

ALPHA(1) = -7.890 BETA(1) = -0.030

## PARAMETRIC DATA

MACH = 1.400 ELEVON = .000  
 RUDDER = .000 SPDRK = .000

## SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW	.000	.1141	.0109	.3997	.6268	.9701	.9063
.020	.020			.6502	.6254	.6483	.6821
.025	.025	.1057	.4422				
.040	.040		.3793				
.045	.045			.4416	.4636	.4785	.4805
.050	.050	.0481	.2660	.2798			
.060	.060		.1842				
.066	.066			.0376	.1080	.1676	.2033
.094	.094	.0566	.1104				
.150	.150	.0000					
.163	.163		.1964				
.177	.177						
.193	.193	.0751					
.229	.229						
.230	.230						
.274	.274						
.339	.339						
.362	.362						
.400	.400						
.402	.402						
.497	.497						
.530	.530						
.565	.565						
.600	.600						
.650	.650						
.700	.700						
.785	.785						
.790	.790						
.775	.775						
.808	.808						
.834	.834						
.850	.850						
.857	.857						
.865	.865						
.900	.900						
.905	.905						



## TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1031)

UPPER WING

DATE 19 JAN 75

ARC11-716 1A14 Q1+T12+312N25

ALPHA(1) = -7.880 BETA(1) = -8.050

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW	
.930	.1806
.933	.1782
.985	.1547

ALPHA(1) = -7.880 BETA(2) = -4.010

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW						
.930	.0337	-.0311	.2367	.5180	.4479	.3834
.933				.5902	.5569	.5842
.985			.3539			
.045		.0378	.3113	.4038	.4268	.4381
.050	-.0221			.2663		
.080			.2314			
.085						
.086		.1218				
.094	.0073			.0570	.1034	.1521
.150		.0000				.1845
.163			.0964			
.177		.1601				
.193						
.229	.0378			-.0714	-.0531	-.0319
.250						-.0500
.274				-.0165		
.339		-.0182				
.362	.1189			-.1895	-.2101	-.2236
.400						
.402			-.1726			
.497	.0000			-.2301	-.2571	
.550			.0000			
.565						-.3136
.600					-.3024	
.650						
.700	-.1072			-.1139		
.725		.0071				-.2777
.750						-.3002
.760			.0426			
.775			.1374	-.0988		
.808			.1680			
.834	.1202					
.850			.1674	.0131	-.1143	

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TABULATED PRESSURE DATA - 1A14A - VOL. 6

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(RBLUS1)

UPPER WING

ALPHAO ( 1 ) = -7.880 BETAO ( 2 ) = -4.010

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.837			.1432				
.865	.1773						
.900	.1584		.1546				-.2095
.905		.1411	.1599				
.950			.1468			-.0377	
.953			.1294				
.965	.1066						

ALPHAO ( 1 ) = -7.870 BETAO ( 3 ) = .020

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7800	.9870
X/CW							
.000	-.0805	-.1153	.1611	.3772	.3084	.2503	.2843
.020			.2920	.5206	.4696	.5015	.5057
.025		-.0158					
.040			.2597	.3678	.3798	.3837	.3823
.045				.2455			
.050	-.0780		.1959				
.080		.0802					
.085							
.086							
.094	-.0405			.0815	.0926	.1421	.1732
.150		.0000					
.163							
.177		.1367					
.193							
.229	.0088						
.250				-.0585	-.0472	-.0283	-.0289
.274							
.339		-.0239					
.362	.0916						
.400							
.402				-.1719			-.2124
.497							
.550	.0000			-.1592			
.565							
.600							
.650				-.2041	-.2358		
.700	-.0555						
.725							
.750							
.760							





TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01051)

UPPER WING

ALPHA( 1 ) = -7.970 BETA( 4 ) = 4.110

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.700	-.0332			.0929	-.1079		
.725						-.1044	-.2689
.750			.0432				
.760				.1255	.1356		
.775			.0940				
.800							
.834	.0855			.1055	.1318	.0638	
.850			.0502				
.857							
.865	.0901			.0750			.0000
.900	.0616		.0412		.1215		
.905				.0581		.1628	
.950			.0416				
.955							
.965	.0109						

ALPHA( 1 ) = -7.990 BETA( 5 ) = 0.100

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2990	.3640	.4270	.5340	.6730	.7600	.8670
X/CW							
.000	-.2211	-.1754	.0603	.1855	.0648	.0526	.0695
.020				.3681	.3070	.3590	.3514
.025			.1874				
.040		-.0818					
.045			.1891				
.050	-.1874			.2651	.2716	.2942	.2827
.060			.1377	.1834			
.065							
.086		.0236					
.094	-.1072			.0403	.0861	.1150	.1254
.150		.0000					
.165			.0495				
.177							
.193		.1073					
.229	-.0435						
.250				-.0528	-.0407	-.0350	-.0306
.274			-.0023				
.339		-.0539					
.362	.0635						
.400				-.1864	-.1531		-.1886
.402			-.1632				
.497	.0000						



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(R81U51)

UPPER WING

ARC11-716 1A14 01-112-512N25

ALPHA( 1 ) = -7.990 BETA( 5 ) = 8.180

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.559						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.834						
.850						
.857						
.865						
.900						
.905						
.950						
.953						
.965						

ALPHA( 2 ) = -3.960 BETA( 1 ) = -8.080

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000						
.020						
.025						
.040						
.045						
.050						
.080						
.085						
.086						
.094						
.150						
.163						
.177						
.193						
.229						
.250						
.274						
.339						

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(R81051)

UPPER WING

ARC11-716 1A14 01+712+512N25

ALPHA( 2 ) = -3.990 BETA( 1 ) = -8.080

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.362	.0911					
.400						
.402						
.497	.2600					
.550						
.565						
.600						
.650						
.700						
.725						
.750						
.760						
.775						
.808						
.824						
.850						
.857						
.865						
.900						
.950						
.953						
.965						

ALPHA( 2 ) = -3.990 BETA( 2 ) = -4.030

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.0322					
.020						
.025						
.040						
.045						
.050						
.080						
.085						
.086						
.094						
.150						
.163						
.177						
.193						



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(R81U51)

UPPER WING

ARC11-716 1A14 01+712+512M25

ALPHA( 2 ) = -3.990 BETA( 2 ) = -4.030

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8670

X/CW

.229 -.0063  
.250  
.274  
.339  
.362 .0548  
.400  
.402  
.497 .0020  
.550  
.565  
.600  
.650  
.700 -.1847  
.725  
.750  
.760  
.775  
.808  
.834 .0607  
.850  
.857  
.865 .1411  
.900 .1301  
.905  
.950  
.953  
.965 .0639

-1.706 -1.546 -1.392 -.1165

-.1215

-.1036

-2.664 -.2931 -.3003

-.2504

-.3167 -.3414

.0000

-.3822 -.3735

-.1312

-1.100 -1.3676 -.3626

-.0235

-.0140 -.2470

.0999

.0471 -.1567 -.2710

.1025

.0797 -.1442 -.2881

.1062

.0974 -.1538

.0915

.0639

ALPHA( 2 ) = -3.050 BETA( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7600 .8670

X/CW

.0000 -.0763 -.0783 .2202 .4578 .4404 .3925 .4402  
.020  
.025  
.040  
.045  
.050  
.060  
.065  
.066  
.094 -.1120

.2337

-.0090

.1853

.2714 .2806 .2887 .2907

.1367

.1046

.0490

(R81U51)

UPPER WING

ARC11-716 1A14 01+112+312M25

ALPHAO ( 2 ) = -3.850 BETA0 ( 3 ) = .040

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.150 .0000 -.0498 -.0175 .0235 .0603

.163 .0000 -.0179

.177 .0598

.193 .0598

.229 -.0273

.250 -.1535 -.1509 -.1391 -.1316

.274 -.1055

.339 -.1058

.362 .0278

.400 -.2454 -.2829 -.3065

.402 .02

.497 .0000

.550 .0000

.565 .0000

.600 .0000

.650 .0000

.700 -.1610

.725 -.0673

.750 -.0237

.760 .0446

.775 .0922

.808 .0358

.834 .0873

.857 .0685

.865 .0892

.900 .0768

.905 .0834

.920 .0749

.933 .0473

.965 .0158

ALPHAO ( 2 ) = -3.970 BETA0 ( 4 ) = 4.090

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000 -.1807 -.1120 .1683 .3991 .3345 .2874 .3311

.080 .080

.025 .025

.040 .040

.045 .045

.050 -.1788



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1US1)

UPPER WING

ARC11-716 1A14 OR+712+S12N25

ALPHA(2) = -3.973 BETA(4) = 4.090

DEPENDENT VARIABLE CP

SECTION (1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.080 .085 .086 .094 .150 .163 .177 .193 .229 .274 .339 .362 .400 .497 .530 .563 .600 .650 .700 .725 .750 .760 .775 .806 .834 .850 .857 .900 .905 .950 .955

.1170

.0837

-.0014

-.1542

.0000

-.0092

.0469

-.0630

-.0737

-.0935

.0183

-.2092

.0000

-.2606

-.3321

-.1365

-.0020

-.0129

.0610

.0246

.0755

.0417

-.1293

.0406

.0179

.0325

.0050

-.0215

-.0940

-.1756

-.3465

-.3673

-.3005

-.2913

-.2606

-.2520

-.1395

-.1330

-.1409

-.0389

-.0191

.0127

.0325



(RB1051)

UPPER WING

ARC11-716 IA14 01+112+512N25

ALPHA(2) = -4.000 BETA(5) = 0.150

SECTION (1) UPPER WING

DEPENDENT VARIABLE CP

V/BW	.2990	.3640	.4270	.5340	.6750	.7850	.8670
X/CW							
.000	-.2843	-.1374	.0463	.3302	.2498	.1842	.2145
.020				.3327	.3036	.3244	.3177
.025			.1531				
.040		-.1297					
.045			.1524				
.050	-.2616			.1982	.2014	.2137	.1998
.080				.1022			
.085			.0883				
.086		-.0529					
.094	-.1829						
.150				-.0302	-.0174	.0077	.0257
.163		.0000					
.177		-.0080					
.193		.0530					
.229	-.0899						
.250				-.1208	-.1239	-.1217	-.1332
.274			-.0563				
.339		-.0869					
.362	.0265						
.400			-.2137	-.2335			-.2807
.452		-.2027					
.497	.0200						
.550			-.2403	-.2768			
.585			.0000				
.620							
.650							-.3243
.700	-.0764			-.1551		-.3099	
.723			.0226				
.750							
.760							
.775			-.0135	.0725	-.0024		
.808			.0376				
.834	.0099						
.850			.0028	.0484	.0631	-.0631	
.857							
.885	.0418						
.900	.0139			.0803			-.0882
.905			-.0141	.0719			
.950				.0010			.0635
.955			-.0242				
.965	-.0453						



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(RR1U51)

UPPER WING

ARC11-716 1A14 01+112+112N25

ALPHA(3) = -.370 BETA(1) = -.8109

SECTION (1) UPPER WING DEPENDENT VARIABLE CP

Y/BW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	.1048	.1387	.4502	.7470	.7250	.6743	.7187
.020			.4528	.4855	.4801	.4926	
.025		.0442					
.040			.1817				
.045							
.050	.0549		.2198	.2326	.2576	.2581	
.060			.0510				
.085		.0316					
.094		.0929					
.094	.0101						
.150							
.165		.0000					
.177							
.193		.0112					
.229	-.0130						
.230							
.274							
.339							
.362	.0273						
.400							
.402							
.497	.0000						
.550							
.565							
.600							
.650							
.700	-.2382						
.725							
.750							
.760							
.775							
.808							
.834	-.0167						
.850							
.857							
.865	.1013						
.900	.1172						
.905							
.950							
.953							
.965	.0653						

DATE 10 JAN 79

(R01051)

## UPPER WING

ARC11-716 1A14 01+Y12+312A25

ALPHA( 3) = -.790      BETA( 2) = -4.010

SECTION (1) UPPER WING

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6750	.7600	.8670
.000	.0167	-.0631	.3672	.6608	.6368	.9667	.6226	.6226
.020			.2453		.4231	.4410	.4312	.4385
.025								
.040	.0025							
.045								
.050	-.0423		.1571		.2041	.2260	.2267	.2261
.060			.0352		.0479			
.066			.0535					
.094	-.0553							
.150					-.1503	-.1227	-.0632	-.0230
.163			.0050	-.0612				
.177			.0117					
.193								
.229	-.0471				-.2626	-.2601	-.2394	-.2131
.250				-.2006				
.274	-.1679							
.339								
.362	.0057			-.3129	-.3537	-.3667		-.3610
.400								
.402				.0000	-.3622	-.4280		
.497	.0000							
.519								
.565								
.600								
.650								
.700	-.2435				-.2066	-.1604	-.4548	-.4156
.725								
.750				-.0986			-.4514	-.4336
.760								
.775				-.0112	-.1567	-.3606		
.806								
.834	-.0023							
.850					-.1047	-.2573	-.3665	
.857				.0235				
.865	.0061							
.900	.0966				-.0554	-.2313		-.3613
.905				.0569				
.950					.0001		-.2222	
.955				.0574				
.965	.0316							

DATE 10 JAN 79

TABUN ATFO PRESSURE DATA - 1A14f - VOL. 6

(R01U51)

**UPPER WING**

AFC11-716 1A14 01 712 512N25

ALPHA( 3) =	BETA( 3) =	.030
-.390		

INIA BADDH (1) NC:11J36

**DEPENDENT VARIABLE CP**

Y/BW	X/CW	.2993	.3640	.4270	.5340	.6730	.7800	.8870
.000	-.0754	-.0913	.3031	.5638	.5350	.4863	.5307	.5307
.020			.2101	.3785	.3378	.3774	.3022	.3022
.040	-.0344		.1355	.1822	.1928	.1923	.1879	.1879
.065	-.1170		.0341	.0357				
.090			.0267					
.110	-.1334		.0000	-.0716	-.1321	-.1296	-.0811	-.0404
.130			.0085					
.150								
.170								
.190	-.0566				-.2555	-.2514	-.2429	-.2232
.210								
.230								
.250								
.274								
.302	-.0028		.1813					
.330								
.362								
.400								
.402								
.497	.0000							
.550								
.565								
.600								
.650								
.700	-.2217							
.725								
.750								
.760								
.775								
.808								
.834								
.850	-.0012							
.857								
.865	.0369							
.900	.0463							
.905								
.950								
.953								
.965	-.0134							

(R81051)

UPPER WING

ALPHA( 3 ) = -.400 BETA( 4 ) = 4.090

ARC11-716 1A14 Q1+T12+812N23

DEPENDENT VARIABLE CP

SECTION ( 1 ) UPPER WING

Y/8W	.2990	.3640	.4270	.5340	.6730	.7800	.8870
X/CW							
.000	-.1084	-.1502	.2580	.4933	.4481	.3935	.4399
.020				.3242	.3182	.3029	.3086
.025			.1532				
.040		-.0823					
.045			.0983				
.050	-.1933			.1509	.1468	.1373	.1356
.060				.0258			
.085			.0214				
.086		-.0090					
.094	-.1886						
.130				-.1318	-.1423	-.1047	-.0722
.163		.0000					
.177			-.0786				
.193		.0150					
.229	-.0824						
.274				-.2207	-.2456	-.2366	-.2341
.339		-.1303					
.362	.0026						
.400				-.2840	-.3391		-.3617
.402			-.2479				
.497	.0000			-.3177	-.3539		
.550			.0000				
.565							
.600							
.630							
.700	-.1678						
.725				-.1232			
.750					-.1532		
.760						-.3745	-.4132
.775			-.0736				
.806			.0221	-.0088	-.2085		
.834	-.0155						
.850				.0321	-.1423	-.2242	
.857			.0021				
.865	.0181						
.900	.0058			.0237			-.2596
.905			-.0062	.0020			
.930				.0141			-.1740
.953			-.0206				
.965	-.0486						



DATE 10 JAN 66 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RB1051)

UPPER WING

ARC11-716 1A14 01-112-512N25

ALPHA(3) = -4.000 BETA(5) = 0.140

DEPENDENT VARIABLE C<sub>p</sub>

SECTION (1) UPPER WING

Y/BW .2990 .3640 .4270 .5340 .6750 .7800 .8870

X/CW

.000	-.3415	-.1732	.0929	.3773	.3554	.3076	.3393
.020				.2478	.2707	.2653	.2555
.025			.0665				
.040		-.1499					
.045			.0325	.1124	.1184	.1177	.1039
.050	-.2889			.0116			
.060			.0295				
.065							
.066		-.0413					
.094	-.2560			-.1174	-.1192	-.0977	-.0657
.150		.0000					
.163			-.0432				
.177		.0257					
.193							
.229	-.0924			-.1803	-.2122	-.2232	-.2118
.250			-.1028				
.274		-.1202					
.339							
.362	.0148			-.2690	-.3133		-.3513
.400			-.2308				
.402							
.497	.0000			-.2938	-.3370		
.550			.0000				
.565							-.3987
.600							
.630							
.700	-.1619			-.1052			
.725					-.1764		
.750							
.760							
.775							
.806							
.834	-.0265			-.0149	-.1348		
.850							
.857			-.0296				
.865	.0017						
.900	-.0177						
.905			-.0344	-.0093	.0477		-.1698
.950				-.0204			
.955							-.1129
.965	-.0728		-.0460				



$$\text{ALPHA}(4) = 4.100 \quad \text{BETA}(2) = -3.980$$

SECTION / UPPER WING

$\pi/\delta W$	$x/C_W$	.2993	.3640	.4270	.5340	.6730	.7830	.8670
.000	.0004	-.0302	.3934	.7342	.7264	.6916	.7169	
.020			.1619	.2741	.3337	.3039	.2830	
.025								
.040		-.0239						
.045			.0564					
.050	-.0281				.0633	.0937	.0941	.0590
.060				-.0856				
.065			-.0724					
.080		.0040						
.085	-.0636				-.2647	-.2431	-.1920	-.1635
.094		.0003						
.150			-.0676	-.1960				
.163								
.177								
.193								
.229	-.0659				-.3565	-.3732	-.3429	-.3174
.250				-.2786				
.274			-.2233					
.339		-.0176			-.4347	-.4736		-.4513
.362				-.3715				
.400		.0000			-.4353	-.5070		
.402								
.497								
.550			.0000					
.565								
.600								
.650								
.700	-.3030				-.3309	-.1914		-.4520
.725								
.750			-.1600				-.3120	-.4520
.760					-.2396	-.4711		
.775								
.806			-.1491					
.834	-.0865							
.850					-.2106	-.3421	-.4441	
.857				-.0991				
.865	.0301							
.903	.0537				-.2078		-.2939	-.4129
.905				-.0373				
.930					-.1723		-.2769	
.933			.0006					
.965	-.0079							



## TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01051)

UPPER WING

ARC11-716 1A14 01+T12+S12N25

ALPHA( 4) = 4.100 BETA( 3) = .060

SECTION ( 1) UPPER WING	DEPENDENT VARIABLE CP			
Y/W	.2990	.3640	.4270	.5340 .6730 .7030 .8070
X/CW				
.000	-.0942	-.1675	.3277	.6426 .6316 .5584 .6217
.020			.2353	.2799 .2414 .2147
.025			.1187	
.040		-.1105		
.045			.0309	
.050	-.1291		.0413	.0590 .0475 .0088
.060			-.1018	
.085		-.0668		
.086		-.0219		
.094	-.1275			-.2853 -.2531 -.2112 -.1891
.150		.0000		
.163			-.1872	
.177		-.0674		
.193				
.229	-.1012			
.250		-.2722		-.3522 -.3732 -.3535 -.3330
.274				
.339	-.0276	-.2270		
.362				
.400		-.3619		-.4213 -.4683 -.4619
.402				
.497	.0000			
.550		-.4332 -.4954		
.565		.0000		
.600				-.4689
.650			-.5079	
.700	-.2915		-.1754	
.725			-.2237	
.750				-.5192 -.4754
.760		-.1577		
.775		-.3696		
.808		-.0920		
.834	-.0315			
.850		-.1935	-.3284	-.3411
.857		-.0497		
.865	.0319			
.880	.0269			
.905		-.1368		-.4032
.950		-.0190	-.2968	
.950		-.0718		-.2747
.955		-.0125		
.965	-.0412			



DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(RBIUS1)

UPPER WING

ARC11-716 1A14 01-712-512N23

ALPHA( 4 ) = 4.100 BETA( 4 ) = 4.130

SECTION ( 1 ) UPPER WING DEPENDENT VARIABLE CF

Y/OW .2990 .3640 .4270 .5340 .6730 .7800 .8870

X/CW

.000	-.2222	-.3484	.2086	.5347	.5324	.4985	.5368
.020				.1486	.1915	.1644	.1508
.025			.0473				
.040		-.1228					
.045			-.0254				
.050	-.2143			-.0167	-.0037	-.0132	-.0447
.060				-.1359			
.065			-.1080				
.066		-.0526					
.094	-.2045			-.2792	-.2785	-.2451	-.2259
.190		.0000					
.163			-.1791				
.177		-.0333					
.193	-.1014			-.3342	-.3635	-.3750	-.3582
.229			-.2475				
.250		-.2037					
.274							
.339	-.0240			-.3822	-.4586		-.4784
.362			-.3889				
.400				-.3911	-.4699		
.402			.0000				-.4892
.497	.0000					-.5154	
.550				-.2004	-.1941		
.565						-.4863	-.3896
.600							
.690	-.2631						
.700							
.725							
.750							
.760			-.1470	-.1553	-.3050		
.775			-.0534				
.808				-.0998	-.2930	-.3227	
.814	-.0245		-.0504				
.850							
.857							
.865	-.0080			-.0634			-.3580
.900	-.0242		-.0486	-.2121			
.905				-.0424			-.2780
.910							
.953			-.0605				
.965	-.0036						



ALPHA( 5) = 0.010 BETA( 1) = -0.050

SECTION ( UPPER WING		DEPENDENT VARIABLE CP			
X/CW					
.000	.2990	.3640	.4270	.5340	.6790 .7800 .8870
.020	.1928	.0761	.5625	.8960	.8447 .7924 .7829
.025			.1843	.2385	.2382 .1610 .1039
.040		.0315			
.045			.0519	.0134	-.0195 -.0428 -.1184
.050	.0394			-.1474	
.060			-.1148		
.065		.0489			
.066					
.094	.0178			-.3584	-.3252 -.3020 -.2912
.150		.0500			
.163			-.2868		
.177		-.1036			
.193					
.229	-.0131			-.4628	-.4464 -.4275 -.4013
.250					
.274			-.3606		
.339		-.2919			
.362	-.0076			-.5337	-.5405
.400			-.4584		
.492	.0000			-.5513	-.5565
.497			.0000		
.550					-.5189
.565					
.600				-.4864	
.650	-.3376				-.4825 -.5138
.700					
.725			-.2859		
.750				-.4478	-.4679
.760			-.2527		
.775					
.806					
.834	-.2664				
.850			-.2900	-.3481	-.4571 -.4731
.857					
.865	-.1378				
.900	-.0300			-.3024	-.4162
.905			-.1859		
.950				-.2577	-.4307
.953			-.1146		
.965	-.0282				

$$\text{ALPHA}( 5 ) = 0.010 \quad \text{BETA}( 2 ) = -4.010$$

SECTION (1) UPPER WING	DEPENDENT VARIABLE CF
1	1
2	2
3	3
4	4
5	5
6	6
7	7
8	8
9	9
10	10
11	11
12	12
13	13
14	14
15	15
16	16
17	17
18	18
19	19
20	20
21	21
22	22
23	23
24	24
25	25
26	26
27	27
28	28
29	29
30	30
31	31
32	32
33	33
34	34
35	35
36	36
37	37
38	38
39	39
40	40
41	41
42	42
43	43
44	44
45	45
46	46
47	47
48	48
49	49
50	50
51	51
52	52
53	53
54	54
55	55
56	56
57	57
58	58
59	59
60	60
61	61
62	62
63	63
64	64
65	65
66	66
67	67
68	68
69	69
70	70
71	71
72	72
73	73
74	74
75	75
76	76
77	77
78	78
79	79
80	80
81	81
82	82
83	83
84	84
85	85
86	86
87	87
88	88
89	89
90	90
91	91
92	92
93	93
94	94
95	95
96	96
97	97
98	98
99	99
100	100

Y/BW	X/CW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
.000	.0172	-.0356	.4268	.4119	.7821	.7102	.7127	
.020			.0920	.1853	.1992	.1171	.0664	
.040		-.0641						
.060			-.0306	-.0269	-.0410	-.0800	-.1424	
.080	-.0374			-.1774	-.1799			
.100			-.0214					
.120	-.0631				-.3824	-.3256	-.3130	
.140		.0000		-.3220				
.160		-.1609						
.180	-.0771			-.3746	-.4743	-.4437	-.4202	
.200			-.3104					
.220	-.0702				-.5393	-.5543	-.5219	
.240			-.4363					
.260	.0000		.0000		-.5905	-.5738		-.5136
.280							-.5228	
.300	-.3449			-.4749	-.2926			
.320			-.2611		-.3879	-.4721	-.4768	-.5078
.340		-.2965						
.360			-.2153		-.2407	-.4647	-.4614	
.380	-.1613							-.4652
.400		-.0905			-.1800			
.420	.0025		-.1261		-.4107	-.4088		
.440			-.0460		-.1804			
.460								
.480	-.0434							

DATE 10 JAN 75 TABULATED PRESSURE DATA - 1A14A - VOL. 6

(R01051)

UPPER WING

ARC11-716 1A14 01+112+S12N25

ALPHA( 5 ) = 0.020 BETA( 3 ) = .030

SECTION ( 1 ) UPPER WING		DEPENDENT VARIABLE CP	
Y/BW	X/CW		
.2990	.3640	.4270	.5340 .6730 .7800 .8870
.000	-.0909	-.2183	.2836 .6726 .6797 .6227 .6232
.020			.0756 .1343 .0660 .0149
.025		-.0160	
.040		-.1633	
.045		-.1108	
.050	-.1326	-.0992	-.0896 -.1178 -.1771
.080		-.2239	
.085		-.2045	
.086		-.1035	
.094	-.1302		
.150		.0000	-.3974 -.3638 -.3455 -.3340
.163		-.2998	
.177		-.1614	
.193	-.1312		-.4694 -.4709 -.4563 -.4362
.229		-.3397	
.250		-.2810	
.274			
.339	-.0761		-.5928 -.5534 -.5315
.362		-.4162	
.400			-.5115 -.5789
.402	.0020	.0200	
.497			-.5256
.530			-.5348
.561			-.2084
.600		-.3010	
.650	-.3403		-.4889 -.5209
.700		-.2373	
.725		-.2646	-.4842
.750		-.2027	
.760			
.775			
.808		-.1454	
.834	-.0819	-.2310	-.4590 -.4526
.850			
.857			
.865	-.0236	-.2181	-.4525
.900	-.0060	-.0870	-.4091
.915		-.1822	-.3791
.940		-.0555	
.975			
.965	-.0651		

DATE 10 JAN 75 TABULATED PRESSURE DATA - IA14A - VOL. 6

(R81U51)

UPPER WING

ARC11-716 IA14 01+T12+S12N25

ALPHA( 5) = 6.010 BETA( 4) = 4.150

SECTION ( 1) UPPER WING DEPENDENT VARIABLE CP

Y/BW	.2995	.3645	.4275	.5340	.6730	.7890	.8870
X/CW							
.000	-.2416	-.4287	.0577	.5332	.5642	.5267	.5298
.025			-.1031	-.0260	.0615	.0073	-.0403
.045		-.2649					
.065			-.1561				
.085	-.2315			-.1641	-.1444	-.1595	-.2171
.105			-.2259	-.2670			
.125		-.1526					
.145	-.2110						
.165		.0000		-.4136	-.3898	-.3706	-.3591
.185			-.2805				
.205		-.1251					
.225	-.1617			-.4583	-.4825	-.4704	-.4565
.245		-.3164					
.265	-.0590	-.2468		-.4610	-.5527		-.5458
.285			-.3719				
.305	.0000			-.4400	-.5678		
.325			.0000				-.5415
.345	-.3132			-.2240			
.365			-.2699				
.385		-.2078				-.5500	-.5491
.405			-.1251	-.2443	-.4839		
.425							
.445	-.0553		-.1062	-.2043	-.3494	-.4470	
.465							
.485	-.0427			-.1586			-.5084
.505	-.0583		-.0922		-.2355		
.525			-.1147			-.3802	
.545		-.0901					
.565	-.1154						



$$\text{ALPHA}(\text{S}) = 7.993 \quad \text{BETA}(\text{S}) = 0.280$$

SECTION (1) UPPER WING

Y/YW	.2990	.3640	.4270	.5340	.6750	.7800	.8870
X/CW							
.000	-.3963	-.3128	-.1025	.1928	.4527	.4664	.5147
.020				-.1603	.0085	-.0047	-.0288
.025			-.1190				
.040		-.2997					
.045			-.1242				
.050	-.3540			-.2235	-.1745	-.1701	-.2054
.060			-.1446	-.2764			
.065							
.066		-.2338					
.094	-.2946						
.150		.0000		-.3260	-.3896	-.3460	-.3419
.163							
.177			-.1970				
.193		-.1221					
.229	-.1308						
.250							
.274			-.2435				
.339	-.1995						
.362	-.0254			-.3738	-.4901		-.3612
.400							
.402	.0000						
.497							
.550				-.3725	-.4690		
.565			.0000				
.601							
.650							
.700	-.2969			-.1837		-.5303	-.5187
.725					-.2214		
.750							
.760			-.1753	-.1216	-.2773		
.775							
.808			-.0876				
.834	-.0746						
.850							
.857			-.1061	-.0968	-.2411	-.4155	
.865	-.0606						
.900	-.0753			-.1010			-.3921
.905			-.1026		-.1491		
.950				-.1019		-.2937	
.953			-.1157				
.965	-.1184						